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CORONA J

PERFORMANCE EVALUATION REPORT

MISSION 1013-1 and 1013-2

FTV 1173; J-15

22 December 1965

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FOREWORD

This report details the performance of the payload system during the operational phase of the Program [REDACTED] Flight Test Vehicle 1173.

Lockheed Missiles and Space Company has the responsibility for evaluating payload performance under the Systems Integration and "J" System contracts.

This document is the final payload test and performance evaluation report for Missions 1013-1 and 1013-2 which was launched on 2 November 1964.

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INTRODUCTION

This report presents the final performance evaluation of Missions 1013-1 and 1013-2 of the Corona Program. The purpose of this report is to define the performance characteristics of the J-15 payload system, to identify the source of in-flight anomalies and recommend the appropriate corrective action.

The performance evaluation was jointly conducted by representatives of Lockheed Missiles and Space Company (LMSC) and ITEK at the facilities of NPIC and AFSPPL. The off-line evaluation using Corona engineering photography acquired over the United States was performed at the individual contractors plants.

The quantitative data used for this report is obtained from government organizations. The diffuse density data, visual RES values and MTF/AIM resolution are produced by AFSPPL. The vehicle attitude error values, frame correlation times are made at NPIC who also supply the Processing Summary and MTF/AIM resolution reports [REDACTED]

Computer programs developed by A/P are utilized to calculate and plot the frequency distribution of the various contributors to image smear to permit analysis and correlation of the conditions of photography to the information content and quality of the acquired pictures. Computer analysis of the exposure, processing and illumination data provides the necessary data to analyze the exposure criteria selected for the mission.

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SECTION 1

SYSTEM PERFORMANCE

A. MISSION OBJECTIVES

The payload section of Mission 1013, placed into orbit by Flight Test Vehicle #1173 and SLV-2A booster #420, consisted of two panoramic cameras, two Stellar-Index cameras, two Mark 5A recovery capsules and a space structure to enclose the cameras and provide mounting surfaces for all equipments. Figure 1-1 presents an inboard profile of the J-15 payload system. This Corona "J" system is designed to acquire search and reconnaissance photography of selected areas of the earth from orbital altitudes. The planned mission was two, four day photographic periods with no deactive period.

B. MISSION DESCRIPTION

The payload was launched from Vandenberg Air Force Base (VAFB) at 2130:20 Z (1:30:20 PST) on 2 November 1964. Ascent and injection were normal and the achieved orbit within nominal tolerances. Tracking and command support was effected by the Air Force Satellite Control Facility consisting of tracking and command stations at [REDACTED]

[REDACTED] under central control of the Satellite Test Center at Sunnyvale, California. Mission 1013-1 consisted of four days operation and was completed by air recovery on 6 November 1964. Telemetry indicated that both panoramic cameras had failed during pass D52. Mission 1013-2 was completed on pass D81, 6 November 1964, with an air recovery.

The comparison of the planned and actual orbit parameters is tabulated as follows:

ORBITAL PARAMETERS

<u>Parameter</u>	<u>Predicted</u>	<u>Orbit 2 Actuals</u>
Period (Min.)	90.87	90.86
Perigee (N. M.)	99.8	100.0
Apogee (N. M.)	244.8	246.0
Inclination (Deg.)	80.0	80.0
Perigee Latitude (Deg. N.)	18.0	25.0
Eccentricity	0.0201	0.0202

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SCHEMATIC I:BOARD PROFILE - CORONA J SYSTEM

MISSION 1013

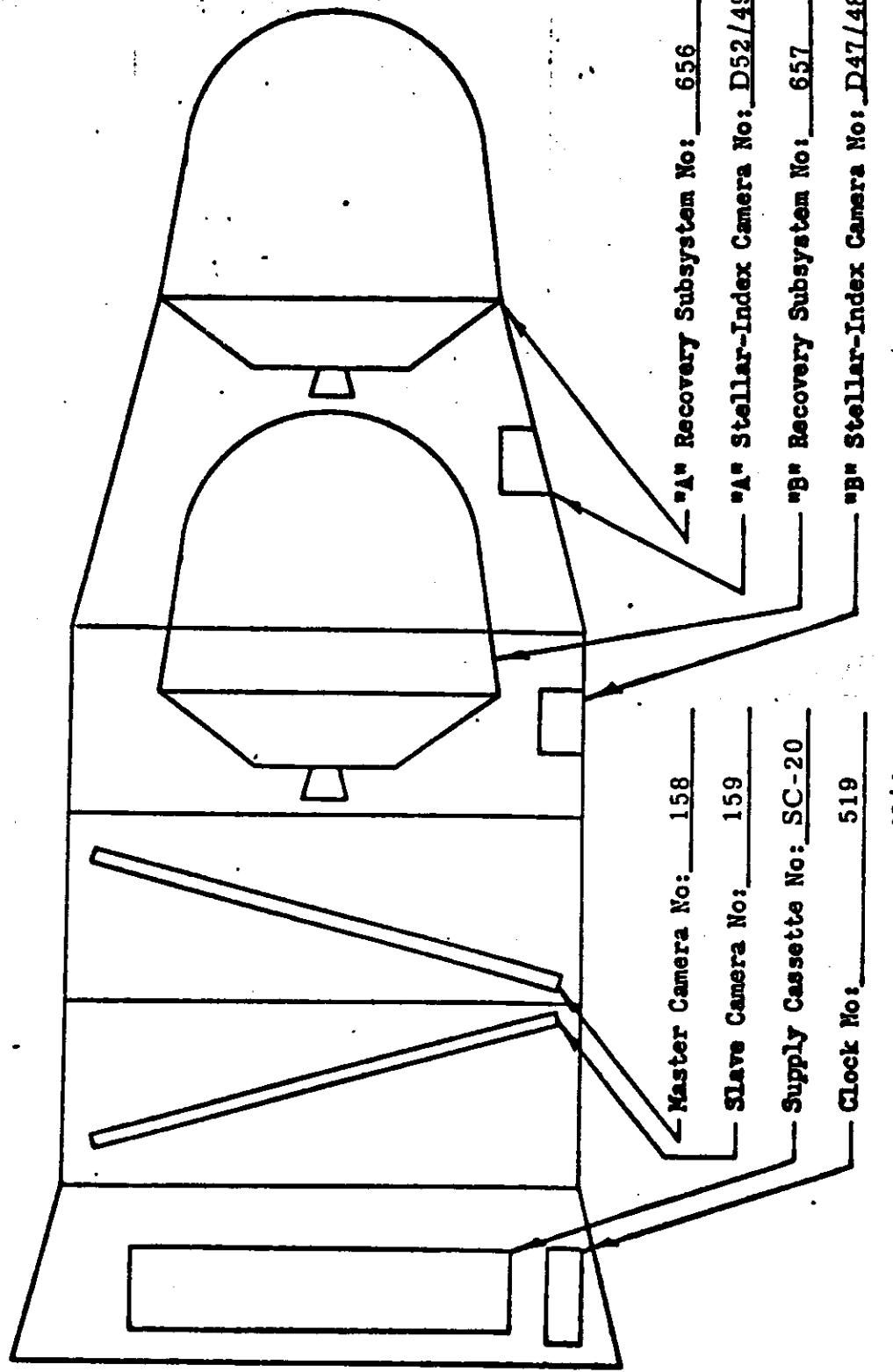


FIGURE 1-1

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Sgt. John J. Farley

C. PANORAMIC CAMERAS

The Master and Slave cameras operated through pass D51. Both cameras failed near the start of a programmed operation of pass D52. The quality and information content of the Master camera photography was very good up to the point of failure. The Slave camera photography was comparable to the Master camera up to frame 90 of pass D24. Subsequent photography was out of focus across half of the format parallel and adjacent to the time track edge. This soft area began at a manufacturers splice and continued up to the point of failure.

D. STELLAR-INDEX CAMERAS

The Stellar-Index cameras of both Mission 1013-1 and 1013-2 operated properly during their respective missions. Both cameras produced excellent stellar and terrain photography.

E. OTHER SUB-SYSTEMS

An erroneous signal from the command programmer caused both the Master and Slave cameras to acquire approximately 400 unprogrammed photographic frames during pass A01 and D01. The clock, instrumentation, pressure make-up and thermal control sub-systems performed satisfactorily through both missions.

F. CONCLUSIONS

Missions 1013-1 and 1013-2 did not fully achieve the objective of acquiring high quality search and reconnaissance photography from orbital altitudes.

G. RECOMMENDATIONS

The evaluation and analysis of the data produced by the missions has resulted in the following recommendations:

1. Visually inspect all optical surfaces following delivery to A/P, prior to and after environmental testing.
2. Review the splicing materials and techniques used by the film manufacturer.

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SECTION 2

PRE-FLIGHT SYSTEMS TESTS

A. ENVIRONMENTAL TESTING

1. Test Objective

As a standard procedure, the J payload systems are subject to thermal/altitude environmental testing which simulates orbital environment. One of the purposes of this test is to demonstrate the system susceptibility to corona discharge. Such discharge fogs the film thus degrading the operational photography.

2. Test Summary

The J-15 Payload System was tested for pressure, thermal and Corona effects in the TASC Chamber starting 14 August 1964. The test consisted of three (3) days active operation with the "A" bucket; one (1) day of de-active soak and two (2) days of active operation with the "B" bucket.

The J-15 system contained a flight pressure make-up system; however, this system failed toward the end of the first day, "A" bucket. Corona markings were reported to be acceptable for flight, without pressure make-up.

Several abnormalities in system performance were evident during the test and are summarized below:

- 1) Aforementioned pressure make-up failure.
- 2) Power turned off during "A" Mission caused the clock to experience a re-start.
- 3) V/h programmer received nine (9) of eleven (11) transmitted commands during "B" Mission, second day - orbit #13.
- 4) Fogging light problem around pressure make-up failure time.
- 5) Uneven shutdown on slave instrument on orbits 3 and 4, second day - "A" Mission.

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J-15 TASC TEST

INTERNAL CAVERA PRESSURE

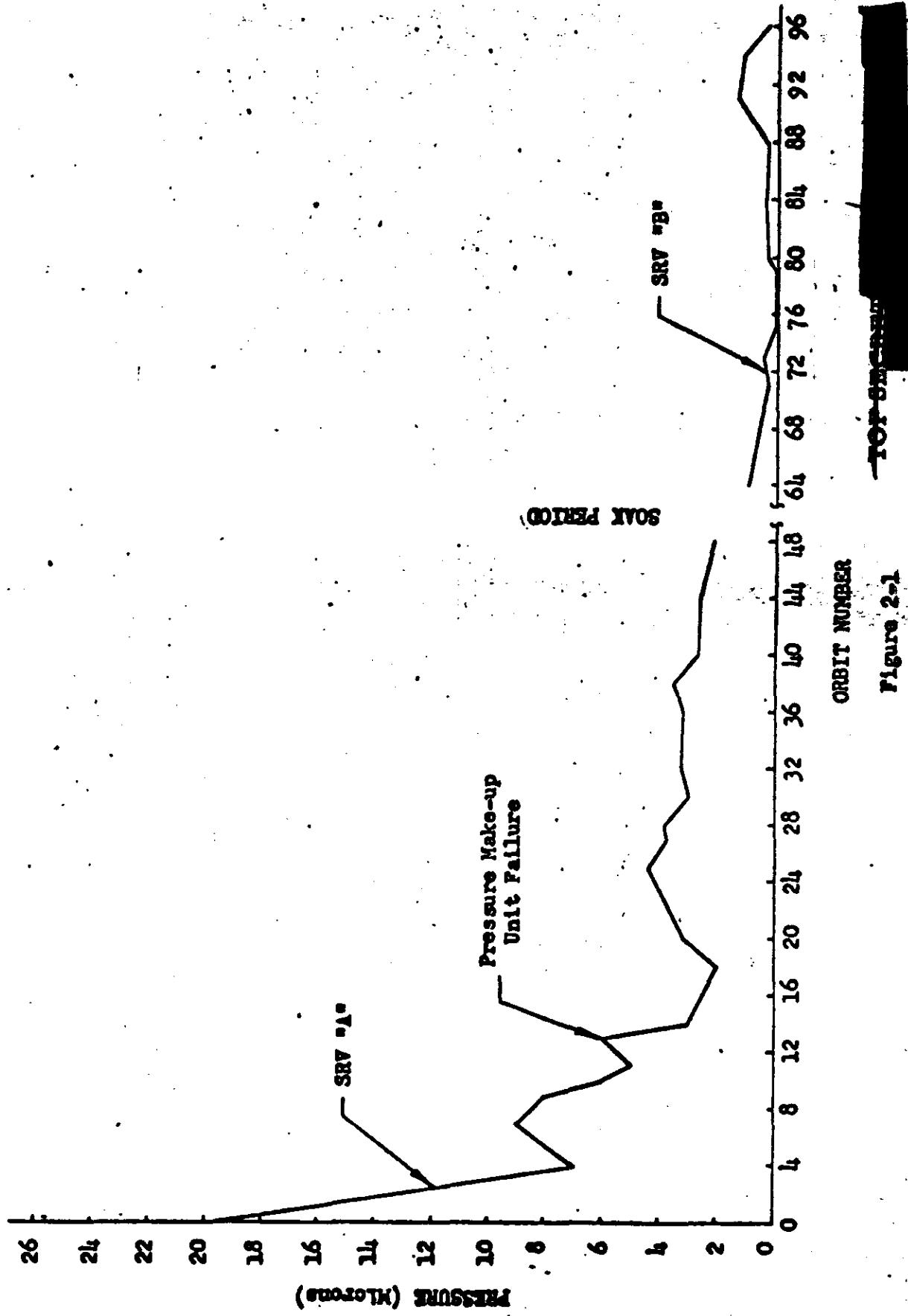


Figure 221

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5. Panoramic Camera Performance

The Panoramic Camera system operation was normal throughout the test, with the exception of the uneven metering at shutdown of the Slave instrument on orbits 3 & 4 of the second day of "A" Mission operation. This uneven metering was found to be due to incorrect tension of the telemetry monitor on the Horizon idler. The cycle periods were within $\pm 1.5\%$ of the calibrated values.

Examination of the film transported through the system during the environmental test showed that minor start-up corona discharge fogging occurred in the Slave camera only. Inspection of the camera systems revealed a film of oil on the field flatteners of both cameras. The source of the oil was not determined hence it is not known whether the oil was present prior to the environmental test. All future systems are to be visually inspected prior to and following the environmental test to assure clean lens surfaces.

6. Stellar-Index Camera Performance

The "A" Mission stellar index operation was normal, with the exception of film movement at platen down time during orbit 15 of the second day.

The "B" mission stellar index metering monitors indicated film motion at platen commands throughout the test. The shutter monitor did not indicate the shutter was opening. The processed film did not indicate any metering anomalies nor shutter failures. The shutter monitor anomaly was caused by insufficient intensity of the fogging lamps, due to instrumentation hook-up.

Post-test examination of the film metered through both cameras showed that the cameras were suitable for flight.

7. Clock Performance

The clock operated throughout both "A" and "B" Missions. The console power was turned off during the "A" Mission, which prevented any clock correlation due to unscheduled clock restart. The clock took four (4) orbits to stabilize during the "B" Mission, after soak,

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and thereafter gained approximately 60 milliseconds between orbits 4 and 15. This was attributed to clock offset.

8. Instrumentation and Commanding System Performance

The V/h programmer brush 27 which indicated a failure to acknowledge two of the eleven commands sent was attributed to operate error in failure to wait approximately seventeen (17) seconds for brush filter delay.

The RTC 6 anomaly was traced to a misaligned relay tension spring. Re-alignment of the relay curbed the problem.

B. RESOLUTION TEST

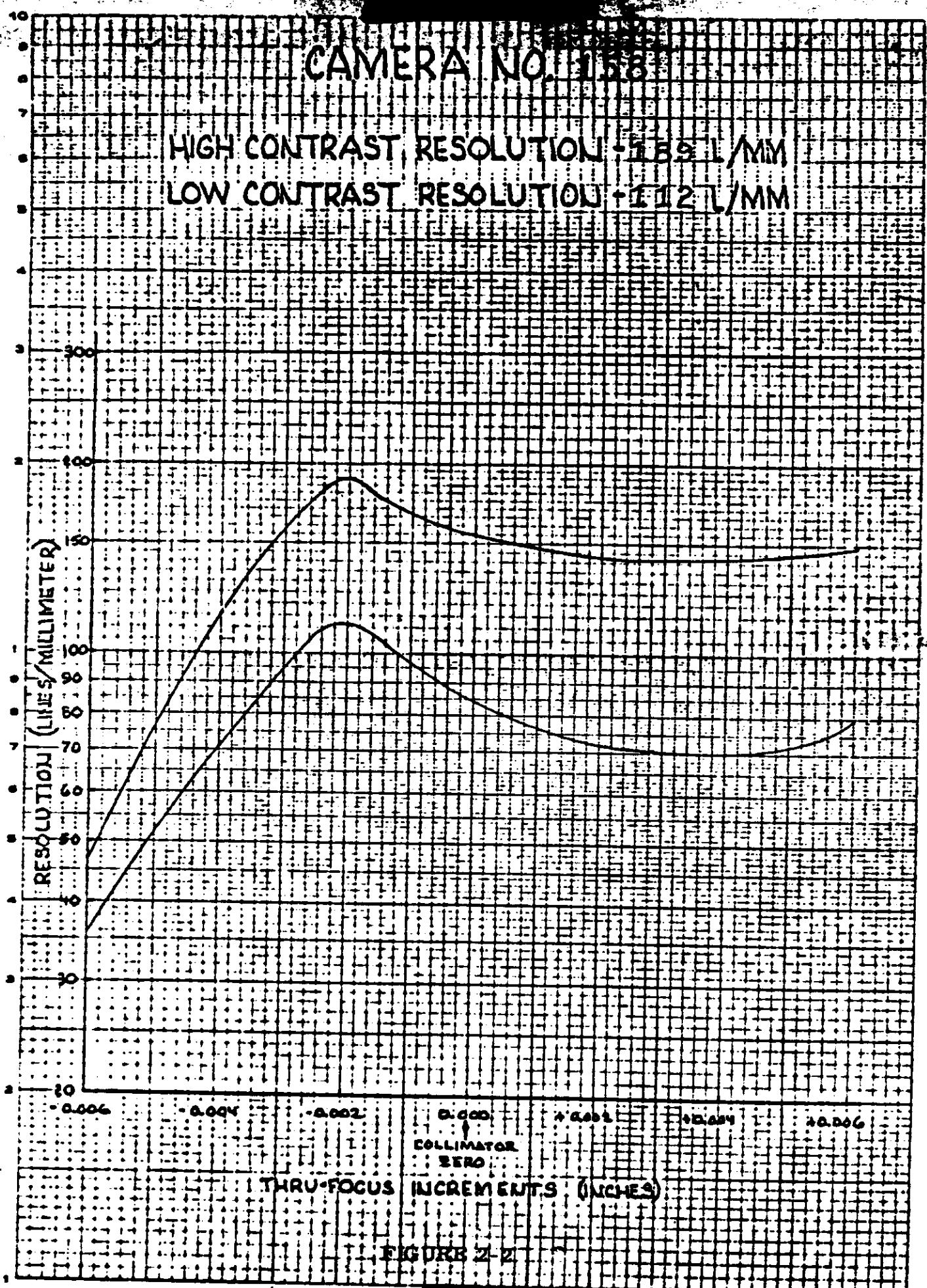
The dynamic resolution test of the J-15 payload system was performed at the A/P facility on 27 August 1964. Each panoramic camera photographed high and low contrast resolution targets. The resulting through focus resolution data is shown in Figure 2-2 for the Master camera and in Figure 2-3 for the Slave camera.

Examination of the resolution data from the Master camera showed that peak performance occurred at -0.002" which is a greater departure from zero focus than normally experienced. Analysis showed that the three sigma focus limits of the A/P collimator are from -0.00225" to +0.00075 hence the peak resolution point of camera #158 is within the expected limits. A re-test of this camera showed no change in the peak resolution setting therefore the test data was approved.

C. LIGHT LEAK TEST

The examination of the film threaded in the J-15 system during the light leak test determined that no film fogging was present. The light leak integrity of the system was considered acceptable for flight.

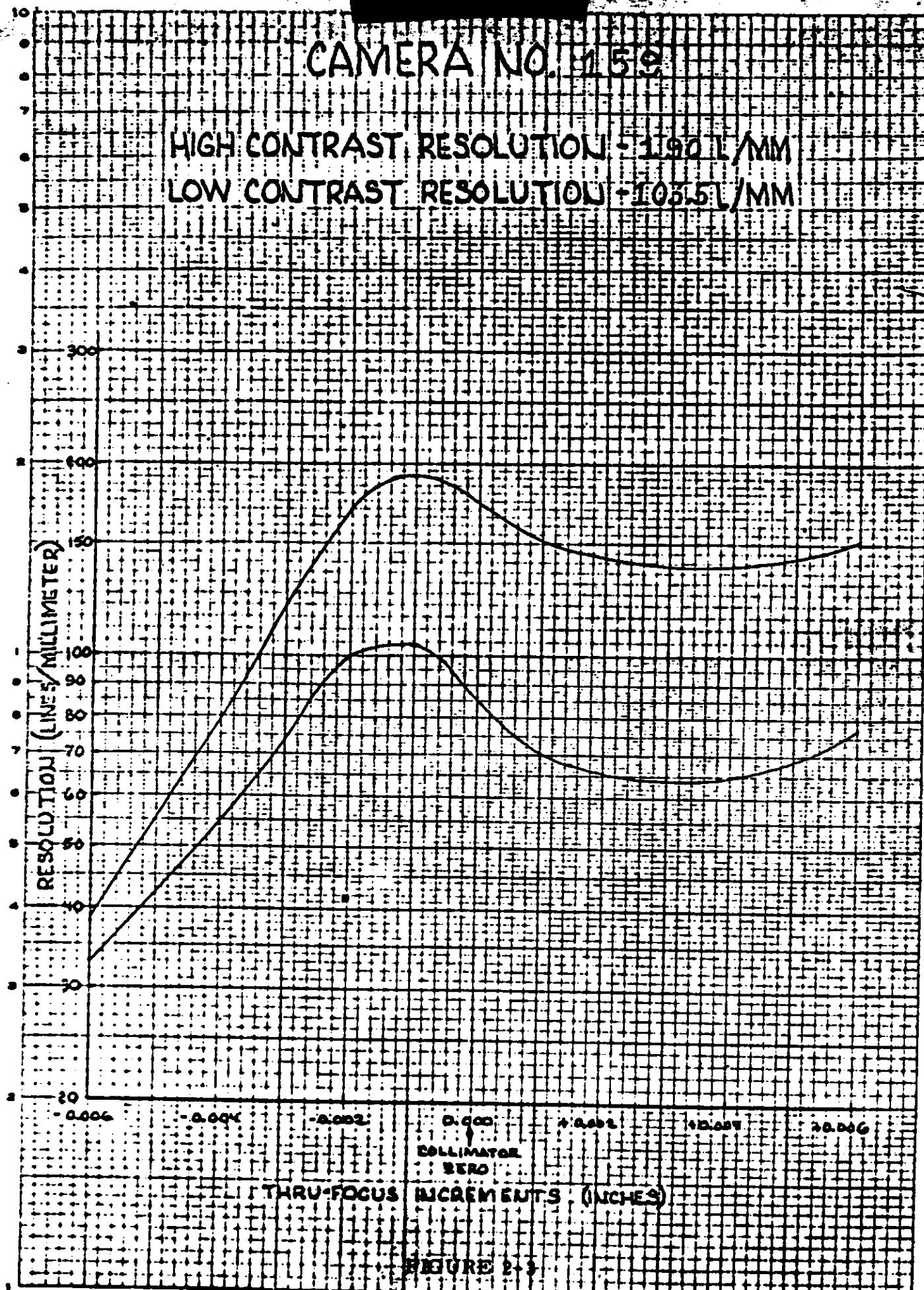
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CAMERA NO. 150

HIGH CONTRAST RESOLUTION - 190 L/MM
LOW CONTRAST RESOLUTION - 105.5 L/MM



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SECTION 3

FLIGHT OPERATIONS

A. INSTRUMENTATION AND COMMAND PERFORMANCE

At acquisition at [REDACTED] on orbit 1 the camera system was operating and had completed 345 cycles from launch. The system continued to run until an OFF command was received at the end of a programmed 14 cycle engineering operation. A total of 416 cycles were completed with the Master camera and 417 cycles with the Slave camera. The vehicle telemetry signal strength was very low resulting in only 50 to 60 seconds of good telemetry data during this pass. This precluded analysis of the complete operation occurring within the normal acquisition range at the [REDACTED] Tracking Station. The only anomaly evident in the data was an indicated OFF position on the ON/OFF command monitor. During the time of the normal programmed engineering operation the telemetry was unusable. Therefore, the status of the ON/OFF monitor was not verified during this programmed operation.

The telemetry monitor relay in the command box is in parallel with the operate relay. From an analysis of the circuitry involved to turn the instrument system on it appears that the operate relay was latched without latching the telemetry relay. This was substantiated by an interrupt to the operate signal to the instrument system by the initiation of 2 intermix commands (RTC 12).

These commands caused a normal OFF of the camera system as evidenced by end of pass markers on the processed film coincident with the time the intermix commands were issued. A sequence of programmed events from launch to orbit 1 is tabulated below.

SEQUENCE OF EVENTS (LAUNCH TO [REDACTED])

<u>Time</u>	<u>Delta Time</u>	<u>Event</u>
T +0		Launch
T +161	161	Inflight reset (door eject)
T +434	273	Burnout (28v reg. to instr. and cassette power switch over)

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SEQUENCE OF EVENTS (LAUNCH TO [REDACTED])
(Continued)

<u>Time</u>	<u>Delta Time</u>	<u>Event</u>
T +3810	3376	1st of 12 programmed brush 27's 1 every 50 sec. (V/h start delay)
T +3910	100	1st recorder center of format on slave
T + 5256	1348	[REDACTED] acq. start sending 2 RTC 12's (intermix mode change)
T + 5260	4	End of pass marker (frame 344M-345S)
T +5401	141	Programmed brush 36 (on command) frame 400
T +5436	35	Programmed brush 37 (off cmd) end of pass mark frame 416- 416S

An attempt to duplicate the "run away" operation was made on the engineering operation on Pass 9. All command settings were the same as the initial command settings for launch. The "run away" operation was not duplicated on this or any other pass during the flight.

The most probable cause of the "run away" operation was a short in the command box operate relay. This short was apparently cleared when the relay was reset by the OFF command. //

The instrumentation system performance was satisfactory with the exception of the low signal strength of the telemetry transmitter. This low signal strength did not impair the payload system operation.

No other command or instrumentation anomalies were evident in the telemetry data.

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B. PANORAMIC CAMERA PERFORMANCE

The panoramic camera system operation appeared normal during the limited data available on the orbit 1 [REDACTED] "run away" operation. Cycle periods were 0.8% faster than the pre-flight calibrations. All camera status monitors indicated normal operation during this operation with the exception of command monitor anomaly noted above.

The camera system operation was normal on the engineering operations on orbits 9, 16, and 31. The operation on orbit 47 was normal with the exception of uneven metering at shutdown of the Slave camera indicating possible slack in the film path. Below is a tabulation of the cycle periods observed on the engineering passes as compared to the pre-flight calibrations.

CYCLE PERIOD DATA

<u>Pass</u>	<u>Cycle Period</u>	<u>Error</u>	<u>Cycle Period</u>	<u>Error</u>
1	2.72	.8% Fast	2.70	.8% Fast
9	5.01	0%	4.96	0.4% Fast
16	2.31	1.7% Slow	2.27	1.3% Slow
*31	2.21	2.2% Fast	2.21	.9% Fast
47	2.20	.9% Fast	2.21	.45% Fast

* V/h ramp changed to increase instrument speed after orbit 16.

At the [REDACTED] acquisition on orbit 56, the film footage monitors indicated the camera system had failed on approximately orbit 52. The engineering operation on pass 63 confirmed the suspected failure. Both cameras were still operating; however, no film was being transported. All status monitors were normal with the exception of the lens rotation/center of format monitor relationship on the Slave camera. This monitor indicated slippage had occurred in the transport system. The cause of the failure was not evident in the telemetry data nor has it been conclusively determined. However, the most probable explanation appears to be a common loss of power to the take-up cassettes. The programmed operations between orbit 47 and the last recovered film are shown below.

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J-15 SEQUENCE

<u>ORBIT</u>	<u>ON TIME</u>	<u>OFF TIME</u>	<u>REMARKS</u>
47	74470	74580	Master - 50 cycles (normal operation) Slave - 51 cycles (uneven shutdown)
51	9259	9378	Master - 48 cycles (metering anomaly on frame 20) Slave - 48 cycles (metering anomaly on frame 10)
52 Operation 1	14560	14692	Master - 50 cycles (metering anomaly throughout pass) Slave - 51 cycles (metering anomaly throughout pass)
52 Operation 2	14730	14834	Master - 59 cycles (frame to frame length anomaly) Slave - 59 cycles (metering anomaly throughout pass)
52	14948	15033	Master - 38 cycles (failed at start up) Slave - 38 cycles (failed after 2-1/2 cycle)

The camera system transferred to the second recovery system and continued to operate without transporting film after the recovery of the second recovery system.

Several operational command modes were utilized in an unsuccessful attempt to duplicate the "run away" operation on orbit 1 after the recovery of the second recovery system.

C. STELLAR-INDEX PERFORMANCE

The Stellar-Index camera operated normally on engineering passes during Mission 1013-1 with no anomalies evident in the telemetry data. The Stellar-Index camera for Mission 1013-2 operated normally on the engineering pass on orbit 72 prior to second recovery and continued operating after second recovery.

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D. CLOCK PERFORMANCE

The clock system operation was satisfactory throughout the mission. The following correlation constants were derived using 8 hand-read data points and performing a least squares linear fit through the data:

<u>SYSTEM TIME I/P</u>	<u>CL TIME I/P</u>	<u>COMP SYS TM</u>
33323. 4335	395863. 1399	33323. 4374
38772. 4478	401312. 1609	38772. 4561
78439. 7810	440979. 4969	78439. 7750
39377. 2445	488316. 9819	39377. 2397
73857. 0135	522796. 7629	73857. 0059
34884. 8462	33353. 7099	34884. 8445
74611. 3319	73080. 2169	74611. 3345
35630. 6569	120499. 5649	35630. 6621

A0 = 362539. 5326

A1 = 0. 999999570617

SIGMA = 0. 00511

RATIO OF CLOCK TIME TO SYS TIME = 0. 100000042938D 01

The following constants were derived by performing a 2nd order fit:

<u>REV.</u>	<u>SYSTEM TIME I/P</u>	<u>CL TIME I/P</u>	<u>COMP SYS TM</u>
8	33323. 4335	395863. 1399	33323. 4325
9	38772. 4478	401312. 1609	38772. 4521
16	78439. 7810	440979. 4969	78439. 7765
25	39377. 2445	488316. 9819	39377. 2448
31	73857. 0135	522796. 7629	73857. 0117
40	34884. 8462	33353. 7099	34884. 8485
47	74611. 3319	73080. 2169	74611. 3346
56	35630. 6569	120499. 5649	35630. 6546

A0 = 362539. 7161

A1 = 1. 00000029943

A2 = 0. 7013757130203D - 12

SIGMA = 0. 00258

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E. PRESSURE MAKE-UP SYSTEM PERFORMANCE.

Operation of the pressure make-up system was satisfactory throughout the mission with a surplus of approximately 1700 PSIA of nitrogen at the end of the Mission 1013-1. Figure 3-1 is a plot showing the consumption during the flight.

F. TEMPERATURE ENVIRONMENT

A tabulation of the real time temperature data recorded at the [redacted] Tracking Station are included as Tables 3-1 and 3-2. These data have been corrected for self heating from data obtained from self-heating tests conducted during environmental testing.

Figures 3-2 through 3-4 are plots showing actual versus predicted temperatures for selected temperature sensors throughout the payload system.

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J-15 PRESSURE MAKE-UP SUPPLY CONSUMPTION
"A" MISSION ONLY

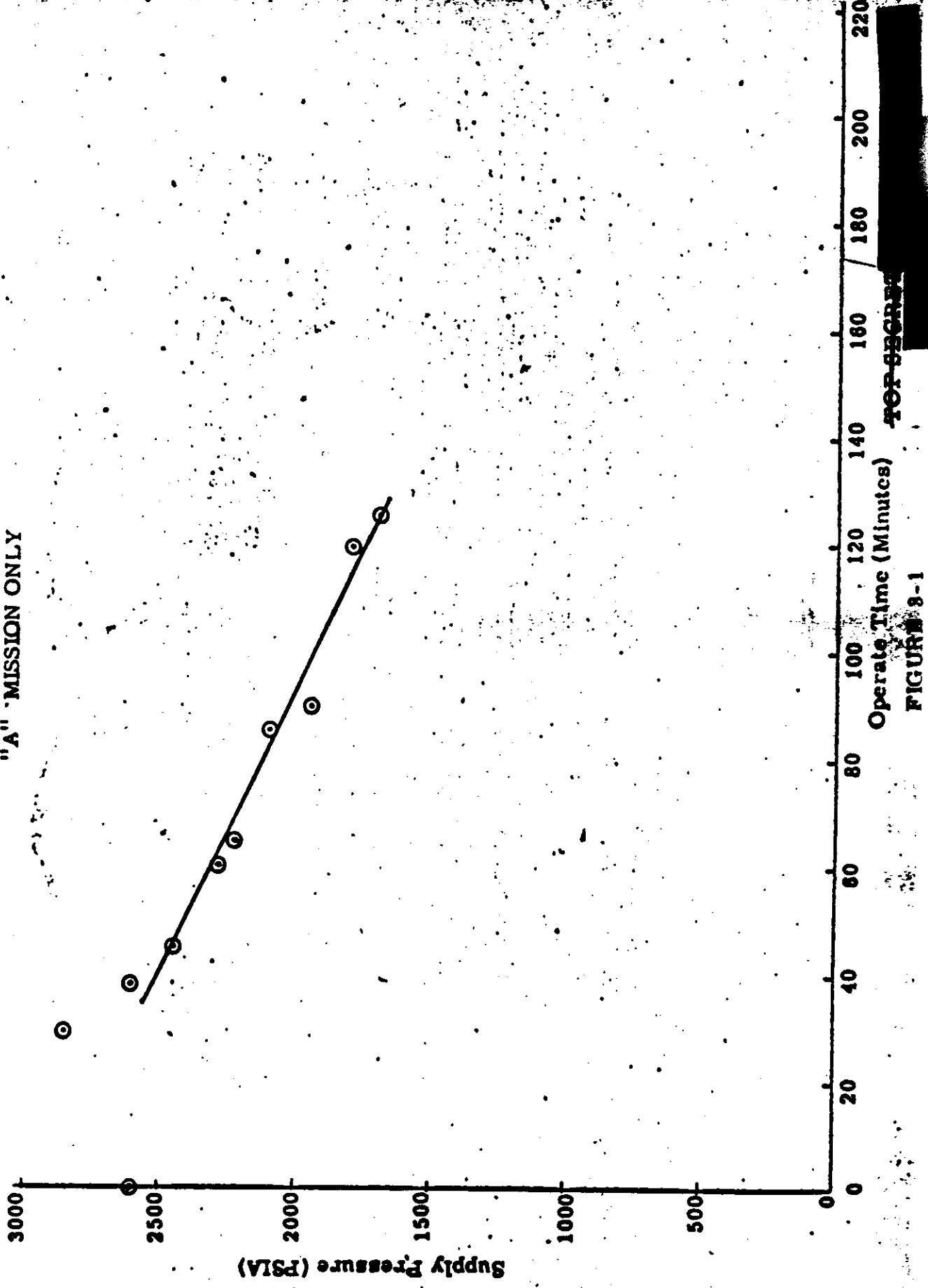


FIGURE 8-1

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J-15 1173 TEMPERATURE SUMMARY

SENSOR	ORBIT											
	Master	08	09	16	25	31	40	47	56	63	72	79
3	70	61	59	54	54	52	56	51	54	49	49	47
4	73	69	68	65	65	61	65	60	66	58	60	54
5	70	72	67	62	63	60	64	58	65	57	59	55
6	68	77	76	69	70	67	70	65	71	64	65	59
7	69	72	71	65	66	63	67	62	66	61	63	57
8	75	77	73	68	68	65	68	64	71	62	65	58
9	73	82	79	72	75	69	74	67	76	66	71	64
10	71	71	67	63	65	59	63	59	64	57	59	55
11	97	81	75	69	72	67	73	65	71	62	65	58
12	77	67	65	58	60	57	62	55	61	52	57	49
13	73	76	71	68	67	65	68	64	66	61	62	57
Avg.	Instr.	Temp.	72	73	70	64	65	62	66	66	59	61

Slave	ORBIT											
	3	4	5	6	7	8	9	10	11	12	13	Supply Spool
68	77	75	69	71	66	69	66	66	66	66	66	59
68	72	68	62	66	59	65	57	66	56	60	60	51
70	72	68	62	64	62	65	59	66	58	60	60	54
68	66	64	58	59	58	60	56	61	53	56	56	50
66	71	68	63	65	61	64	60	65	58	60	58	54
69	67	65	59	61	58	63	56	64	55	58	58	50
71	64	61	56	58	56	59	53	59	51	54	49	
69	68	67	62	64	59	63	58	64	57	59	53	
92	63	60	55	58	53	58	52	54	50	52	49	
71	74	71	63	68	62	67	59	68	57	64	54	
76	69	66	62	62	61	63	60	62	58	57	56	
70	69	67	62	64	60	64	58	64	57	59	53	

TABLE 3-1

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NOTE: All data corrected for self-heating, except injection

