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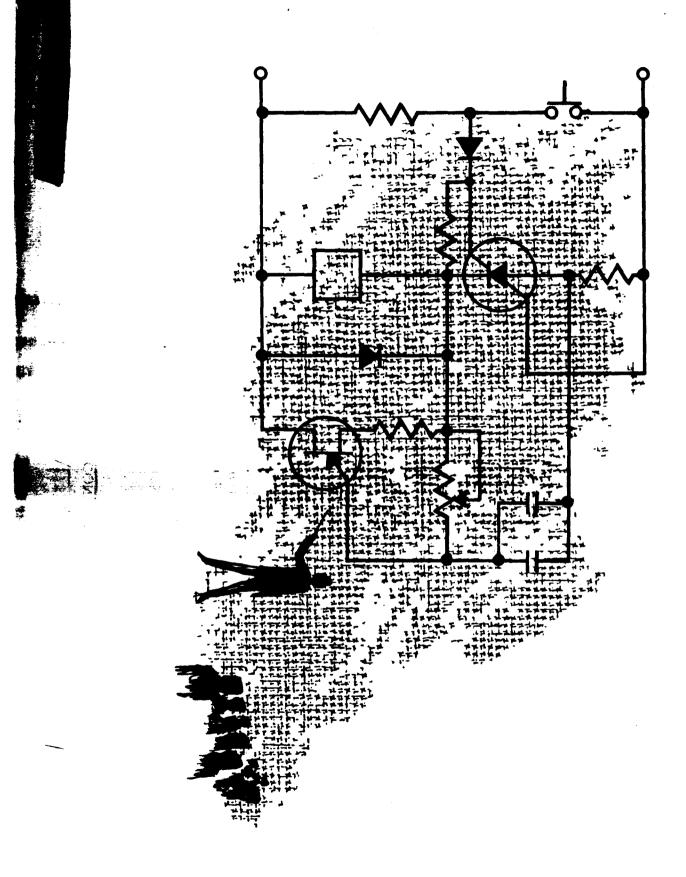
In Accordance with E. O. 12958 on NOV 26 1997

CORONA/J-3 DESIGN REVIEW

PAGES E-I thru E-50

(ELECTRICAL)

DECEMBER 1966



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Chi

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COMMAND SYSTEM

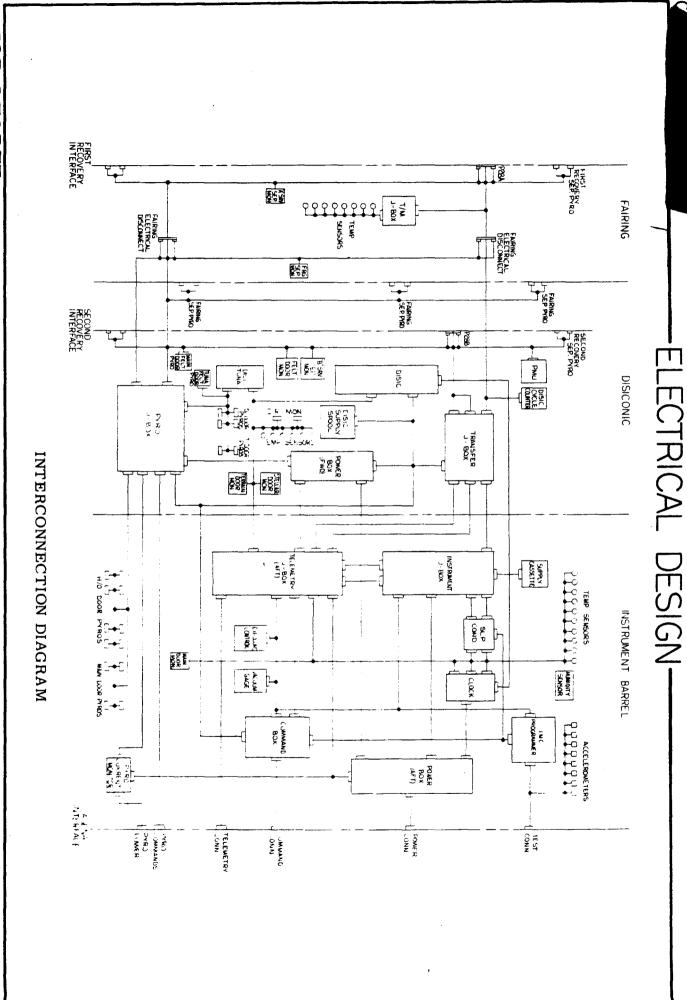
EXPOSURE CONTROL

EXPOSURE CONTROL	EXPOSURE CONTROL TIMING DIAGRAM	EXPOSURE CONTROL COMMAND DESCRIPTION	EXPOSURE CONTROL SIMPLIFIED SCHEMATIC	EXPOSURE CONTROL	EXPOSURE CONTROL PROGRAMMER BLOCK DIAGRAM
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J-3 DESIGN REVIEW

NO.



J-3 ELECTRICAL IMPROVEMENTS

COMMAND SYSTEM

- MORE CAPABILITY FOR BYPASSING VARIOUS PORTIONS OF CAMERA OPERATE CIRCUITS
- ALL CAMERA OPERATE PROGRAMS ARE AVAILABLE FOR EMERGENCY OPERATE MODE,
- RELIABILITY. FEWER SERIES ELEMENTS IN PRIMARY COMMAND AND CONTROL LINES INCREASES
- EXPOSURE CONTROL FOR CR PAN CAMERA AND DISIC PROVIDED
- RUGGEDIZED STEPPER SWITCHES ELIMINATE NEED FOR REMOVAL OF POWER TO SWITCHES DURING ASCENT.
- REDUNDANT CIRCUITS PROVIDED FOR A TO B TRANSFER CONTROL CIRCUITS

IMC SYSTEM

- EXPANDED SELECTION OF AVAILABLE CURVES TO MATCH ORBIT
- EARTH OBLATENESS CORRECTION FACTOR ADDED TO ORBIT ECCENTRICITY FUNCTION
- HAS SLEW TO HOME CAPABILITY IF START DELAY TIME PLUS FUNCTION PERIOD EXCEEDS ORBIT PERIOD.

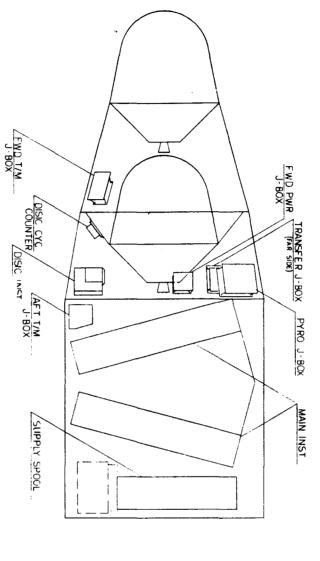
T/M SYSTEM

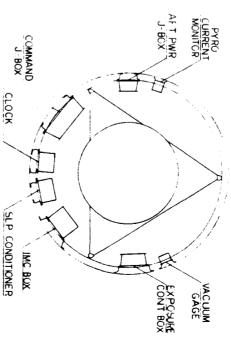
INCREASED BITS OF INFORMATION BY ADDING MORE COMMUTATED CHANNELS,

PYRO SYSTEM

- SEPARATE PYRO VOLTAGE BUSS IS PROVIDED.
- EACH PYRO DEVICE IS ACTIVATED BY INDEPENDENT CIRCUITS FROM COMMAND TO SQUIB EXCITATION
- FUSISTOR LOCATED BETWEEN PYRO POWER SOURCE & RELAY CONTACTS PROTECTS BUSS FROM POSSIBLE PLASMA ARCING IN RELAYS
- ADDED RECOVERABLE TAPE RECORDER



