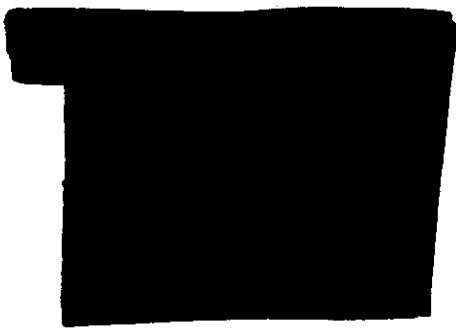




LAUNCH REQUIREMENTS-VEHICLE 1173/J-15

Attached is the launch requirements and limitations for Vehicle 1173/J-15



Declassified and Released by the NRO  
In Accordance with E. O. 12958  
on NOV 26 1997



LAUNCH REQUIREMENTS - VEHICLE 1173/N-15

1.0 SCOPE

1.1 General

The following requirements govern the condition under which the vehicle will be launched with the A/P Payload System. Any deviation from the prescribed limits shall be cause for hold. Any status changes must be reported to System Integration immediately after occurrence for evaluation. All discrepancies and deviations must be corrected prior to resumption of vehicle launch count-down.

2.0 PAYLOAD INTERNAL TEMPERATURE

$65 \pm 10^{\circ}$  F from mating to T-4 hours.

$65 \pm 5^{\circ}$  F from T-4 hours thru launch.

Temperature on Payload System shall be monitored and recorded every one-half hour from mating to launch.

3.0 RELATIVE HUMIDITY

50% or less at all times.

4.0 H<sub>2</sub> PRESSURE

The H<sub>2</sub> must be connected and flowing when the system is on the launch pad. Maximum off time is one-half hour per day.

LAUNCH REQUIREMENTS - VEHICLE 1173/J-15 (Cont'd)

5.0 SRV TRANSMISSION FREQUENCY

Recovery T/M                    223.2 ± 0.1 MC

Recovery Beacon                235.0 ± 0.1 MC

6.0 POWER

Power must be applied to the payload interface whenever the payload is raised or lowered.

7.0 THERMAL BLANKET

The thermal blanket must remain on the payload from mating until launch.

8.0 PRIMARY TLM READOUTS

Payload checkout (must be verified thru vehicle TLM).

8.1 Lens Rotation, Horizon idler, and center of format Instrument No. 1 channel 09 Link I. No backup channel is available.

8.2 Lens Rotation, Horizon idler, and center of format Instrument No. 2 channel 10 Link I. No backup channel is available.

8.3 Ring A Commutator (.4 x 60' channel 13 Link I endpoints listed below. The primary commutated points must be verified during payload confidence after mating and payload checkout task.

LAUNCH REQUIREMENTS - VEHICLE 1179/J-13(Cont'd)