J-15 1173 TEMPERATURE SUMMARY

<u>SENSOR</u>	<u>ORBIT</u>						
Fairing/Barrel #1							
1	L	08	09	16	25	31	40
	OBH	49	42	29	52	33	46
2	OBH	12	9	-1	12	-1	-5
3	OBH	3	0	3	-1	3	-4
4	256	63	60	51	57	48	57
5	OBH	87	84	75	90	71	78
6	OBH	78	72	75	81	75	68
	FAIRING						
Barrel #2							
1	156	59	59	72	62	69	56
2	155	56	56	84	66	84	53
3	180	23	20	30	27	36	20
4	189	6	3	-4	6	-4	6
5	178	17	14	11	11	11	-10
	CONIC ADAPTER						
Conic Adapter							
1	163	63	63	66	60	57	54
Clock							
1	97	71	67	65	67	62	60
2	105	82	80	73	77	73	77
Thrust Cone "A" to "B" SRV							
1	110	52	49	41	42	40	43
		80	72	72	62	63	59
2							
Stellar/Index "A" to "B"							
1	84	66	62	56	56	59	53
2	69	66	60	57	57	53	57
Recovery Batt. "B" SRV							
1	72	76	75	72	68	69	67
Master Cassette "A" SRV							
1	--	--	--	--	--	--	--
2	--	--	--	--	--	--	--

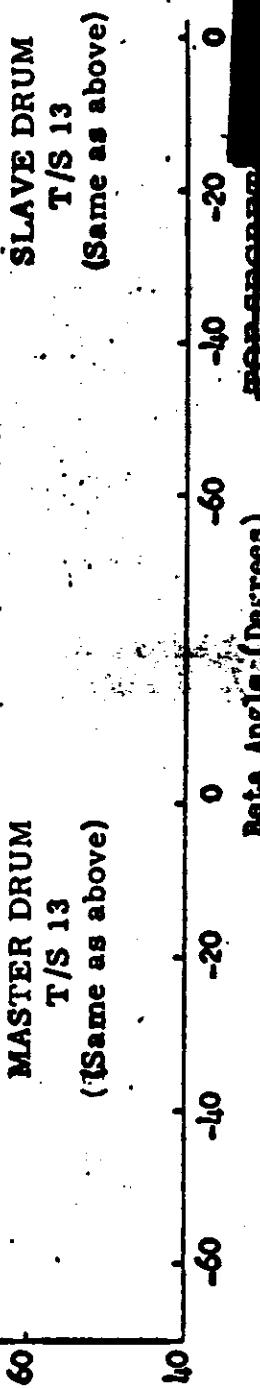
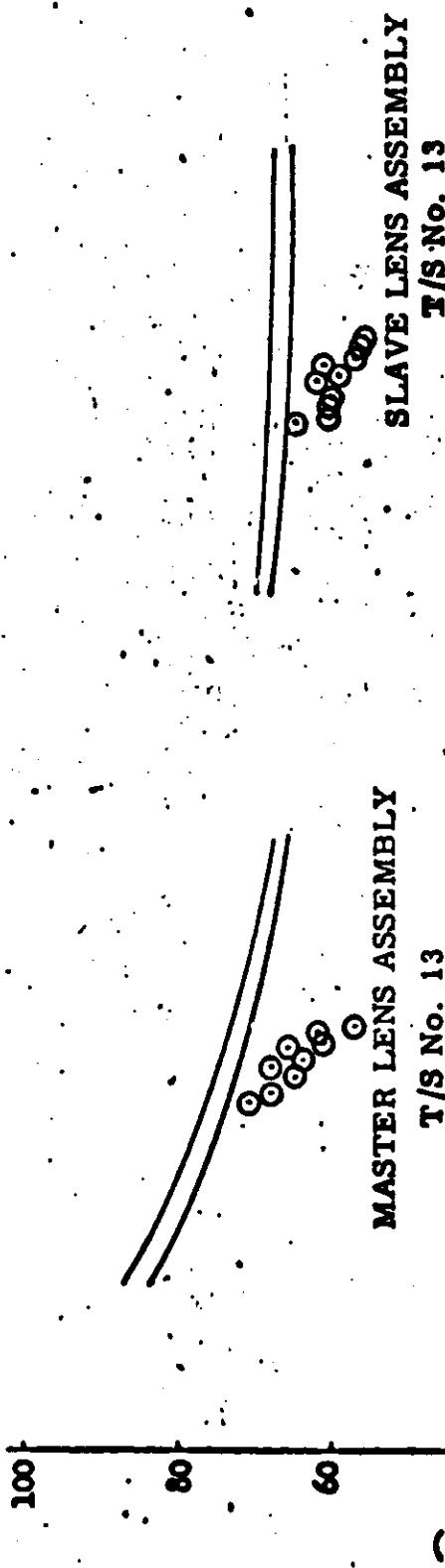
NOTE: Only thrust Cone Data corrected for self-heating

TABLE 3-2

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J-15 PREDICTED AND ACTUAL TEMPERATURES

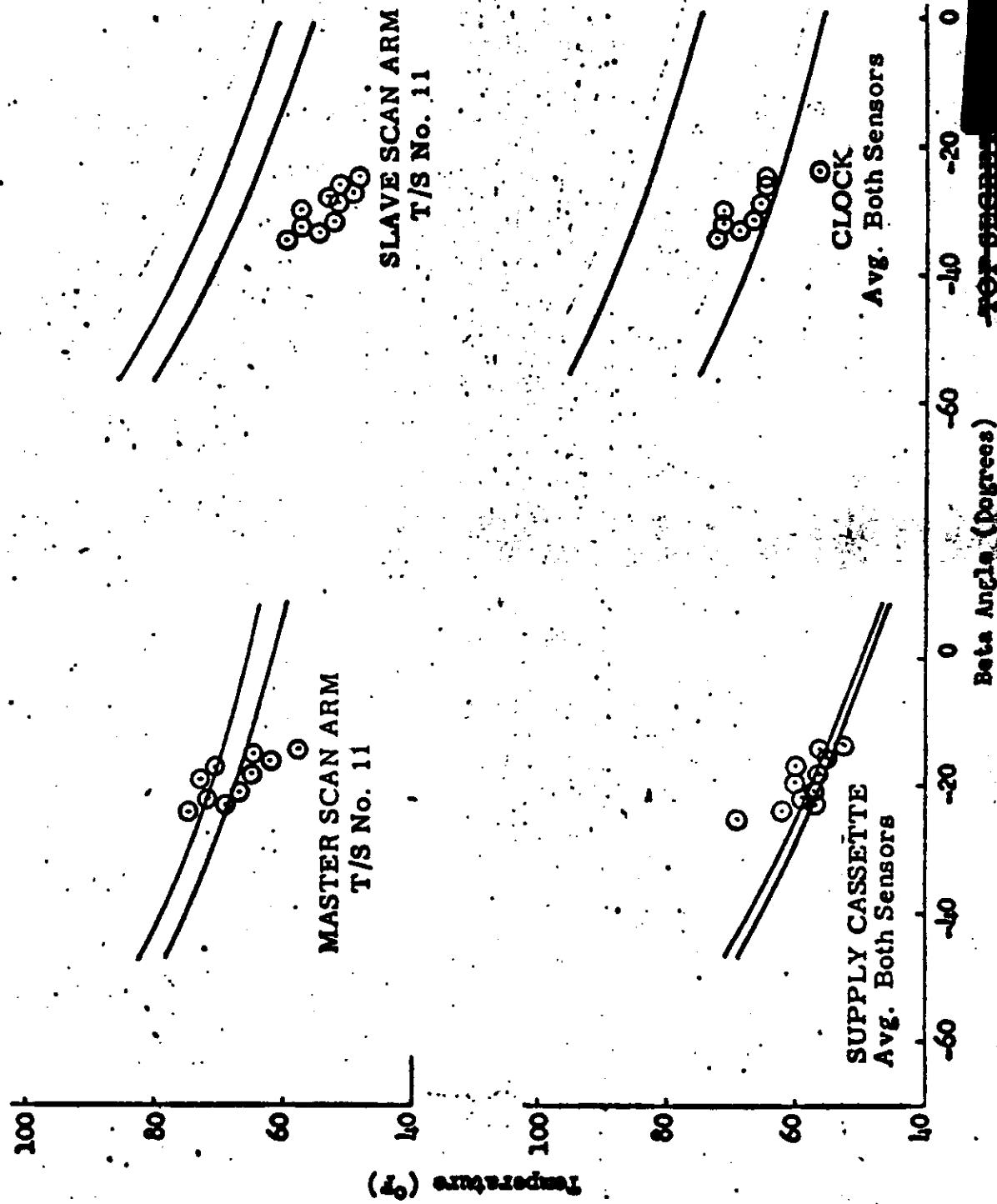


MASTER DRUM
T/S 13
(Same as above)

FIGURE J-15

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J-15 PREDICTED AND ACTUAL TEMPERATURES



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J-15 PREDICTED AND ACTUAL TEMPERATURES

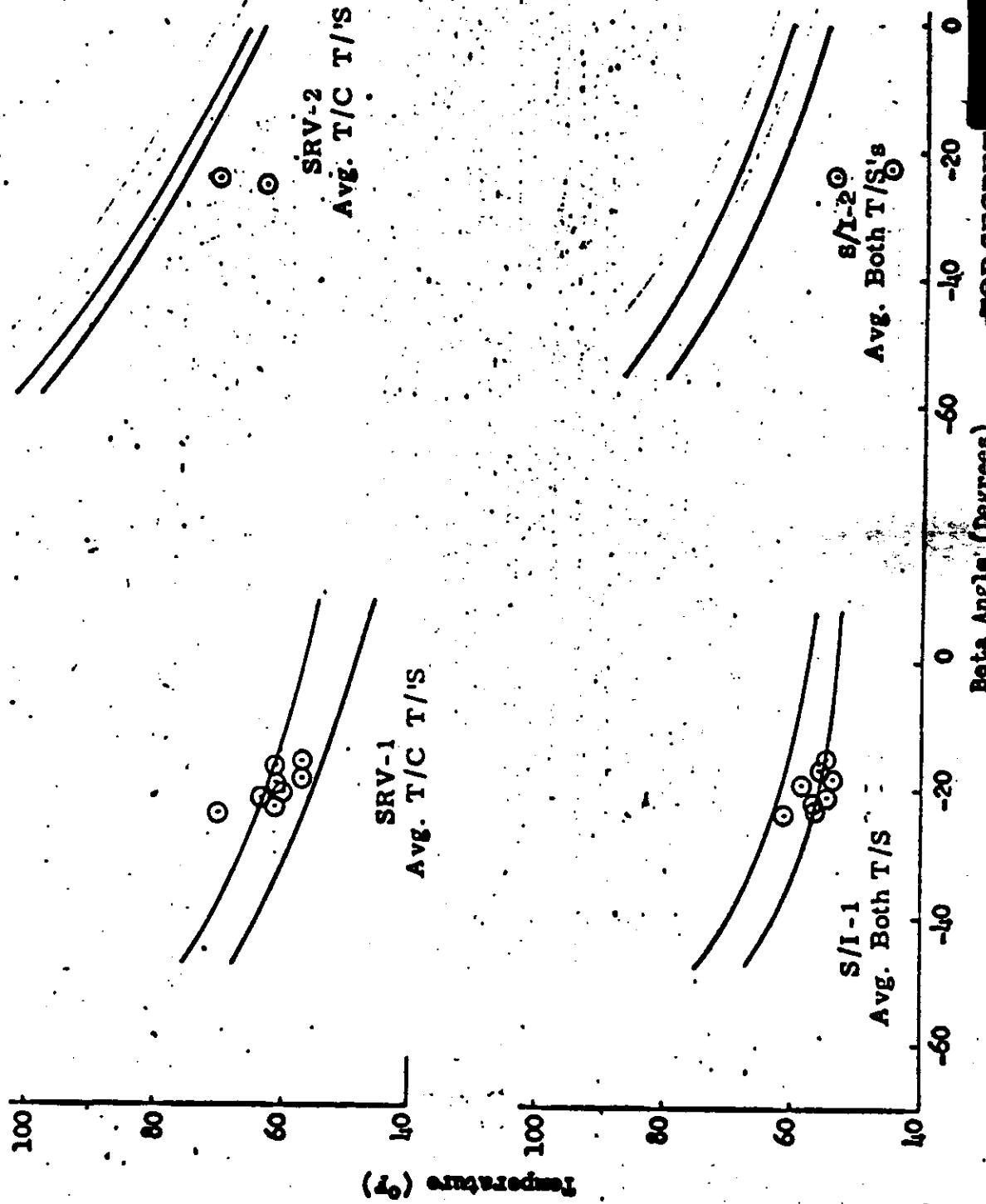


FIGURE 3-4

SECTION 4

MISSION 1013-1 RECOVERY SYSTEM

SRV #656 was received at A/P on 30 December 1963. The receiving inspection weight was 152.2 pounds. After modifications and incorporation of outstanding E.O.'s, the SRV was delivered to Systems Test for incorporation into the J-10 system. The SRV was subsequently found defective and after repair was assigned to J-15.

The Recovery System was shipped to VAFB on 16 September 1964 and finally mated to the J-15 system on 29 October 1964. A successful air catch was made during orbit 65 on 6 November 1964. Table 4-1 gives a tabulation of the sequence of recovery events.

The condition of the recovered capsule was satisfactory with damage limited to normal paint blistering. Figures 4-1 and 4-2 show the location and temperatures encountered by Templates attached to the capsule cover for the recovery system.

A complete post flight recovery sequence was conducted on the recovered capsule including cassettes and all parameters were within specification.

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MISSION 1013-1
RE-ENTRY SEQUENCE OF EVENTS

<u>EVENT</u>	<u>SYSTEM TIME</u>	<u>DELTA TIME</u>	
		<u>Actual</u>	<u>Nominal</u>
Transfer	260.15	-	--
Electrical Disconnect	261.18	1.03	.90 \pm .43 -.40
*Separation	262.15	2.00	2.0 \pm .25
**Spin	264.60	3.42	3.4 \pm .30
Retro	272.13	7.53	7.55 \pm .45
Despin	282.82	10.69	10.75 \pm .54
T/C Separation	284.49	1.67	1.5 \pm .15
"G" Switch Open	677.12	392.63	--
Parachute Cover Off	710.80	33.68	34.0 \pm 1.5
Drogue Chute Deployed	711.38	0.58	.63 \pm .08
Drogue Chute Release	721.41	10.03	10.05 \pm 1.0
Main Chute Deployed	722.04	0.63	.52 \pm .12
Main Chute Disreefed	726.40	4.36	4.0 \pm 1.7

* From Transfer

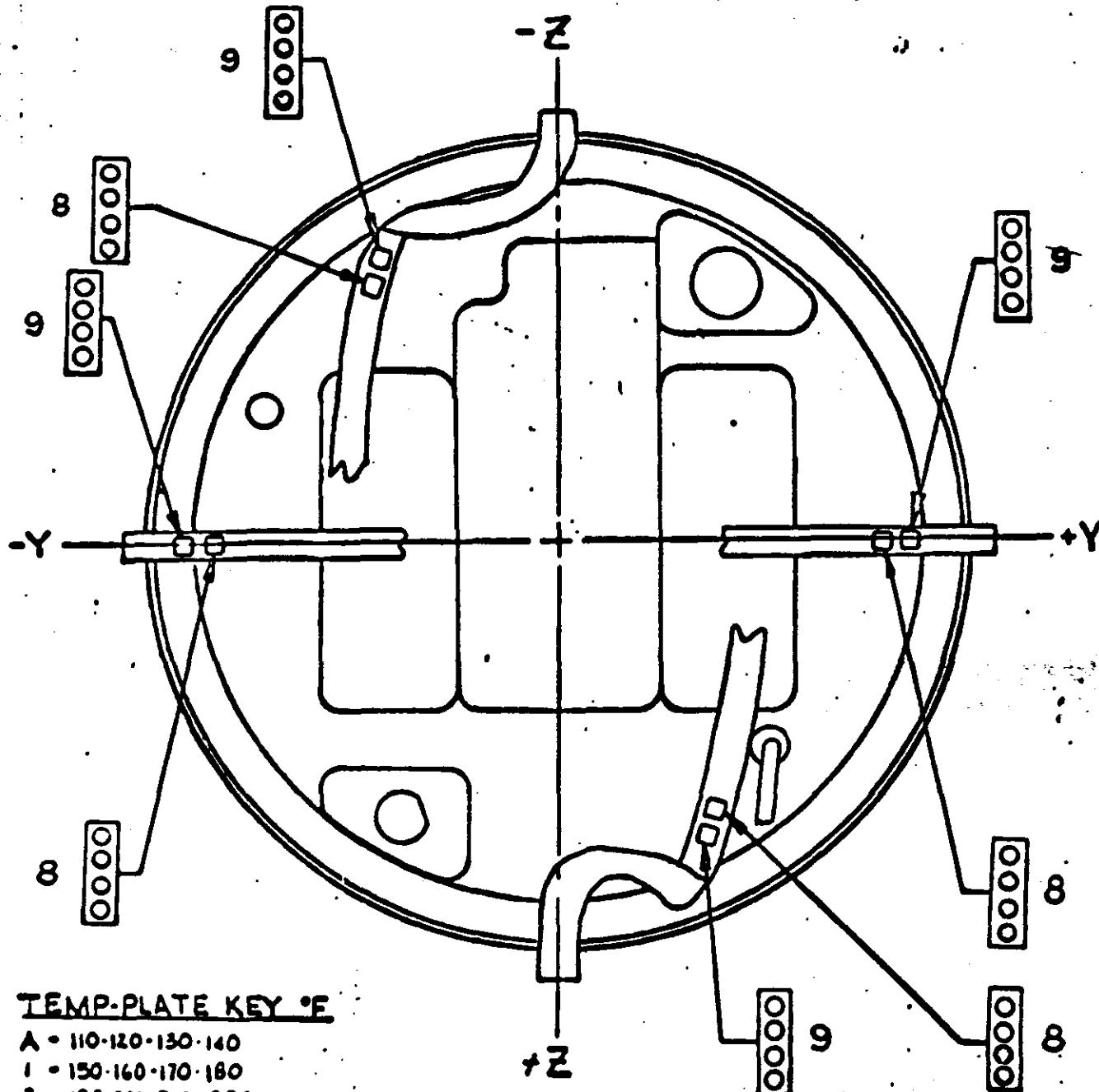
** From Elect. Disc.

TABLE 4-1

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TEMP-PLATE INSTALLATION - MK V-A CAPSULE



TEMP-PLATE KEY °F

- A • 110-120-130-140
- 1 • 150-160-170-180
- 2 • 190-200-210-220
- 3 • 230-240-250-260
- 4 • 270-280-290-300
- 5 • 310-320-330-340
- 6 • 350-360-370-380
- 7 • 390-410-435-450
- 8 • 460-480-500-520
- 9 • 530-550-570-590

LOOKING FORWARD

USE OF TEMP PLATES
ON PARACHUTE SHROUDS

- INDICATOR TURNED BLACK
TEMP REACHED OR EXCEEDED
INDICATED LEVEL

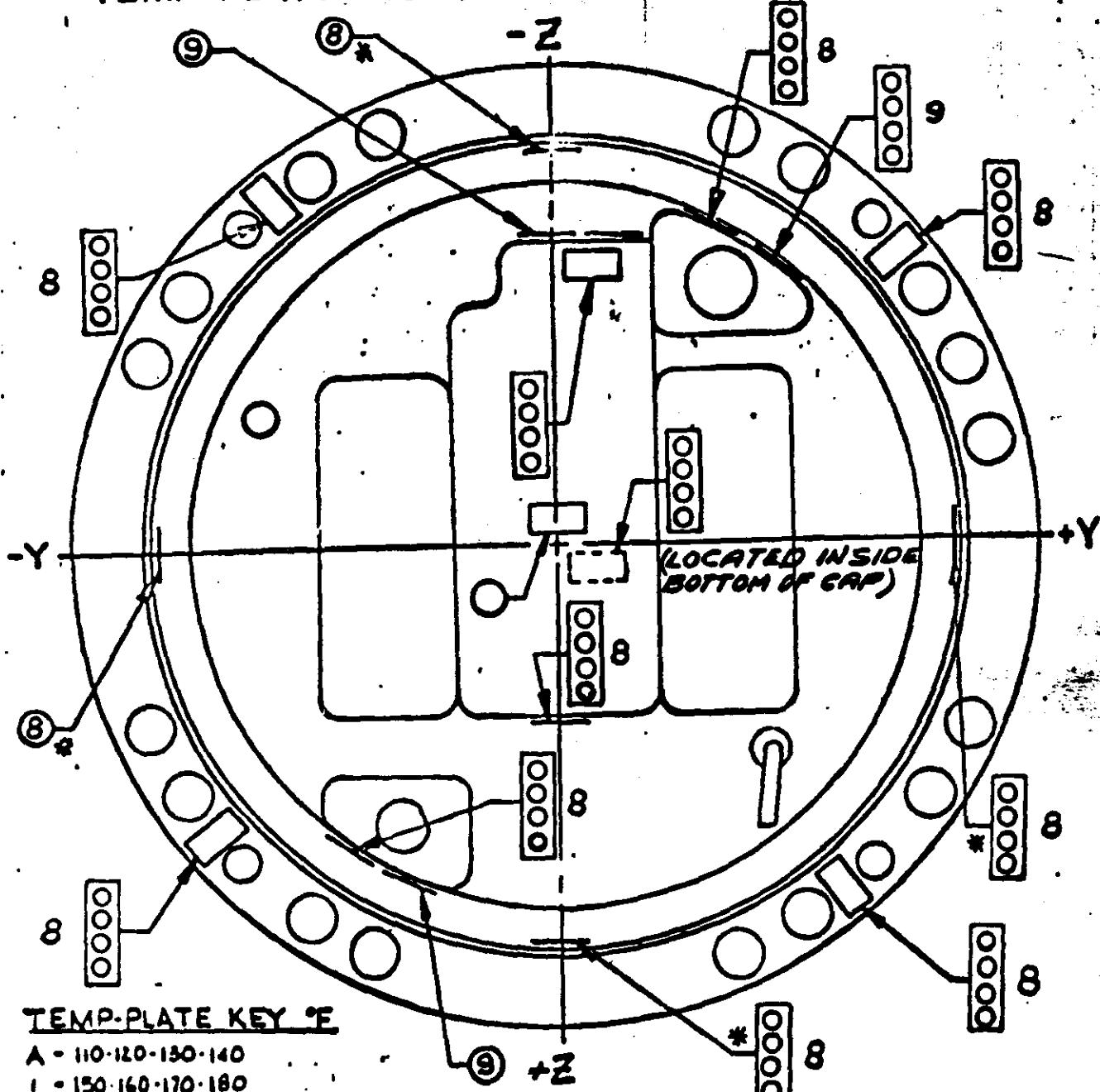
1013-1

FIGURE 4-1

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TEMP-PLATE INSTALLATION - MK II-A CAPSULE



TEMP-PLATE KEY *

- A - 110-120-130-140
- B - 150-160-170-180
- C - 190-200-210-220
- D - 230-240-250-260
- E - 270-280-290-300
- F - 310-320-330-340
- G - 350-360-370-380
- H - 390-410-435-450
- I - 400-450-500-550

* LOCATED INSIDE CAPSULE
ON NOSE WALL

• INDICATOR TURNED BLACK
TEMP REACHED OR EXCEEDED
INDICATED LEVEL

1013-1

FIGURE 4-2

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SECTION 5

MISSION 1013-2 RECOVERY SYSTEM

SRV #657 was received at A/P on 20 January 1964 at a receiving weight of 149.4 pounds. After modification and incorporation of outstanding E.O.'s the capsule was delivered to systems test for incorporation into the J-10 system. An over-rich resin concentration resulted in cracks in the forebody during TASC testing. Following repair the SRV was assigned to the J-15 payload.

The capsule was shipped to VAFB on 16 September 1964. A successful air catch was made during orbit 81 on 7 November 1964. Table 5-1 lists the recovery sequence of events.

The condition of the recovered capsule was satisfactory with damage limited to normal paint blistering. Figures 5-1 and 5-2 show the location and temperatures encountered by the Templates attached to the capsule.

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MISSION 1013-2
RE-ENTRY SEQUENCE OF EVENTS

<u>EVENT</u>	<u>SYSTEM TIME</u>	<u>DELTA TIME</u>	
		<u>Actual</u>	<u>Nominal</u>
Transfer	85881.45	-	--
Electrical Disconnect	85882.56	1.11	.90 + .43 - .40
*Separation	85883.44	1.99	2.0 + .25
** Spin	85885.98	3.42	3.4 + .30
Retro	85893.52	7.54	7.55 + .45
Despin	85904.19	10.67	10.75 + .54
T/C Separation	85905.74	1.55	1.5 + .15
"G" Switch Open	86369.46	463.72	--
Parachute Cover Off	003.71	34.25	34.0 + 1.5
Drogue Chute Deployed	004.37	0.76	.63 + .08
Drogue Chute Release	014.43	10.06	10.05 + 1.0
Main Chute Deployed	014.93	0.50	.52 + .12
Main Chute Disreefed	019.68	4.75	4.0 + 1.7

* From Transfer

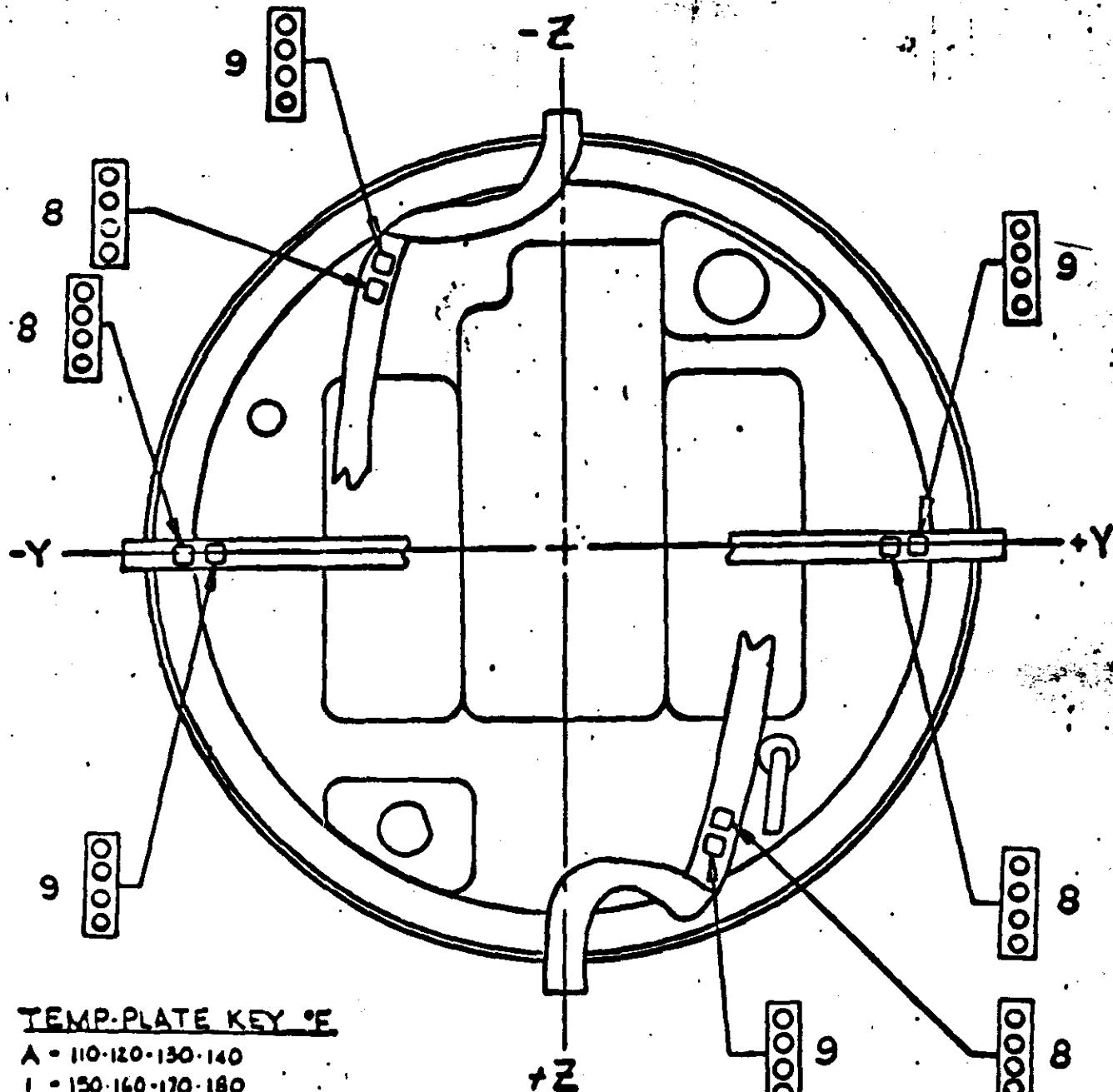
** From Elect. Disc.

TABLE 5-1

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TEMP-PLATE INSTALLATION - Mk V-A CAPSULE



TEMP-PLATE KEY 'E'

- A - 110-120-130-140
- 1 - 150-160-170-180
- 2 - 190-200-210-220
- 3 - 230-240-250-260
- 4 - 270-280-290-300
- 5 - 310-320-330-340
- 6 - 350-360-370-380
- 7 - 390-410-435-450
- 8 - 460-480-500-520
- 9 - 530-550-570-590

LOOKING FORWARD

USE OF TEMP PLATES
ON PARACHUTE SHROUDS

● INDICATOR TURNED BLACK
TEMP REACHED OR EXCEEDED
INDICATED LEVEL

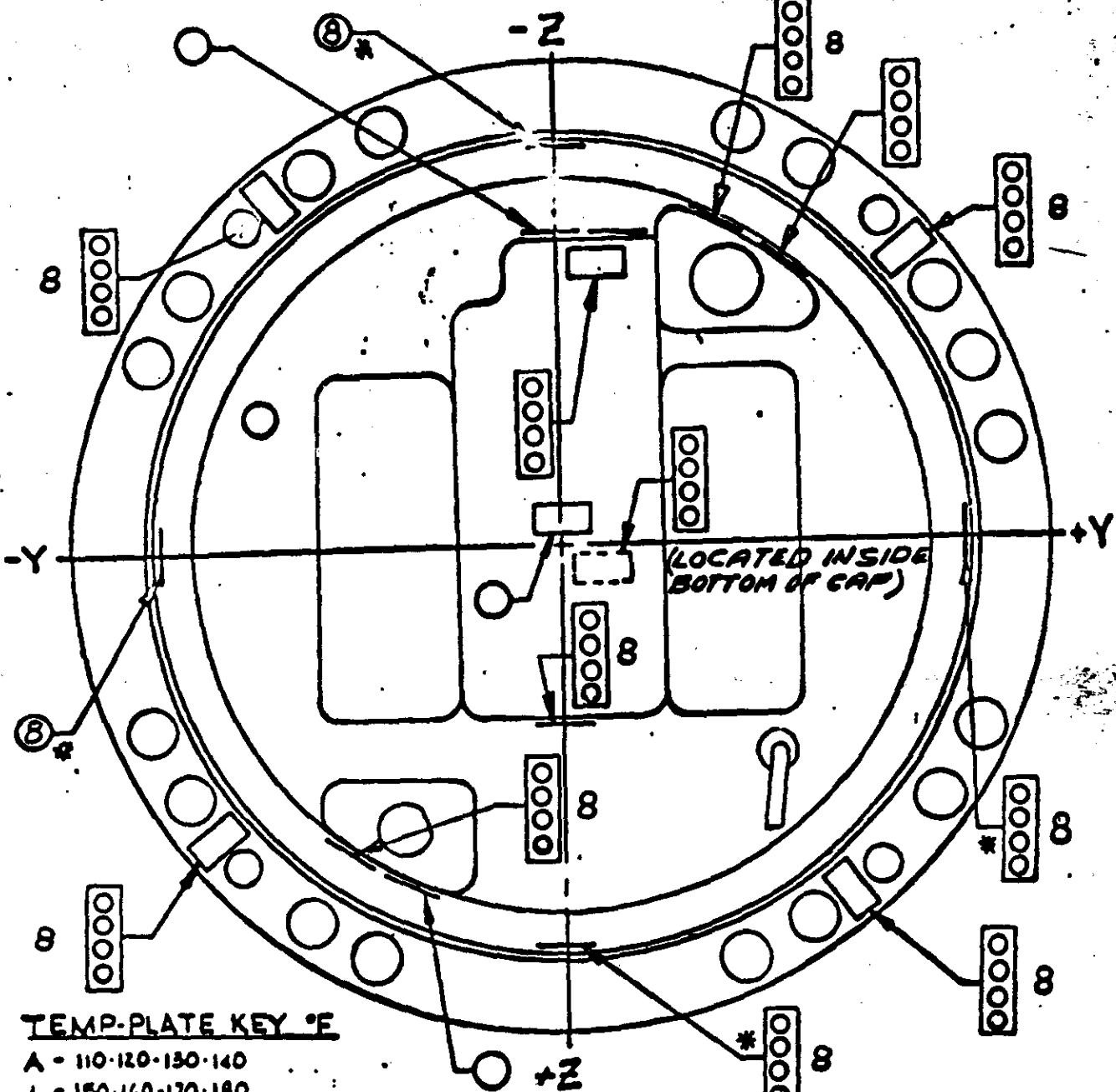
1013-2

FIGURE 5-1

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TEMP-PLATE INSTALLATION - Mk V-A CAPSULE



TEMP-PLATE KEY °F

- A - 110-120-130-140
- 1 - 150-160-170-180
- 2 - 190-200-210-220
- 3 - 230-240-250-260
- 4 - 270-280-290-300
- 5 - 310-320-330-340
- 6 - 350-360-370-380
- 7 - 390-410-435-450
- 8 - 460-480-500-520
- 9 - 530-550-570-590

LOOKING FORWARD

* LOCATED INSIDE CAPSULE
ON NOSE WALL

● INDICATOR TURNED BLACK
TEMP REACHED OR EXCEEDED
INDICATED LEVEL

1013-2

FIGURE 5-2

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SECTION 6

MASTER (FWD) PANORAMIC CAMERA

A. COMPONENT ASSIGNMENT

<u>Component</u>	<u>Serial Number</u>
Main Camera	158
Main Camera Lens	1302435
Supply Horizon Camera	173B
Supply Horizon Camera Lens	813550
Take-up Horizon Camera	164A
Take-up Horizon Camera Lens	813512
Supply Cassette	SC-20

B. CAMERA DATA AND FLIGHT SETTINGS

Main Camera:

Lens	24" f/3.5
Slit Width	0.325"
Filter Type	Wratten 21
Film Type	Eastman Type 4404

Supply (Port) Horizon Camera:

Lens	55mm f/6.8
Aperture Setting	f/6.8
Exposure Time	1/100 second
Filter Type	Wratten 25

Take-up (Starboard) Horizon Camera:

Lens	55mm f/6.8
Aperture Setting	f/8.0
Exposure Time	1/100 second
Filter Type	Wratten 25

C. POST FLIGHT PERFORMANCE EVALUATION

The camera operated properly up to pass D47E of Mission 1013-1. During passes D47E, D51 and D52 film metering was unstable causing intermittent out of focus frames. The soft imagery was present in frames that were longer than nominal by as much as 0.090". The camera failed at or near the start of the third programmed operate sequence of pass D52.

Table 6-1 lists the known data at the time that film metering ceased in both panoramic cameras. It is from this data that the time of failure is calculated. The exact cause of the failure is unknown however the most probable cause is a temporary loss of power to the take-up cassettes. The fact that the cameras did not fail at the same time does not fully support this postulate although it is possible for a camera to meter some film without take-up tension. The camera continued to operate properly through Mission 1013-2, as evidenced by the normal operation of both Stellar-Index cameras, however no panoramic camera film was metered after the failure during pass D52.

The second major anomaly encountered by the Master camera was the acquisition of 416 frames of photography during passes A-01 and D-01 of which only the last 14 frames were programmed. The failure and probable cause is discussed in Section 3 of this report.

The overall photographic quality up to pass D47E was considered to be somewhat lower than recently observed. This quality reduction was attributed to an increased cloud and atmospheric haze coverage. In those areas not affected by these conditions the photographic quality and information content rated among the best observed to date.

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MISSION 1013-1
PANORAMIC CAMERA FAILURES - PASS D52

	MASTER CAMERA	SLAVE CAMERA
MISSION 1013-1 LAST FRAME	#104 + 3-1/8" of #105	#107 + 15-1/2" of #108
MISSION 1013-2 LAST FRAME	#107 + 10-1/2" of #108	#110 + 22-1/2" of #111
AT END OF PASS MARK AT FAILURE	18-1/2" of #110	1/4" of #114
TIME WORD AT EXPOSURE	#104 - 99957397	#105 - 99957777
PROGRAMMED CAMERA OFF	#109	#110
PROGRAMMED CAMERA ON	#110	#111
CAMERA FAILURE		Apx. 2-1/2 cycles after camera On

TABLE 6-1

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There were some minor light leaks near the beginning and end of most camera operations. The resulting degradation was far less severe than resulted during recent missions. No dendritic static or corona discharge fogging was observed on the recovered film.

The horizon cameras, binary data block and 200 cycle time track operated properly throughout the mission.

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SECTION 7

SLAVE (AFT) PANORAMIC CAMERA

A. COMPONENT ASSIGNMENT

<u>Component</u>	<u>Serial Number</u>
Main Camera	159
Main Camera Lens	1362435
Supply Horizon Camera	166B
Supply Horizon Camera Lens	813553
Take-up Horizon Camera	129A
Take-up Horizon Camera Lens	812272
Supply Cassette	SC-20

B. CAMERA DATA AND FLIGHT SETTINGS

Main Camera:

Lens	24" f/3.5
Slit Width	0.225"
Filter Type	Wratten 21
Film Type	Eastman Type 4404

Supply (Starboard) Horizon Camera:

Lens	55mm f/6.8
Aperture Setting	f/8.0
Exposure Time	1/100 second
Filter Type	Wratten 25

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Take-up (Port) Horizon Camera:

Lens	55mm f/6.8
Aperture Setting	f/6.8
Exposure Time	1/100 second
Filter Type	Wratten 25

C. POST FLIGHT PERFORMANCE EVALUATION

The camera operated properly up to frame 89 of pass D24. During the exposure of frame 90 one half of the format, along the major film axis, went out of focus as the scan head passed a manufacturers splice. A sketch of frames 89, 90 and 91 acquired during pass D24 is shown in Figure 7-1. Approximately 65% of all frames following frame 91 exhibited this same out of focus condition.