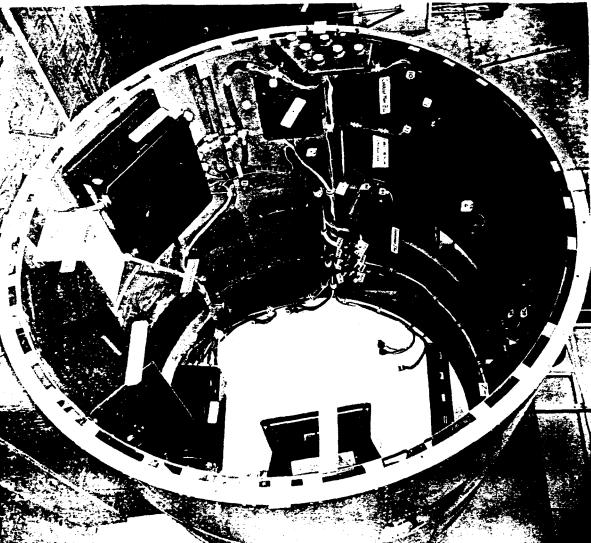




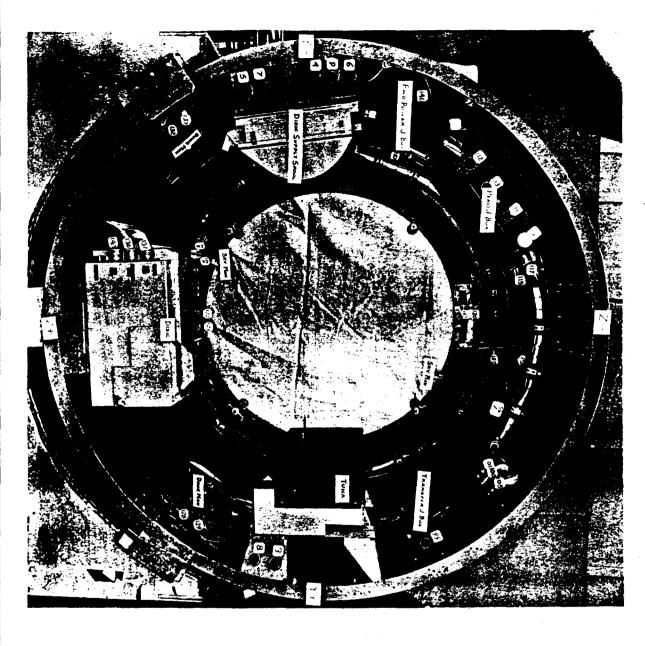
EQUIPMENT INSTALLATION

PROFILE

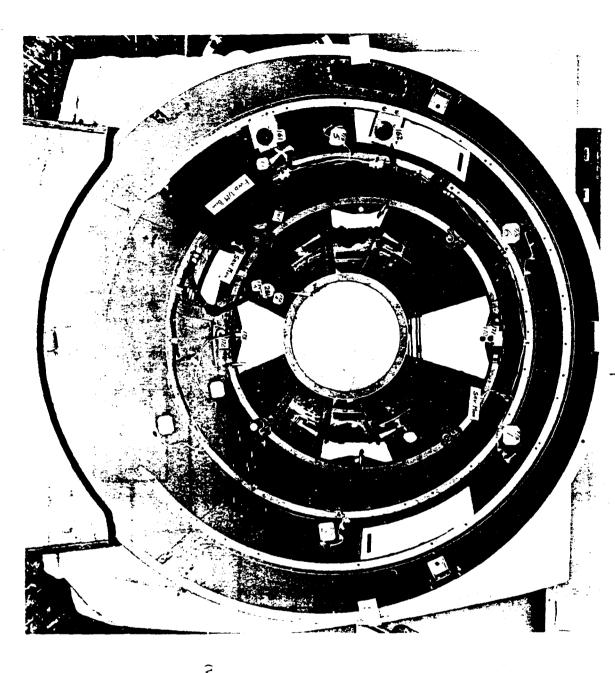




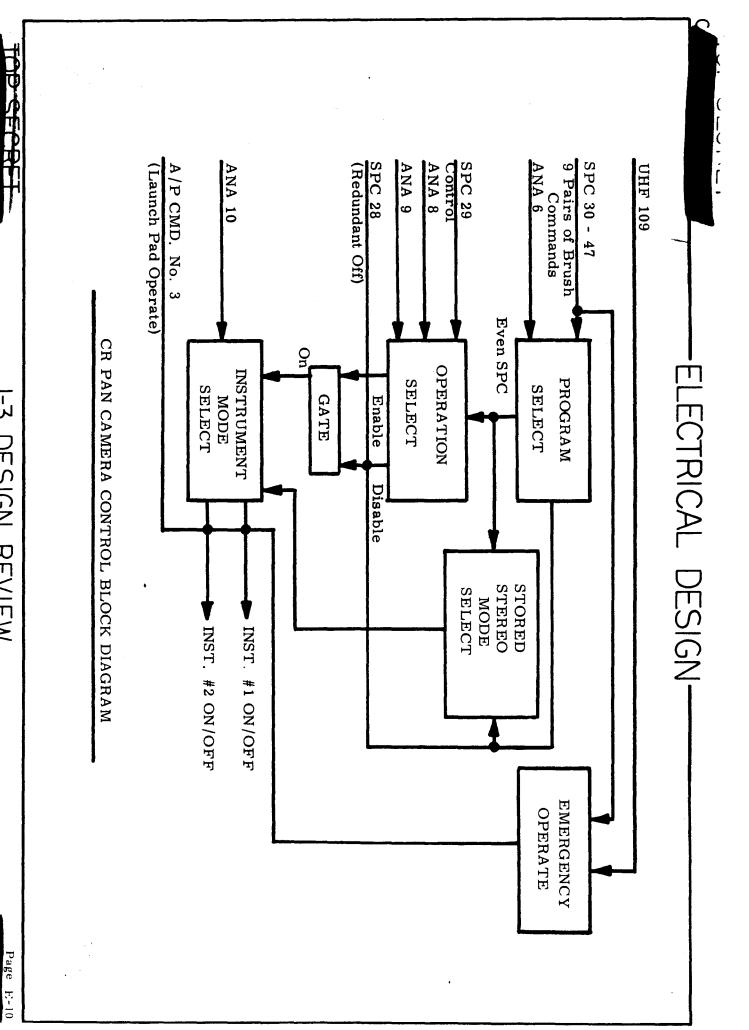
(VIEW LOOKING FORWARD INTO IN THE SITUAL INTERNAL MONT INSTRUMENT BARROLL, VOLNATIN OTREACE