MONITOR	VERIFICATION					
	PRIMARY			ALTERNATE		
Function	CH-LK-PT	Volts	Tol.	CH-LK-PT	Volts	Tol.
Inst. 1 Cycle Count 1	13-1-22	.55 step	± .15	13-1-25	4.05 or greater	4.05 or greater
Inst. 1 Cycle Count 10	13-1-23	.55 step	± .15	13-1-25	"	"
Inst. 1 Cycle Count 100	13-1-24	.55 step	± .15	13-1-25	"	"
Film Footage Pot Inst. 1	13-1-25	4.05 or greater	4.05 or greater	8-2-56	"	"
Inst. 2 Cycle Count 1	13-1-27	.55 step	± .15	13-1-31	9 or less	9 or less
Inst. 2 Cycle Count 10	13-1-23	.55 step	± .15	13-1-31	"	"
Inst. 2 Cycle Count 100	13-1-23	.55 step	± .15	13-1-31	"	"
Calibrate Plus	13-1-30	5	--	13-1-07	5	--
Calibrate Plus	13-1-30	5	--	13-1-14	5	--
Inst. 2 Film Footage Pot	13-1-31	9 or less	9 or less	8-2-47	9 or less	9 or less
Inst. 1 Door Eject	13-1-33	1.3	± .2			
Inst. 2 Door Eject	13-1-43	1.3	± .2	"	--	--
Film Door Closure	13-1-47	4.7	± .2	"	--	--
Fairing Separation	13-1-49	1.3	± .2	"	--	--
Continuity Loop SRV-1	13-1-51	5.38	± .2	"	--	--
Separation Monitor-SRV's	13-1-52	0.5	+ .1 - .1	16-1-25	0.5	+ .1 - .1
Recovery Battery SRV-1	13-1-53	0	± .2	None	--	--
Continuity Loop SRV-2	13-1-54	5.38	± .2	"	--	--
Recovery Battery SRV-2	13-1-55	0	± .2	"	--	--
Calibrate Zero	13-1-57	0	--	13-1-04	0	--
Calibrate Zero	13-1-57	0	--	13-1-11	0	--
Calibrate Zero	13-1-57	0	--	13-1-17	0	--
Sync.	13-1-58	5.5	± .2	13-1-59	5.5	± .2
Sync.	13-1-58	5.5	± .2	13-1-60	5.5	± .2
Booster Pressure	13-1-37	3.0 or greater		11-1-11	3.0 or greater	--

All Command Selector points as listed per launch requirements list in Appendix - I.

LAUNCH REQUIREMENTS - VEHICLE 1173/7-15 (Cont'd)

9.0 HOMING OF CAMERA STOVES

9.1 Both camera stoves must be properly homed prior to terminal count.

10.0 LAUNCH REQUIREMENTS COMMAND SETTINGS

10.1 All stepping switches must be positioned in accordance with the Command Settings List prior to terminal count. (See Appendix - I)

11.0 FILM CONSUMPTION PRIOR TO LAUNCH

11.1 Minimum film consumption prior to launch shall be 100 cycles on both instruments.

11.2 Should the Payload System stay in a loaded condition for an extended period, the System shall be operated 10 cycles per applicable procedure every other day.

12.0 RESPONSIBILITY

12.1 It shall be the responsibility of the senior A/P Payload Engineer to ensure the implementation of the restrictions and requirements listed herein. In addition, he is charged with the responsibility of supplying the following system information immediately prior to launch to Flight Operations and Computer Services by telephone.

LAUNCH REQUIREMENTS - VEHICLE 1173/-15 (Continued)

12.1.1 Final System Weight, in Lbs. \_\_\_\_\_

12.1.2 Cycle counter film footage pot and leader length readings for both instruments, in accordance with J1210 Para. 18, Item 5.4.1.

	<u>MASTER</u>	<u>SLAVE</u>
a. Cycle Counter	_____	_____
b. Film Footage Pot in Volts	_____	_____
c. Leader Lengths in Feet	_____	_____

12.1.3 Cycle counter, film footage pot and leader length readings for both instruments, in accordance with J1210, Para. 18, Item 5.4.10.

	<u>MASTER</u>	<u>SLAVE</u>
a. Total length of leader from "A" bucket hub to flight payload splice.	_____ Ft.	_____ Ft.
b. Cycle Counter Readings	_____ Cts.	_____ Cts.
c. Take-up Pot Voltage	_____ V	_____ V

12.1.4 Cycle counter and film footage pot readings for both instruments, at launch, in units.

	<u>Cycle Counter</u>	<u>Film Footage Pot</u>
a. Master	_____	_____
b. Slave	_____	_____

12.1.5 Clock error, static run, in microseconds: \_\_\_\_\_

12.1.6 Length of off-spool of both instruments, including control strips, in feet:

a. Master	_____ Ft.
b. Slave	_____ Ft.

12.1.7 Length of off-spool of both S/I Instruments, in feet:

a. S/I A Stellar	_____ Ft.
Index	_____ Ft.
b. S/I B Stellar	_____ Ft.
Index	_____ Ft.