A small out of focus area was noted at the supply end of random frames starting in pass D04. The soft imagery corresponds in area and magnitude to similar anomalies in previous missions. It is not considered that there is any relationship between this degradation and the anomaly that began during pass D24.

The quality of the Slave camera photography was essentially the same as the Master camera in areas that were not degraded by out of focus imagery.

The binary data block and horizon cameras operated properly during the mission. The 200 cycle time track ranged from light to non-existent. The frequency marks were imaged inside the format area and could only be detected where background density conditions were ideal.

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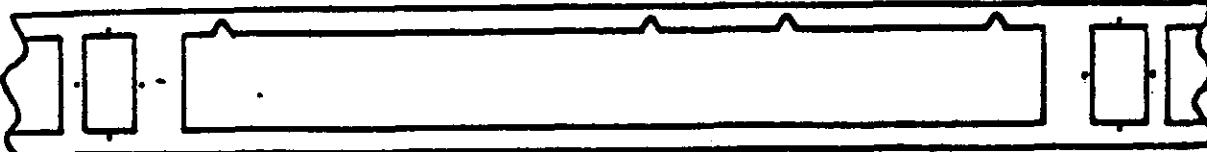
MISSION 1013-1

SLAVE CAMERA FORMATS

CAMERA NO. 159

PASS NO. D24

FRAME NO. 89



CAMERA NO. 159

PASS NO. D24

FRAME NO. 90



CAMERA NO. 159

PASS NO. D24

FRAME NO. 91



DATA KEY:

Sharp focus.

Soft focus.

Manufacturers splice.

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FIGURE 7-1

SECTION 8

PANORAMIC CAMERA EXPOSURE

The exposure parameters of both the panoramic cameras were the normal 0.225 inch wide slit and Wratten 21 filter. These conditions place the nominal exposure on the intermediate level processing curve, as published by [REDACTED] for their 3404 emulsion.

The illumination conditions during the mission were relatively broad as the flight was conducted near the autumnal equinox. The frequency distributions of the solar elevations and solar azimuths encountered during the photographic operations are shown in Figures 8-1 and 8-2.

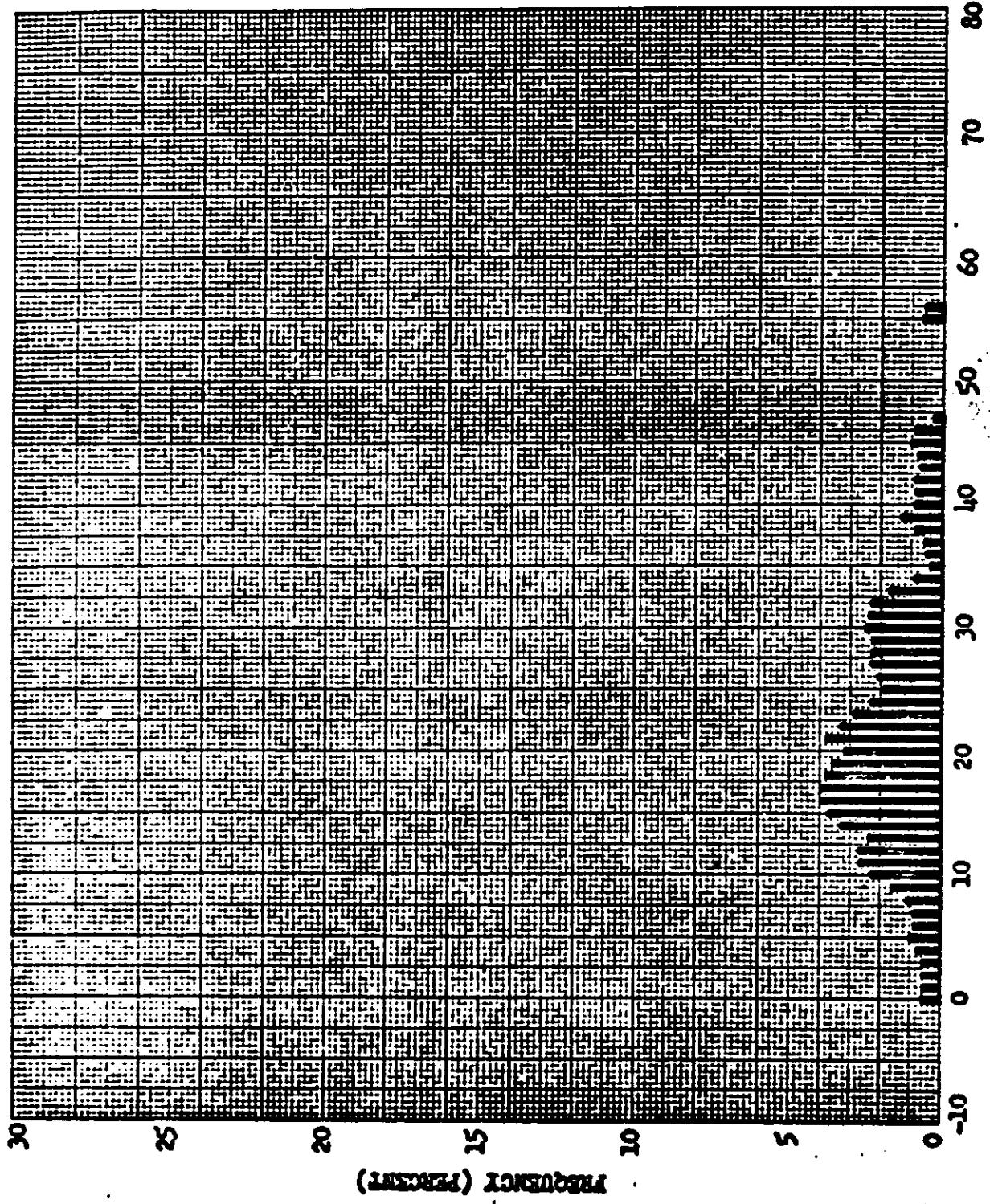
The nominal exposure times are shown as a function of latitude for passes D08 and D56 in Figures 8-3 and 8-4. The predicted level of processing for the original negative is based on the in-flight performance estimate and is tabulated below with the processing levels reported by [REDACTED]

<u>Mission</u>	<u>Camera</u>		<u>Primary</u>	<u>Intermediate</u>	<u>Full</u>
1013-1	FWD	Predicted	0	64	36
		Reported	0	42	58
1013-1	AFT	Predicted	0	63	37
		Reported	2	7	91

The variation in the predicted and reported processing levels is generally consistent with the data observed from recent missions. The use of greater percentages of full processing has been experienced throughout the Corona program. Further analysis and calculations are in process to attempt to ascertain the optimum exposure-processing conditions.

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SOLAR ELEVATION FREQUENCY DISTRIBUTION

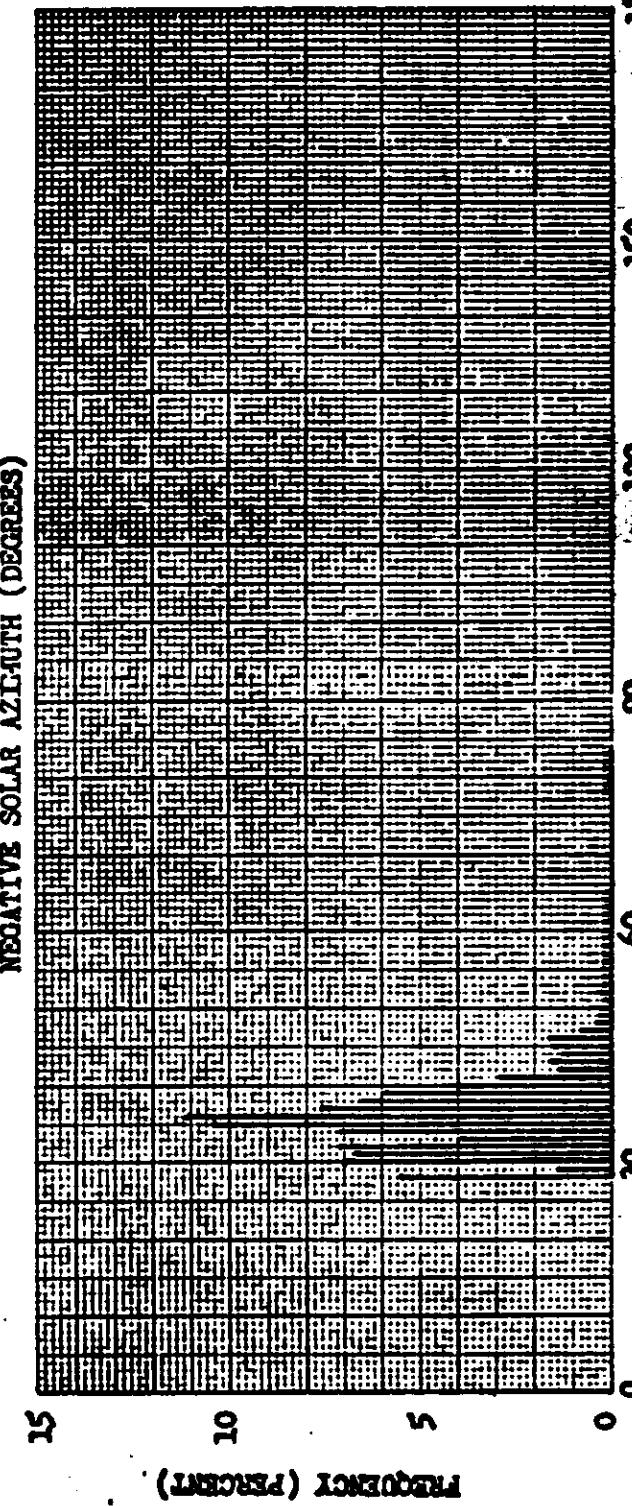
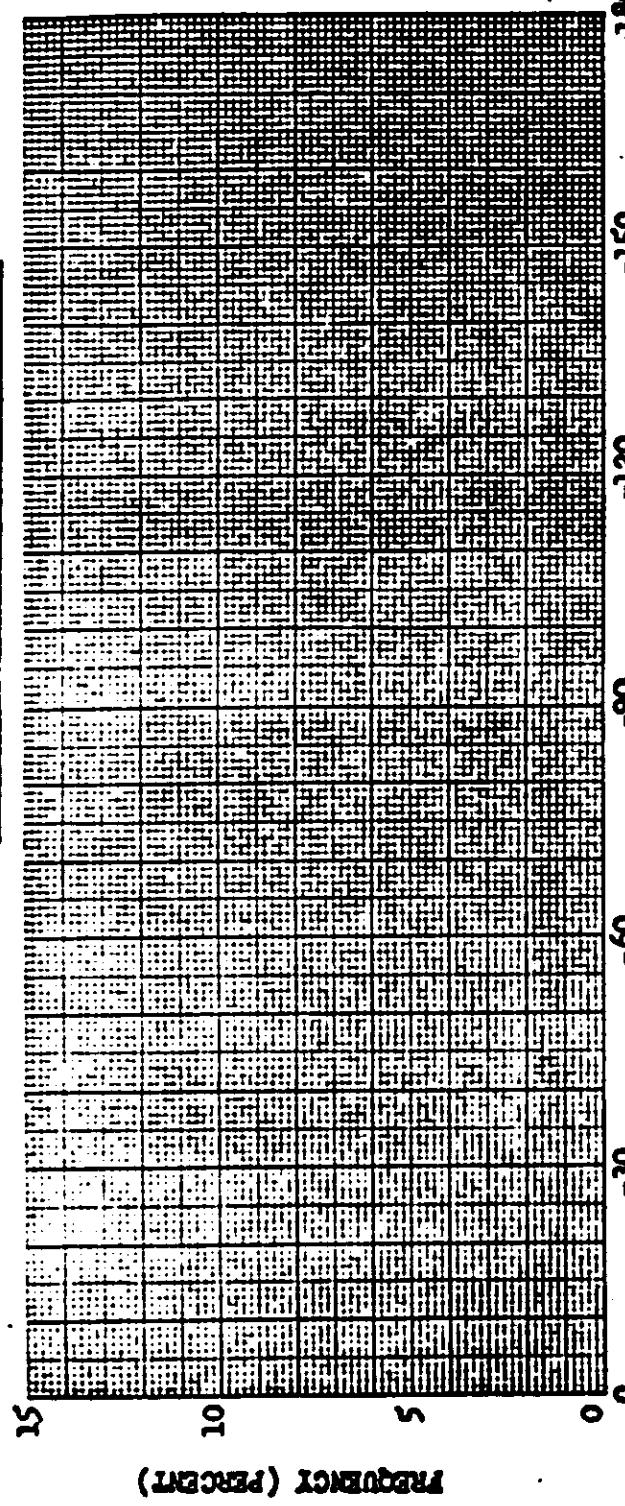
Mission No: 1013-1
Payload No: J-15
Camera No: 158
Launch Date: 11/2/64
Launch Time: 2130 Z
Inclination: 75°



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FIGURE 8-1

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SOLAR AZIMUTH FREQUENCY DISTRIBUTION



Mission No: 1013-1

Payload No: J-15

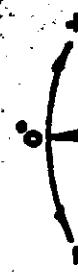
Camera No: 158

Launch Date: 11/2/64

Launch Time: 2130 Z

Inclination: 75°

SIGN MOTION

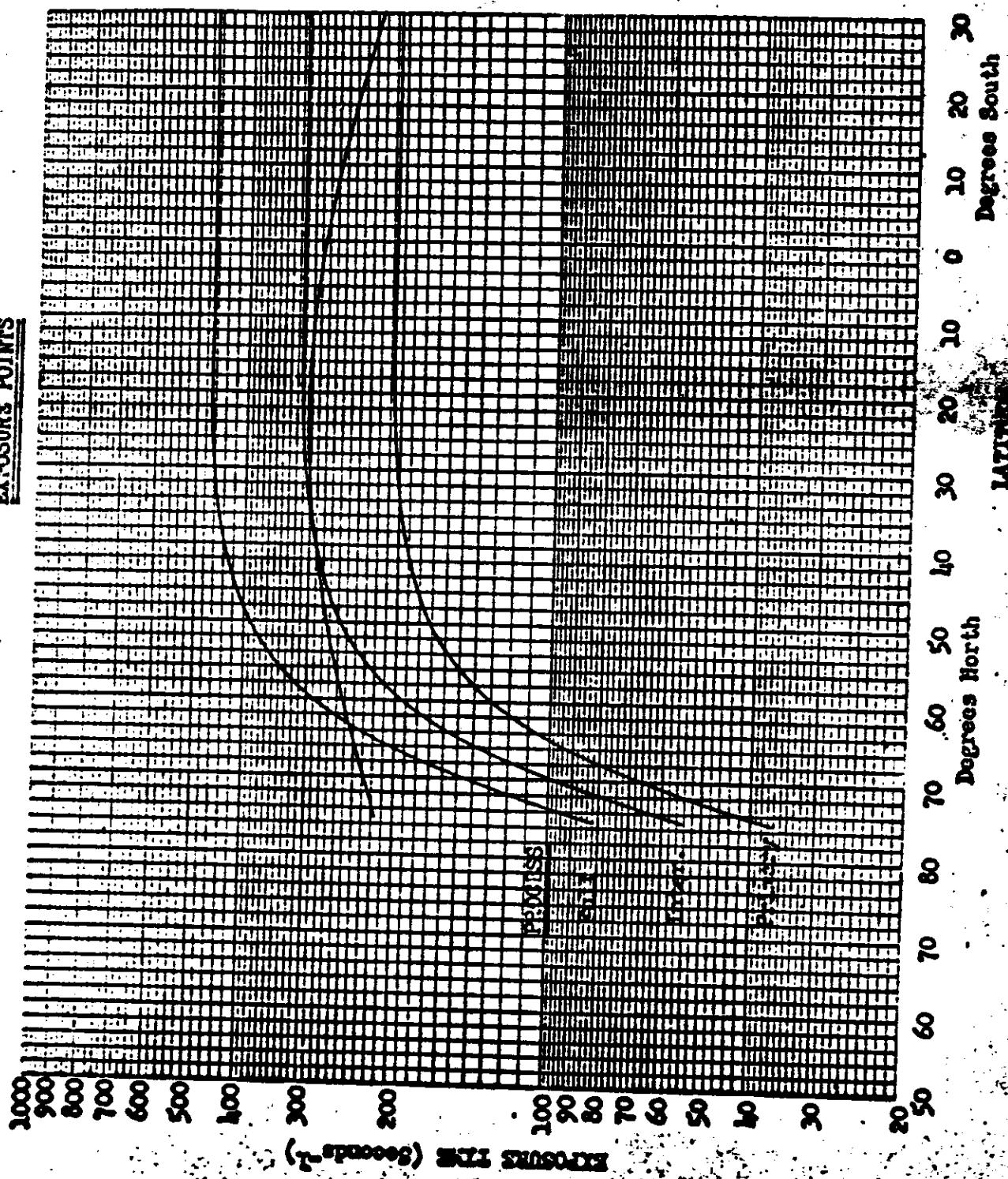


DIRECTION OF MOTION

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FIGURE 8-3



Mission No: 1013

Payload No: J-15

Camera No: 158 & 159

Pass No: 56

Launch Date: 11/2/64

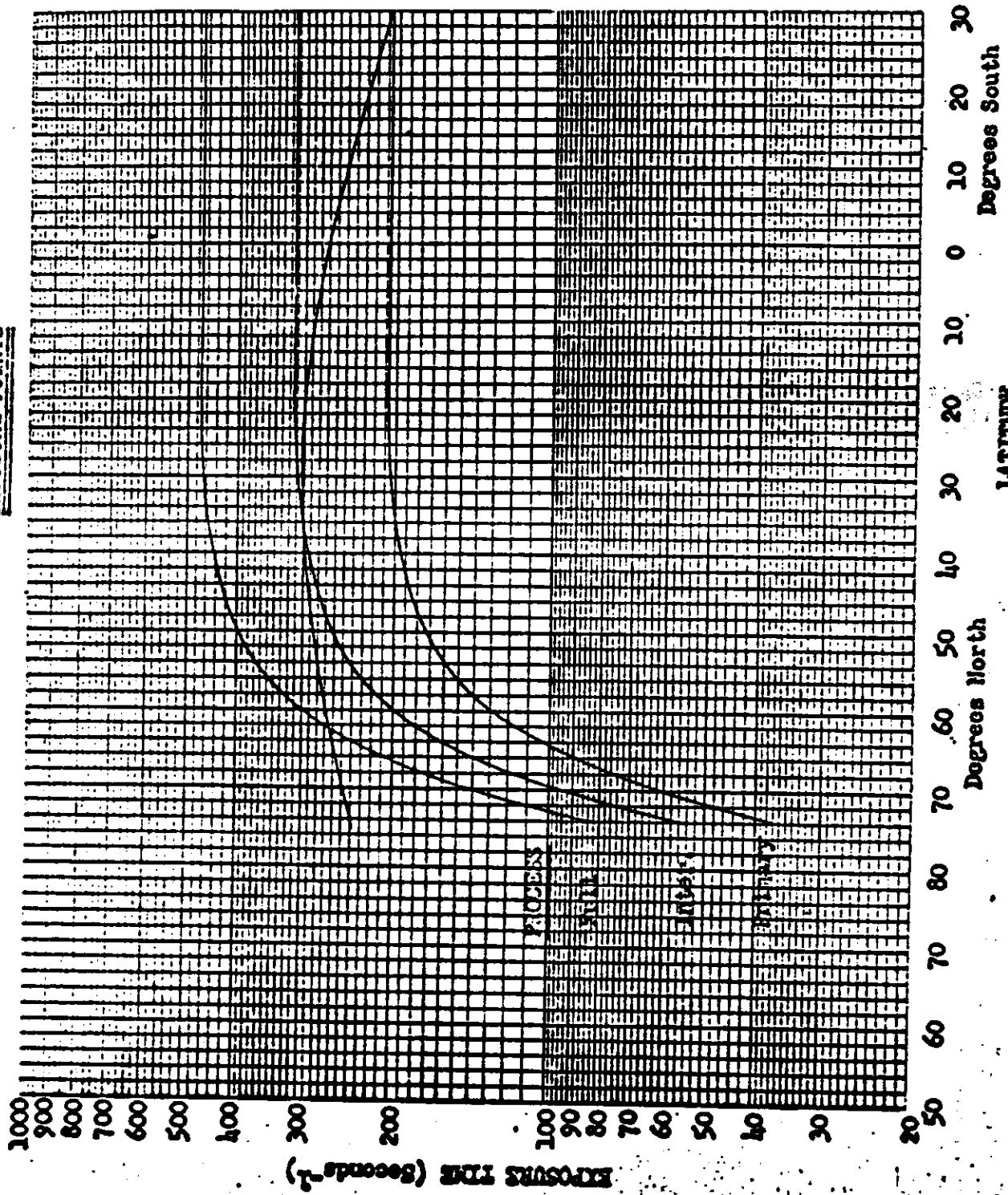
Launch Time: 2130 Z

0.225" width:

Filter Type: Wratten 21

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FIGURE 8-4



ΕΛΛΗΝΟΦΩΝΙΑ

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The camera slit and filter combination selected for Mission 1013 produced an exposure that was somewhat greater than selected for previous missions. The predominant factor that leads to this choice was the extensive camera operations that were programmed near the solar terminator.

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SECTION 9

DIFFUSE DENSITY MEASUREMENTS

Table 9-1 lists mission data supplied by AFSPPPL. This data includes the visual Reciprocal Edge Spread (RES) values, the area on the format in which the value was obtained and the general characteristics of the edge as shown on the data key page. The densitometric measurements of the base plus fog, minimum and maximum terrain densities and the maximum cloud densities are also listed with other general data such as solar elevation, altitude and overlap.

The columns are arranged in the following order:

<u>COLUMN NUMBER</u>	<u>HEADING</u>	<u>DATA</u>
1		Ascending or Descending pass
2-4	Pas Nbr	Pass Number
5		FWD or AFT camera
6-8	Frm Nbr	Frame Number
9-17	Area 1 RES	RES data in area 1
9-11	WWW	With flight RES value
12-14	AAA	Across flight RES value
15	S	Subject - see key
16	T	Terrain - see key
17	Q	Qualifiers - see key
18-26	Area 2 RES	RES data in area 2
27-35	Area 3 RES	RES data in area 3
36-44	Area 4 RES	RES data in area 4
45-53	Area 5 RES	RES data in area 5
54-56	D min	Terrain minimum density
57-59	D max	Terrain maximum density
60-62	D B+F	Base plus fog density
63-65	LIM max	Cloud maximum density

<u>COLUMN NUMBER</u>	<u>HEADING</u>	<u>DATA</u>
66-68	LAT.	Latitude
68	T.	0 = North, 1 = South
69-71	Sun Elevation	Solar Elevation
73-74	CLD	Percent cloud cover
75-76	OL	Percent overlap

The data key for the listings of the "Subject", "Terrain" and "Qualifiers" is shown below.

I SUBJECT

1. Buildings
2. Roads, runways
3. Tanks, A/C, other man-made
4. Non-cultural

II TERRAIN

1. Flat
2. Hilly
3. Mountains
4. Flat and snow
5. Hilly and snow
6. Mountains and snow

III EDGE QUALIFIERS

1. Clear
2. Snow
3. Hazy
4. Shadow
5. Snow and Haze
6. Snow and Shadow
7. Haze and Shadow
8. Snow, Haze and Shadow

1013-1

PAS FRMAREAL RESAREA2 RESAREA3 RESAREA4 RESAREA5 RES D D D LIM SUN

NBR NBRWWWAAA~~S~~TQWWHAA~~S~~TQWWHAA~~S~~TQWWHAA~~S~~TQWWHAA~~S~~TQMINMAXB+FMAXLATECLDOL

DC03F005			02021459N+1410099
D003F010	099085412		12016002021858N+1409805
D003F020			16621602022257N+1609099
D003F027			02022055N+1710099
DC04F005		072072431	08016502121065N+ 902599
DC04F015			05221001718063N+1001099
DC04F025	078085432		06220801517062N+1200507
U004F035			05219201518760N+1301099
DC04F045	104094431		05022201923158N+1501510
CC04F058			06721802222853N+1901099
DC04F068	085085411		05620002222451N+2001010
CC05F005	C85072422		14421801421457N+1509099
CC05F015			04621001400056N+1600099
D005F025	104104111		03019401400054N+1800005
CC05F035			04518201418053N+1901599
CC05F050104094112			08016001421245N+2609099
DC05F060			01422044N+2709899
CC05F070			01322042N+2810099
D005F080			01322240N+2910099
CC05F087			01322639N+3010099
CC08F007	094090111		05709001922059N+1406003
CCC8F017			04210601917857N+1501099
D008F027	099094111		04010401921056N+1701005
DC08F041			01318049N+2210099
CC08F051	078072412		05815901800048N+2300005
U008F061			06017101700046N+2400099
CC08F071	C78090431		06919801200044N+2500005
D008F081			07419001100043N+2700099
U008F091	104104431		07216501200041N+2800007
E008F101			06016101100040N+2900099
D008F111	C67072431		04016901100038N+3100005
CC09F001			02817901100036N+3200099
AC1			01200039N-4600009
DC09F005			01620854N+1910099
DC09F015			01721853N+2010099
DC09F025			01621051N+2110099
DC09F035	118104111		04205201221249N+3308099
CC09F045			04809201622247N+2409599
DG09F055			07414001821846N+2509599
D009F056		07207811205413401821645N+2509099	
DC09F066			01821344N+2705099
CC09F076			01822442N+2808099
DC10F005094082412			098149019239035+5507599
D010F015			07413601224105S+5507599
D010F024	075094412		08712401323906S+5508099
U016F005			01619437N+3204099
CC16F012			02012836N+3309099
DC19F005	099090431		07519002023057N+1606099
DC19F015			05620601621856N+1708599
DC19F025	111090431		05620701521854N+1805004

1013-1

PAS F R M A R E A 1 R E S A R E A 2 R E S A R E A 3 R E S A R E A 4 R E S A R E A 5 R E S D D D L I M S U N

NBR NBR WWWHAAASTQWWHAAASTQWWHAAASTQWWHAAASTQWWHAAASTQMINMAXB+FMAXLATECLDCL		
DC01F035		05816401722053N+2003099
DC19F047		02424051N+2104099
DC21F005	104099111	062192020251N+2106099
DC21F015		04218801420650N+2206099
DC21F025	094094111	03518201319848N+24003C5
DC21F035		05421401400043N+2800099
DC21F045	094082131	02819201400042N+2900005
DC21F055		04220401421640N+3008599
DC21F064	067063112	07214001420239N+3105099
DC21F071		08015401421038N+3205099
DC21F081	082082212	04514801622429N+3802099
DC21F091		06016502023228N+4007099
DC21F101	090085212	06218001923326N+4105099
DC21F111		01823025N+4209099
DC21F118	094099111	05013801823024N+4308506
DC23F005		01620156N+1610099
DC23F014	104094111	06510601612155N+1707599
DC23F024		06112201217053N+1804099
DC23F034		01017452N+2010099
DC23F044		01118850N+2110099
DC23F050	085094422	05811901518749N+2209506
DC23F060		07421601616047N+2301599
DC23F070	078078421	07815901620846N+2507507
DC23F080		07220001519744N+2605099
DC23F091	111094111	04611101100040N+3000003
DC23F101		04120801020039N+3107599
DC23F111	099104433	05421101021037N+3200504
DC23F121		04721401623536N+3300599
DC24F005	094090411	05209002015070N+ 403001
DC24F015		05908902018268N+ 508099
DC24F025	072085412	05811801917967N+ 708003
DC24F035		05717402019865N+ 903099
DC24F045	085082411	03617402019463N+1002505
DC24F055		05415102021061N+1209599
DC24F066		02021551N+2110099
DC24F074	078085111	07912001821850N+2209899
DC24F084		06909401419849N+2309099
DC24F094	134125111	04613001516847N+2401006
DC25F005	099090111	03806001918856N+1602099
DC25F015		06708401820855N+1709099
DC25F025		01821053N+1910099
DC30F005		01422023N+4306099
DC30F008	099085111	04810001421223N+4407099
DC30F018		01222021N+4501599
DC30F025		01222020N+4503599
DC31F005	094085111	03218001600032N+3700003
DC31F015		04615201919831N+3800299
DC31F020	094099131	03814001921830N+3800507
DC35F005	090085433	10822402321262N+1100508
DC35F015		08922002216061N+1200599

1013-1

PAS FRMAKEA1 RESAREA2 RESAREA3 RESAREA4 RESAREA5 RES O O O LIM SUN

NBR NBRWWHAAASTQWWHAAASTQWWHAAASTQWWHAAASTQWWHAAASTQMINMAXB+FMAXLATECLDOL

0035F025078078433		07222902321459N+1409099
0035F035		02320858N+1510099
D036F005	061065311	05613601800063N+1100006
DC36F015		05214001614062N+1201599
D036F025		09011601616060N+1409899
DC36FC35		09009441105213201618258N+1508099
CC36F043		09818501719257N+1608599
0036F053		07112001720053N+2010099
QC36F062	085094412	08317001621052N+2109099
0036F072		08414401721150N+2209999
C036F087	104094111	04911802223145N+2604007
DC36FG97		03311102120743N+2700299
CC36F107		02221742N+2910099
DC36F113		02223141N+2910099
CC37F005	067070421	05012402017265N+908008
C037F015		04714601918064N+1007099
D037F025		05115401918562N+1206099
CC37F035		05914601921060N+1309099
CC37F043		07807841105617001920259N+1409599
CC37F055		01522054N+1909999
DC37FC65	085090111	02118801319052N+2004009
CC37F075		03518901916951N+2200099
D037F085	094085431	03020401917649N+2301506
DC37F095		01922031N+3800099
DC37F105		01522729N+4010099
DC37F115		01222227N+4100099
DC37F125	085078432	06112201222926N+4208599
DC37F135		03310901320624N+4300099
DC37F145	082081412	06020801916023N+4400499
DC37F155		01923221N+4505099
DC37F165		01223020N+4708099
D038F005		01116357N+1610099
DC38F015	000090211	04409201118956N+1709507
D038F025		02615201116054N+1802099
C038F035	094094433	03516101100042N+2900008
DC38F045		09215601317441N+3004099
CC38F055		05213601321039N+3105508
CC39F005		04413202118564N+ 903507
D039F015		05413002121863N+1007099
DC39F025		07817002122061N+1109899
DC39F035		05404321204419502122960N+1309599
DC39F045		04514602422858N+1408599
D039F055		07614002223056N+1609899
CC39F065		07210802221855N+1709599
DC39F075		05711001921153N+1909899
DC39F085	049047211	04612601500047N+2500099
CC39F095		05014501417845N+2602099
DC39F105		07207231206512901519044N+2802510
D039F115		06216601518542N+2901599
0039F125	072075431	03014801512941N+3001007

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1013-1

PAS FERMAREA1 RESAREA2 RESAREA3 RESAREA4 RESAREA5 RES D D D LIM SUN

NBR NBRWWWA
AASTQWWWA
AASTQWWWA
AASTQWWWA
AASTQWWWA
AASTQMINMAXB+FMAXLATECLDCL

0039F135		067072111	02616001400039N+3100099
0C39F145		094099111	04018701400038N+3300010
DC40E005			04014802121059N+1405006
0040F015			03814802122558N+1503099
DC40F025		094099111	03416202121056N+1702510
DC40F035			04409202121354N+1804599
0041F005094104111			07410502020451N+2109099
DC41F015			06609902021450N+2309099
DC41F025			<u>1181111106008401821648N+2408599</u>
DC41F035			01621846N+2510099
DC47F005		085072211	02019001200037N+3300007
DC47F015			03817001200036N+3500099
DC47F025		111099111	03213801100034N+3600007
DC47F035			03016801300032N+3700099
0047F041	075078432		02415701100031N+3800099
DC51F005		082075433	03818001200063N+ 900006
DC51F015			05019601200062N+1000099
DC51F025		094090433	04020401200060N+1200008
DC51F035			04719601200059N+1400009
CC51F044			01619458N+1510099
DC52F005		070067421	03805801913669N+404099
CC52F015			04012101800068N+ 500099
DC52F025		078085421	07013801800066N+ 600006
DC52F035			06014101800064N+ 800099
CC52F045		085085421	06113901900063N+ 900005
DC52F055			04617101900059N+1400099
DC52F065		085078421	05216401900058N+1500005
DC52F075			07019101500056N+1600099
DC52F085		094090421	03819601300055N+1800006
DC52F095			04415001300053N+1900099
DC52F104		070067121	03916601300052N+2000006
CC03A005			01720660N+1310099
DC03A015			01820558N+1410099
CC03A021	055075411		12516001721257N+1509899
DC03A025			12019501821557N+1609099
DC03A028	082061413		09520501921156N+1608099
DC04A005		078075431	07216201816266N+ 800205
DC04A015			09020901800065N+ 900007
DC04A025		090085431	09621301800063N+1100005
DC04A035			09221201817461N+1200208
DC04AC45		063067431	08221501719259N+1400208
DC04AQ58			06613801821854N+1806599
CC04A068	090094413		05420201819253N+2000507
DC04A074			06016801820152N+2100707
CC05AG05			13821402021458N+1409899
DC05AG07		082078411	11818501921058N+1409006
CC05A015			09522002020457N+1503007
DC05A025		070067421	05419801800055N+1700006
CC05A035			06219001800054N+1800006
DC05AG41		082075121	05516801921653N+1900007

1013-1

~~PAS FRMAREAL RESAREA2 RESAREA3 RESAREA4 RESAREAS RES D D O LIM SUN~~

NBR NBRWWHAAASTQWWHAAASTQWWHAAASTQWWHAAASTQMINMAXB+FMAXLATECLDOL

DC05A050		01921846N+2510099
DC05A060	075072411	07215001922044N+2707599
DC05A070		01822143N+2810099
DC05A080		01920841N+2910099
DC05A090		02022640N+3010099
DC08AC05		01818460N+1310099
CC08A015	094104111	05908501616558N+1402506
DC08A025		04412201616656N+1601599
CC08AC32	094094111	04009401619055N+1701006
CC08A042		01720850N+2210099
DC08A052		01721249N+2310099
CC08A054085090111		09813101621048N+2308599
CC08A064		06013601600047N+2400099
CC08AC74	070067411	09720801800045N+2600007
CC08A084		07521001900044N+2700099
CC08A094	049063411	09617401200042N+2800006
DC08A104		05216001200040N+2900099
CC08A114	072065421	04917601100039N+3100007
CC08A127		03019001500037N+3200099
AC09A005		01500038N-4700099
CC09AC05		01017055N+1810099
CC09A015		00715854N+1910099
DC09A025		00815952N+2010099
CC09AC35		08508211204311400814750N+2209899
DC09A045		06911401621749N+2309099
DC09AC55	104099112	05609101621847N+2409599
CC09A065		01521745N+2507599
CC09A075		01622044N+2705099
DC10A005	C72067422	10617501622802S+5406099
CC10AC15		11016401623204S+5405099
DC10A025	070070122	10015501623305S+5506010
CC16AC08		01621638N+3206099
CC19AC05		01521058N+1506099
DC19A015	075078412	09019501621157N+1604007
CC19A025		07818901621155N+1809099
CC19A035	078085422	08521801521853N+1902007
CC19A045		08616501521051N+2007599
DC21A005	094099421	05218201820352N+2000504
CC21A015		05416001618651N+2109099
DC21A024	C90082421	05614601619249N+2205007
CC21A034		06815801700044N+2700099
CC21AC44	094090111	05616601600043N+2800007
CC21A054		03815601600041N+2900099
CC21A064	078082111	07015201700039N+3100008
DC21A074		01621C38N+3209999
DC21AC84	078075112	08120001920630N+3807506
CC21AC94		06015601821829N+3907599
CC21A104	C85085111	06816201821227N+4003007
CC21A114		09016001821826N+4209099
CC21A119063059411		07312601822225N+4208099

1013-1

PAS FRMAREA1 RESAREA2 RESAREA3 RESAREA4 RESAREA5 RES D D D LIM SUN

NBR NBRWWHAAASTQWHWAAASTQWHWAAASTQWHWAAASTQWHWAAASTQMINMAXB+FMAXLATECLDOL

DC23A005		01917857N+1510099
DC23A015		01819156N+1610099
DC23A020		07207011205819801817855N+1708099
DC23A030		08413801815253N+1801099
CC23A040		01820852N+2010099
DC23A050		01820250N+2110099
DC23A055	063057412	07311101820849N+2209599
CC23A065		05513601920847N+2404599
DC23A075	072067411	05412001819646N+2505099
DC23A082		10814601817845N+2607599
DC23A092	094078111	04218601800041N+3000004
CC23A102		04420001821240N+3101099
DC23A112	094085411	06413001822438N+3209599
CC23A121		06021801822037N+3304099
DC24A005067070411		03206401812271N+302599
DC24A015		02806401815069N+406599
CC24A025	082085411	04010001816068N+605099
DC24A035		05611501717566N+809599
CC24A045	085085411	03816301820664N+901504
CC24A055		05815001817562N+1108599
DC24A067		01520052N+2010099
DC24A077		01219451N+2110099
CC24A080		01220151N+2109806
DC24A090		01821249N+2308599
CC24A099090085111		06911402121248N+2407099
CC25A005	090094111	03508502017457N+1500299
DC25A015		04508002019456N+1700408
CC25A025		01920554N+1810099
CC25A030		02021254N+1910099
DQ30A005		02122424N+4201599
CC30A015	078067112	07216402023623N+4304099
CC30A025		02122221N+4401599
DC31A005	099104111	04420002118033N+3600204
CC31A015		052194020 32N+37 99
DC31A020		04420802122631N+3700508
DC35A005	045052432	08017501917063N+1007099
C035A015		070204019 62N+1100099
C035A025	052055432	075210019 60N+1300018
DC35A035		01919359N+1408099
CC36A005	057049421	072190020 63N+1000010
D036A015		054152020 62N+1100099
D036A025049054411		09414202016860N+1308099
DC36A035		12016802017059N+1409599
D036A044		02018058N+1509899
D036A055		10619602020053N+1906099
C036A065065		01820252N+2010099
DC36A074	067055411	05818201419250N+2206099
L036A084		06211001821247N+2508099
CC36A094	085070111	05810002021445N+2602010
C036A104		03411402020844N+2800599

1013-1

PAS FFMAREAL RESAREA2 RESAREA3 RESAREA4 RESAREA5 RES D D D LIM SUN

NBR NBRWWWAAA~~STQWWAAA~~STQWWAAA~~STQWWAAA~~STQMINMAXB+FMAXLATECLDOL~~~~~~~~

DC36A114		02020842N+2910099
D037A005		03604343105816201915065N+ 801009
D037A015		05818401913064N+ 901599
DC37A025	078067421	06016401916462N+1102011
D037A035		06614502016261N+1204599
DC37A045		03604142206813001917659N+1407099
C037A055		01920454N+1809899
D037A065		10817402019053N+1906599
CC37A075		046190020 51N+2100099
DC37A085	028032431	04418501817250N+2200210
DC37A095		01821831N+3810099
CC37A105		01823029N+3910099
C037A115		01221828N+411C099
D037A125		06410001021226N+4209899
CC37A135	041038422	07217501622425N+4307007
CC37A145		15617802020324N+4401599
D037A155		05615202013022N+4500199
CC37A165		01921221N+4607099
D037A167		01822221N+4708099
D038A005		01518857N+1510099
CC38A015		01318056N+1710099
CC38A022		06917901418455N+1809099
D038A032		051228020 54N+1900099
DC38A042	052063433	068207018 54N+1900099
D038A052		01919554N+1999999
D039A005	067085412	04514001811065N+ 802015
CC39A015		03614401819264N+ 903599
DC39A025		01818662N+1010099
CC39A035		01820161N+1210099
CC39A039	118099411	05612002019960N+1208599
C039A049		05016201819759N+1405099
D039A059	078067412	06816801821257N+1508513
CC39A069		01821655N+1710099
D039A084	104094411	07014801800047N+2400010
CC39A094		08915201800046N+2500099
DC39A104	094099111	10018001917844N+2602010
CC39A114		06817401819043N+2702099
CC39A124	072063411	08216401818442N+2901012
CC39A134		04616201600040N+3000099
DC39A144	072065111	03815901400039N+31000C8
D039A152		03016301200038N+3200099
DC40AC05078085111		04209201919060N+1308099
DC40A015		04112601921059N+1404599
CC40AC25085090111		03608201817457N+1502511
DC40A035		04409401819855N+1705099
DC40A039	070065111	05213901919655N+1707515
C041A005		01821051N+2110099
CC41A011		08209011105417001820251N+2109099
DC41A021		08012001721049N+2309599
CC41A031		09009011108211801821448N+2408599

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1013-1

PAS F R M A R E A 1 R E S A R E A 2 R E S A R E A 3 R E S A R E A 4 R E S A R E A 5 R E S D D D L I M S U N

NBR N B R W W W A A A S T Q W W H A A A S T Q W W H A A A S T Q W W H A A A S T Q W M I N M A X B + F M A X L A T E C L D O L

DC41A041		01722047N+2510099
DC47A005	094090431	05218601820037N+3202507
DC47A015		05020401700036N+3400099
D047A025	082078431	06219001800035N+3500007
DC47A035		07519801700033N+3600099
DC47A043	075072431	07020001700032N+3700008
D051A005	075072431	07817001916464N+ 901007
D051A015		06220001900063N+1000099
DC51A025	041047432	09621201900061N+1200010
D051A035		09622001900060N+1300099
DC51A044	C39045432	08222001918259N+1401599
DC52A005	063061421	03509201700070N+ 300004
D052A015		04208301700069N+ 4000099
DC52A025	061054431	04010001800067N+ 600010
DC52A035		05011401800066N+ 700099
D052A045	052054431	05012201700064N+ 900010
DC52A055		03715901700061N+1200099
DC52A065	050048431	03413801700059N+1400011
D052A075		06008501800057N+1500099
DC52A085	067070431	07820001600056N+1700010
DC52A095		05709901700054N+1800099
DC52A105	078085431	06220001700053N+1900011
DC52A107		04519001600052N+2000007
8 F		N+-0

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TABLE 9-1

The diffuse density measurements made by AFSPP were computer sorted at A/P to permit analysis of the density ranges encountered at the three processing levels. A study of sorting techniques showed that no absolute method was available to separate the density values as the accuracy of the Processing History published by [REDACTED] appears rather low and processing transition phases are not accounted for. The sorting technique selected uses the base plus fog density values where measurements up to 0.09 density are considered as having received Primary processing, 0.10 to 0.17 as Intermediate and above 0.17 density as Full. The percentage of original negative that was processed at each level, based on the computer sort, is tabulated below with the predicted and reported processing percentages.