(VIEW LOOKING FORWARD INTO DISIC CONIC MOCK-UP) DISIC CONIC/INSTRUMENT BARREL INTERFACE

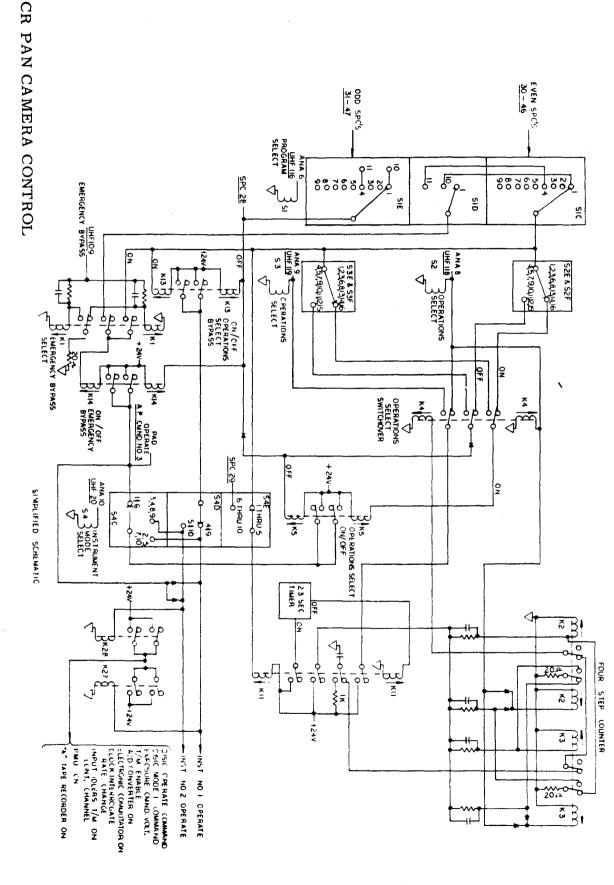


FAIRING/DISIC CONIC INTERFACE
(VIEW LOOKING FORWARD INTO FAIRING MOCK-UP)



CR PAN CAMERA CONTROL

- UTILIZES TWO STEPPER SWITCHES FOR OPERATION SELECT CAPABILITY.
- STORED PROGRAM COMMANDS OPERATION SELECT ALLOWS IN-FLIGHT EDITING OF SELECTED INSTRUMENT
- SPC ENABLES OR DISABLES. PROVIDES SELECTION OF 256 DIFFERENT COMBINATION SEQUENCES OF EIGHT
- NINE INSTRUMENT OPERATE PROGRAMS AVAILABLE.
- & INSTRUMENT MODE SELECTOR RTC CAPABILITY TO BYPASS PROGRAM SELECTOR, OPERATIONS SELECTOR,



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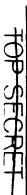
J-3 DESIGN REVIEW

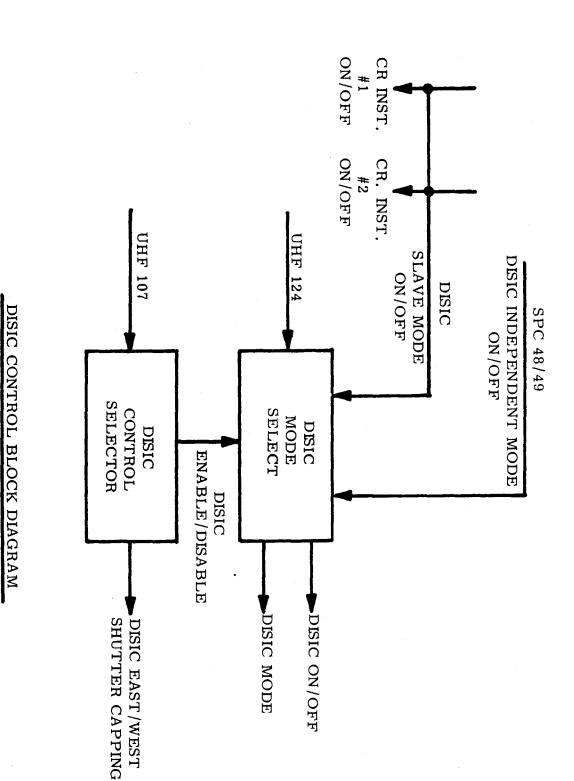
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CR PAN CAMERA CONTROL COMMAND DESCRIPTION

- ANA-6/UHF-116 SELECTS ONE OF NINE AVAILABLE SPC PAIRS FOR CAMERA ON/OFF CONTROL. SUBSEQUENT IN-FLIGHT PROGRAM EDITING ACCOMPLISHED BY ANA-8/UHF-118 AND ANA-9/UHF-119
- ANA-8/UHF-118 SELECTS ONE OF SIXTEEN AVAILABLE COMBINATIONS OF FOUR SEQUENTIAL STEPS TO PROVIDE ENABLING OR DISABLING OF THE SELECTED "ON" SPC.
- THE RESULTING CAPABILITY IS 256 COMBINATIONAL SEQUENCES OF CAMERA ON/OFF COMMANDS ANA-9/UHF-119 PROVIDES IDENTICAL CAPABILITY AS ANA-8/UHF-118. THIS SEQUENCE IS AUTOMATICALLY STARTED ON THE COMPLETION OF THE FOUR STEPS SELECTED BY ANA-8/UHF-118
- ADVANCES OPERATIONS SELECT LOGIC. THE TWO SOURCES ARE THE SELECTED "ON" SPC OR COMMANDS RESULTING FROM OPERATIONS SELECT. SELECTS SOURCE OF COMMAND WHICH ANA-10/UHF-120 SELECTS EITHER, BOTH OR NEITHER INSTRUMENT TO RESPOND TO ON/OFF
- UHF-109 PROVIDES CAPABILITY OF SELECTING EMERGENCY CAMERA OPERATIONS BYPASSING PROGRAM SELECT, OPERATION SELECT & INSTRUMENT MODE SELECT. IT IS ALSO USED AS A BACKUP RESET FOR TIMERS IN A TO B TRANSFER CIRCUITS.
- SPC 30 THROUGH 47: CAMERA PROGRAMS 1 THROUGH 9. EVEN NUMBERED SPC'S ARE ON CONVERTED TO AN ACTIVE "OFF" COMMAND BY OPERATIONS SELECT DISABLE LOGIC. COMMANDS, ODD NUMBERED SPC'S ARE OFF COMMANDS. EVEN NUMBERED SPC'S MAY BE
- SPC 28 CAMERA PROGRAMS REDUNDANT OFF COMMAND. ALL OF THE VARIOUS CAMERA OPERATE CONTROL CIRCUITS WILL RESPOND TO AN SPC 28.
- SPC 29 WHEN SELECTED BY ANA-10/UHF-120 ADVANCES THE ACTIVE OPERATIONS SELECT LOGIC SEQUENCE ONE STEP. PROVIDES BLOCK EDITING OF CAMERA OPERATE PROGRAMS
- AP CMD #3 ALLOWS BLOCKHOUSE OPERATION OF CR INSTRUMENTS IN STEREO MODE