12.1.8 DOOR MONITOR CALIBRATIONS:

<u>CONDITION</u>	<u>INST. NO. 1</u>	<u>T/M VOLTAGE</u>	
		<u>INST. NO. 2</u>	<u>FILM DOOR</u>
DOOR ON (CLOSED)	_____	_____	_____
DOOR OFF (OPEN)	_____	_____	_____

12.1.9 SEPARATION MONITOR CALIBRATIONS:

<u>CONDITION</u>	<u>SRV-A&amp;B</u>	<u>SRV A (ONLY)</u>	<u>T/M VOLTAGE</u>	
			<u>SRV E*</u>	<u>FAIRING</u>
MATED	_____	_____	_____	_____
SEPARATED	_____	_____	_____	_____

\* = AFTER FIRST RECOVERY

12.1.10 CONTINUITY LOOP MONITOR CALIBRATIONS:

<u>CONDITION</u>	<u>T/M VOLTAGE</u>			<u>SRV - A</u>	<u>SRV - B</u>
	<u>CONTINUITY LOOP</u>	<u>S/I SEAL</u>	<u>MAIN SEAL</u>		
CLOSED	OPEN	OPEN	1.84 ± 5%	_____	_____
CLOSED	OPEN	CLOSED	4.30 ± 5%	_____	_____
CLOSED	CLOSED	CLOSED	5.38 ± 5%	_____	_____
OPEN	CLOSED	CLOSED	4.57 ± 5%	_____	_____
OPEN	CLOSED	OPEN	2.16 ± 5%	_____	_____
OPEN	OPEN	OPEN	.75 ± 5%	_____	_____
OPEN	OPEN	CLOSED	3.42 ± 5%	_____	_____
CLOSED	CLOSED	OPEN	3.12 ± 5%	_____	_____



LAUNCH REQUIREMENTS: The following settings/requirements are specified for the  
J-15 1173 payload. Date of issue (R-20) (R-7).

PANORAMIC LENS SETTINGS:

	<u>Instrument No. 1 (Master)</u>	<u>Instrument No. 2 (Slave)</u>
Slit Dimensions	<u>.200 X 2.278</u>	<u>.200 X 2.278</u>
Filter Type	<u>WRATTEN 21</u>	<u>WRATTEN 21</u>

NOTE: SLIT LENGTH AND WIDTH MUST BE MEASURED AT TIME OF INSTALLATION

Measured by \_\_\_\_\_

Verified by \_\_\_\_\_

HORIZON OPTICS SETTINGS:

	<u>Instrument No. 1 (Master)</u>	<u>Instrument No. 2 (Slave)</u>
<b>Supply Horizons:</b>		
Aperture	<u>F6.8</u>	<u>F8.0</u>
Speed	<u>1/100 Sec.</u>	<u>1/100 Sec.</u>
Filter	<u>WRATTEN 25</u>	<u>WRATTEN 25</u>
<b>Take-up Horizons:</b>		
Aperture	<u>F8.0</u>	<u>F6.8</u>
Speed	<u>1/100 Sec.</u>	<u>1/100 Sec.</u>
Filter	<u>WRATTEN 25</u>	<u>WRATTEN 25</u>

STELLAR INDEX OPTICS SETTINGS:

	<u>Stellar Index A</u>	<u>Stellar Index B</u>
<b>Stellar Lens:</b>		
Aperture	<u>F1.8</u>	<u>F1.8</u>
Speed	<u>2 Sec.</u>	<u>2 Sec.</u>
Filter	<u>None</u>	<u>None</u>

LAUNCH REQUIREMENTS: CONTINUED Date of Issue \_\_\_\_\_ (R-20) \_\_\_\_\_ (R-7)  
STELLAR INDEX OPTICS SETTINGS:

	<u>Stellar Index A</u>	<u>Stellar Index B</u>
Index Lens:		
Aperture	<u>F4.5</u>	<u>F4.5</u>
Speed	<u>1/500 Sec.</u>	<u>1/500 Sec.</u>
Filter	<u>WRATTEN 21</u>	<u>WRATTEN 21</u>

FILM NOMENCLATURE:

Panoramic Instruments:

Instrument No. 1 (Master)      Instrument No. 2 (Slave)

Primary:

Type	<u>7J40 - 16000</u>	<u>7J40 - 16000</u>
Emul. Data	<u>77-7-9-4</u>	<u>77-7-9-4</u>
Wt. & Spool No.	<u>89.0-79.2-48-102T</u>	<u>88.7-79.1-48-115B</u>
Box No.	<u>23</u>	<u>23</u>

Secondary:

Type	<u>7J40-16000</u>	<u>7J40-16000</u>
Emul. Data	<u>77-7-9-4</u>	<u>77-7-9-4</u>
Wt. & Spool No.	<u>89.4-79.4-48-3T</u>	<u>88.5-79.1-48-69B</u>
Box No.	<u>2</u>	<u>2</u>

PANORAMIC OFFSPOOLING REQUIREMENTS:

	<u>Master</u>	<u>Slave</u>
Original Length - Ft.	<u>16000</u>	<u>16000</u>
Length to Offspool - Ft.	<u>None + 10</u>	<u>None + 10</u>
Length to Load - Ft.	<u>16000</u>	<u>16000</u>

TITLE

LAUNCH REQUIREMENTS  
- CAMERA SYSTEM -

LOCKHEED MISSILES & SPACE COMPANY

TEST PROCEDURE  
APPENDIX - I  
PAYLOAD LAUNCH  
REQUIREMENTS LETTER

SHEET OF SHEETS

LAUNCH REQUIREMENTS: CONTINUED Date of Issue \_\_\_\_\_ (R-20) \_\_\_\_\_ (R-7)

FILM NOMENCLATURE: CONTINUED

Stellar Index:

	<u>Stellar Index A</u>		<u>Stellar Index B</u>	
	<u>Stellar</u>	<u>Index</u>	<u>Stellar</u>	<u>Index</u>
Primary:				
Type	<u>3J-34-75</u>	<u>7J-33-135</u>	<u>3J-34-75</u>	<u>7J-33-135</u>
Emul. Date	<u>44-30-7-4</u>	<u>31-4-7-4</u>	<u>44-30-7-4</u>	<u>44-30-7-4</u>
Secondary:				
Type	<u>3J-34-75</u>	<u>7J-33-135</u>	<u>3J-34-75</u>	<u>7J-33-135</u>
Emul. Date	<u>44-30-7-4</u>	<u>31-4-7-4</u>	<u>44-30-7-4</u>	<u>44-30-7-4</u>

STELLAR INDEX FILM OFFSPOOLING REQUIREMENTS

	<u>Stellar A</u>		<u>Stellar B</u>	
	<u>Stellar</u>	<u>Index</u>	<u>Stellar</u>	<u>Index</u>
Original Length - Ft.	<u>75</u>	<u>135</u>	<u>75</u>	<u>135</u>
Length to Offspool - Ft.	<u>30</u> <sup>+1</sup> <sub>-0</sub>	<u>45</u> <sup>+1</sup> <sub>-0</sub>	<u>30</u> <sup>+1</sup> <sub>-0</sub>	<u>45</u> <sup>+1</sup> <sub>-0</sub>
Length to Load - Ft.	<u>45</u>	<u>90</u>	<u>45</u>	<u>90</u>

CYCLE RATIO STELLAR INDEX TO PANORAMIC 1 to 7

STELLAR BAFFLE TYPE:

<u>Stellar A</u>	<u>Stellar B</u>
<u>11.500</u>	<u>11.500</u>
_____	_____
_____	_____

LAUNCH WINDOW: 21:30 - 22:30 Z