<u>Mission</u>	<u>Camera</u>		<u>Primary</u>	<u>Intermediate</u>	<u>Full</u>
1013-1	FWD	Predicted	0	64	36
		Reported	0	42	58
		Computed	0	55	45
1013-1	AFT	Predicted	0	63	37
		Reported	2	7	91
		Computed	0	33	67

The correlation of the reported and computed percentages at the three processing levels is very good for Mission 1013-1 and is more consistent than the normal mission values.

The tabulations of density frequency distributions for Mission 1013-1 are shown in Tables 9-2 and 9-3. The graphical presentation of the density distribution are computer plotted in Figures 9-1 through 9-21. Analysis of these plots and the associated mean and median density values show that the Master camera density values were essentially the same as the Slave camera and the density ranges of both cameras are within the nominal range.

Table 9-4 shows the distribution of the minimum terrain density measurements that are within the outside of the desired control range of 0.40 to 0.90 density. The percentage of values below 0.30 density is approximately 3%.

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An extensive study is in process to ascertain the inter-relationship of the conditions of illumination, resulting densities and exposure-processing parameters. The apparent low density of the Master camera photography is of some concern however the high information content rating for the camera indicates that this approach to exposure is valid. It is also noteworthy that no information was lost in shadow areas.

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ФОР-СВЕРД

MISSION • 1013-1 • INSTRUMENT • FRWD 01/18/65 DENSITY FREQ DISTR

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TABLE 9-2

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-TOP SECRET-

TABLE 9-2

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MISSION * 1013-1 * INSTRUMENT * FRW0 01/18/65 DENSITY FREQ DISTR

DENSITY VALUE	PRIMARY MIN MAX LIM	INTERMEDIATE MIN MAX LIM	FULL MIN MAX LIM	ALL LEVELS MIN MAX LIM
1.01	00	00	00	00
1.02	00	00	00	00
1.03	00	00	00	00
1.04	00	00	00	00
1.05	00	00	00	00
1.06	00	00	00	00
1.07	00	00	00	00
1.08	00	00	00	00
1.09	00	00	00	00
1.10	00	00	00	00
1.11	00	00	00	00
1.12	00	00	00	00
1.13	00	00	00	00
1.14	00	00	00	00
1.15	00	00	00	00
1.16	00	00	00	00
1.17	00	00	00	00
1.18	00	00	00	00
1.19	00	00	00	00
1.20	00	00	00	00
1.21	00	00	00	00
1.22	00	00	00	00
1.23	00	00	00	00
1.24	00	00	00	00
1.25	00	00	00	00
1.26	00	00	00	00
1.27	00	00	00	00
1.28	00	00	00	00
1.29	00	00	00	00
1.30	00	00	00	00
1.31	00	00	00	00
1.32	00	00	00	00
1.33	00	00	00	00
1.34	00	00	00	00
1.35	00	00	00	00
1.36	00	00	00	00
1.37	00	00	00	00
1.38	00	00	00	00
1.39	00	00	00	00
1.40	00	00	00	00
1.41	00	00	00	00
1.42	00	00	00	00
1.43	00	00	00	00
1.44	00	00	00	00
1.45	00	00	00	00
1.46	00	00	00	00
1.47	00	00	00	00
1.48	00	00	00	00
1.49	00	00	00	00
1.50	00	00	00	00
SUBTOTAL	00	00	00	00

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TABLE 9-2

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MISSION • 1013-1

• INSTRUMENT • FRWD

01/18/65

DENSITY FREQ DISTR

-TOP SECRET

TABLE 9-2

~~TOP SECRET~~

MISSION • 1013-1

* INSTRUMENT * FRWD.

01/18/65

DENSITY FREQ DISTR

DENSITY VALUE	PRIMARY MIN MAX LIM	INTERMEDIATE MIN MAX LIM	FULL MIN MAX LIM	ALL LEVELS MIN MAX LIM
2.01	0	0	0	0
2.02	0	0	0	0
2.03	0	0	0	0
2.04	0	0	0	0
2.05	0	0	0	0
2.06	0	0	0	0
2.07	0	0	0	0
2.08	0	0	0	0
2.09	0	0	0	0
2.10	0	0	0	0
2.11	0	0	0	0
2.12	0	0	0	0
2.13	0	0	0	0
2.14	0	0	0	0
2.15	0	0	0	0
2.16	0	0	0	0
2.17	0	0	0	0
2.18	0	0	0	0
2.19	0	0	0	0
2.20	0	0	0	0
2.21	0	0	0	0
2.22	0	0	0	0
2.23	0	0	0	0
2.24	0	0	0	0
2.25	0	0	0	0
2.26	0	0	0	0
2.27	0	0	0	0
2.28	0	0	0	0
2.29	0	0	0	0
2.30	0	0	0	0
2.31	0	0	0	0
2.32	0	0	0	0
2.33	0	0	0	0
2.34	0	0	0	0
2.35	0	0	0	0
2.36	0	0	0	0
2.37	0	0	0	0
2.38	0	0	0	0
2.39	0	0	0	0
2.40	0	0	0	0
2.41	0	0	0	0
2.42	0	0	0	0
2.43	0	0	0	0
2.44	0	0	0	0
2.45	0	0	0	0
2.46	0	0	0	0
2.47	0	0	0	0
2.48	0	0	0	0
2.49	0	0	0	0
2.50	0	0	0	0
SUBTOTAL		13	39	53

-TOP SECRET-

TABLE 9-2

~~TOP SECRET~~

MISSION • 1013-1 • INSTRUMENT • FRWD 01/18/65 DENSITY FREQ DISTR

DENSITY VALUE	PRIMARY MIN MAX LIM	INTERMEDIATE MIN MAX LIM	FULL MIN MAX LIM	ALL LEVELS MIN MAX LIM
2.51	0 0 0	0 0 0	0 0 0	0 0 0
2.52	0 0 0	0 0 0	0 0 0	0 0 0
2.53	0 0 0	0 0 0	0 0 0	0 0 0
2.54	0 0 0	0 0 0	0 0 0	0 0 0
2.55	0 0 0	0 0 0	0 0 0	0 0 0
2.56	0 0 0	0 0 0	0 0 0	0 0 0
2.57	0 0 0	0 0 0	0 0 0	0 0 0
2.58	0 0 0	0 0 0	0 0 0	0 0 0
2.59	0 0 0	0 0 0	0 0 0	0 0 0
2.60	0 0 0	0 0 0	0 0 0	0 0 0
2.61	0 0 0	0 0 0	0 0 0	0 0 0
2.62	0 0 0	0 0 0	0 0 0	0 0 0
2.63	0 0 0	0 0 0	0 0 0	0 0 0
2.64	0 0 0	0 0 0	0 0 0	0 0 0
2.65	0 0 0	0 0 0	0 0 0	0 0 0
2.66	0 0 0	0 0 0	0 0 0	0 0 0
2.67	0 0 0	0 0 0	0 0 0	0 0 0
2.68	0 0 0	0 0 0	0 0 0	0 0 0
2.69	0 0 0	0 0 0	0 0 0	0 0 0
2.70	0 0 0	0 0 0	0 0 0	0 0 0
SUBTOTAL	0 0 0	0 0 0	0 0 0	0 0 0
TOTAL	0 0 0	79 79 71	65 65 71	144 144 142

MISSION 1013-1 INSTR - FRWD 01/18/65 PROCESSING AND EXPOSURE ANALYSIS

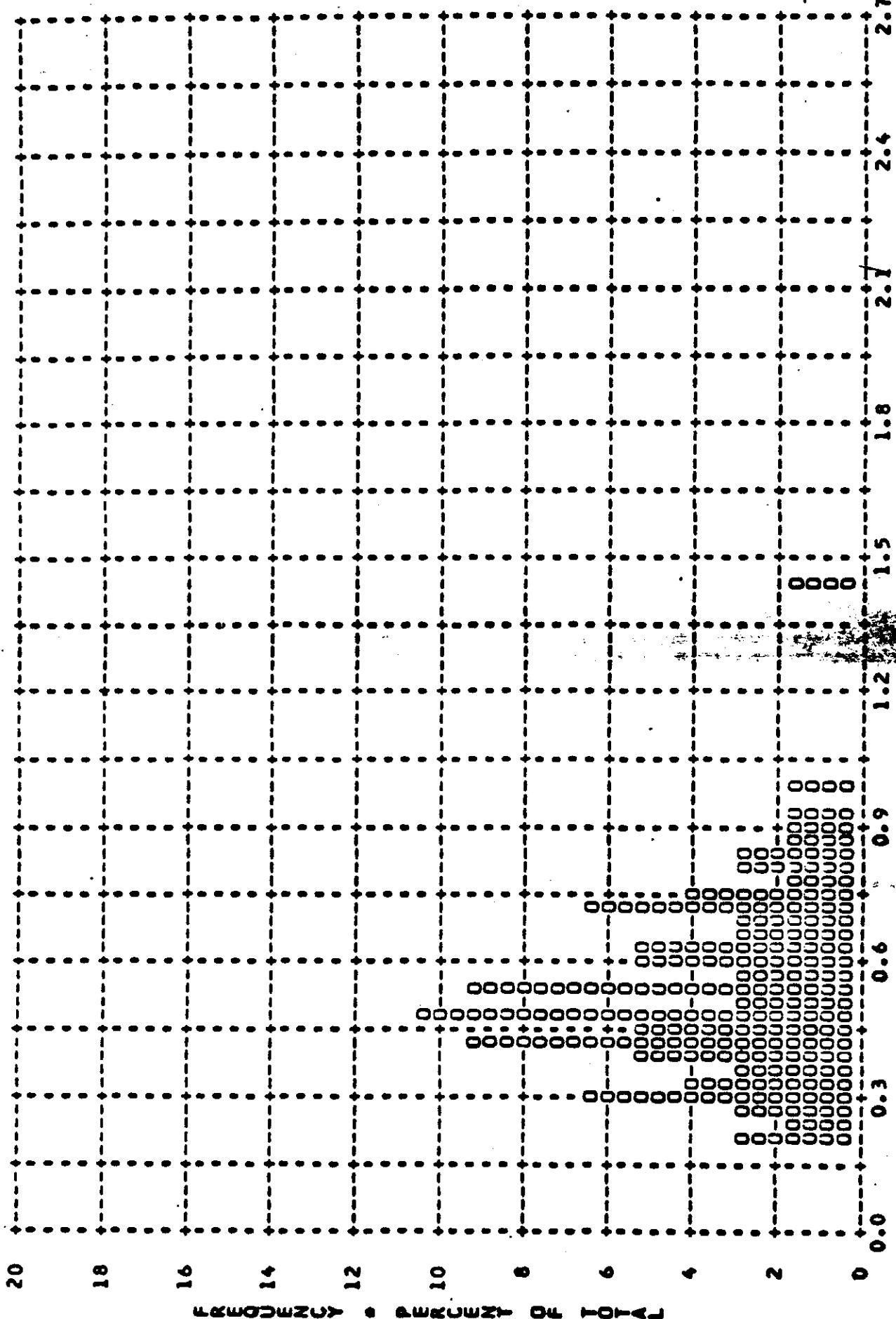
PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC
INTERMEDIATE	79	1 PC	23 PC	72 PC	3 PC	1 PC
FULL	65	14 PC	0 PC	80 PC	6 PC	0 PC
ALL LEVELS	144	7 PC	13 PC	76 PC	4 PC	1 PC
PROCESS LEVEL	BASE + FOG	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0.01-0.09	0.01-0.13	0.14-0.39	0.40-0.90	-----	0.91 AND UF
INTERMED	0.10-0.17	0.01-0.20	0.21-0.39	0.40-0.90	0.91-1.34	1.35 AND UF
FULL	0.18 AND UP	0.01-0.39	-----	0.40-0.90	0.91-1.69	1.70 AND UF

~~TOP SECRET~~

TABLE 9-2

-TOP SECRET

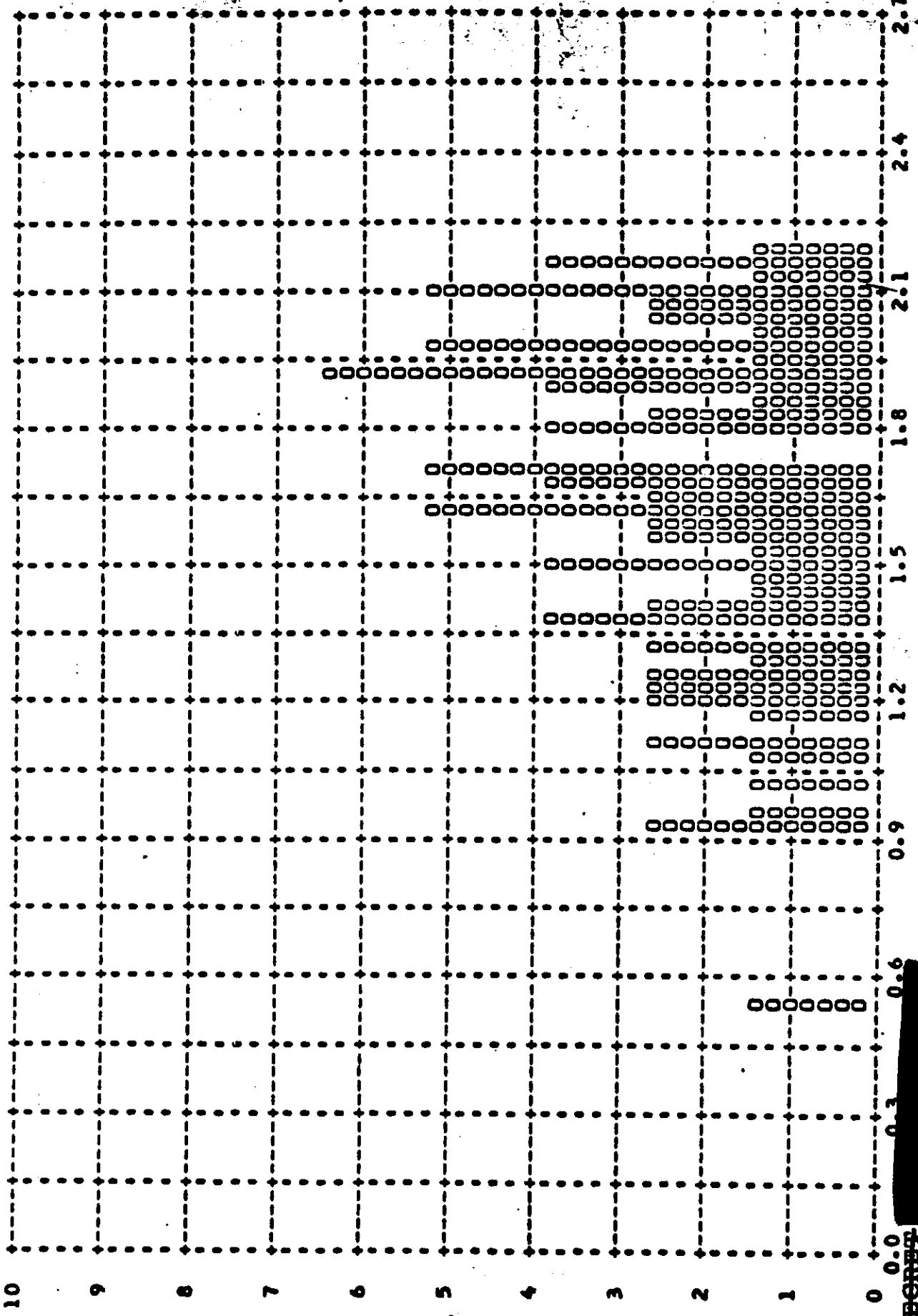
MISSION • 1013-1 • INSTR • FRWD • 01/18/65 PLOT OF U MIN • TERRAIN • PROCESSING • INTERMEDIATE
ARITH MEAN • 0.54 • MEDIAN • 0.50 • STD DEV • 0.21 • RANGE • 0.20 TU 1.44 WITH 79 SAMPLES

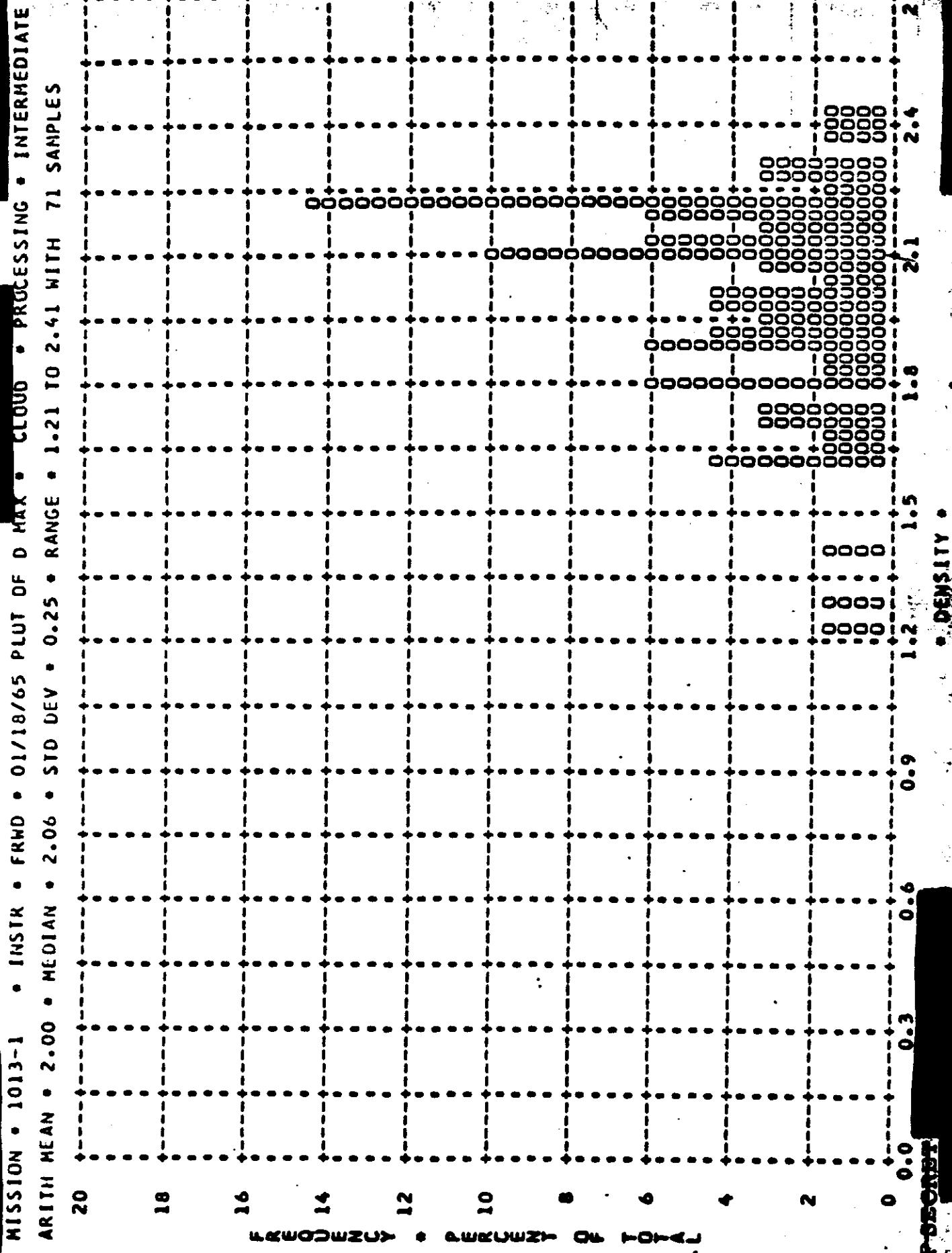


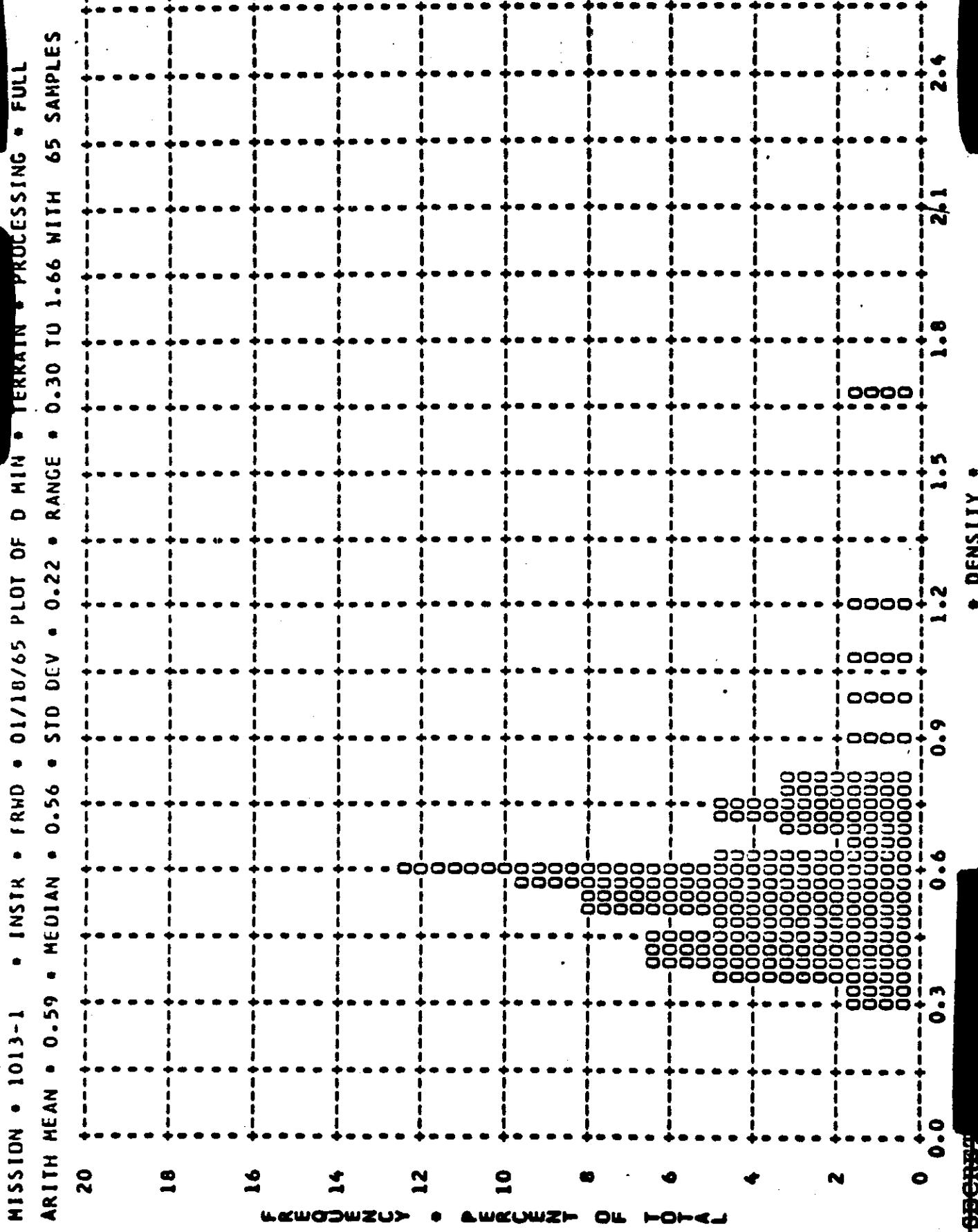
-TOP SECRET

MISSION • 1013-1 • INSTR • FRWD • 01/18/65 PLUT OF D MAX • TERRAIN • PROCESSING • INTERMEDIATE

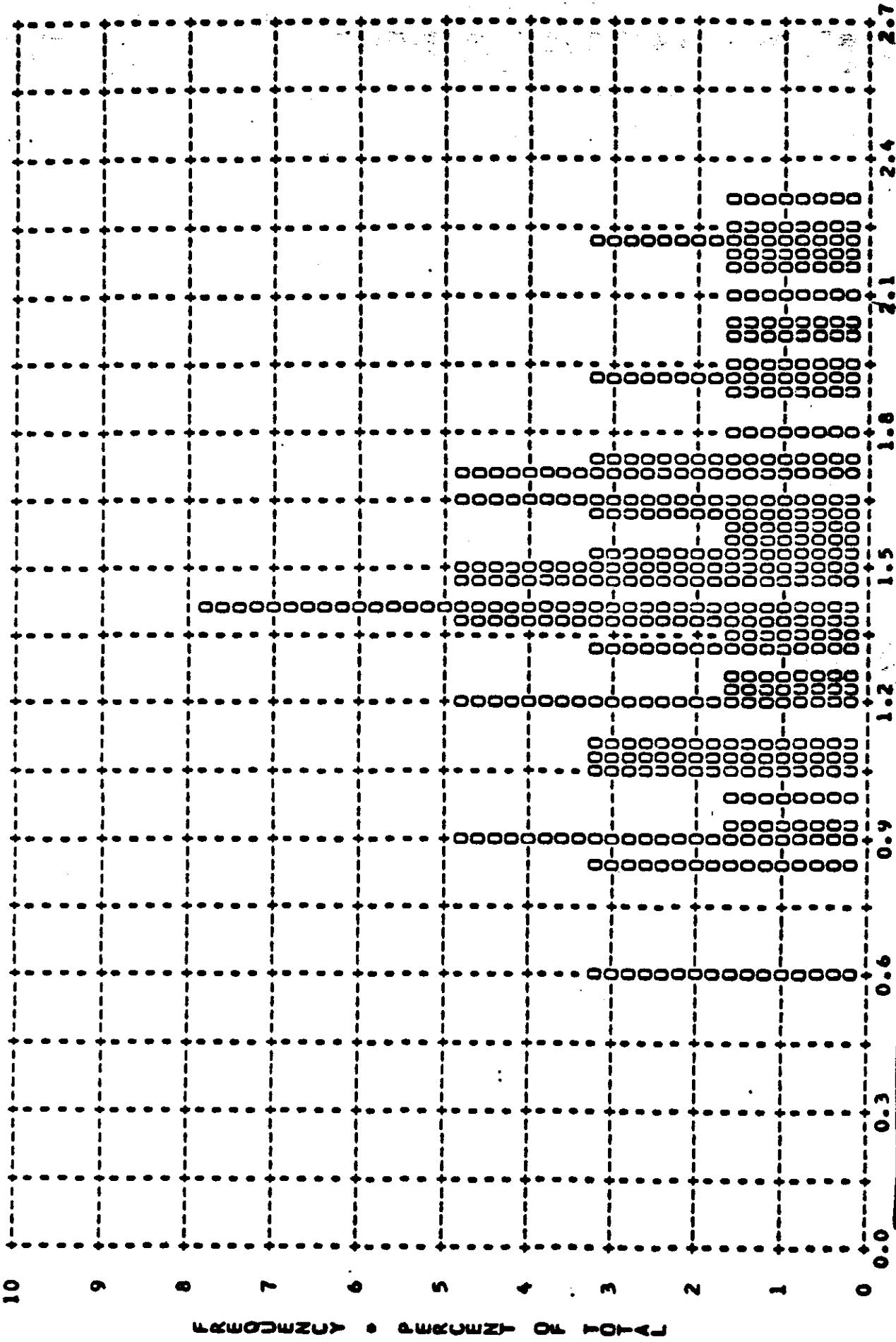
AKITH MEAN • 1.63 • MEDIAN • 1.66 • STD DEV • 0.37 • RANGE • 0.52 TO 2.18 WITH 79 SAMPLES





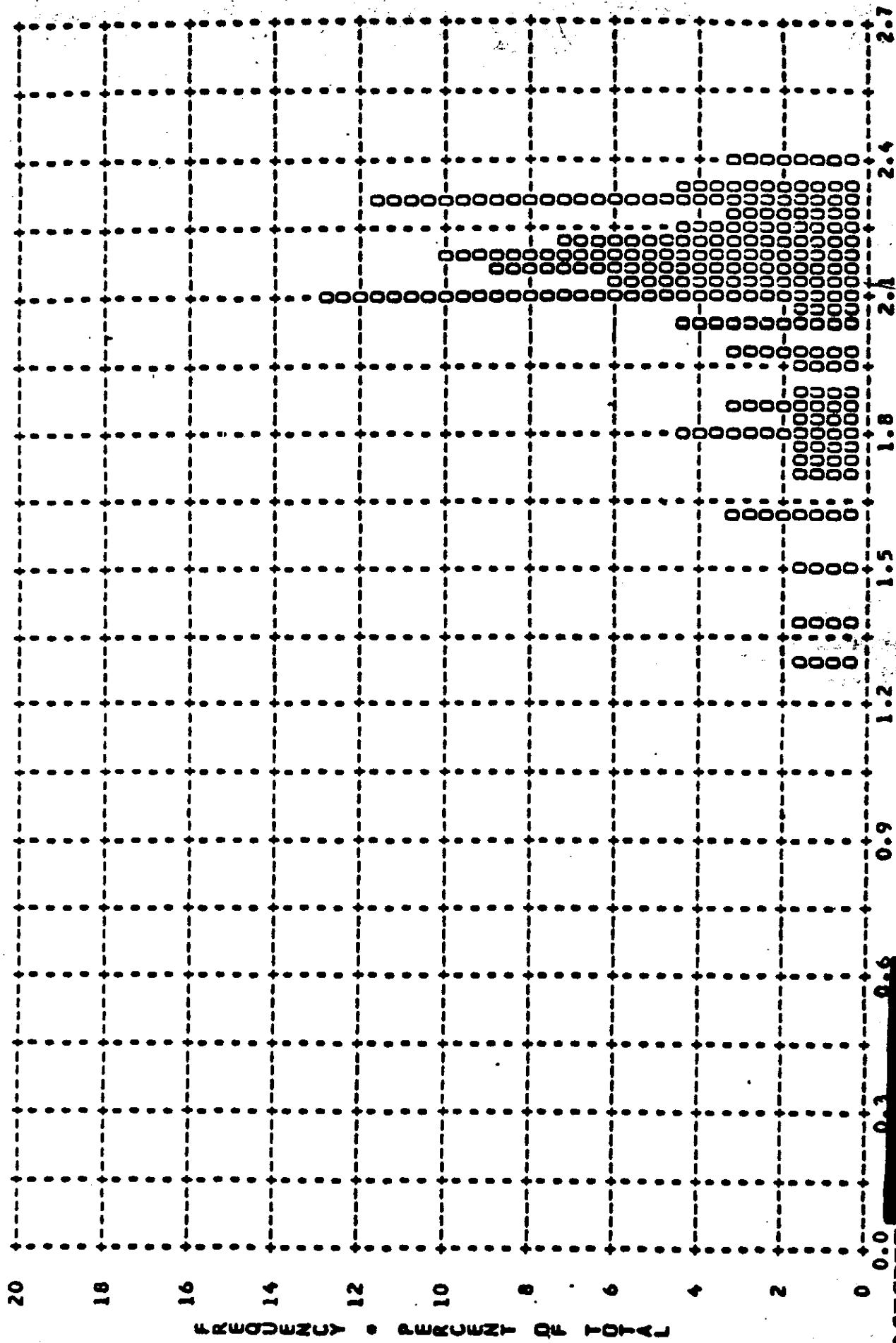


MISSION • 1013-1 • INSTR • FRWD • 01/18/65 PLOT OF D MAX - TERRAIN • PROCESSING • FULL
ARITH MEAN • 1.47 • MEDIAN • 1.46 • STD DEV • 0.41 • RANGE • 0.58 TO 2.29 WITH 65 SAMPLES



-TOP SECRET

MISSION • 1013-1 • INSTR • FRWD • 01/18/65 PLOT OF U MAX • CLOUD • PROCESSING • FULL
ARITH MEAN • 2.07 • MEDIAN • 2.14 • STD DEV • 0.24 • RANGE • 1.28 TO 2.40 WITH 71 SAMPLES