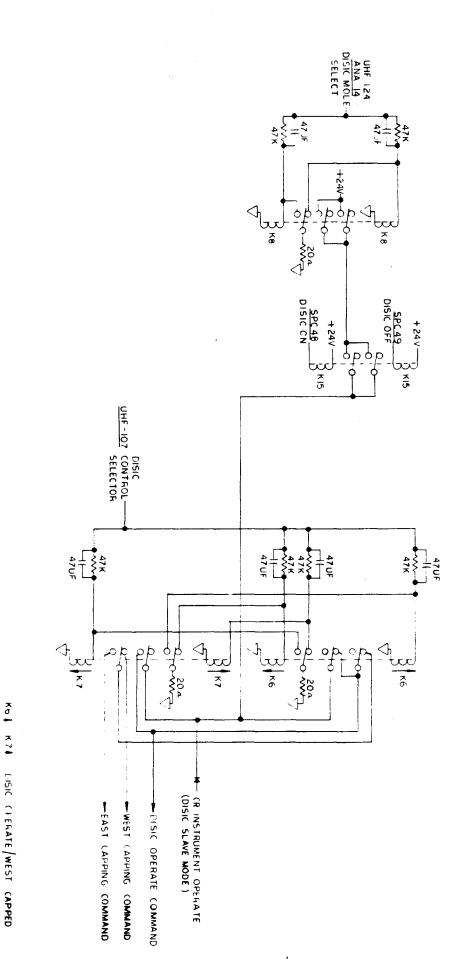




J-3 DESIGN REVIEW

DISIC CONTROL

- HAS INDEPENDENT SPC CONTROL WHICH IS ENABLED BY RTC.
- SLAVE CONTROL MODE OVERRIDES INDEPENDENT CONTROL MODE
- IN SLAVE CONTROL MODE THREE PAIR OF STELLAR EXPOSURES AR E MADE TO ONE TERRAIN EXPOSURE.
- IN INDEPENDENT CONTROL MODE, ONE PAIR OF STELLAR EXPOSURES ARE MADE TO ONE TERRAIN EXPOSURE.
- SHUTTER RTC CONTROL IS USED TO INHIBIT UNCAPPING OF EITHER STELLAR
- RTC CONTROL ALSO PROVIDES COMPLETE SYSTEM DISABLING.



DISIC CONTROL

X6↓ X7↑ CISIC OPERATE NONE CAPPED LISIC OPENATE/EAST CAPPED

K61 K71 DISIC DISABLE

J-3 DESIGN REVIEW

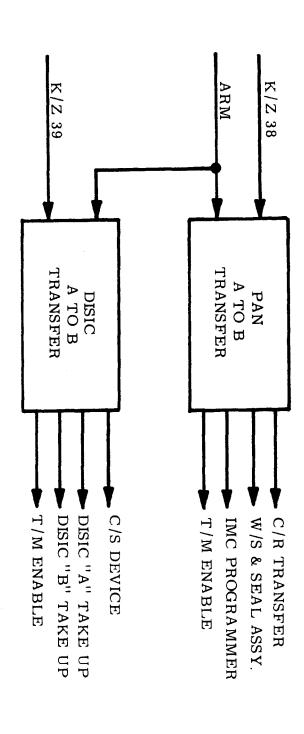
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DISIC CONTROL COMMAND DESCRIPTION

ANA-14/UHF-124 ALTERNATELY DISABLES OR ENABLES SPC-48 & SPC-49 FOR USE AS AN INDEPENDENT DISIC OPERATE PROGRAM.

OPERATE COMMAND FROM BEING SENT TO THE DISIC. UNCAPPING CIRCUITRY. IT IS ALSO USED TO DISABLE ANY



A TO B TRANSFER BLOCK DIAGRAM

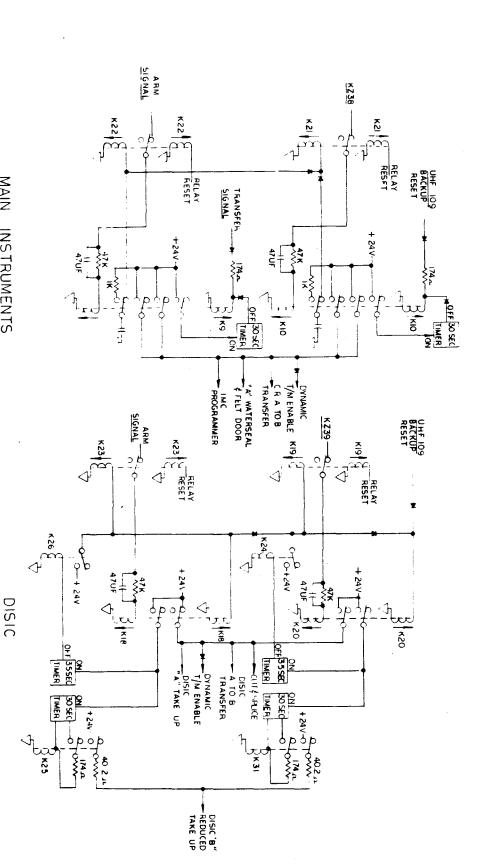
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J-3 DESIGN REVIEW

A TO B TRANSFER

SEQUENCES BY SECURE RTC'S. INDEPENDENT EXECUTION OF DISIC OR CR "A" TO "B" TRANSFER

BOTH SEQUENCES BACKED-UP BY REDUNDANT CIRCUIT EXECUTED BY "A" RECOVERY COMMANDS.



MAIN INSTRUMENTS

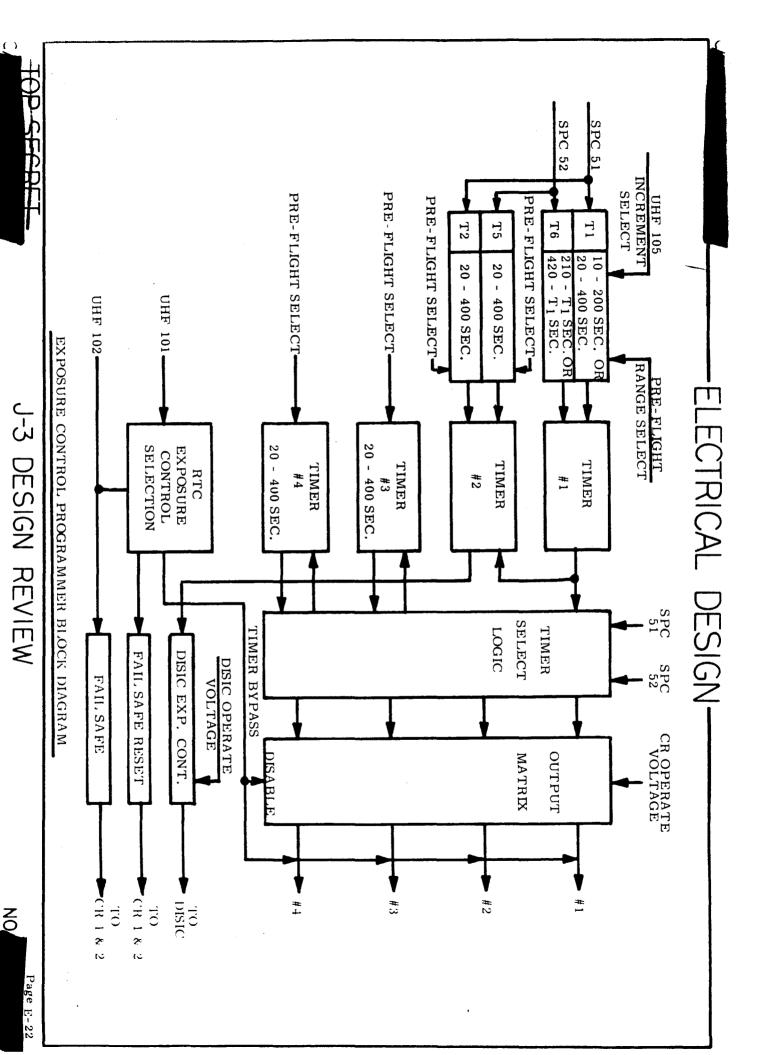
A TO B TRANSFER

J-3 DESIGN REVIEW



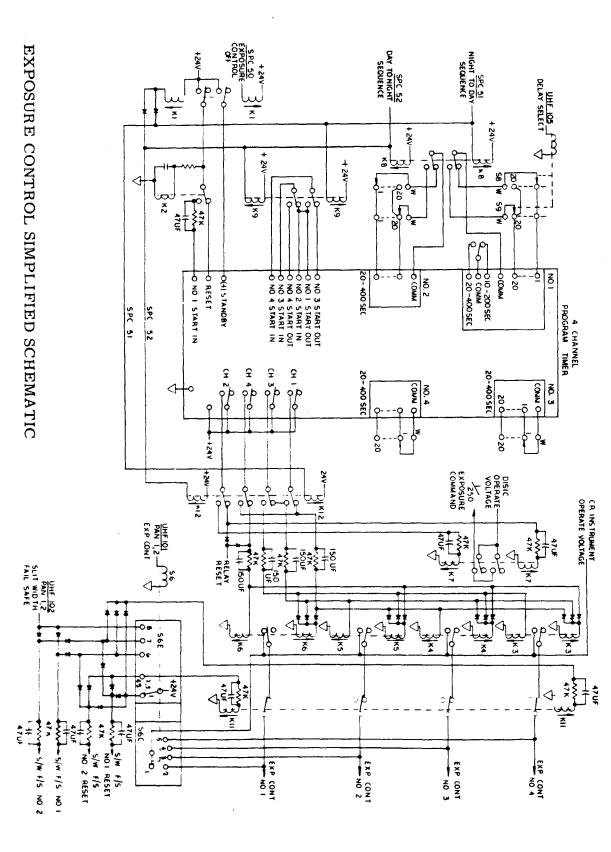
A TO B TRANSFER COMMAND DESCRIPTION

- KZ-38 INITIATES A SEQUENCE OF EVENTS RESULTING IN THE TRANSFER OF THE CR INSTRUMENT'S FILM TAKE-UP CONTROL & INSTRUMENTATION FROM THE "A" SRV TO THE "B" SRV.
- KZ-39 INITIATES A SIMILAR SEQUENCE FOR THE DISIC FILM TRANSFER.
- UHF-109: BACKUP RESET FOR TIMERS INITIATED BY KZ-38 & KZ-39
- "A" RECOVERY ARM SIGNAL INITIATES IDENTICAL SEQUENCES TO THOSE INITIATED BY KZ-38 & KZ-39. IT RESULTS IN SIMULTANEOUS TRANSFER OF CR AND DISIC FILM FROM SRV "A" TO "B" SRV, IF NOT PREVIOUSLY COMPLETED BY KZ-38 OR KZ-39.
- BY THE ARM SIGNAL. "A" RECOVERY TRANSFER SIGNAL: BACKUP RESET FOR TIMERS INITIATED



EXPOSURE CONTROL

- PROVIDES CONTROL OF FOUR EXPOSURE POSITIONS AND A FAIL SAFE POSITION FOR THE CR INSTRUMENT AND ONE EXPOSURE CHANGE FOR
- SEQUENCES THROUGH EXPOSURE CHANGES FROM NIGHT TO DAY & DAY TO NIGHT POSITIONS.
- SPECIFIC TARGETING REQUIREMENTS, ONE OF THE FOUR EXPOSURE POSITIONS FOR THE CR INSTRUMENTS FOR HAS RTC CONTROL FOR BYPASS OF SEQUENCER AND SELECTION OF
- PROVIDES RTC CONTROL OF TIMING SYSTEM TO COMPENSATE FOR LAUNCH WINDOW UNCERTAINTIES AND/OR DELAYS.



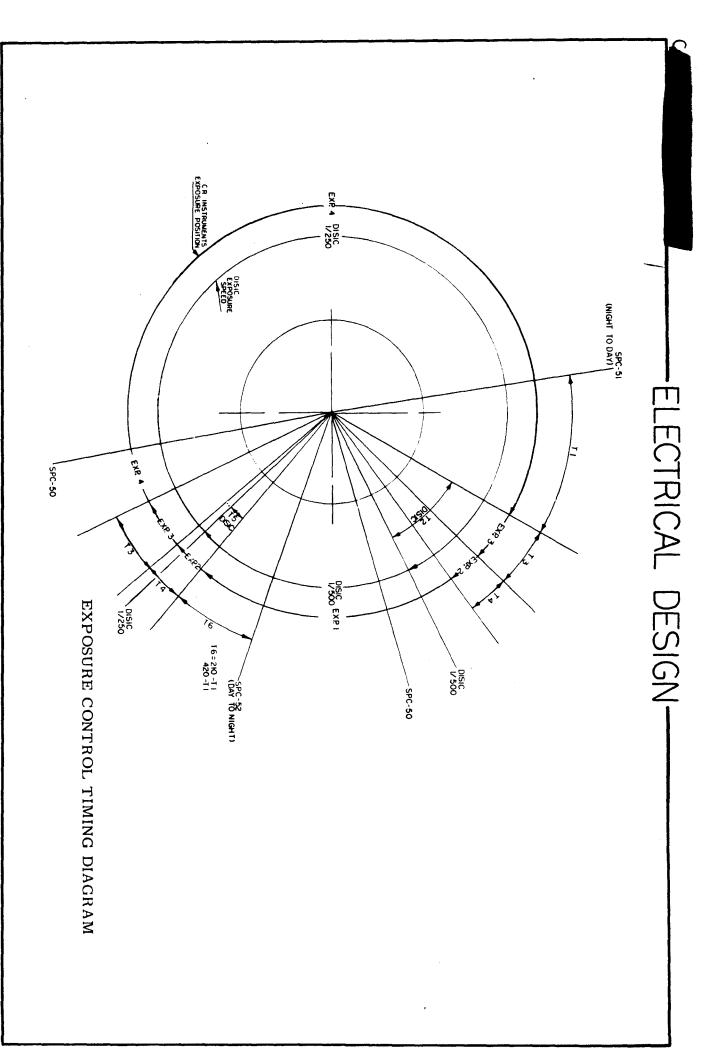
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J-3 DESIGN REVIEW

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EXPOSURE CONTROL COMMAND DESCRIPTION

- COMPENSATE FOR LAUNCH WINDOW UNCERTAINTIES AND/OR DELAYS. UHF-105 SELECTS ONE OF TWENTY AVAILABLE TIME DELAYS TO
- SLIT WIDTHS ON THE CR INSTRUMENTS AND INHIBITS NORMAL SEQUENCING. EXPOSURE CONTROL SYSTEM. UHF-101 PROVIDES INDIVIDUAL SELECTION OF ANY ONE OF THE FOUR IT ALSO PROVIDES INDIVIDUAL FAIL SAFE RESET OF EITHER INSTRUMENT'S
- UHF-102 BACKS UP UHF-101 FOR PLACING BOTH INSTRUMENTS IN THE EXPOSURE FAIL SAFE POSITION.
- SPC-51 INITIATES A NIGHT TO DAY SEQUENCE OF EXPOSURE SETTINGS FOR THE CR INSTRUMENTS AND THE DISIC.
- SPC-52 INITIATES A DAY TO NIGHT SEQUENCE OF EXPOSURE SETTINGS FOR THE CR INSTRUMENTS AND THE DISIC
- SPC-50 REMOVES POWER FROM TIMERS BETWEEN SEQUENCES.