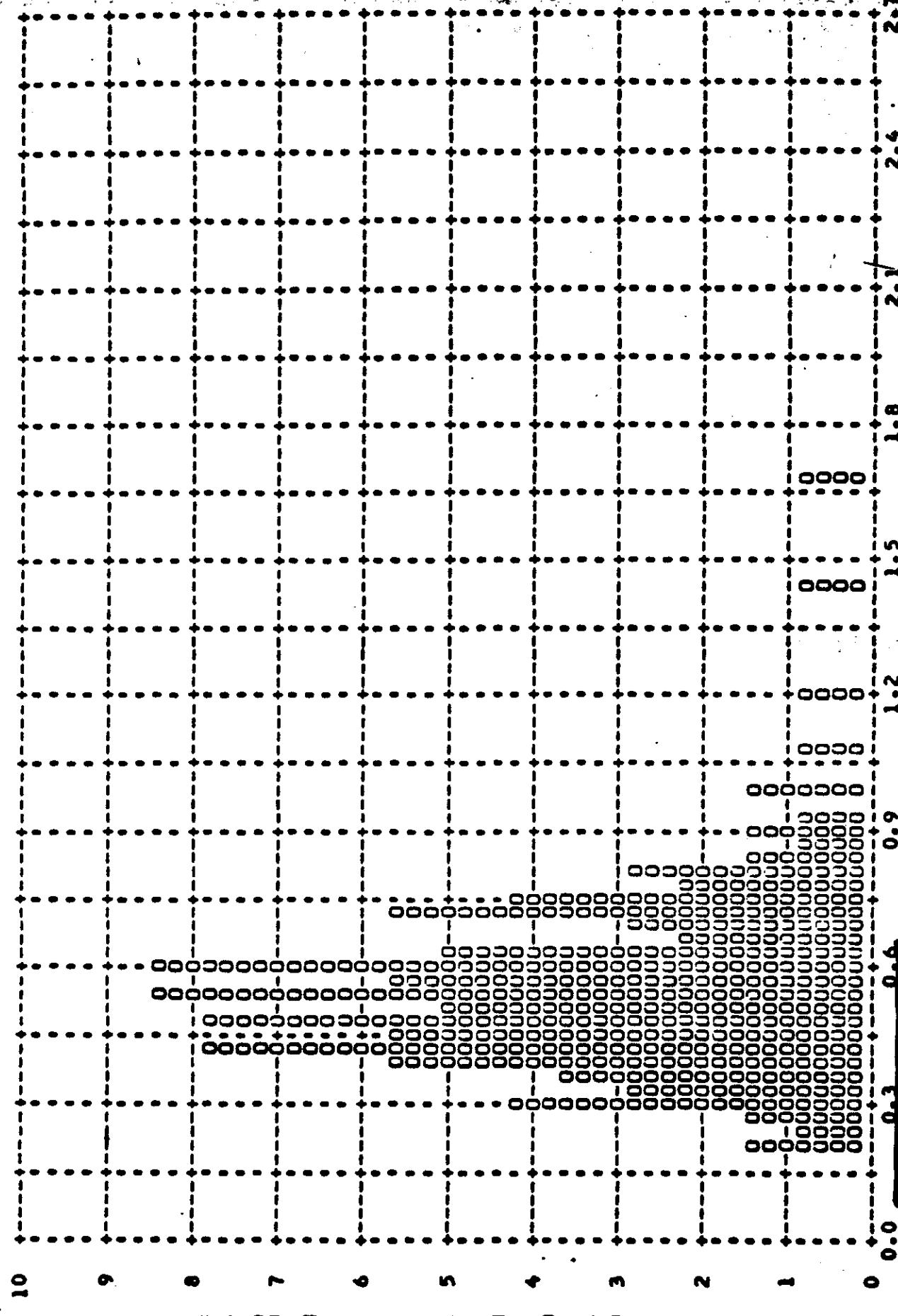


-TOP SECRET

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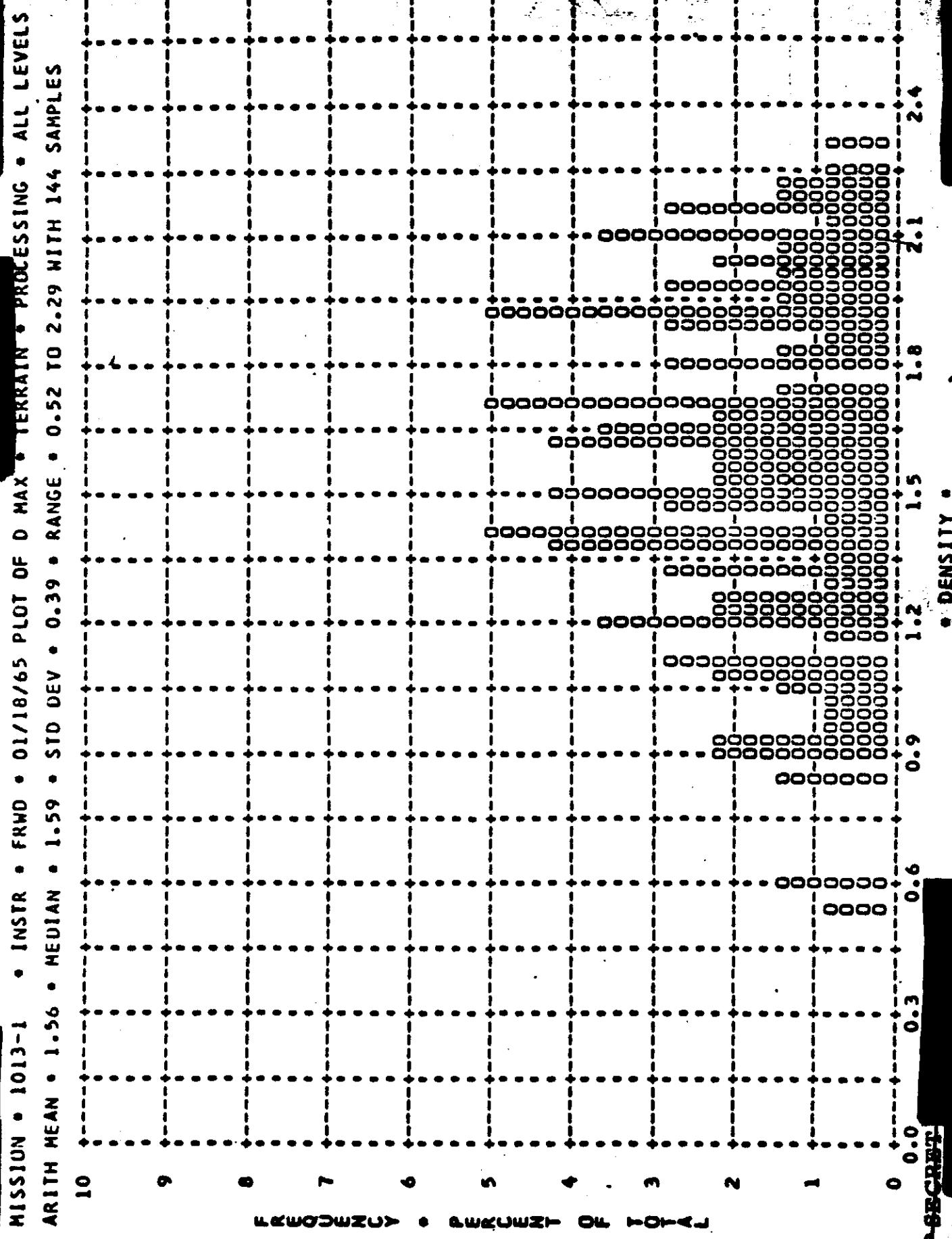
MISSION • 1013-1 • INSTR • FRWD • 01/18/65 PLOT OF D MIN • TERRAIN • PROCESSING • ALL LEVELS

ARITH MEAN • 0.56 • MEDIAN • 0.54 • STD DEV • 0.22 • RANGE • 0.20 TO 1.66 WITH 144 SAMPLES



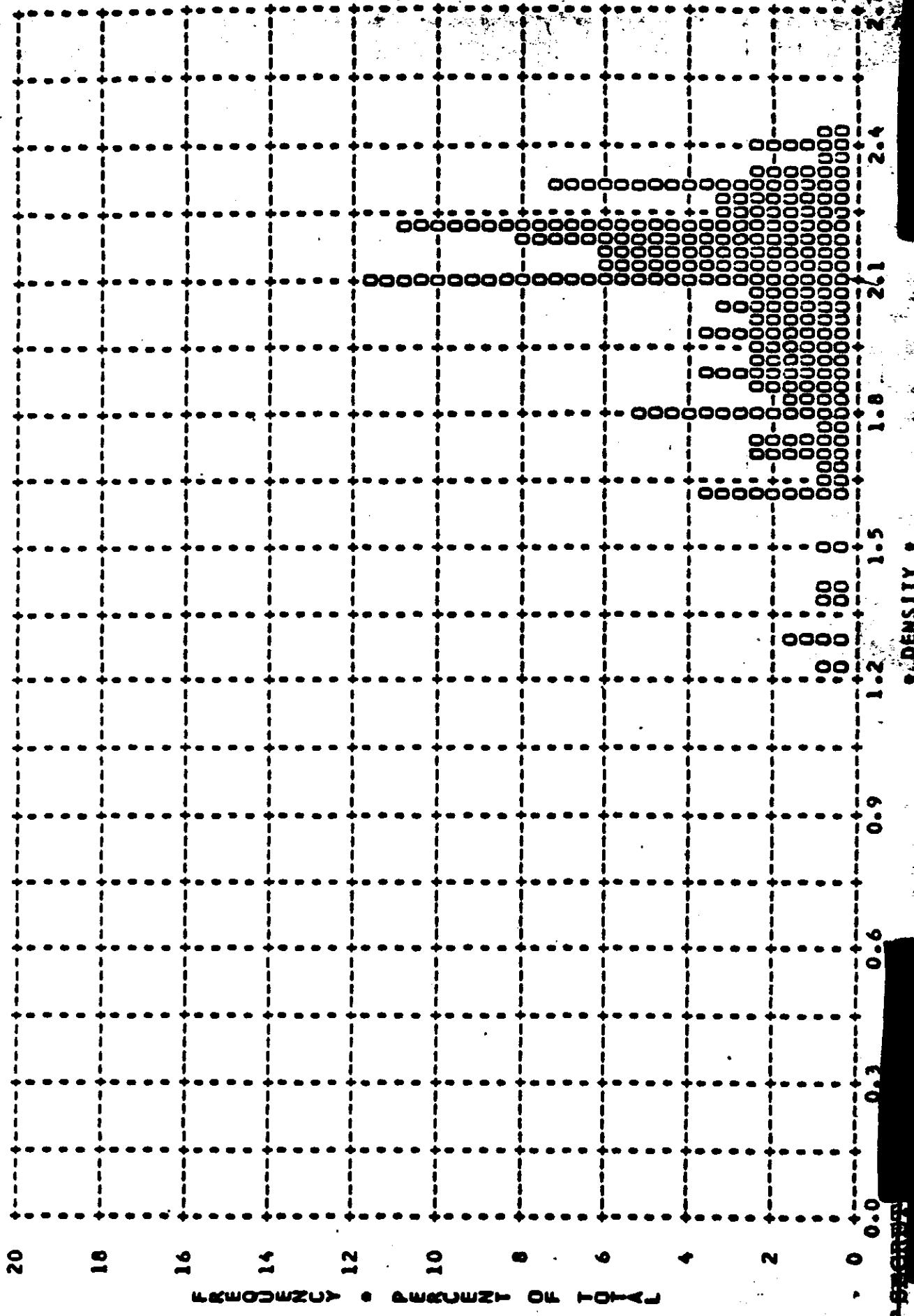
NUMBER OF POINTS • DENSITY OF POINTS

TOP SECRET



~~TOP SECRET~~

MISSION • 1013-1 • INSTR • FRWD • 01/18/65 PLOT OF D MAX • GROUPS • PROCESSING • ALL LEVELS
ARITH MEAN • 2.03 • MEDIAN • 2.10 • STD DEV • 0.25 • RANGE • 1.21 TO 2.41 WITH 142 SAMPLES



~~TOP SECRET~~

MISSION • 1013-1

• INSTRUMENT • AFT

01/18/6

DENSITY FREQ DISTR

TOP SECRET

TABLE 9-3

TOP SECRET

MISSION • 1013-1 • INSTRUMENT • AFT 01/18/65 DENSITY FREQ DISTR

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TABLE 9-3

-TOP SECRET-

MISSION • 1013-1

• INSTRUMENT • AFT

01/18/65

DENSITY FREQ DISTR

TABLE 9-3

-TOP SECRET-

MISSION • 1013-1 • INSTRUMENT • AFT 01/18/65 DENSITY FREQ DISTR

DENSITY VALUE	PRIMARY MIN	PRIMARY MAX	PRIMARY LIM	INTERMEDIATE MIN	INTERMEDIATE MAX	INTERMEDIATE LIM	FULL MIN	FULL MAX	FULL LIM	ALL LEVELS MIN	ALL LEVELS MAX	ALL LEVELS LIM
1.51	0	0	0	0	0	0	0	0	0	0	0	0
1.52	0	0	0	0	0	0	0	0	0	0	0	0
1.54	0	0	0	0	0	0	0	0	0	0	0	0
1.55	0	0	0	0	0	0	0	0	0	0	0	0
1.56	0	0	0	0	0	0	0	0	0	0	0	0
1.57	0	0	0	0	0	0	0	0	0	0	0	0
1.58	0	0	0	0	0	0	0	0	0	0	0	0
1.59	0	0	0	0	0	0	0	0	0	0	0	0
1.60	0	0	0	0	0	0	0	0	0	0	0	0
1.61	0	0	0	0	0	0	0	0	0	0	0	0
1.62	0	0	0	0	0	0	0	0	0	0	0	0
1.63	0	0	0	0	0	0	0	0	0	0	0	0
1.64	0	0	0	0	0	0	0	0	0	0	0	0
1.65	0	0	0	0	0	0	0	0	0	0	0	0
1.66	0	0	0	0	0	0	0	0	0	0	0	0
1.67	0	0	0	0	0	0	0	0	0	0	0	0
1.68	0	0	0	0	0	0	0	0	0	0	0	0
1.69	0	0	0	0	0	0	0	0	0	0	0	0
1.70	0	0	0	0	0	0	0	0	0	0	0	0
1.71	0	0	0	0	0	0	0	0	0	0	0	0
1.72	0	0	0	0	0	0	0	0	0	0	0	0
1.73	0	0	0	0	0	0	0	0	0	0	0	0
1.74	0	0	0	0	0	0	0	0	0	0	0	0
1.75	0	0	0	0	0	0	0	0	0	0	0	0
1.76	0	0	0	0	0	0	0	0	0	0	0	0
1.77	0	0	0	0	0	0	0	0	0	0	0	0
1.78	0	0	0	0	0	0	0	0	0	0	0	0
1.79	0	0	0	0	0	0	0	0	0	0	0	0
1.80	0	0	0	0	0	0	0	0	0	0	0	0
1.81	0	0	0	0	0	0	0	0	0	0	0	0
1.82	0	0	0	0	0	0	0	0	0	0	0	0
1.83	0	0	0	0	0	0	0	0	0	0	0	0
1.84	0	0	0	0	0	0	0	0	0	0	0	0
1.85	0	0	0	0	0	0	0	0	0	0	0	0
1.86	0	0	0	0	0	0	0	0	0	0	0	0
1.87	0	0	0	0	0	0	0	0	0	0	0	0
1.88	0	0	0	0	0	0	0	0	0	0	0	0
1.89	0	0	0	0	0	0	0	0	0	0	0	0
1.90	0	0	0	0	0	0	0	0	0	0	0	0
1.91	0	0	0	0	0	0	0	0	0	0	0	0
1.92	0	0	0	0	0	0	0	0	0	0	0	0
1.93	0	0	0	0	0	0	0	0	0	0	0	0
1.94	0	0	0	0	0	0	0	0	0	0	0	0
1.95	0	0	0	0	0	0	0	0	0	0	0	0
1.96	0	0	0	0	0	0	0	0	0	0	0	0
1.97	0	0	0	0	0	0	0	0	0	0	0	0
1.98	0	0	0	0	0	0	0	0	0	0	0	0
1.99	0	0	0	0	0	0	0	0	0	0	0	0
2.00	0	0	0	0	0	0	0	0	0	0	0	0
SUBTOTAL	0	0	0	0	0	0	0	28	14	1	43	41
										71		57

-TOP SECRET-

TABLE 9-3

~~TOP SECRET~~

MISSION • 1013-1

* INSTRUMENT * AFT

01/18/62

DENSITY FREQ DISTR

DENSITY VALUE	PRIMARY MIN	PRIMARY MAX	PRIMARY LIM	INTERMEDIATE MIN	INTERMEDIATE MAX	INTERMEDIATE LIM	FULL MIN	FULL MAX	FULL LIM	ALL LEVELS MIN	ALL LEVELS MAX	ALL LEVELS LIM
2.01	0	0	0	0	0	0	1	0	0	2	0	0
2.02	0	0	0	0	0	0	0	0	0	0	0	0
2.03	0	0	0	0	0	0	0	0	0	0	0	0
2.04	0	0	0	0	0	0	0	0	0	0	0	0
2.05	0	0	0	0	0	0	0	0	0	0	0	0
2.06	0	0	0	0	0	0	0	0	0	0	0	0
2.07	0	0	0	0	0	0	0	0	0	0	0	0
2.08	0	0	0	0	0	0	0	0	0	0	0	0
2.09	0	0	0	0	0	0	0	0	0	0	0	0
2.10	0	0	0	0	0	0	0	0	0	0	0	0
2.11	0	0	0	0	0	0	0	0	0	0	0	0
2.12	0	0	0	0	0	0	0	0	0	0	0	0
2.13	0	0	0	0	0	0	0	0	0	0	0	0
2.14	0	0	0	0	0	0	0	0	0	0	0	0
2.15	0	0	0	0	0	0	0	0	0	0	0	0
2.16	0	0	0	0	0	0	0	0	0	0	0	0
2.17	0	0	0	0	0	0	0	0	0	0	0	0
2.18	0	0	0	0	0	0	0	0	0	0	0	0
2.19	0	0	0	0	0	0	0	0	0	0	0	0
2.20	0	0	0	0	0	0	0	0	0	0	0	0
2.21	0	0	0	0	0	0	0	0	0	0	0	0
2.22	0	0	0	0	0	0	0	0	0	0	0	0
2.23	0	0	0	0	0	0	0	0	0	0	0	0
2.24	0	0	0	0	0	0	0	0	0	0	0	0
2.25	0	0	0	0	0	0	0	0	0	0	0	0
2.26	0	0	0	0	0	0	0	0	0	0	0	0
2.27	0	0	0	0	0	0	0	0	0	0	0	0
2.28	0	0	0	0	0	0	0	0	0	0	0	0
2.29	0	0	0	0	0	0	0	0	0	0	0	0
2.30	0	0	0	0	0	0	0	0	0	0	0	0
2.31	0	0	0	0	0	0	0	0	0	0	0	0
2.32	0	0	0	0	0	0	0	0	0	0	0	0
2.33	0	0	0	0	0	0	0	0	0	0	0	0
2.34	0	0	0	0	0	0	0	0	0	0	0	0
2.35	0	0	0	0	0	0	0	0	0	0	0	0
2.36	0	0	0	0	0	0	0	0	0	0	0	0
2.37	0	0	0	0	0	0	0	0	0	0	0	0
2.38	0	0	0	0	0	0	0	0	0	0	0	0
2.39	0	0	0	0	0	0	0	0	0	0	0	0
2.40	0	0	0	0	0	0	0	0	0	0	0	0
2.41	0	0	0	0	0	0	0	0	0	0	0	0
2.42	0	0	0	0	0	0	0	0	0	0	0	0
2.43	0	0	0	0	0	0	0	0	0	0	0	0
2.44	0	0	0	0	0	0	0	0	0	0	0	0
2.45	0	0	0	0	0	0	0	0	0	0	0	0
2.46	0	0	0	0	0	0	0	0	0	0	0	0
2.47	0	0	0	0	0	0	0	0	0	0	0	0
2.48	0	0	0	0	0	0	0	0	0	0	0	0
2.49	0	0	0	0	0	0	0	0	0	0	0	0
2.50	0	0	0	0	0	0	0	0	0	0	0	0
SUBTOTAL	0	0	0	0	0	0	0	0	0	0	0	0
							25	18	54	21	0	79

-TOP SECRET-

TABLE 9-3

~~TOP SECRET~~

MISSION * 1013-1 * INSTRUMENT * AFT 01/18/65 DENSITY FREQ DISTR

DENSITY VALUE	PRIMARY MIN MAX LIM	INTERMEDIATE MIN MAX LIM	FULL MIN MAX LIM	ALL LEVELS MIN MAX LIM
2.51	0 0	0 0	0 0	0 0
2.52	0 0	0 0	0 0	0 0
2.53	0 0	0 0	0 0	0 0
2.54	0 0	0 0	0 0	0 0
2.55	0 0	0 0	0 0	0 0
2.56	0 0	0 0	0 0	0 0
2.57	0 0	0 0	0 0	0 0
2.58	0 0	0 0	0 0	0 0
2.59	0 0	0 0	0 0	0 0
2.60	0 0	0 0	0 0	0 0
2.61	0 0	0 0	0 0	0 0
2.62	0 0	0 0	0 0	0 0
2.63	0 0	0 0	0 0	0 0
2.64	0 0	0 0	0 0	0 0
2.65	0 0	0 0	0 0	0 0
2.66	0 0	0 0	0 0	0 0
2.67	0 0	0 0	0 0	0 0
2.68	0 0	0 0	0 0	0 0
2.69	0 0	0 0	0 0	0 0
2.70	0 0	0 0	0 0	0 0
SUBTOTAL	0 0	0 0	0 0	0 0
TOTAL	1 1	3 47	47 39	96 101 144 144 143

MISSION 1013-1 INSTR - AFT 01/18/65 PROCESSING AND EXPOSURE ANALY

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPDSEC
PRIMARY	1	0 PC	0 PC	100 PC	0 PC	0 PC
INTERMEDIATE	47	0 PC	15 PC	72 PC	13 PC	0 PC
FULL	96	7 PC	0 PC	75 PC	18 PC	0 PC
ALL LEVELS	144	5 PC	5 PC	74 PC	16 PC	0 PC
PROCESS LEVEL	BASE + FOG	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPPOSED
PRIMARY	0.01-0.09	0.01-0.13	0.14-0.39	0.40-0.90	-----	0.91 AND
INTERMED	0.10-0.17	0.01-0.20	0.21-0.39	0.40-0.90	0.91-1.34	1.35 AND
FULL	0.18 AND UP	0.01-0.39	-----	0.40-0.90	0.91-1.69	1.70 AND

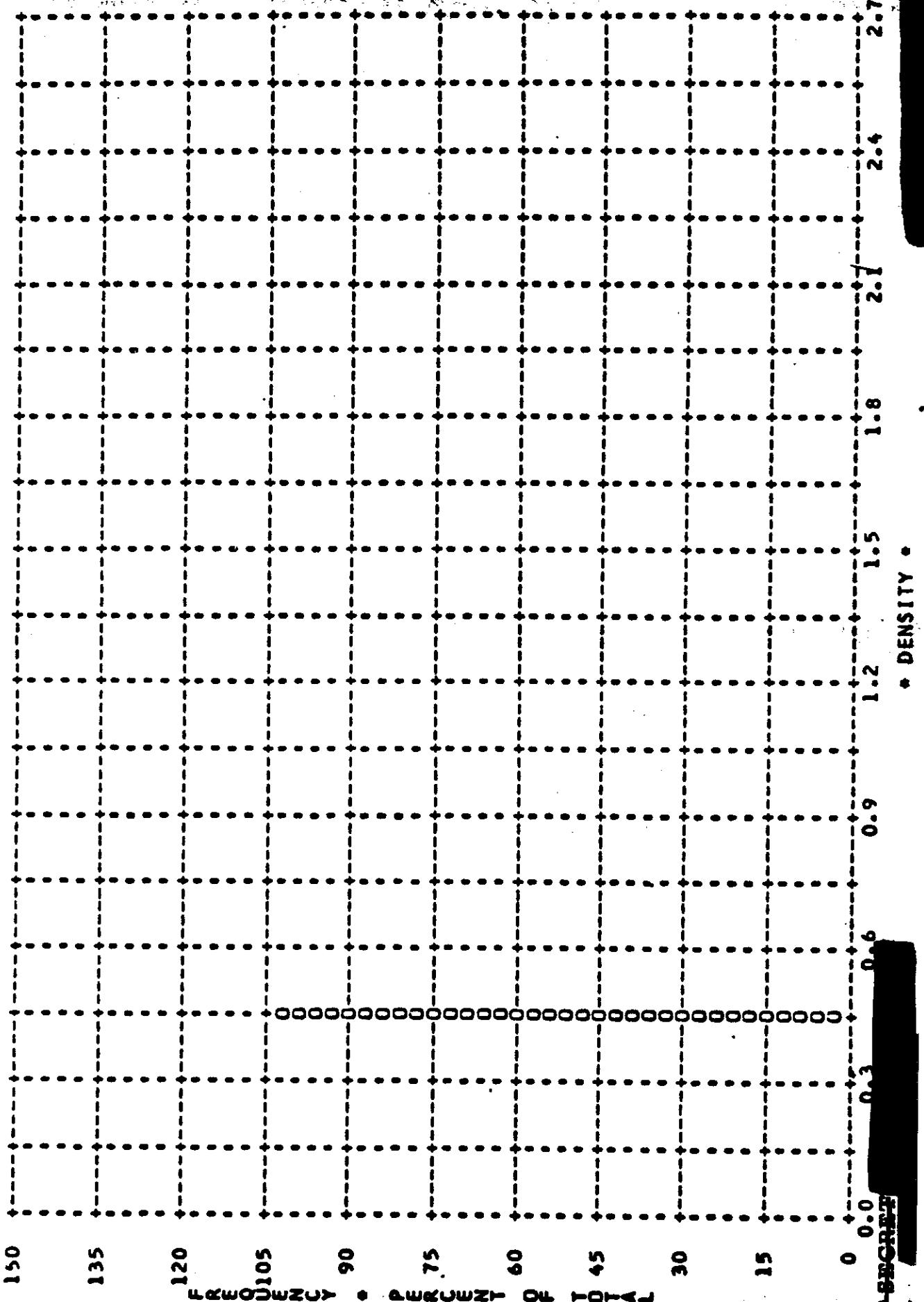
~~TOP SECRET~~

TABLE 9-3

-TOP SECRET

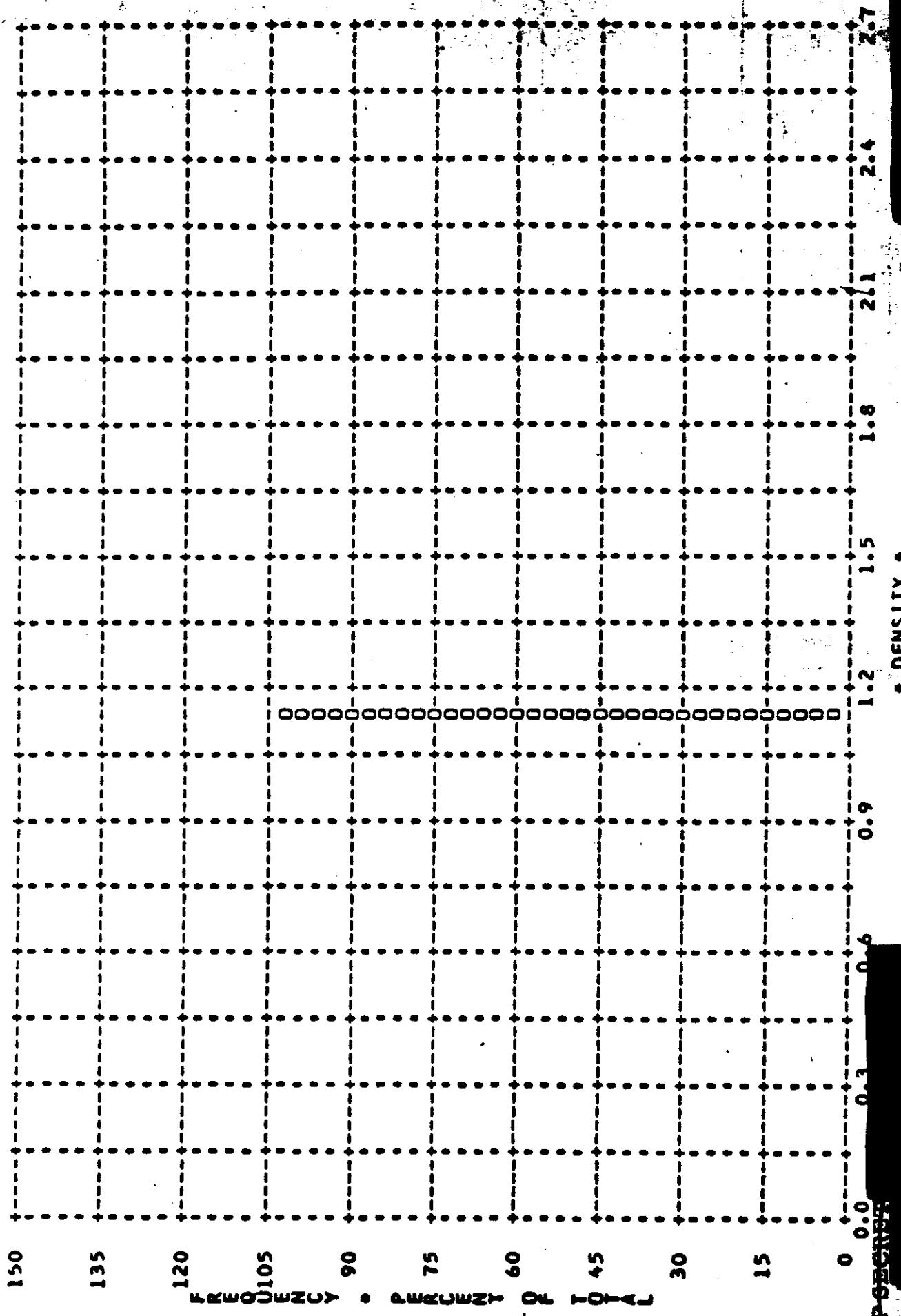
MISSION • 1013-1 • INSTR • AFT • 01/18/65 PLOT OF 0 MIN TERRAIN PROCESSING • PRIMARY

ARITH MEAN • 0.43 • MEDIAN • 0.43 • STD DEV • 0.00 • RANGE • 0.43 10 0.43 WITH 1 SAMPLES

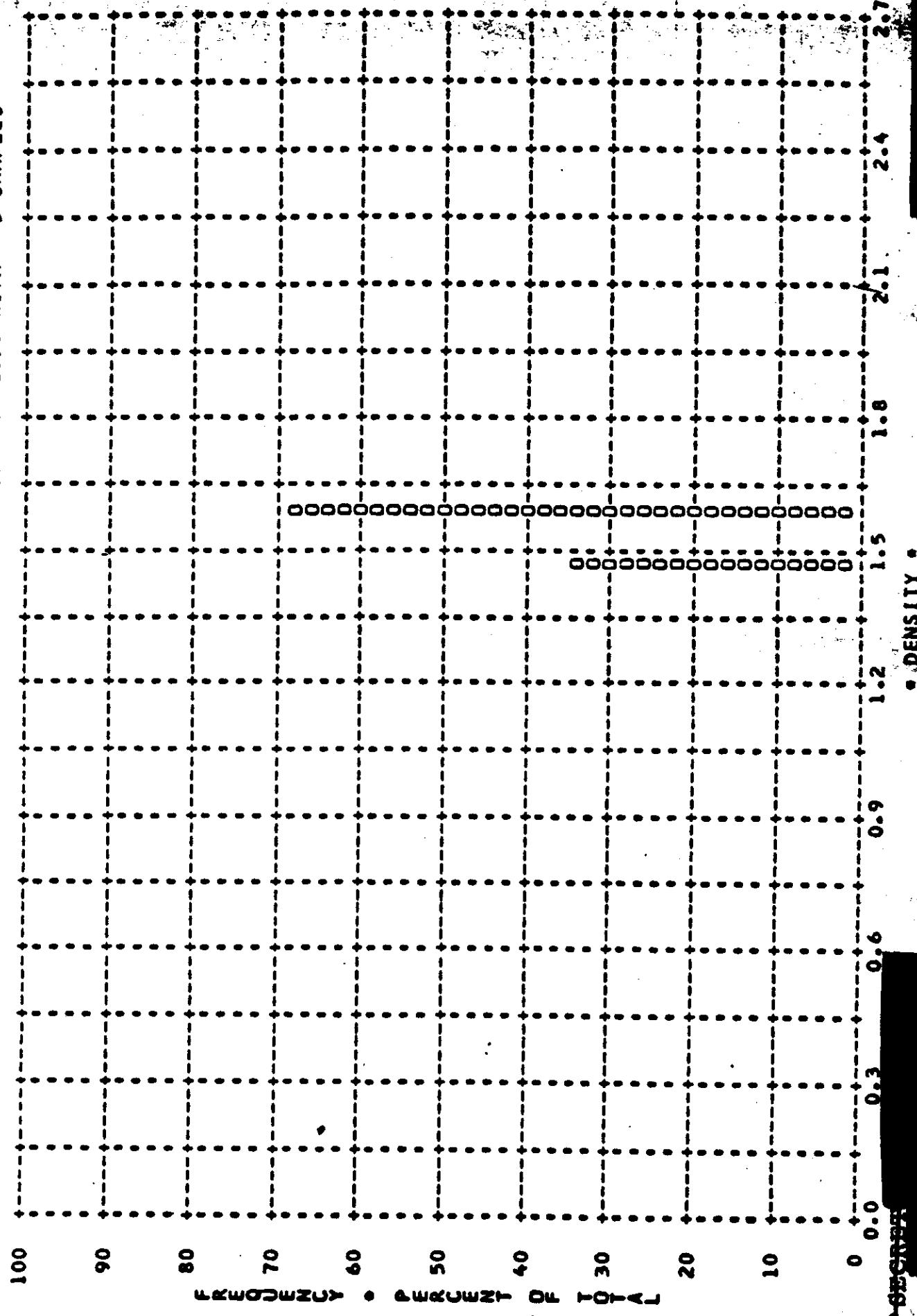


REFERENCES

MISSION • 1013-1 • INSTR • AFT • 01/18/65 PLOT OF D MAX • TERRAIN • PROCESSING • PRIMARY
ARITH MEAN • 1.14 • MEDIAN • 1.14 • STD DEV • 0.00 • RANGE • 1.14 TO 1.14 WITH 1 SAMPLES



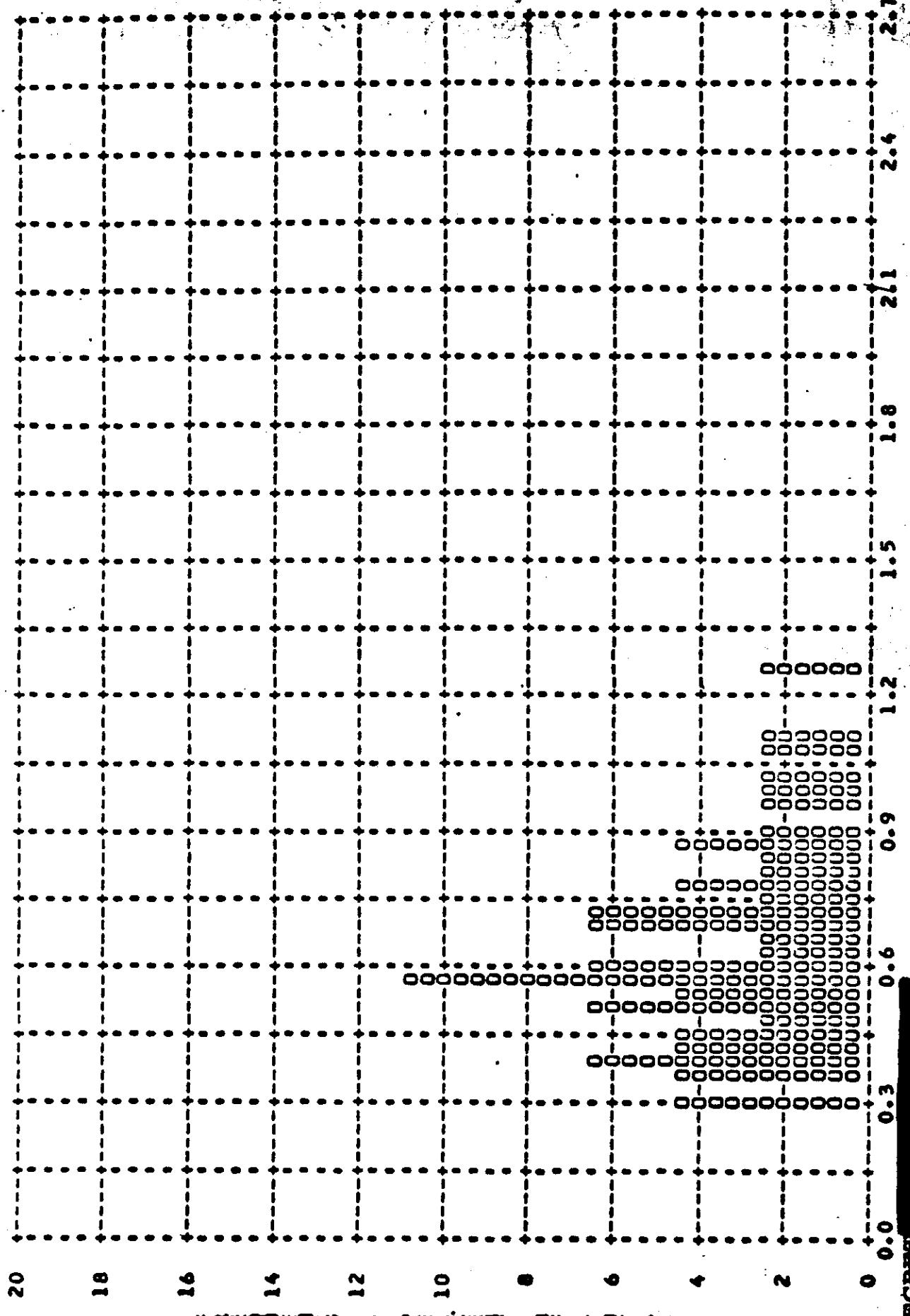
MISSION • 1013-1 • INSTR • AFT • 01/16/65 PLOT OF D MAX • CLOUD • PROCESSING • PRIMARY
ARITH MEAN • 1.55 • MEDIAN • 1.58 • STD DEV • 0.07 • RANGE • 1.47 TO 1.59 WITH 3 SAMPLES