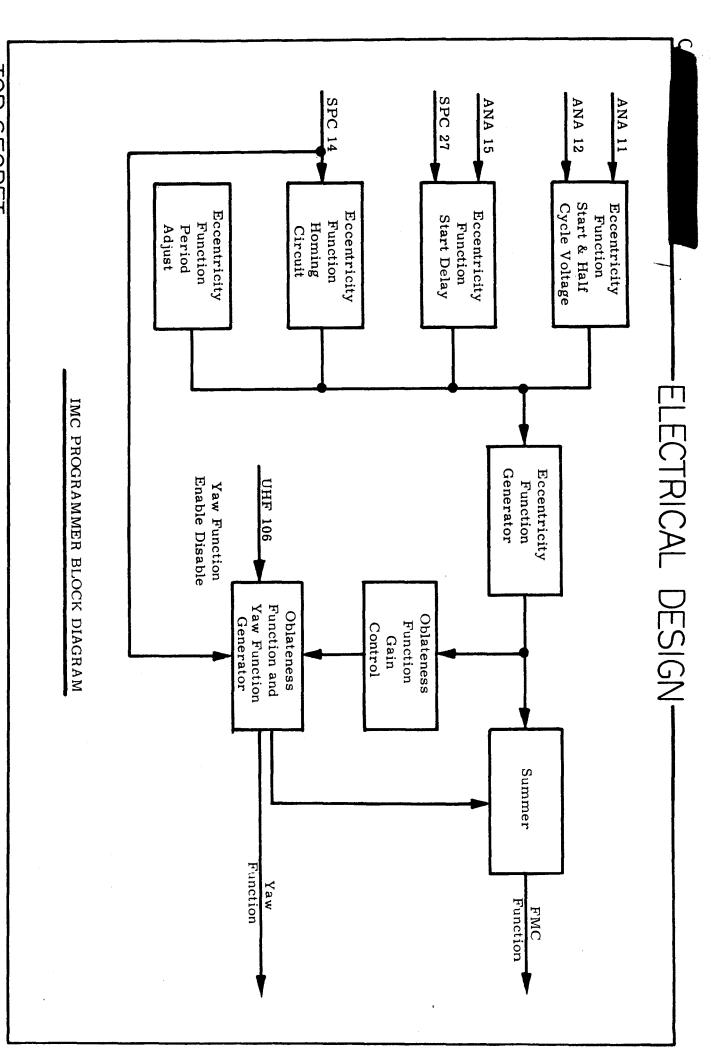


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J-3 DESIGN REVIEW

EXPOSURE CONTROL

- T₁ SELECTABLE IN-FLIGHT WITH UHF 105
- T_2 , T_3 , T_4 , T_5 , T_6 SELECTED PRE-FLIGHT
- EACH HAS A RANGE OF 20 TO 400 SECONDS IN 20 SECOND INCREMENTS
- EXP. 1 < EXP. 2 < EXP. 3 < EXP. 4



<u>N</u>0

IMC PROGRAMMER

- CYCLE & NODDING RATES. PROVIDES FMC FUNCTION TO CR INSTRUMENTS FOR CONTROL OF
- ERROR SIGNAL TO AGENA GUIDANCE. PROVIDES CROSS TRACK MOTION COMPENSATION IN FORM OF YAW
- & EARTH OBLATENESS PARAMETERS. FMC FUNCTION IS INSTANTANEOUS SUMMATION OF ORBIT ECCENTRICITY
- ECCENTRICITY FUNCTION HAS CAPABILITY FOR RTC SELECTION OF:
- 20 START VOLTAGE LEVELS
- 20 HALF CYCLE VOLTAGE LEVELS
- 20 START DELAY TIMES
- OBLATENESS FUNCTION & YAW FUNCTION USE COMMON DRIVE SYSTEM INITIATED BY SPC



IMC PROGRAMMER COMMAND DESCRIPTION

- ANA-15/UHF-125 SELECTS ONE OF TWENTY AVAILABLE TIME DELAYS TO FMC FUNCTION. RECEIPT OF A SPC #27 AND START OF THE ECCENTRICITY PORTION OF THE COMPENSATE FOR ORBIT UNCERTAINTIES. CONTROLS THE TIME BETWEEN
- START VOLTAGE LEVELS. ANA-11/UHF-121 SELECTS ONE OF TWENTY AVAILABLE FMC FUNCTION
- HALF CYCLE VOLTAGE LEVELS. ANA-12/UHF-122 SELECTS ONE OF TWENTY AVAILABLE FMC FUNCTION
- GENERATOR, UHF-106 ALTERNATELY ENABLES AND DISABLES OUTPUT OF YAW FUNCTION
- SPC-27 INITIATES INFLIGHT VARIABLE TIME DELAY SYSTEM WHICH STARTS ECCENTRICITY FUNCTION GENERATOR.
- SPC-14 INITIATES START OF OBLATENESS FUNCTION GENERATOR AND YAW FUNCTION GENERATOR. IT IS ALSO USED TO SLEW THE ECCENTRICITY FUNCTION GENERATOR TO HOME POSITION.

SILICON LIGHT PULSER (SLP) CONDITIONER

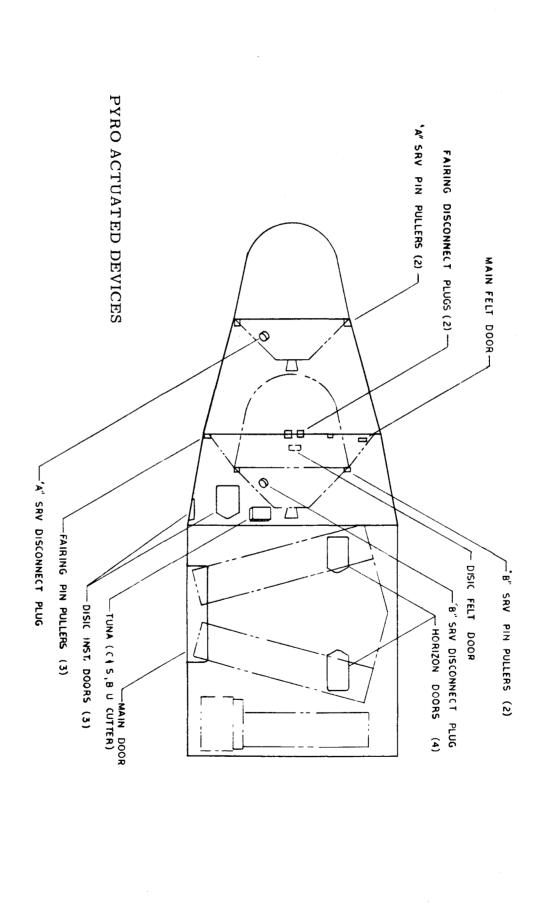
- SLP DATA HEADS IN CR. CONDITIONS & SEQUENCES CLOCK PARALLEL BINARY TIME WORD TO
- PARITY UTILIZING THREE COLUMNS OF DATA HEAD FOR EACH INSTRUMENT. PROVIDES 30 BIT INDEX, TIME WORD, COMPLEMENT OF TIME WORD &
- TO PROVIDE PROPER DOT INTENSITY FOR VARIOUS FILM ASA SPEEDS HAS CAPABILITY FOR TWO DISTINCT VARIABLE COLUMN DRIVE DURATIONS
- AND SEQUENCING. UTILIZES SOLID STATE MICRO LOGIC INTEGRATED CIRCUITS FOR GATING



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J-3 DESIGN REVIEW

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J-3 DESIGN REVIEW

PYRO SYSTEM

- ONE PLUG ON THE PAYLOAD/AGENA INTERFACE IS USED EXCLUSIVELY FOR SUPPLYING PYRO POWER AND PYRO RETURN TO THE PAYLOAD.
- COMMAND TO SQUIB EXCITATION. EACH PYRO DEVICE IS ACTIVATED BY INDEPENDENT CIRCUITS FROM
- A FUSISTOR PROTECTS EACH PYRO CIRCUIT.