-TOP SECRET

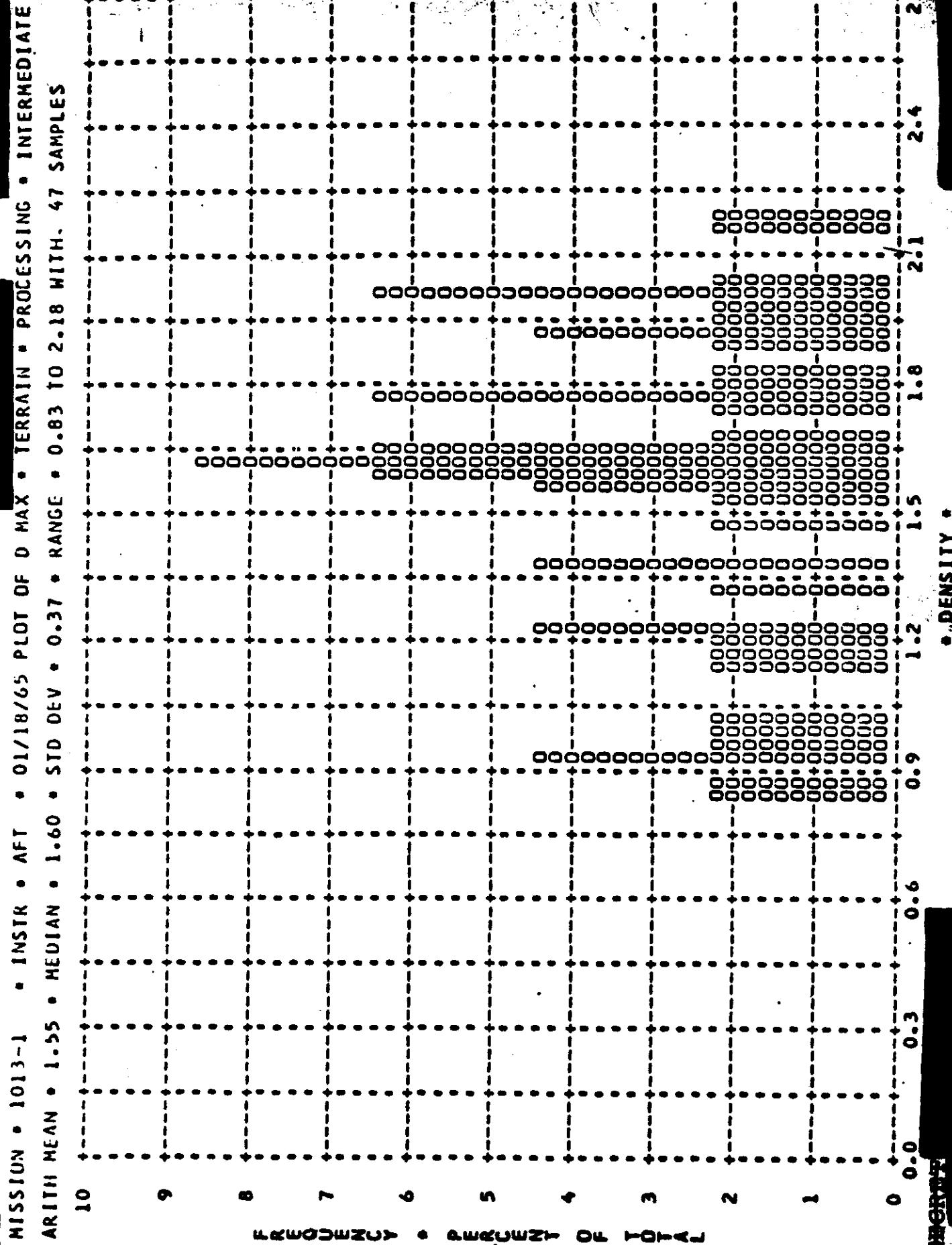
MISSION • 1013-1 • INSTR • AFT • 01/18/65 PLOT OF D MIN • TERRAIN • PROCESSING • INTERMEDIATE

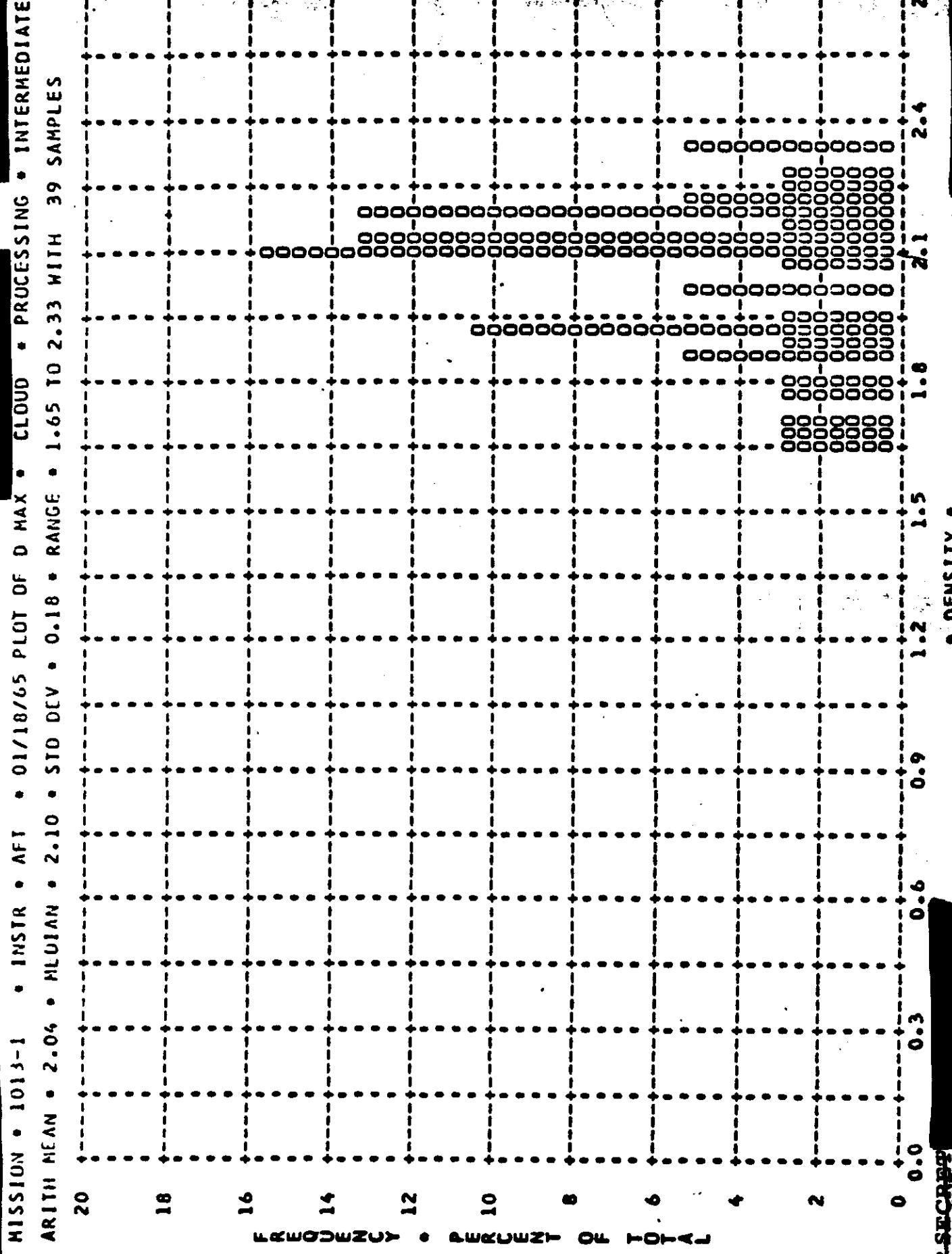
ARITH MEAN • 0.64 • MEDIAN • 0.59 • STD DEV • 0.23 • RANGE • 0.30 TO 1.25 WITH 47 SAMPLES



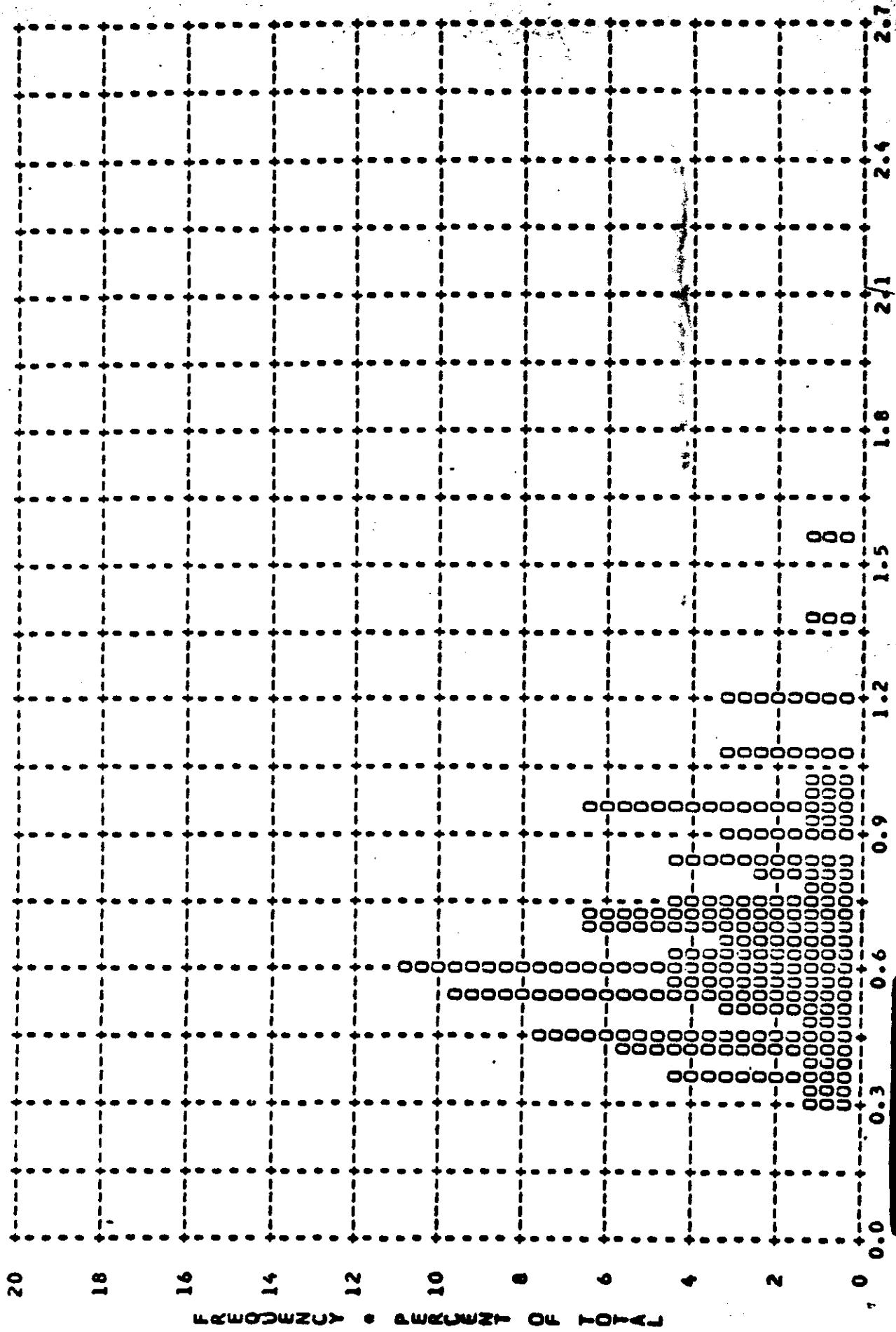
-TOP SECRET

-TOP SECRET





MISSION • 1013-1 • INSTR • AFT • 01/18/65 PLOT OF D MIN • TERRAIN • PROCESSING • FULL
ARITH MEAN • 0.68 • MEDIAN • 0.62 • STD DEV • 0.24 • RANGE • 0.28 TO 1.56 WITH 96 SAMPLES.



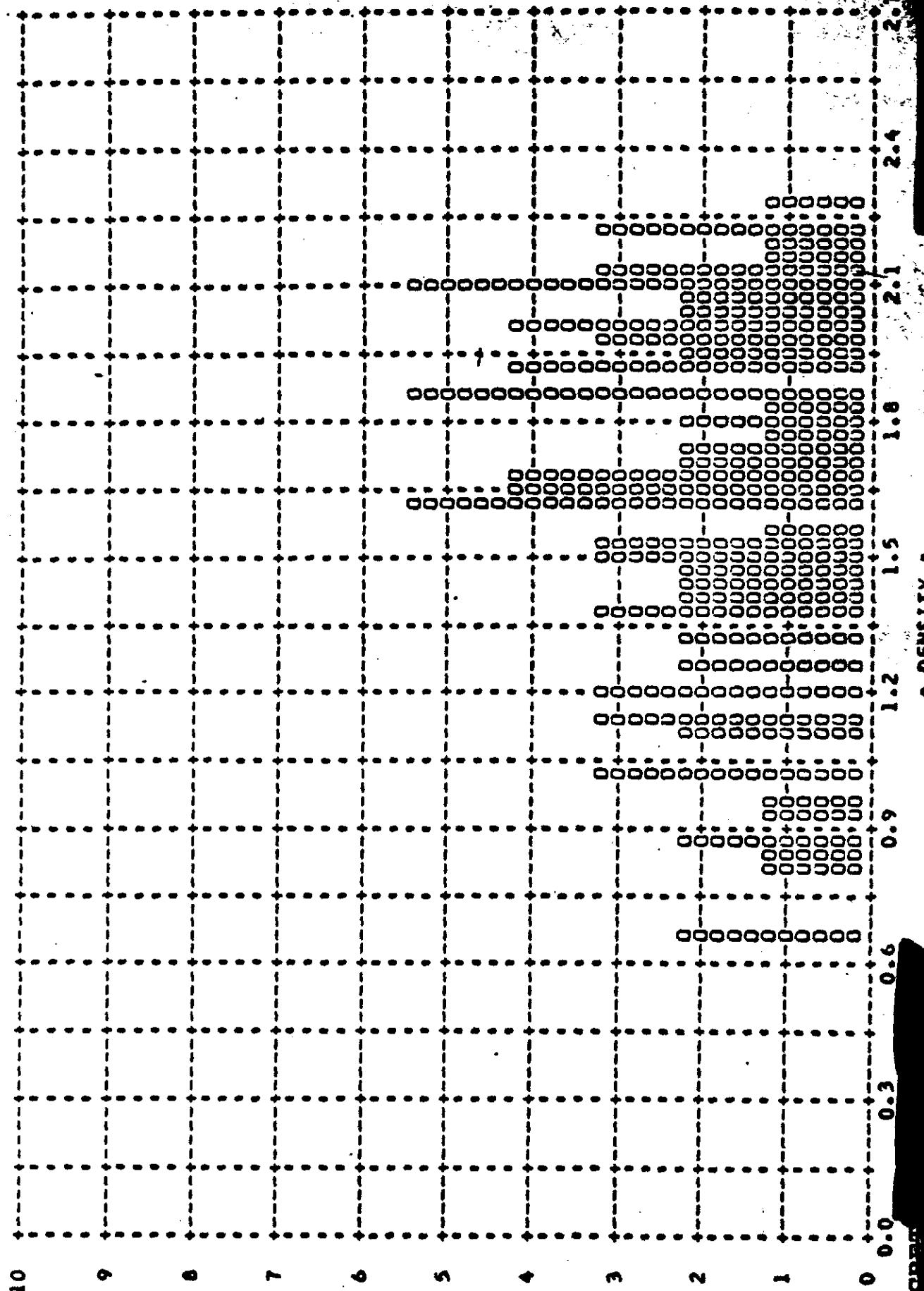
TOP SECRET

• DENSITY •

ENCLOSURE

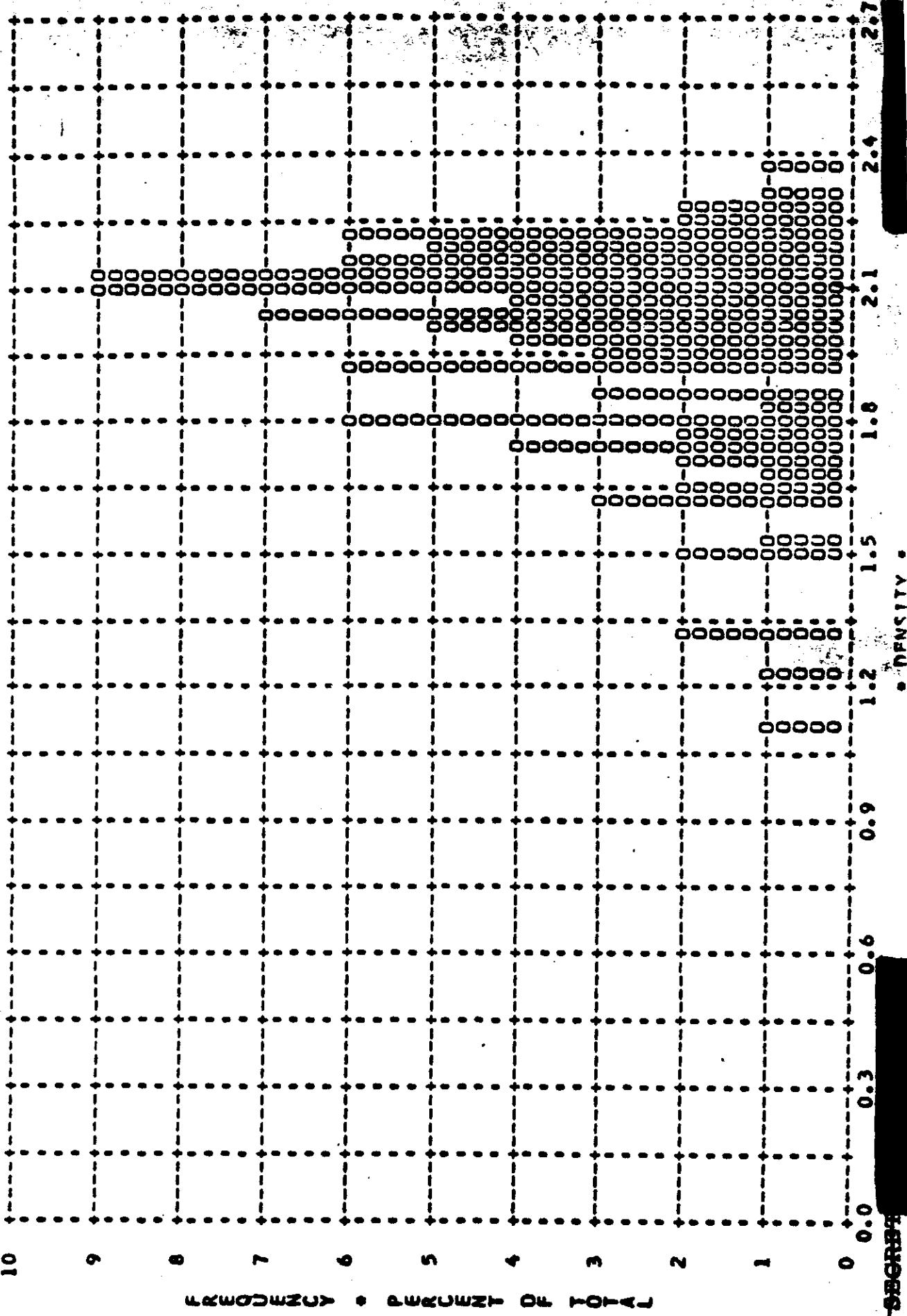
TOP SECRET

MISSION • 1013-1 • INSTR • AFI • 01/18/65 PLOT OF D MAX • TERRAIN • PROCESSING • FULL
ARITH MEAN • 1.62 • MEDIAN • 1.68 • STD DEV • 0.41 • RANGE • 0.64 TO 2.28 WITH 96 SAMPLES



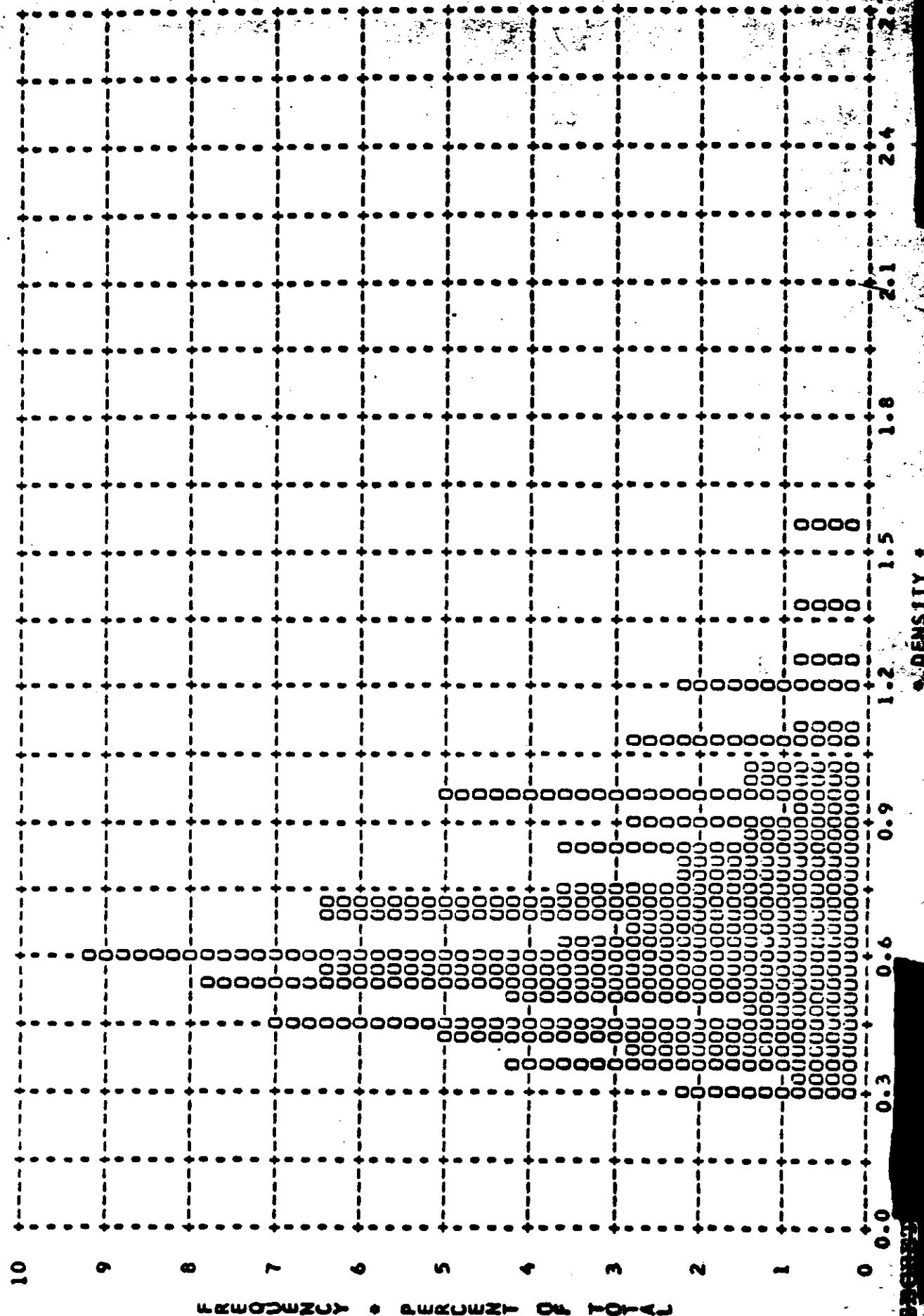
REF ID: A6250257 • BUREAU OF THE ARMY

MISSION • 1013-1 • INSTR • AFT • 01/18/65 PLUT OF D MAX • CLOUD • PROCESSING • FULL
ARITH MEAN • 1.95 • MEDIAN • 2.02 • STD DEV • 0.24 • RANGE • 1.10 10 2.36 WITH 101 SAMPLES



MISSION • 1013-1 • INSTR • AFI • 01/18/65 PLUT OF D MIN • TERRAIN • PROCESSING • ALL LEVELS

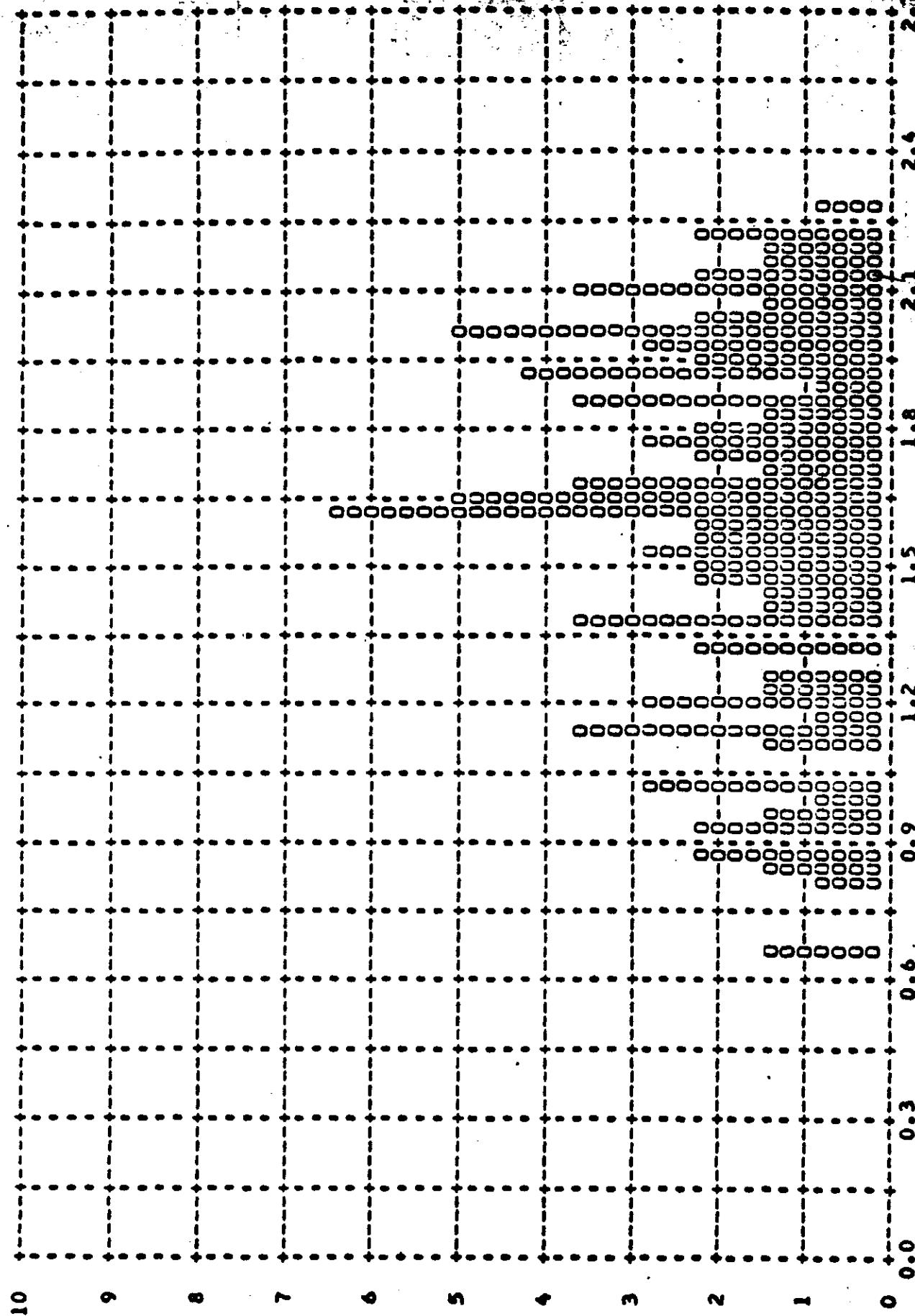
ARITH MEAN • 0.66 • MEDIAN • 0.62 • STD DEV • 0.24 • RANGE • 0.28 TO 1.56 WITH 144 SAMPLES



LEWISBURG • MARYLAND ON RIVER

MISSION • 1013-1 • INSTR • AFI • 01/18/65 PLOT OF D MAX • TERRAIN • PROCESSING • ALL LEVELS

ARITH MEAN • 1.59 • MEDIAN • 1.63 • STD DEV • 0.40 • RANGE • 0.64 TO 2.28 WITH 144 SAMPLES



FREQUENCY • NUMBER OF POINTS

TOP SECRET

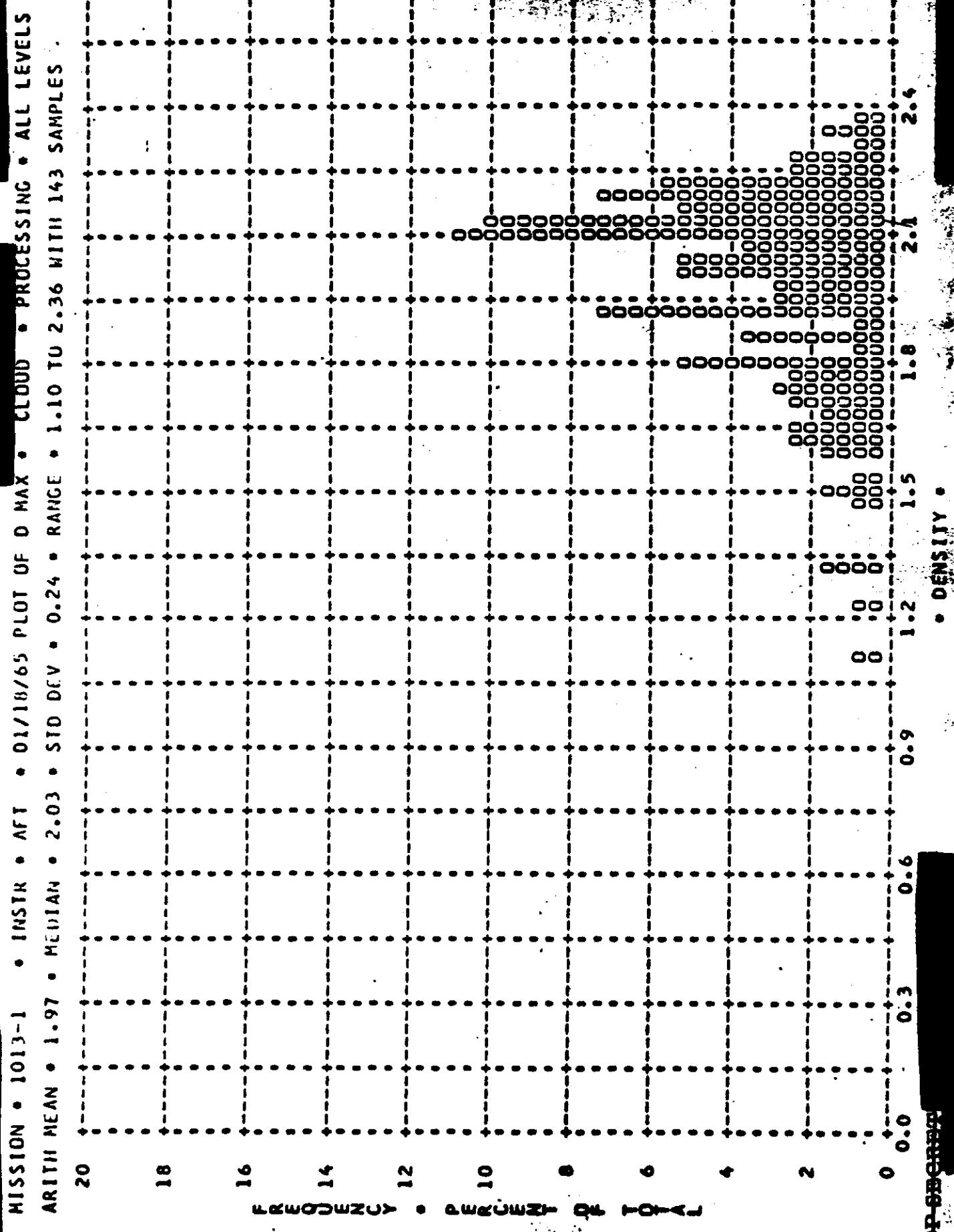


TABLE 9-4

		PROCESSING AND EXPOSURE ANALYSIS							
		UNDER PROCESSED			OVER PROCESSED			OVER EXPOSED	
		SAMPLE SIZE	EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED		
MISSION	1013-1	INSTR - FRWD	01/18/65						
PROCESS LEVEL									
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC	0 PC		
INTERMEDIATE	79	1 PC	23 PC	72 PC	3 PC	0 PC			
FULL	65	14 PC	20 PC	80 PC	6 PC	1 PC			
ALL LEVELS	144	17 PC	13 PC	76 PC	4 PC	1 PC			
MISSION	1013-1	INSTR - AFT	01/18/65						
PROCESS LEVEL									
PRIMARY	1	0 PC	0 PC	100 PC	0 PC	0 PC	0 PC		
INTERMEDIATE	47	9 PC	15 PC	72 PC	13 PC	0 PC	0 PC		
FULL	96	7 PC	10 PC	75 PC	18 PC	0 PC	0 PC		
ALL LEVELS	144	5 PC	5 PC	74 PC	16 PC	0 PC	0 PC		
PROCESS LEVEL									
PRIMARY	0.01-0.09	0.01-0.13	0.14-0.39	0.40-0.90	-----	-----	0.91 AND UP		
INTERMEDIATE	0.10-0.17	0.01-0.20	0.21-0.39	0.40-0.90	0.91-1.34	1.35 AND UP			
FULL	0.18 AND UP	0.01-0.39	0.40-0.90	0.40-0.90	0.91-1.69	1.70 AND UP			

SECTION 10

PERFORMANCE MEASUREMENTS

The photography acquired by both panoramic cameras during Mission 1013-1 received an initial MIP rating of 85. A summary is tabulated below of the average visual RES values and MTF/AIM resolution values measured by AFSPPL and the MTF/AIM resolution values computed by [REDACTED]. The microdensimeter slit used by AFSPPL and [REDACTED] was 80 microns by 1 micron.

Mission	Camera	Visual		[REDACTED]	
		RES	AFSPPL	All	High
1013-1	FWD	89	94	85	99
1013-1	AFT	77	97	81	103

The [REDACTED] data normally contains two readings of the same edge the tabulation shows both the average of all the readings and the average of the highest readings of each edge. The value of the average of all readings is questionable as no valid reason can be ascertained for a measurement being greater than the resolution recorded however many factors can reduce the reading.

The details of the measurement and computing techniques, targets measured and target locations are fully reported in the evaluation report published by AFSPPL and are not normally included in this report.

[REDACTED] has recently completed the re-calibration of the MTF/AIM values from Mission 1007-2 and up. Since this data has not been published in a previous report the corrected measurements for Mission 1013-1 are included in this report.

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Analysis of Photographic Image to Evaluate System Performance

Mission 1013-1

Resolution in lines/mm based on the aerial image modulation - 4404 curve from edge trace data reduced by computer techniques.

Arithmetic Mean	83.3 l/mm
Standard Deviation	27.3 l/mm
Coefficient of Dispersion	33%
Number of Edges	49
 M. I. P. Frame	 95 l/mm

Spread function width at 50% amplitude in microns from edge trace data reduced by computer techniques.

Arithmetic Mean	10.8 μ
Standard Deviation	4.1 μ
Coefficient of Dispersion	38%
Number of Edges	49
 M. I. P. Frame	 9.2 μ

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Analysis of Photographic Image to Evaluate System Performance

Mission 1013-1

Resolution in lines/mm based on the aerial image modulation ~ 4404 curve from edge trace data reduced by computer techniques.

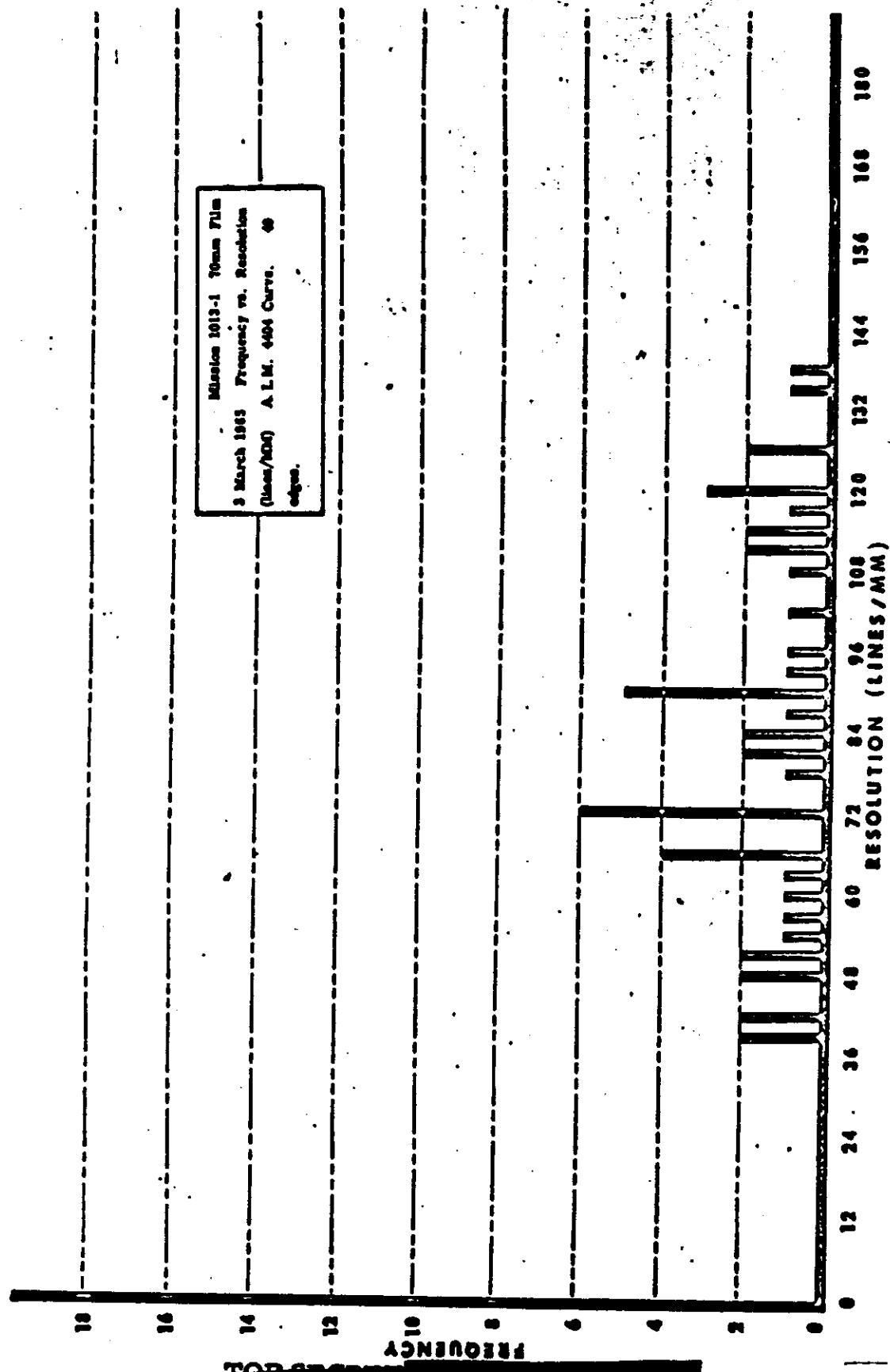
	FWD Camera	AFT Camera	Airfields	Buildings
Arithmetic Mean	84.8 l/mm	81.1 l/mm	79.3 l/mm	90.7 l/mm
Standard Deviation	23.1 l/mm	33.0 l/mm	24.9 l/mm	30.8 l/mm
Coefficient of Dispersion	27%	41%	31%	34%
Number of Edges	29	20	32	17

Spread function width at 50% amplitude in microns from edge trace data reduced by computer techniques.

	FWD Camera	AFT Camera	Airfields	Buildings
Arithmetic Mean	10.2 μ	11.8 μ	11.4 μ	9.7 μ
Standard Deviation	2.7 μ	5.4 μ	4.2 μ	3.8 μ
Coefficient of Dispersion	27%	46%	36%	39%
Number of Edges	29	20	32	17

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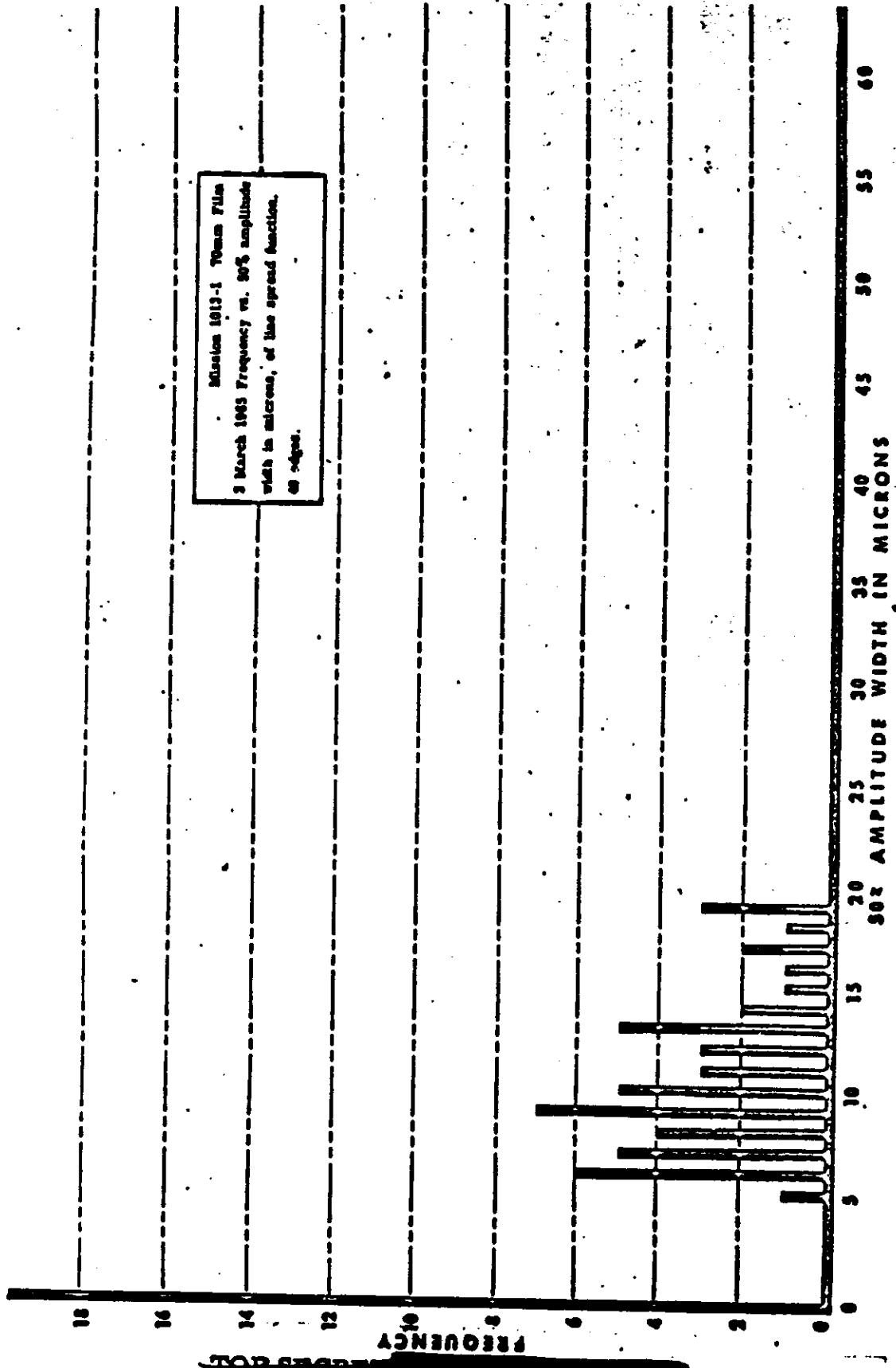
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March 1943 Frequency vs. 30° amplitude
width in microns, of the spread function.
at 400.



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~~TOP SECRET~~Analysis of Photographic Image to Evaluate System Performance

Mission 1013-1

FORWARD CAMERA

<u>Pass</u>	<u>Frame</u>	<u>Location</u>	<u>Orientation</u>	<u>Subject</u>	<u>50% Amplitude Spread Function Width (Microns)</u>	<u>A.L.M. Resolution</u>
D-08	024	X35.3 Y11.0	035	Breakwater	9.4	93
D-09	047	X67.5 Y11.3	060	Airfield	12.4	47
D-09	047	X67.5 Y11.3	060	Airfield	7.6	91
D-21	057	X14.5 Y13.6	090	Airfield	15.6	55
D-21	057	X14.5 Y13.6	090	Airfield	13.0	65
D-21	060	X20.2 Y12.8	080	Airfield	9.5	90
D-21	060	X20.2 Y12.8	080	Airfield	8.3	90
D-21	062	X24.3 Y13.7	070	Airfield	14.3	80
D-21	062	X24.3 Y13.7	070	Airfield	12.4	72
D-21	114	X48.0 Y11.0	060	Airfield	8.0	103
D-21	114	X48.0 Y11.0	060	Airfield	11.3	78
D-23	090	X51.8 Y12.8	140	Buildings	12.6	65
D-23	090	X51.8 Y12.8	140	Buildings	6.5	120
D-39	130	X35.3 Y12.8	010	Dam	10.0	73
D-39	130	X35.3 Y12.8	010	Dam	12.0	67
D-39	131	X45.3 Y11.1	035	Buildings	5.8	135
D-39	131	X45.3 Y11.1	035	Buildings	7.4	115
D-39	132	X54.5 Y13.3	065	Buildings	9.5	89
D-39	132	X54.5 Y13.3	065	Buildings	7.0	115
D-40	020	X18.1 Y12.3	055	Buildings	6.4	126
D-40	020	X18.1 Y12.3	055	Buildings	4.8	145
D-40	020	X15.6 Y12.7	088	Airfield	12.6	72
D-47E	009*	X50.5 Y14.5	045	Buildings	11.4	65
D-47E	009*	X50.5 Y14.5	045	Buildings	13.3	61

*M.I.P. Frame

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Mission 1013-1

FORWARD CAMERA

50%
 Amplitude
 Spread
 Function
 Width
 (Microns)

A.L.M.
Resolution

<u>Pass</u>	<u>Frame</u>	<u>Location</u>	<u>Orientation</u>	<u>Subject</u>		
D-47E	009*	X50.3 Y12.8	120	Airfield	10.3	71
D-47E	009*	X50.3 Y12.8	120	Airfield	9.5	72
D-47E	025	X48.7 Y10.3	010	Airfield	9.5	80
D-47E	025	X48.7 Y10.3	010	Airfield	9.0	86
D-47E	029	X18.6 Y10.6	060	Airfield	6.2	121
D-47E	029	X18.6 Y10.6	060	Airfield	14.0	62

*M.I.P. Frame

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Mission 1013-1

AFT CAMERA

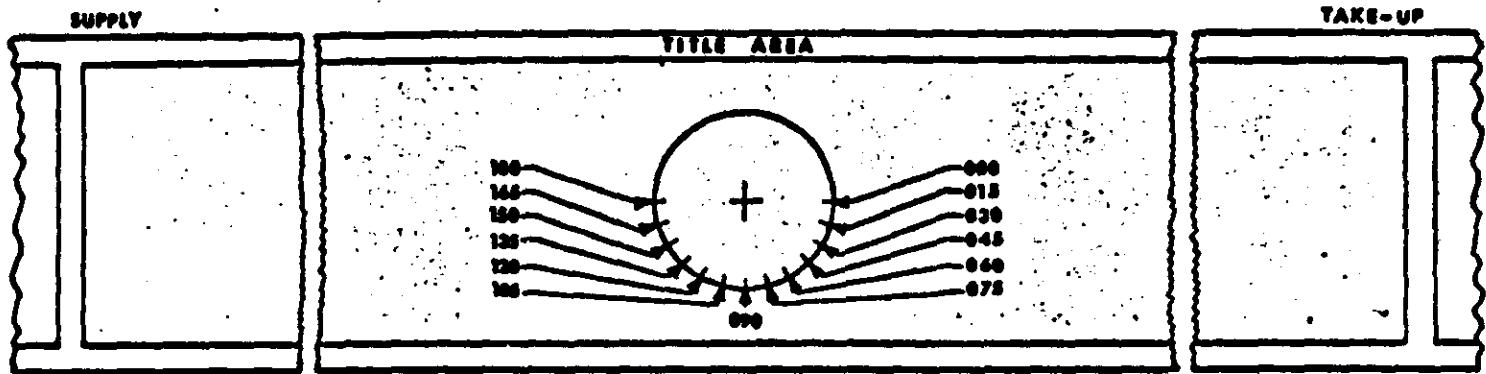
<u>Pass</u>	<u>Frame</u>	<u>Location</u>	<u>Orientation</u>	<u>Subject</u>	<u>Width (Microns)</u>	<u>A. I. M.</u>	<u>Resolution</u>
D-08	030	X55.6 Y14.7	040	Breakwater	4.8	50%	137
D-21	063	X76.5 Y12.5	100	Airfield	20.6	Amplitude	39
D-21	063	Y76.5 Y12.5	100	Airfield	20.7	Spread	42
D-21	066	X70.4 Y13.5	088	Airfield	15.4	Function	50
D-21	066	X70.4 Y13.5	088	Airfield	13.4	Width	51
D-21	068	X66.2 Y13.2	085	Airfield	17.2	(Microns)	57
D-21	068	X66.2 Y13.2	085	Airfield	20.6	A. I. M.	42
D-21	121	X41.8 Y14.7	140	Airfield	7.5	Resolution	118
D-21	121	X41.8 Y14.7	140	Airfield	5.7		119
D-39	138	X55.3 Y13.3	020	Dam	17.8		40
D-39	138	X55.3 Y13.3	020	Dam	16.7		49
D-40	026	X75.7 Y14.0	100	Airfield	6.3		111
D-40	026	X75.7 Y14.0	100	Airfield	9.0		125
D-47E	015*	X39.8 Y10.3	055	Buildings	6.6		108
D-47E	015*	X39.8 Y10.3	055	Buildings	8.3		84
D-47E	015*	X39.8 Y12.2	080	Airfield	9.2		95
D-47E	015*	X39.8 Y12.2	080	Airfield	10.9		72
D-47E	035	X72.3 Y13.7	055	Airfield	10.1		83
D-47E	035	X72.3 Y13.7	055	Airfield	3.6		159
D-47E	038	X43.8 Y11.1	055	Airfield	8.6		90
D-47E	038	X43.8 Y11.1	055	Airfield	7.4		110

*M.I.P. Frame

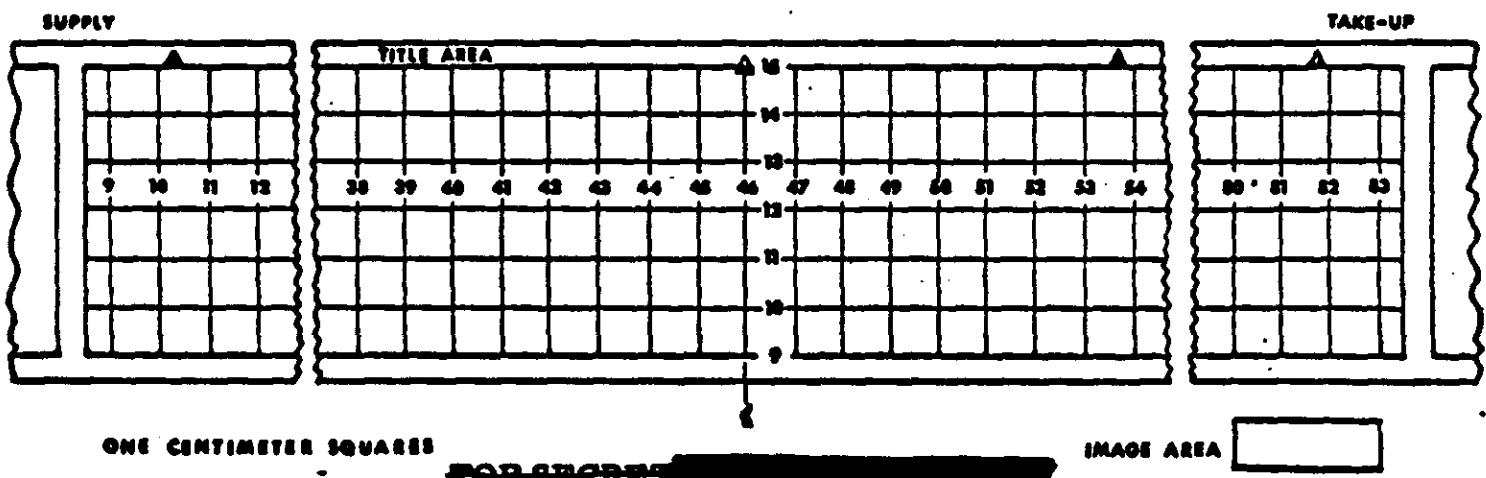
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TOP SECRET

Reference System For Orientation Of C/M/J Mission Edges
original negative — emulsion down



Coordinate Locator Grid For C/M/J Mission Edges
original negative - emulsion down



ONE CENTIMETER SQUARES

TOP SECRET

IMAGE AREA

SECTION 11

OBSERVED DATA

Engineering operation on Mission 1013 provided daylight photography on descending passes 16, 31 and 47.

D16 was off the coast of California and the panoramic frames contained only clouds and water. However, the eastward looking horizon photographs were relatively cloud free and afforded excellent views of the mainland from Moro Bay to Clear Lake and inland beyond Mono and Pyramid Lakes.

The horizon photos are frequently useful for a rough orientation and location of vehicle position by identifying gross geographical features, such as San Francisco Bay Area, Baja, California; Cape Cod, the Great Lakes and other distinctive features. The center of the earth image will be approximately 300 miles from vehicle nadir.

The out-of-focus or soft imagery of much film from this mission has been discussed elsewhere in this report.

The Aft photography on D31 supplies an example of the extent of degradation from soft imagery, whatever it's source. In the overlap area of Frames 4 and 5 is a copper mining-smelting complex at Morenci, Arizona, 2-1/2" west of the C/F notch. In the South edge of Frame 4, a group of settling basins with catwalks and flumes is quite sharp, indicating that this system was capable of detecting dimensions of 3 feet or less on objects with a high aspect ratio (greater than 5:1). In the North edge of Frame 5 these basins could scarcely be seen even though the largest one is 300 feet in diameter.

Fort Huachuca fixed targets appeared in the bonus area of Frame 13 Aft. The largest group could not be resolved. The bonus area is always out of focus.

Evaluation of the CORN targets at Indian Springs and Pahrump, Nevada indicates a bar-target resolution of 7 to 10 feet both foreward and aft for prevailing conditions.

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SECTION 12

MISSION 1013-1 STELLAR-INDEX CAMERA

A. COMPONENT ASSIGNMENT

<u>Component</u>	<u>Serial Number</u>
Camera	D52
Index Reseau	49
Stellar Reseau	55

B. CAMERA DATA AND FLIGHT SETTINGS

Stellar Camera:

Lens	85mm f/1.8
Exposure Time	2 seconds
Filter Type	None
Film Type	Eastman Type 4401

Index Camera:

Lens	38mm f/4.5
Exposure Time	1/500 second
Filter Type	Wratten 21
Film Type	Eastman Type 4400

C. POST FLIGHT EVALUATION

The camera operated properly throughout the mission. The stellar and index portions each produced 421 photographic frames. The quality of all photography was excellent.

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SECTION 13

MISSION 1013-2 STELLAR-INDEX CAMERA

A. COMPONENT ASSIGNMENT

<u>Component</u>	<u>Serial Number</u>
Camera	D47
Index Reseau	48
Stellar Reseau	54

B. CAMERA DATA AND FLIGHT SETTINGS

Stellar Camera:

Lens	85mm f/1.8
Exposure Time	2 seconds
Filter Type	None
Film Type	Eastman Type 4401

Index Camera:

Lens	38mm f/4.5
Exposure Time	1/500 second
Filter Type	Wratten 21
Film Type	Eastman Type 4400

C. POST FLIGHT EVALUATION

The camera operated properly throughout the mission. Both portions produced 102 frames. This small quantity was the result of the early recovery of the mission. The photographic quality was the same as Mission 1013-1.

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SECTION 14

VEHICLE ATTITUDE

The vehicle attitude errors for Mission 1013-1 were derived from the reduction of the Stellar camera photography. This attitude data is supplied to A/P by NPIC.

The attitude errors for each frame and the attitude control rates are calculated at the A/P computer facility. The computer also plots the frequency distribution of the rates and errors. Figures 14-1 through 14-6 show these distributions for Mission 1013-1.

The summary table below lists the maximum attitude errors and rates that were experienced during 90% of the photographic operations, excluding the first six frames of each operation, and the total range of the errors and rates.

<u>Value</u>	<u>Mission 1013-1</u>	
	<u>90%</u>	<u>Range</u>
Pitch Error ($^{\circ}$)	0.64	-0.60 to +1.25
Roll Error ($^{\circ}$)	0.31	-0.35 to +1.05
Yaw Error ($^{\circ}$)	1.33	-1.75 to +0.35
Pitch Rate ($^{\circ}/hr$)	36.9	-75 to +90
Roll Rate ($^{\circ}/hr$)	29.0	-50 to +85
Yaw Rate ($^{\circ}/hr$)	32.3	-90 to +100

The performance of the attitude control system is comparable to the control systems used on recent missions. The panoramic photography was not degraded by the attitude control system.

J-15 A BUCKET - FWD INSTR FRAMES 1-6 OF EACH OF OMITTED 90 PERCENT = 0.6

Y = PITCH ANGLE ERROR - DEGREES (X) VERSUS FREQUENCY - PERCENT (Y)

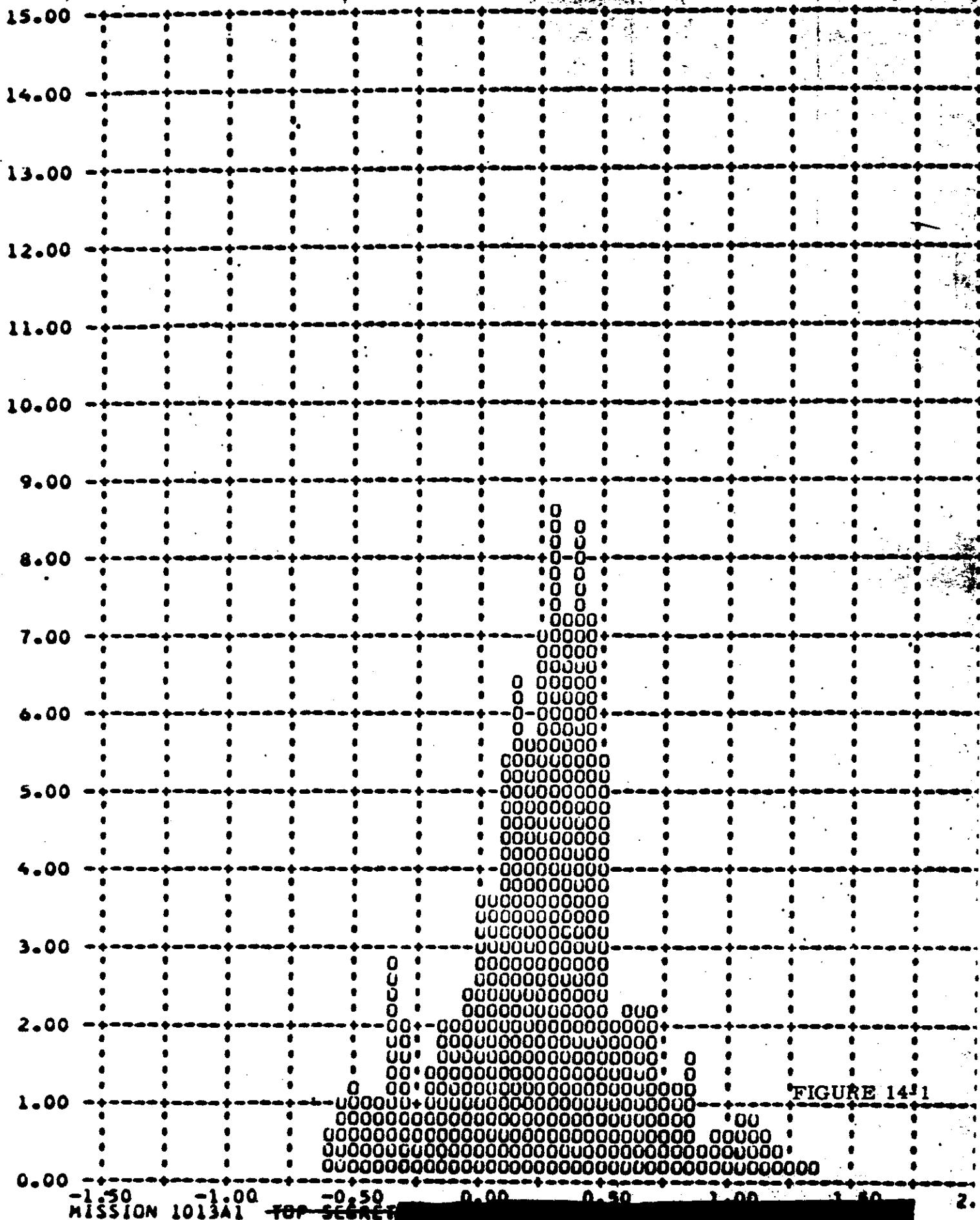


FIGURE 14-1

MISSION 1013A1 TOP SECRET

J-15 A BUCKET - FWD INSTR FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT = 0.

Y = ROLL ANGLE ERROR - DEGREES (X) VERSUS FREQUENCY - PERCENT (Y)

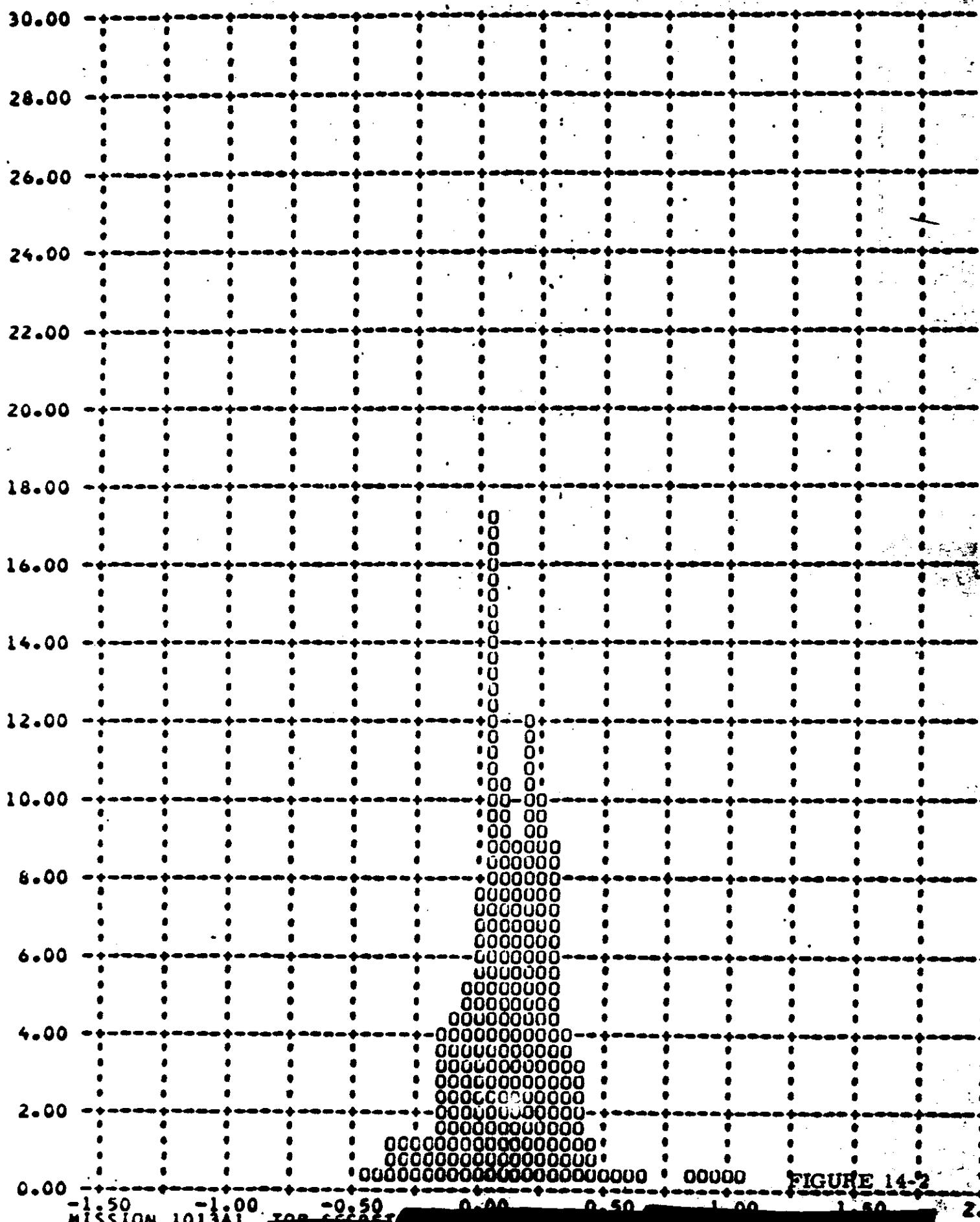


FIGURE 14-2

MISSION 1013A1 TOP SECRET

MISSION 1013A1 -TOP SECRET

J-15 A BUCKET - FWD INSTR FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT
Y YAW ANGLE ERROR - DEGREES (X) VERSUS FREQUENCY - PERCENT (Y)

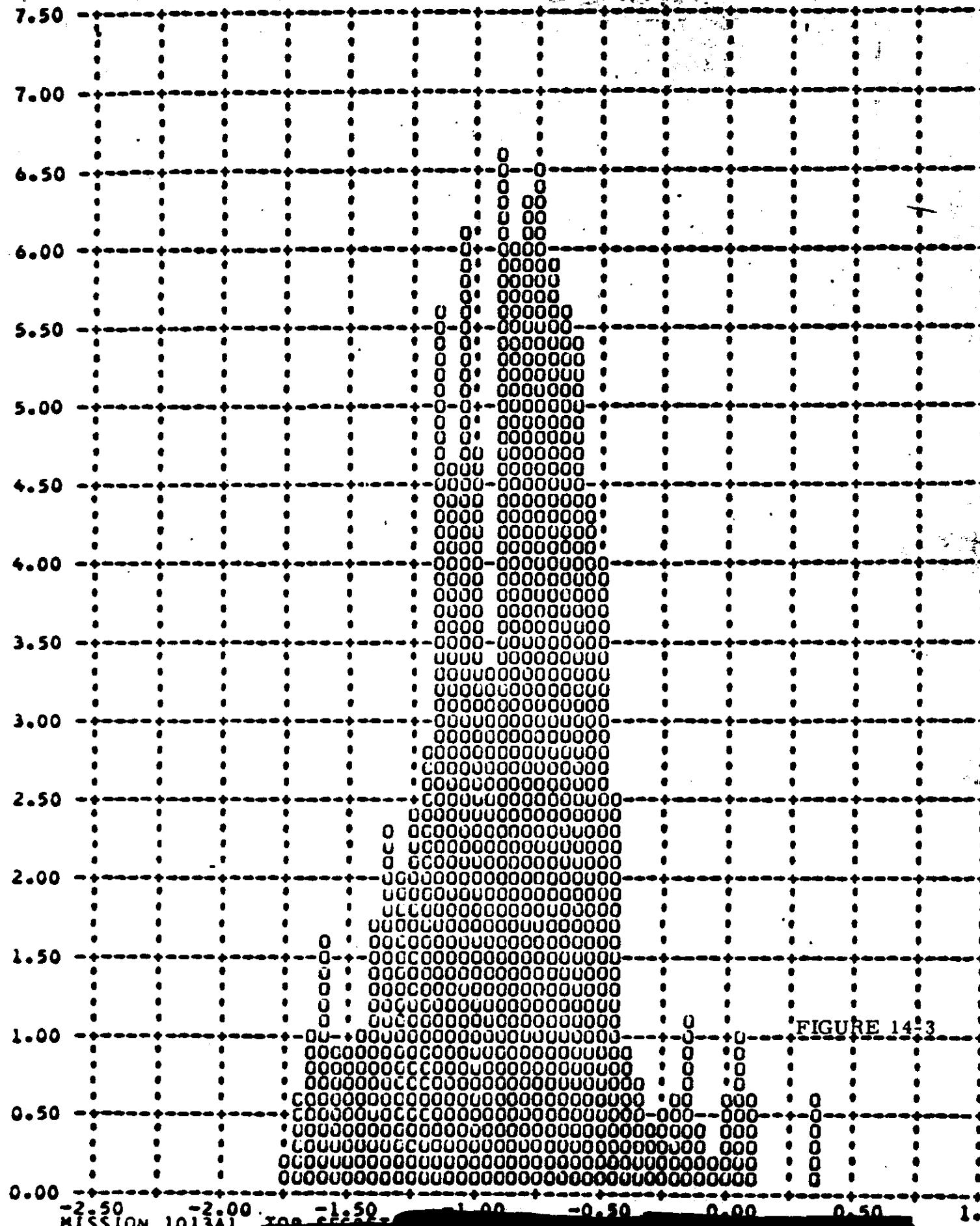


FIGURE 14-3

MISSION 1013A1 -TOP SECRET

MISSION 1013A1 TOP SECRET

J-15 A BUCKET - FWD INSTR FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT = 36.5
Y PITCH RATE ERROR - DEG/HOUR (X) VERSUS FREQUENCY - PERCENT (%)

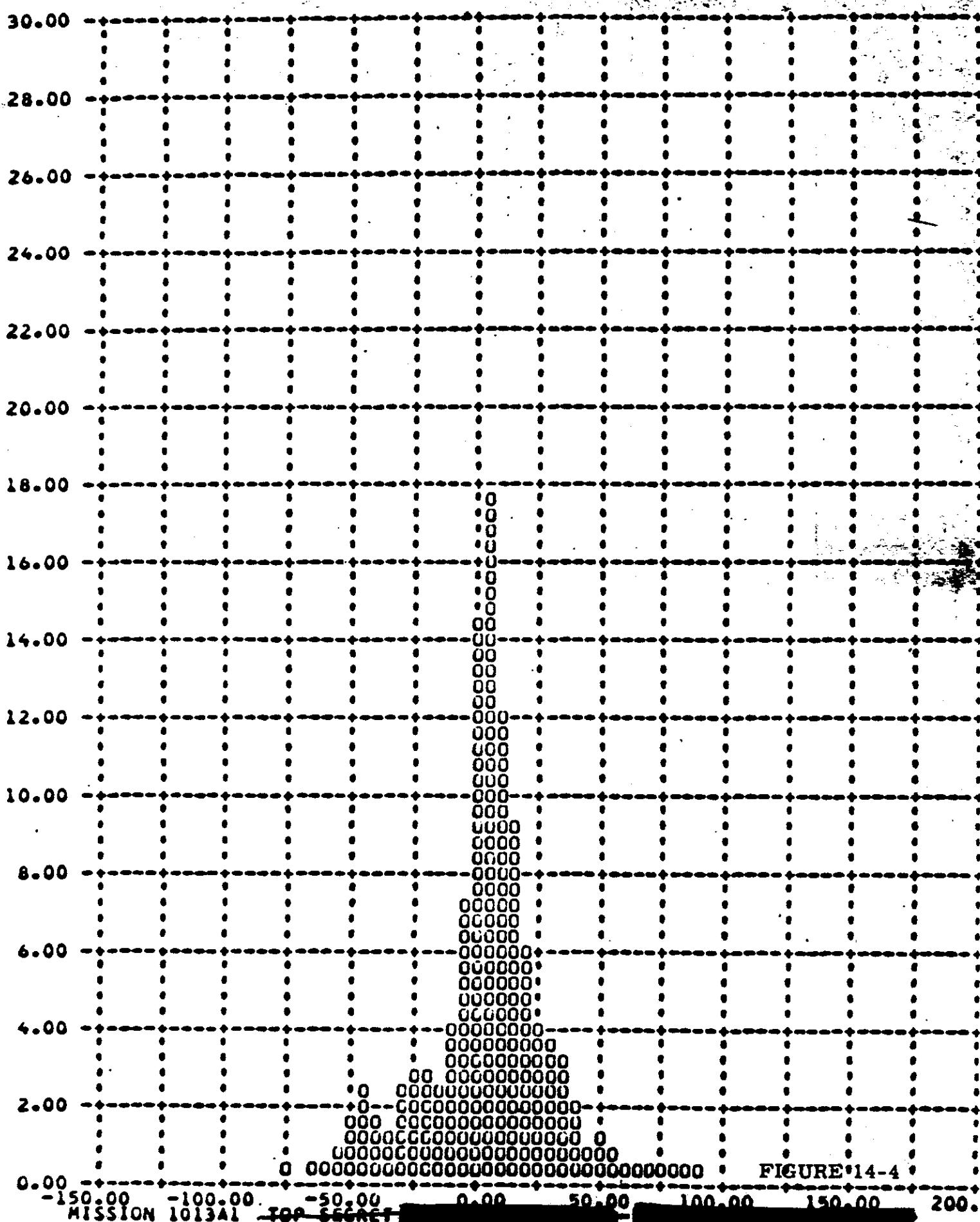


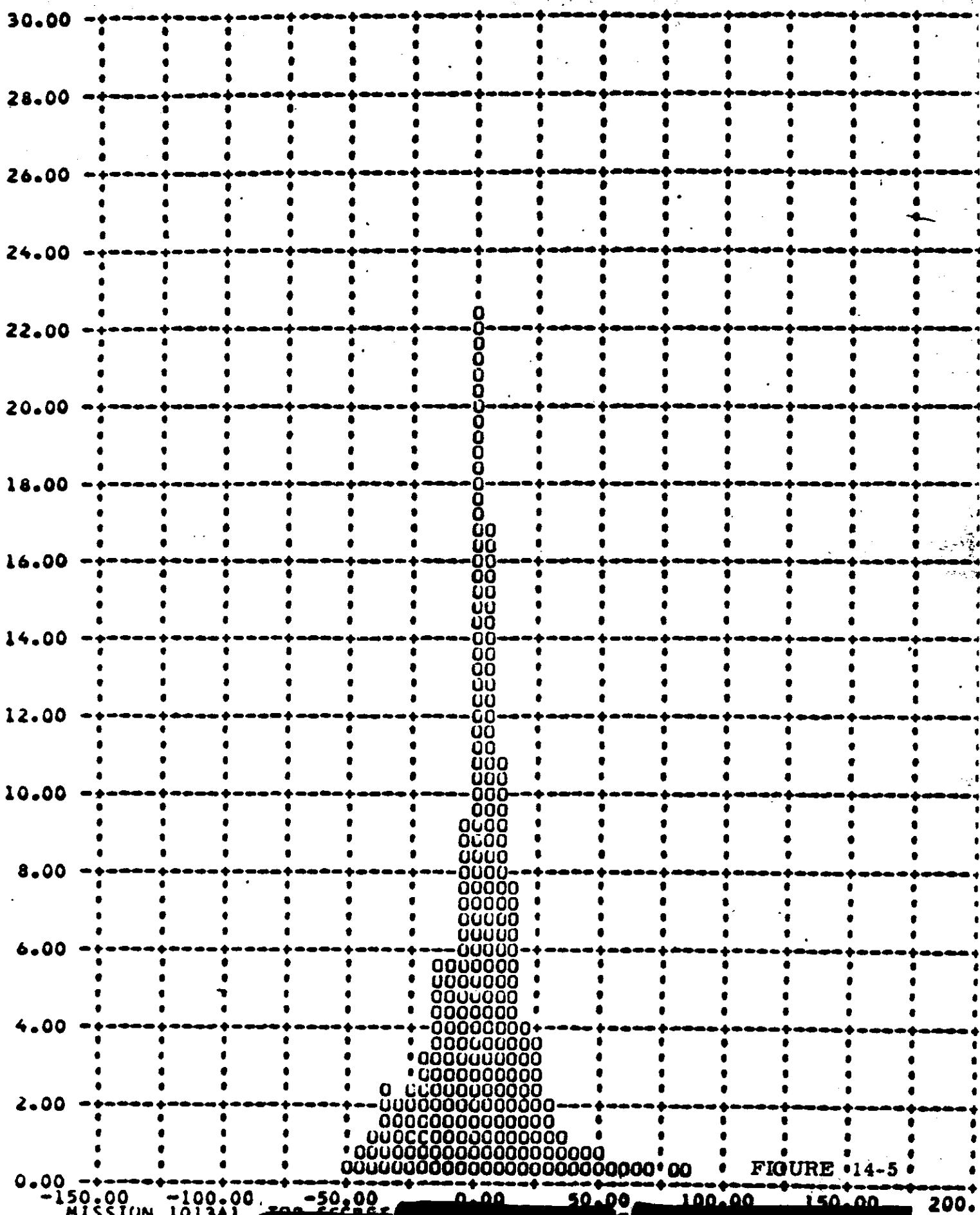
FIGURE 14-4

MISSION 1013A1 TOP SECRET

MISSION 1013A1 - TOP SECRET

J-15 A BUCKET - FWD INSTR FRAMES 1-6 OF EACH DP OMITTED 90 PERCENT = 29.0

Y ROLL RATE ERROR - DEG/HOUR (X) VERSUS FREQUENCY - PERCENT (Y)



MISSION 1013A1 -TOP SECRET

J-15 A BUCKET - FWD INSTR FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT • 32.