PYRO EVENT & COMMAND DESCRIPTION

IN-FLIGHT RESET (ORBIT MODE SIGNAL BACKUP)

MAIN DOOR
DISIC TERRAIN DOOR
DISIC STELLAR #1 DOOR

DISIC STELLAR #1 DOOR (R.H. LOOKING FORWARD)

IN-FLIGHT RESET + 100 MILLISECONDS (ORBIT MODE SIG. BACKUP)

LEFT HORIZON DOOR #1
RIGHT HORIZON DOOR #1
LEFT HORIZON DOOR #2
RIGHT HORIZON DOOR #2
DISIC STELLAR #2 DOOR (L. H. LOOKING FORWARD)

KZ 38 (A TO B TRANSFER

MAIN SEAL ASSEMBLY (LIGHT SEAL)
MAIN WATER SEAL

KZ 39 CUT & SPLICE (DISIC)

"A" ARM

"A" T/M BATTERY (SRV) "A" MAIN WATER SEAL "A" RECOVERY BATTERY (SRV) MAIN SEAL ASSEMBLY DISIC CUT & SPLICE (IF NOT PREVIOUSLY ACTUATED BY KZ 39) IF NOT PREVIOUSLY ACTUATED BY KZ 38)

"A" TRANSFER

DISIC SEAL ASSEMBLY (LIGHT SEAL)
"A" THRUST CONE THERMAL BATTERIES (SRV)
"A" DISIC WATER SEAL (SRV)

"A" DISCONNECT

SPIN OFF DISCONNECT P28A



PYRO EVENT & COMMAND DESCRIPTION (CONT'D)

"A" SEPARATE

"A" SRV PIN PULLERS

NOTE: TEN SECONDS AFTER REMOVAL OF THE SEPARATE SIGNAL THE RECOVERY SWITCH-OVER CIRCUIT TRANSFERS THE RECOVERY

COMMANDS TO THE "B" SRV.

"B" ARM

FAIRING SPIN OFF DISCONNECT PLUGS "B" T/M BATTERY (SRV)

"B" RECOVERY PROGRAMMER BATTERY (SRV)

BACKUP CUTTER (DISIC)

"B" ARM + 100 MILLISECONDS

FAIRING SEPARATE PIN PULLERS
"B" MAIN WATER SEAL

"B" TRANSFER

"B" DISIC WATER SEAL (SRV)

"B" THRUST CONE THERMAL BATTERIES (SRV)

"B" DISCONNECT

SPIN OFF DISCONNECT P28B

"B" SEPARATE

"B" SRV PIN PULLERS

POWER ESTIMATE

- MAXIMUM AVERAGE POWER (ASSUMING ALL PAYLOAD SUBSYSTEMS ARE OPERATED AT THEIR AVERAGE CURRENT SIMULTANEOUSLY)
- UNREGULATED 24 VDC

(APPROX.	TOTAL	A/P COMMAND SYSTEM	CLOCK CONDITIONER	CLOCK	IMC SYSTEM	T/M SYSTEM	DISIC	CR INSTRUMENTS	
30.0 AMPS)	27.35	1.00	0.50	0.35	1.00	1.00	3.50	20.00	AMPS

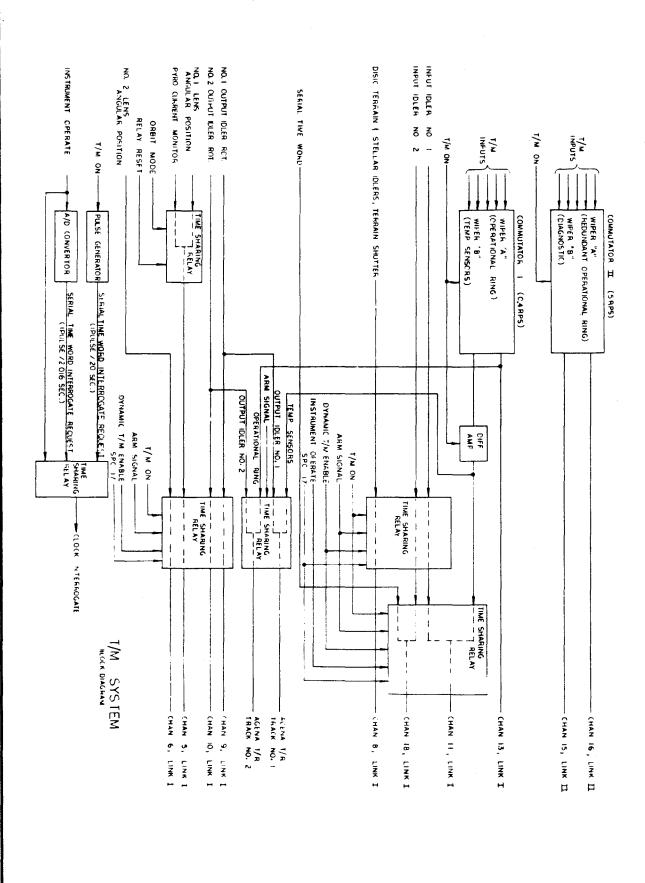
115 V 400

		CR INSTRUMENTS IMC SYSTEM
(APPROX.	TOTAL	
0.6 AMPS)	0.57	0.50



POWER DISTRIBUTION PLAN

- A POWER BUSS IS SUPPLIED FOR EACH POWER SOURCE.
- FORWARD END OF THE PAYLOAD. A HIGH CAPACITY POWER BUSS RUNS FROM THE AGENA INTERFACE TO THE
- IN THE PAYLOAD. FEEDER CABLING IS PROVIDED FROM THE MAIN BUSS TO EACH EQUIPMENT
- A COMPATIBLE POWER RETURN SYSTEM IS USED THROUGHOUT THE SYSTEM.
- POWER SUPPLIED FROM THE AGENA INCLUDES + 24 VDC UNREGULATED AND
- 5 V DC-DC CONVERTERS ARE USED FOR TELEMETRY EXCITATION & SLP CONDITIONER.
- A + 15 VDC CONVERTER IS USED IN THE IMC PROGRAMMER



J-3 DESIGN REVIEW

TELEMETRY SYSTEM

PROVIDES INSTRUMENT OPERATIONAL DATA

- TEMPERATURE
- OPERATIONAL MODES
- COMMAND MODES
- CYCLES OF OPERATION
- PAYLOAD CONSUMPTION
- CLOCK SERIAL TIME WORD

T/M CONTROL AND COMMAND DESCRIPTION

- T/M ON (AGENA INTERFACE)
- ENERGIZES ALL T/M CIRCUITS
- RESETS CONTINUOUS T/M ENABLE
- T/M DATA ENABLE
- ENERGIZES ALL T/M CIRCUITS
- INSTRUMENT OPERATE
- ENERGIZES ALL T/M CIRCUITS
- ENERGIZES RECOVERY TAPE RECORDER SYSTEM
- DYNAMIC T/M ENABLE (SPC 17, KZ 38 & 39, ARM SIG.)
- ENABLES CONTINUOUS T/M IF T/M ON IS PRESENT