Y YAW RATE ERROR - DEG/HOUR (X) VERSUS FREQUENCY - PERCENT (Y)

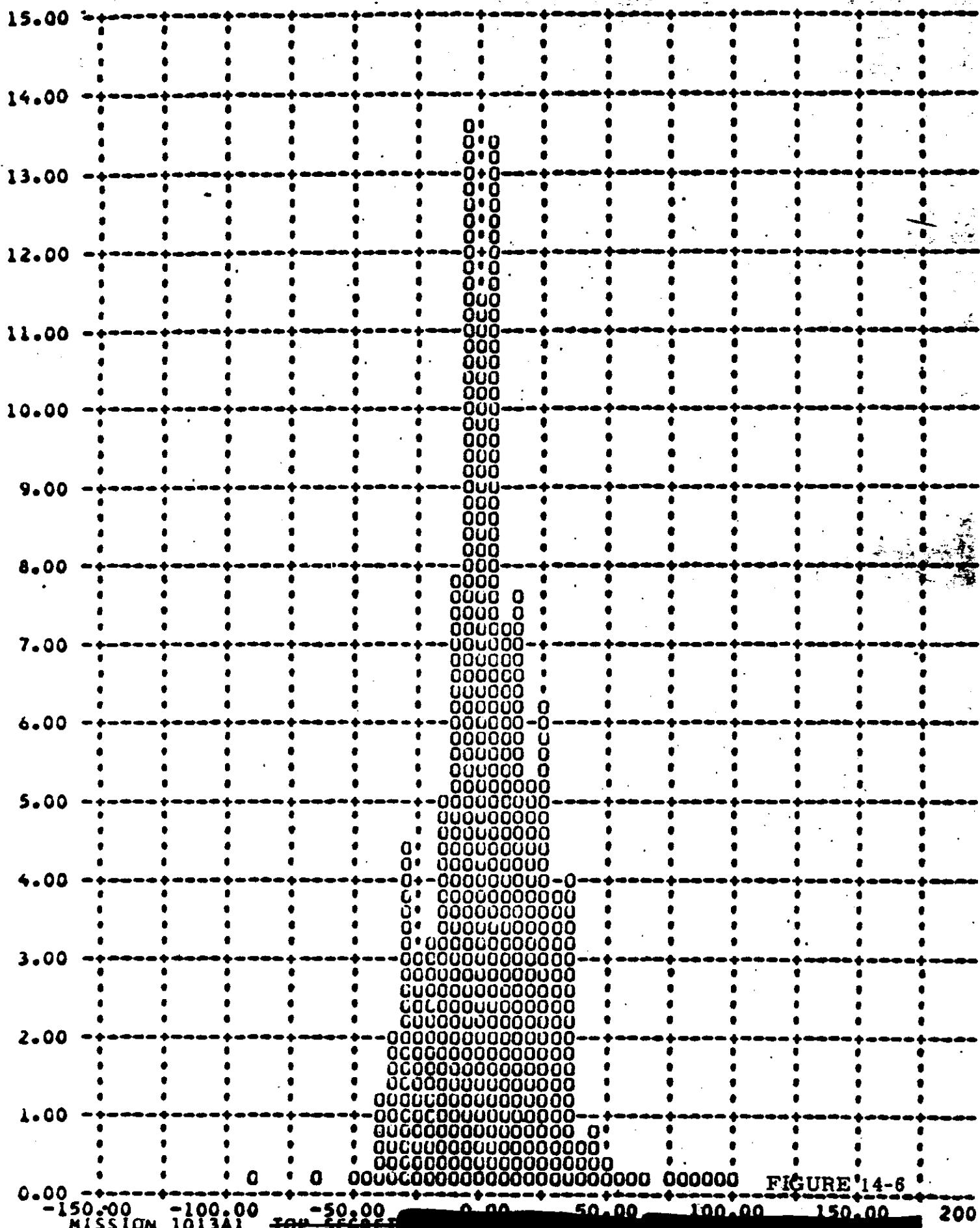


FIGURE 14-6

MISSION 1013A1 -TOP SECRET

SECTION 15

IMAGE SMEAR ANALYSIS

The frame correlation tape supplied to A/P by NPIC contains the binary time word of each frame of photography. A computer program has been assembled at A/P which calculates the exposure time of each frame and compares the camera cycle rate with the ephemeris to calculate the V/h mismatch. This data is combined with the vehicle attitude error and rate values of each frame and the crab error caused by earth rotation at the latitude of each frame. The program outputs the total along track and cross track IMC error and the limit of ground resolution that can be acquired by a camera regardless of focal length and system capabilities.

The computer rejects the first six frames of all operations as the large V/h error induced by camera start-up is not representative of the overall system operations. The frequency distribution of the V/h errors and resolution limits for both panoramic cameras are computer plotted and are shown in Figures 15-1 through 15-6.

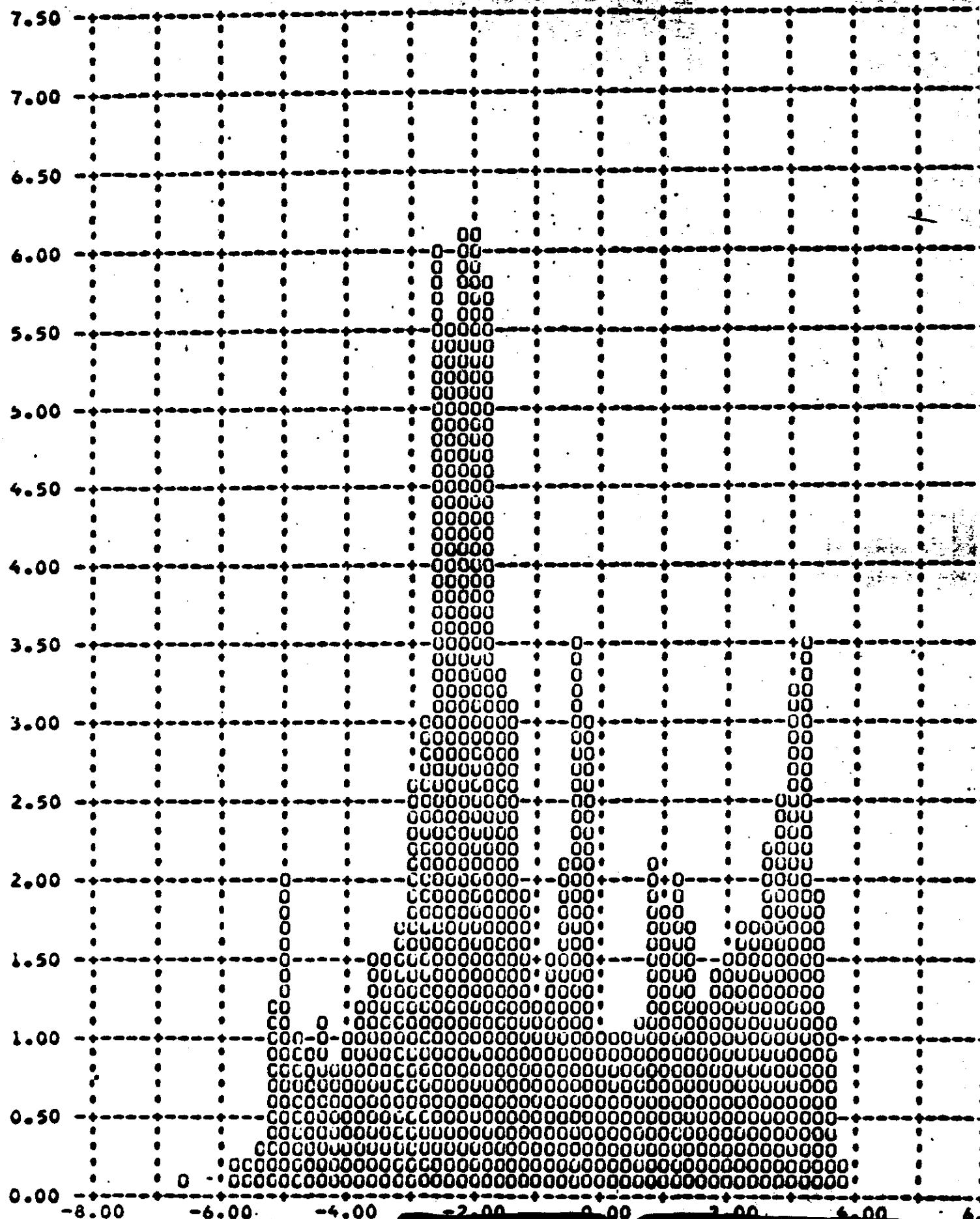
The summary table below presents the maximum V/h ratio error and resolution limits that existed during 90% of the photographic operations of Mission 1013-1 and the total range of values during all operations that were computed.

<u>Value</u>	FWD Camera		AFT Camera	
	<u>90%</u>	<u>Range</u>	<u>90%</u>	<u>Range</u>
V/h Ratio Error (%)	3.7	-6.6 to +3.8	4.5	-5.6 to +5.2
Along Track				
Resolution Limit (ft.)	7.8	0 to 18.0	9.6	0 to 21.0
Cross Track				
Resolution Limit (ft.)	8.3	0 to 9.8	8.2	0 to 9.8

The calculations show that the panoramic cameras were not smear limited during the mission. It should be noted that the 90% V/h mismatch value is the lowest that has been experienced during the last several missions.

MISSION 1013A1 TOP SECRET

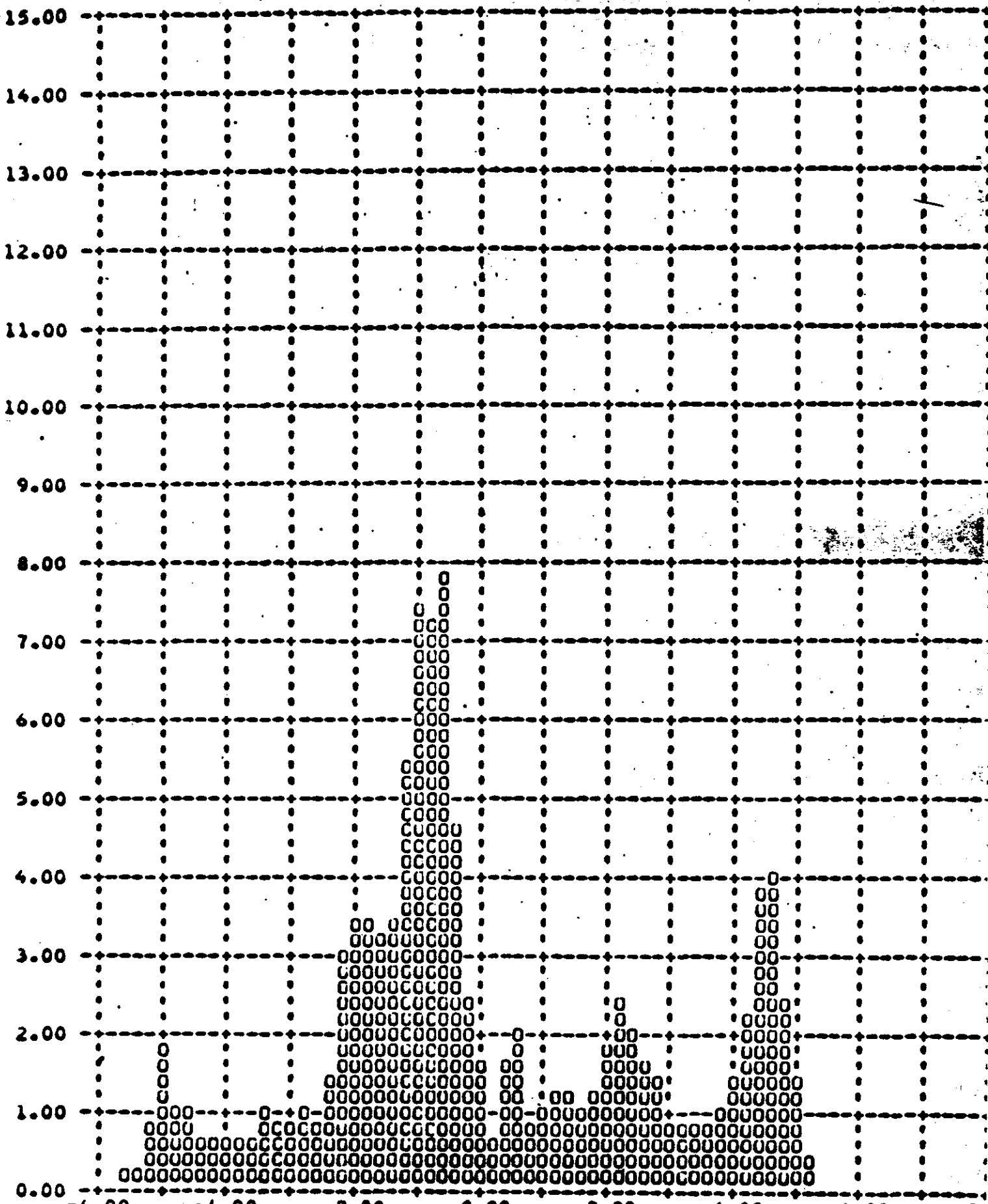
J-15 A BUCKET - FWD INSTA FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT
Y V/H RATIO ERROR - PERCENT (X) VERSUS FREQUENCY - PERCENT (Y)



MISSION 1013A1 TOP SECRET

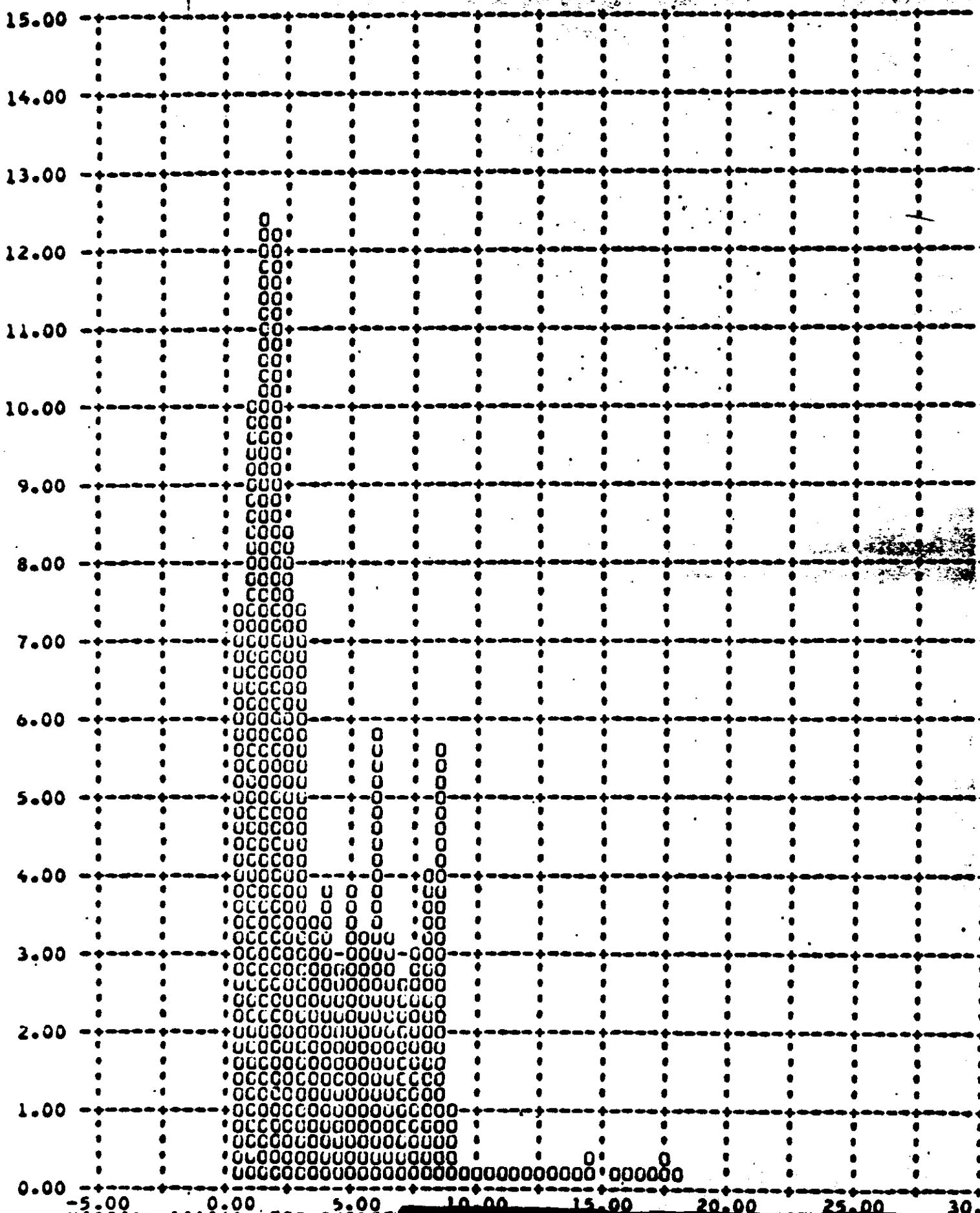
J-15 A BUCKET - AFT INSTR FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT 6.5

Y V/H RATIO ERROR - PERCENT (X) VERSUS FREQUENCY - PERCENT (Y)



MISSION 1013A1 TOP SECRET

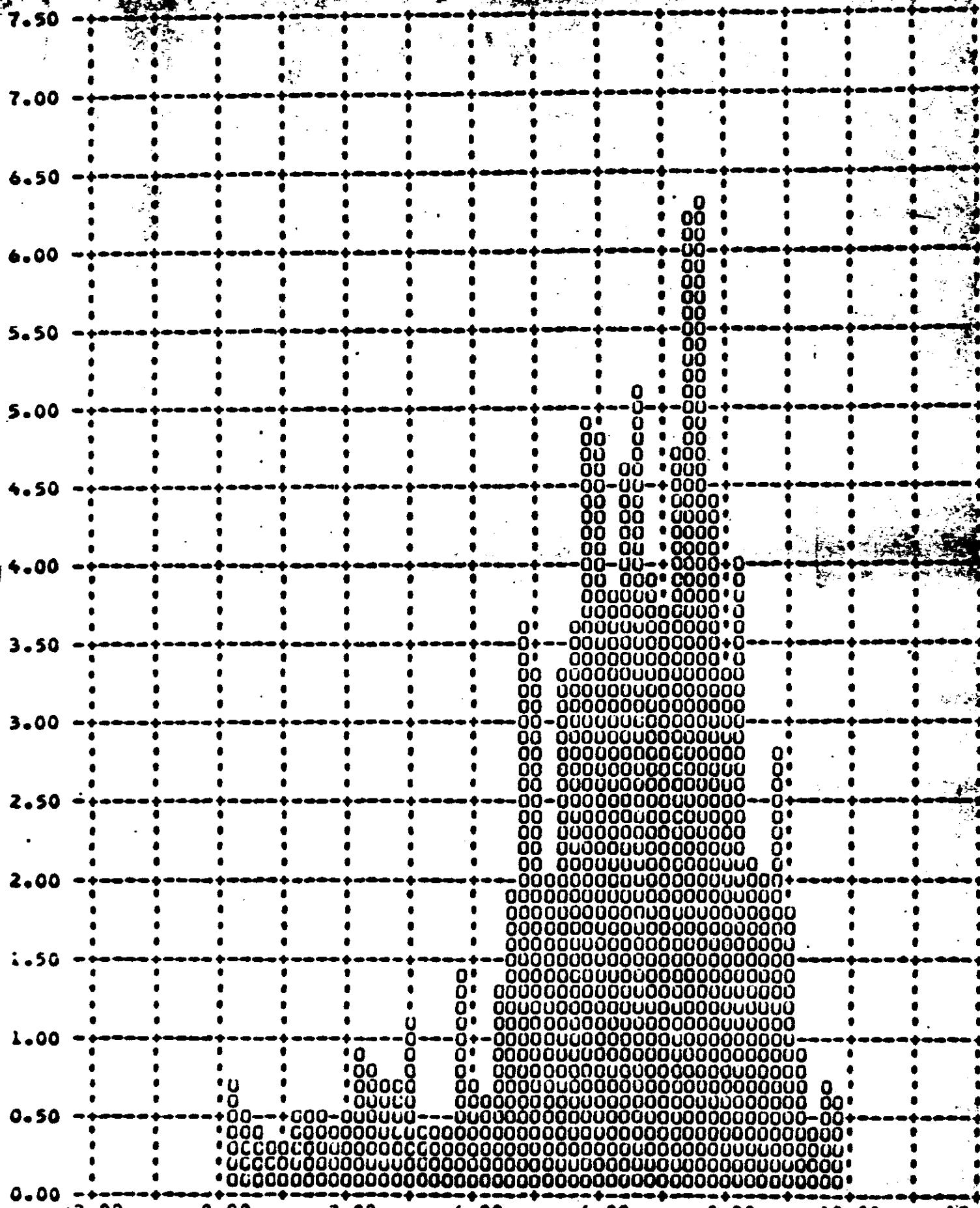
J-15 A BUCKET - FWD INSTR FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT = 7.
Y ALONG TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT



MISSION 1013A1 TOP SECRET

MISSION 1013A1 TOP SECRET

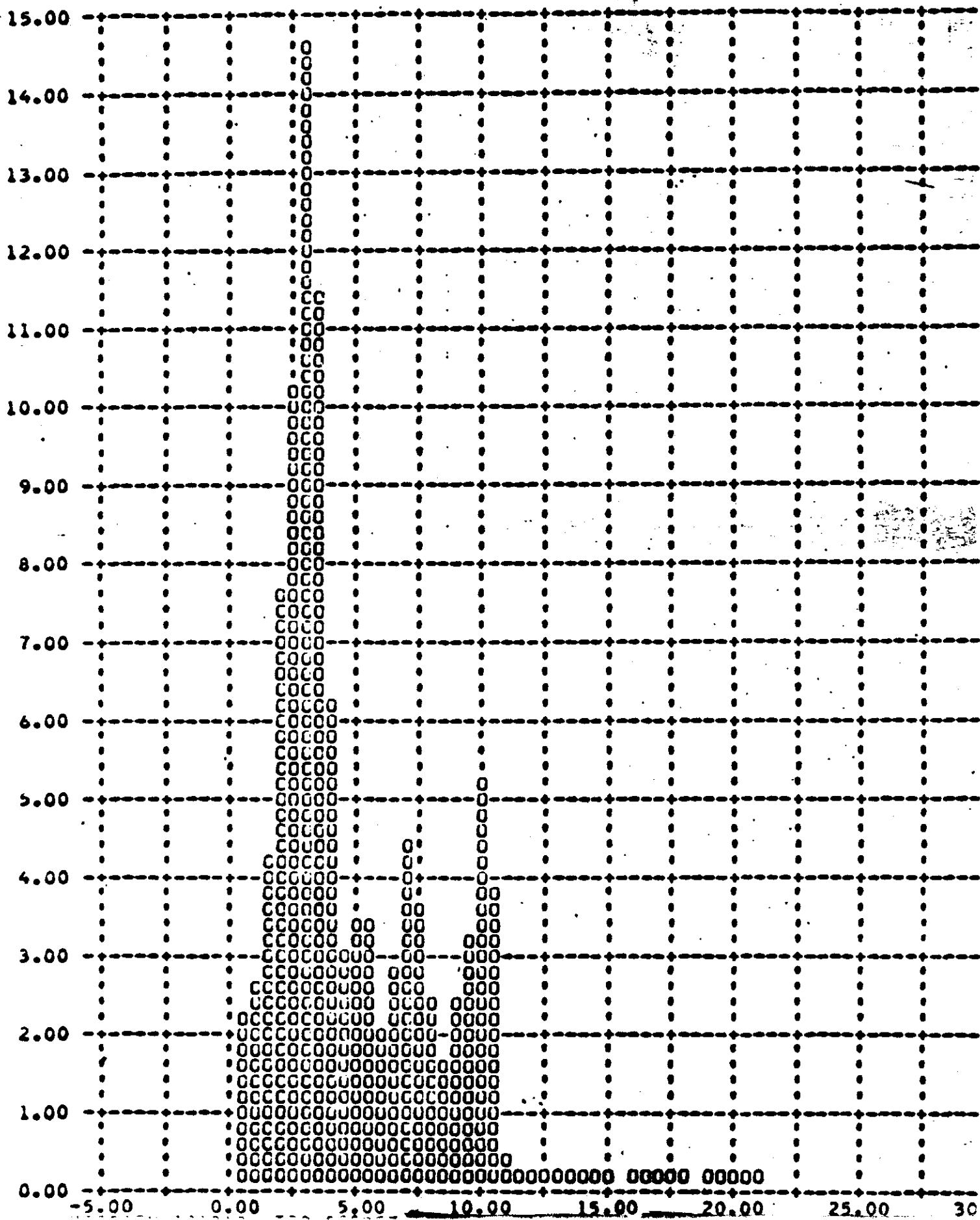
BUCKET TWO INSTRUMENT FRAMES 1-6 OR EACH OF OMITTED 90 PERCENT - 8.2
CROSS TRACK RESOLUTION LIMIT - PERCENT VERSUS FREQUENCY - PERCENT



MISSION 1013A1 TOP SECRET

MISSION 1013A2 ~~TOP SECRET~~

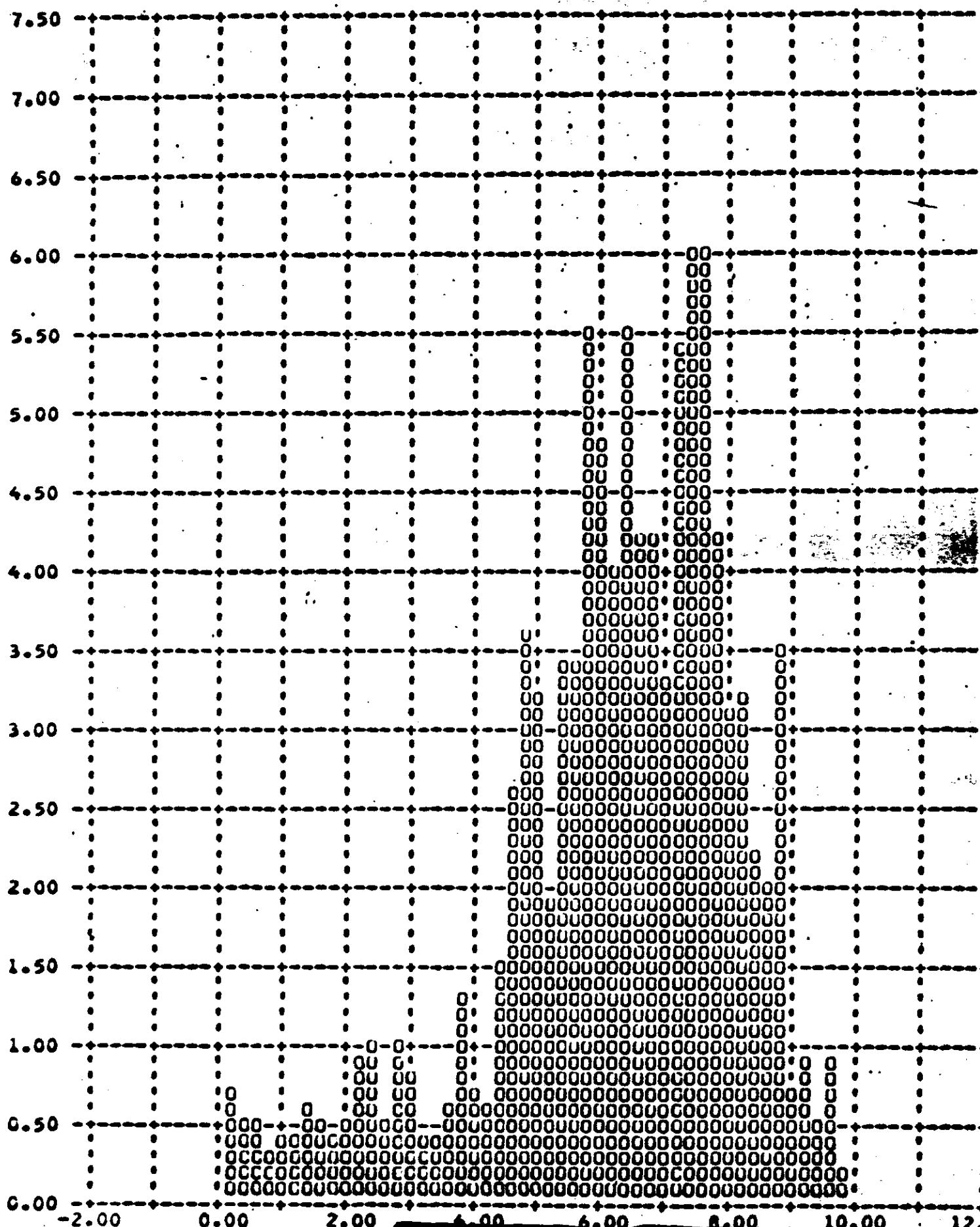
J-15 A BUCKET - AFT INSTR FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT
Y ALONG TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT (



MISSION 1013A2 ~~TOP SECRET~~

J-15 A BUCKET - AFT INSTR. FRAMES 1-6 OF EACH OP OMITTED 90 PERCENT

Y CROSS TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT (%)



MISSION 1013A2 ~~TOP SECRET~~

12.

~~TOP SECRET~~

SECTION 16

RADIATION DOSAGE

Each recovery system flown on a Corona mission contains a sealed packet of Eastman Type 3401 and Royal X Pan emulsions to determine the total radiation received at the take-up cassette. Both film types have been irradiated by LMSC at various levels and the base plus fog densities recorded after controlled processing.

Following recovery the film dosimeter packets are removed at A/P and processed with a pre-flight sample of the same film type and sensitometric control film. The resulting base plus fog density measurement of the dosimeter strips is used to ascertain the total radiation level. The table below presents the base plus fog readings for the dosimeter strips and the radiation level equivalents.

<u>Emulsion</u>	Mission 1013-1	
	<u>B + F</u>	<u>Radiation</u>
Type 3401	0.17	0.5 R
Royal X Pan	0.20	0.3 R

The mean total radiation seen by the take-up cassettes during both missions was approximately 0.5 roentgens. This level is essentially the same as received during recent missions and is below the level that will degrade the panoramic photography.

~~TOP SECRET~~

SECTION 17

SYSTEM RELIABILITY

Reliability calculations for the payload are based on a sample beginning with M-7. Hence both the major part of the Mural Program and the "J" Program are covered in the calculation. For certain auxiliaries, i.e., the stellar-index camera and the horizon cameras, the sample size is changed to recognize incorporation of modified equipment or new designs where reliability was one of the principal reasons for the modification. However, for primary mission function, the sample size is consistent with reliability reporting for the vehicle.

The reliability estimates of this section deal exclusively with the payload. Failures to achieve orbit or vehicle induced failures are thereby excluded. Recoveries before a complete mission has been completed are considered as full missions providing that early termination was caused by reasons not connected with payload operation. Film quality is not considered in the reliability estimate calculation. Hence, only electrical and mechanical functioning are considered.

The reliability estimate is also divided into primary and secondary functions. The primary functions are operation of the panoramic cameras, main camera door operation, operation of the payload clock, and recovery operations. The secondary mission functions are horizon camera operation excluding catastrophic open shutter failure mode, auxiliary data recording, and stellar-index camera operation.

Panoramic Camera Reliability

Sample Size - 78 opportunities to operate.

One failure - capping shutter on slave instrument on system M-7.

Assume - 3000 cycles per camera per mission.

Estimated Reliability = 97.8% at 50% confidence level.

Main Camera Door Reliability

Sample Size - 31 vehicles x 2 doors = 62 opportunities to operate.

Estimated Reliability = 99.0% at 50% confidence level.

~~TOP SECRET~~

Payload Command and Control

Sample Size - 3940 hours operation

1 failure.

Estimated Reliability = 95.9% at 50% confidence level

Payload Clock Reliability

Sample Size - 3940 hours operation

No failures

Estimated Reliability = 93.3% at 50% confidence level.

Estimated Reliability of Payload Functioning on orbit = 96.0% at 50% confidence level.

Recovery System Reliability

28 opportunities to recover

1 failure - improper separation due to water seal - cutter failure.

Estimated Reliability = 94.0% at 50% confidence level.

Stellar-Index Camera Reliability

Sample begins with J-5

Sample Size = 5950 cycles

Number of failures = 0

Estimated Reliability = 95.1% at 50% confidence

Horizon Camera Reliability

Sample includes J5 and up

Sample Size - 25,500 cycles

Estimated Reliability of Single Camera = 96.0% at 50% confidence level.

Estimated Reliability of Four Horizon Cameras at a Parallel

Redundant System = 99.8% at 50% confidence level

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SECTION 18

SUMMARY DATA

The comparison of the operating parameters and the performance achieved by previous missions has been difficult due to the large volume of data that results from each mission. Some of the pertinent characteristics from prior missions have been summarized in Tables 18-1 through 18-3.

The summary data was started with Mission 1004 as the J-05 camera system was the first to incorporate the major modifications of the titanium drum and scan arm, four roller scan head and Corona J capabilities. Only those missions that culminated in the recovery of some photography have been listed, therefore Missions 1003 and 1005 are deleted.

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MISSION SUMMARY

PERFORMANCE SUMMARY

MISSION NUMBER	CAMERA SERIAL NUMBER	M 1 P value	VISUAL MRS	SLIT MRS	APSPPI MRS/AM	SLIT SURFACE (μ)	SLIT (μ)	AVERAGE ALTITUDE (km)	20% ATTITUDE RANGE (°)		20% ALTITUDE RANGE (km)		90% V/H ERROR (m)	90% V/H CROSS TRACK	
									ALL	HIGH	PITCH	ROLL			
1004-1	FWD 124	83	76	350	97	109	115	127	0.48	0.42	1.04	30.0	21.0	8.1	7.7
1004-2	FWD 125	84	78	350	88	43	113	124	0.48	0.42	1.04	30.0	21.0	8.1	6.1
1004-3	AFT 127	85	75	350	85	104	117	120	0.48	0.42	1.04	30.0	21.0	8.1	6.1
1004-4	FWD 144	90	78	350	74	43	95	90	0.74	0.50	0.91	44.0	30.0	4.9	6.8
1004-5	FWD 145	90	75	350	71	43	95	97	0.41	0.42	1.14	26.8	23.5	15.4	6.7
1004-6	AFT 145	90	85	350	64	61	90	92	0.41	0.42	1.14	26.8	27.6	13.0	6.7
1007-1	FWD 144	85	85	350	63	43	97	92	0.48	0.44	1.43	37.6	25.9	20.9	2.6
1007-2	FWD 145	85	85	350	72	77	90	90	0.48	0.44	0.47	—	43.0	25.0	2.6
1008-1	FWD 150	85	80	350	73	43	95	91	0.39	0.39	0.39	43.0	23.9	23.9	—
1008-2	AFT 151	85	81	350	84	43	95	92	0.39	0.39	0.39	43.0	24.0	32.5	2.6
1009-1	FWD 154	85	85	350	85	82	92	91	0.38	0.38	0.71	42.9	24.0	24.0	4.3
1009-2	AFT 155	85	84	350	82	87	92	91	0.38	0.38	0.71	42.9	24.0	24.0	4.3
1010-1	FWD 152	85	85	350	85	80	90	95	0.38	0.38	0.71	22.7	27.6	3.3	8.3
1010-2	AFT 153	85	85	350	82	85	90	95	0.38	0.38	0.71	22.7	27.6	3.3	8.3
1011-1	FWD 160	90	90	350	75	80	90	95	0.48	0.48	0.65	33.6	23.9	27.6	2.6
1012-1	FWD 161	90	90	350	77	80	90	95	0.48	0.48	0.65	33.6	23.9	27.6	2.6
1012-2	AFT 162	90	90	350	75	80	90	95	0.48	0.48	0.65	33.6	23.9	27.6	2.6
1013-1	FWD 163	90	90	350	77	80	90	95	0.48	0.48	0.65	33.6	23.9	27.6	2.6

FIGURE 18-2

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TOP SECRET

EXPOSURE - PROCESSING SUMMARY

MISSION NUMBER	SOLAR ELEVATION RANGE, DEG. LOW HIGH	CAMERA RANGE, DEG. LOW HIGH	PREDICTED PROCESSING TIME	COMPUTED PROCESSING TIME	TERRAIN Q-MIN			TERRAIN Q-MAX			CLOUD Q-MIN			CLOUD Q-MAX			OVER EXPOSED (%)	UNDER EXPOSED (%)	CLOUD COVER (%)
					P	V	T	LOW	HIGH	MEAN	LOW	HIGH	MEAN	LOW	HIGH	MEAN			
					P	V	T	P	V	T	P	V	T	P	V	T			
1004-1	2.5	12.4	0.1	0.1	7.9	17	0.79	2.1	0.26	1.82	0.83	0.78	0.63	2.43	2.04	2.08	0	0	35
1004-2	2.1	12.4	0.1	0.1	8.0	17	0.80	2.0	0.22	1.56	0.78	0.70	0.93	2.43	1.98	2.03	0	0	35
1004-3	1.0	13.1	0.1	0.1	8.3	13	0.83	1.8	0.23	1.80	0.83	0.78	0.36	2.30	1.90	2.03	0	0	35
1004-4	1.0	13.1	0.1	0.1	7.6	17	0.77	1.8	0.29	1.91	0.81	0.73	0.36	2.38	1.93	2.46	1.88	1.96	35
1005-1	1.0	13.1	0.1	0.1	9.1	18	0.91	4.0	0.23	1.61	0.71	0.68	0.80	2.31	1.58	1.52	1.3	2.40	2.20
1005-2	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1005-3	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1005-4	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1006-1	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1006-2	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1006-3	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1006-4	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1007-1	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1007-2	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
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1008-1	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
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1008-3	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1008-4	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1009-1	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1009-2	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1009-3	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1009-4	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1010-1	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1010-2	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1010-3	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1010-4	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1011-1	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1011-2	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
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1011-4	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1012-1	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1012-2	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1012-3	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1012-4	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1013-1	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1013-2	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1013-3	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20
1013-4	1.0	13.1	0.1	0.1	9.0	18	0.90	3.9	0.23	1.60	0.70	0.68	0.80	2.30	1.57	1.52	1.2	2.40	2.20

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