T/M INSTRUMENTATION LIST

SIX CONTINUOUS CHANNELS

- TERRAIN IDLER, TERRAIN SHUTTER & STELLAR IDLER
- PAN #1 OUTPUT IDLER ROTATION, 99/101 CLUTCH COMMAND
- CURRENT MONITOR DURING ASCENT PAN #1 LENS ANGULAR POSITION, CENTER OF FORMAT COMMAND, PYRO
- PAN #2 OUTPUT IDLER ROTATION, 99/101 CLUTCH COMMAND
- PAN #2 LENS ANGULAR POSITION, CENTER OF FORMAT COMMAND
- CLOCK SERIAL WORD (TIME SHARED WITH PAN #2 INPUT IDLER)

FOUR COMMUTATED CHANNELS

- TEMPERATURE MEASUREMENTS
- TIME SHARED WITH PAN #1 INPUT IDLER
- 0.4 RPS X 60 POINT COMMUTATOR RING B
- 22 TEMP. SENSORS ON STRUCTURE & SKIN
- 16 INSTRUMENT TEMP. SENSORS
- 4 SRV TEMP. SENSORS
- 2 DISIC TEMP. SENSORS
- 2 BLAST SHIELD TEMP. SENSORS
- 1 TEMP. SENSOR ON IMC PROGRAMMER
- 1 PMU TEMP. SENSOR
- OPERATIONAL CONTROL
- 0.4 RPS X 60 POINT COMMUTATOR RING A
- RTC & SPC STATUS MONITORS
- PAN #1 & #2 CYCLE COUNTER MONITOR



T/M INSTRUMENTATION LIST (CONT'D)

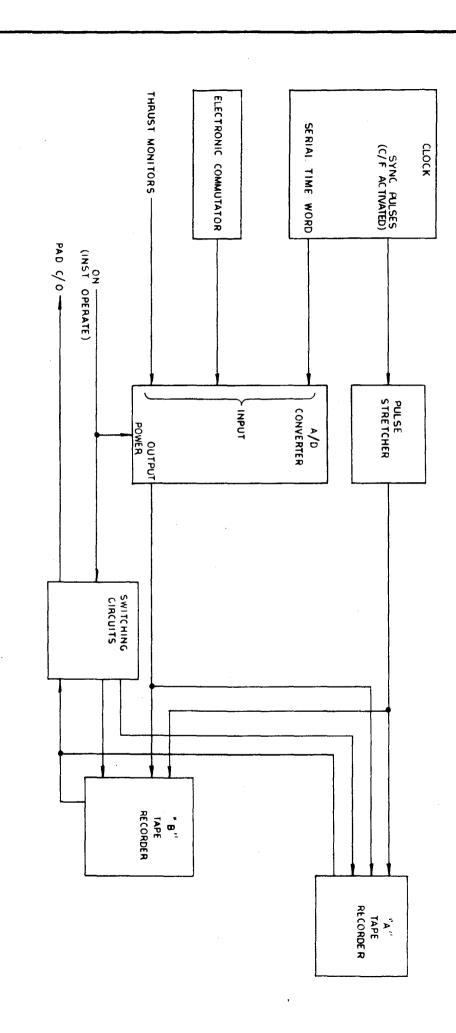
- TERRAIN CYCLE COUNTER
- SRV SEPARATIONS
- FAIRING SEPARATIONS
- PAN #1 & #2 FILM CHANGE DETECTOR
- SRV RECOVERY BATTERY MONITOR
- STELLAR & TERRAIN TAKE-UP DIAMETER
- PAN #1 & PAN #2 TAKE-UP DIAMETER
- REDUNDANT OPERATIONAL CONTROL
- 5.0 RPS X 60 POINT COMMUTATOR RING A
- DIAGNOSTIC MEASUREMENTS

5.0 RPS X 60 POINT COMMUTATOR RING B

METERING MONITORS

WATER SEALS

- DOOR SEPARATIONS
- PLATEN POSITIONS
- YAW POSITION
- CURRENT MONITORS
- VOLTAGE MONITORS



TAPE RECORDER

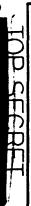


TAPE RECORDER SYSTEM

- OPERATED FROM INSTRUMENT OPERATE COMMAND
- RECORDS DATA FROM TWO MAIN SOURCES:
- DIGITAL DATA #1
- OUTPUT OF AN A/D MULTIPLEXER
- A/D MULTIPLEXER INPUTS:
- a/D Wolling DESSETT INTO 10.
- SERIAL TIME WORD FROM CLOCK (2 CHANNELS)

AGENA THRUST MONITOR DATA (12 CHANNELS)

- TWO ELECTRONIC COMMUTATOR OUTPUTS (2 CHANNELS)
- DIGITAL DATA #2
- TWO PULSE STRETCHER OUTPUTS
- PULSE STRETCHERS DRIVEN BY CLOCK SYNC OUTPUTS
- CLOCK SYN. OUTPUTS DRIVEN BY MAIN INSTRUMENTS CENTER OF FORMAT PULSES
- DIGITAL CLOCK SYNC
- TIME BASE FROM A/D CONVERTER
- 10 KC PULSE RECORDED ON BOTH CHANNELS



ENGINEERING ANALYSIS

IN THE FOLLOWING DOCUMENTS: FAILURE MODES ARE DISCUSSED IN DETAIL FOR EACH OF THE MAJOR SUBSYSTEMS

•	ENGINEERING ANALYSIS REPORT - T/M SUBSYSTEM	T9-6-041
•	ENGINEERING ANALYSIS REPORT - EXPOSURE CONTROL SUBSYSTEM	T9-6-037
•	ENGINEERING ANALYSIS REPORT - COMMAND SUBSYSTEM	T9-6-042
•	ENGINEERING ANALYSIS REPORT - J-3 PYROTECHNIC SUBSYSTEM	T9-6-036
•	ENGINEERING ANALYSIS REPORT - PMU SUBSYSTEM	T9-6-039
•	ENGINEERING ANALYSIS REPORT - DIGITAL TAPE RECORDER SUBSYSTEM	T9-6-038
•	ENGINEERING ANALYSIS REPORT - IMC SUBSYSTEM	T9-6-034
•	ENGINEERING ANALYSIS REPORT - SLP DATA CONDITIONER	T9-6-035
•	ENGINEERING ANALYSIS REPORT - POWER SUBSYSTEM	T9-6-040



BIBLIOGRAPHY

A/P INTERFACE DOCUMENTS

ELECTRICAL INTERFACE SPECIFICATION FOR THE J-3 CONSTANT ROTATOR SYSTEM

T3-5-020 ELECTRICAL INTERFACE SPECIFICATION FOR THE J-3 SRV SYSTEM

T3-5-021 ELECTRICAL INTERFACE SPECIFICATION FOR THE J2-J3 DISIC SYSTEM

ELECTRICAL INTERFACE SPECIFICATION FOR ADVANCED PROJECTS PAYLOAD AND AGENA ORBITAL VEHICLE

A/P DESIGN CONTROL SPECIFICATIONS

T3-5-028 SYSTEM ADVANCED PROJECTS DESIGN CONTROL SPECIFICATION FOR J-3

J-3 DESIGN REVIEW

NO.