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17 November 1967

TO:

[REDACTED]
C. Murphy
A. Johnson
[REDACTED]

THRU:

FROM:

SUBJECT: MISSION 1038-1 and 1038-2 FINAL REPORT (J-34)

Enclosed is the Final Performance Evaluation Report for
Mission 1038-1 and 1038-2.

[REDACTED]
Manager
Advanced Projects

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CCRONA J

PERFORMANCE EVALUATION REPORT

Mission 1038-1 and 1038-2

FIV 1629, J-34

October 10, 1967

Approved

Advanced Projects

Approved

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FOREWARD

This report details the performance of the payload system during the operational phase of the [REDACTED] Flight Test Vehicle 1629.

Lockheed Missiles and Space Company has the responsibility for evaluating payload performance under the Level-of-Effort and "J" System contracts.

This document is the final payload test and performance evaluation report for Mission 1038-1 and 1038-2 which was launched on 14 January 1967.

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INTRODUCTION

This report presents the final performance evaluation of Missions 1038-1 and 1038-2 of the Corona Program. The purpose of this report is to define the performance characteristics of the J-34 payload system and to identify the source of in-flight anomalies.

The performance evaluation was jointly conducted by representatives of Lockheed Missiles and Space Company (LMSC) and ITEK at the facilities of NPIC and AFSPPF. The off-line evaluation using Corona engineering photography acquired over the United States was performed at the individual contractors plants.

The quantitative data used for this report is obtained from government organizations. The diffuse density data, and MTF/AIM resolution are produced by AFSPPF. The vehicle attitude error values, frame correlation times are made at NPIC who also supply the Processing Summary reports published by [REDACTED]

Computer programs developed by A/P are utilized to calculate and plot the frequency distribution of the various contributors to image smear to permit analysis and correlation of the conditions of photography to the information content and quality of the acquired pictures. Computer analysis of the exposure, processing and illumination data provides the necessary data to analyze the exposure criteria selected for the mission.

SECTION 1

SYSTEM PERFORMANCE

A. MISSION OBJECTIVES

The payload section of Mission 1038, placed into orbit by Flight Test Vehicle #1629 and SLV-2A booster #495, consisted of two panoramic cameras, two Stellar-Index cameras, two Mark 5A recovery capsules and a space structure to enclose the cameras and provide mounting surfaces for all equipments. Figure 1-1 presents an inboard profile of the J-3^{1/4} payload system. This Corona "J" system is designed to acquire search and reconnaissance photography of selected areas of the earth from orbital altitudes. The planned missions were 4/5/6 days -1 and 7/6/5 -2 with no inactive period.

B. MISSION DESCRIPTION

The payload was launched from Vandenberg Air Force Base (VAFB) at 2128:18 Z (1328:18 PST) on 14 January 1967. Ascent and injection were normal and the achieved orbit was within nominal tolerances. Tracking and command support was effected by the Air Force Satellite Control Facility consisting of tracking and command stations at [REDACTED]

[REDACTED] under central control of the Satellite Test Center at Sunnyvale, California. Mission 1038-1 consisted of five days operation and was completed by air recovery on 19 January 1967. Mission 1038-2 was completed with an air recovery on 26 January 1967 following seven days of photographic operations.

The comparison of the planned and actual orbit parameters is tabulated as follows:

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ORBITAL PARAMETERS

<u>Parameter</u>	<u>Predicted</u>	Orbit 42 <u>Actuals</u>	Orbit 120 <u>Actuals</u>
Period (Min.)	90.104	90.088	89.919
Perigee (N. M.)	99.834	96.944	98.505
Apogee (N. M.)	204.925	208.05	204.100
Inclination (Deg.)	80.00	80.075	80.074
Perigee Latitude (Deg. N.)	20.35	29.232	46.518
Eccentricity	0.014623	0.01546	0.01471

C. PANORAMIC CAMERAS

Both instruments operated satisfactorily. The image quality was degraded by atmospheric conditions. The slave camera starboard H.O. imagery was veiled.

D. STELLAR-INDEX CAMERAS

Both instruments produced acceptable imagery for data reduction.

E. OTHER SUBSYSTEMS

The clock, instrumentation, command, PMU and thermal control subsystems performed satisfactorily throughout both missions.

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SCHEMATIC INBOARD PROFILE - COPORA J SYSTEM

MISSION 1038

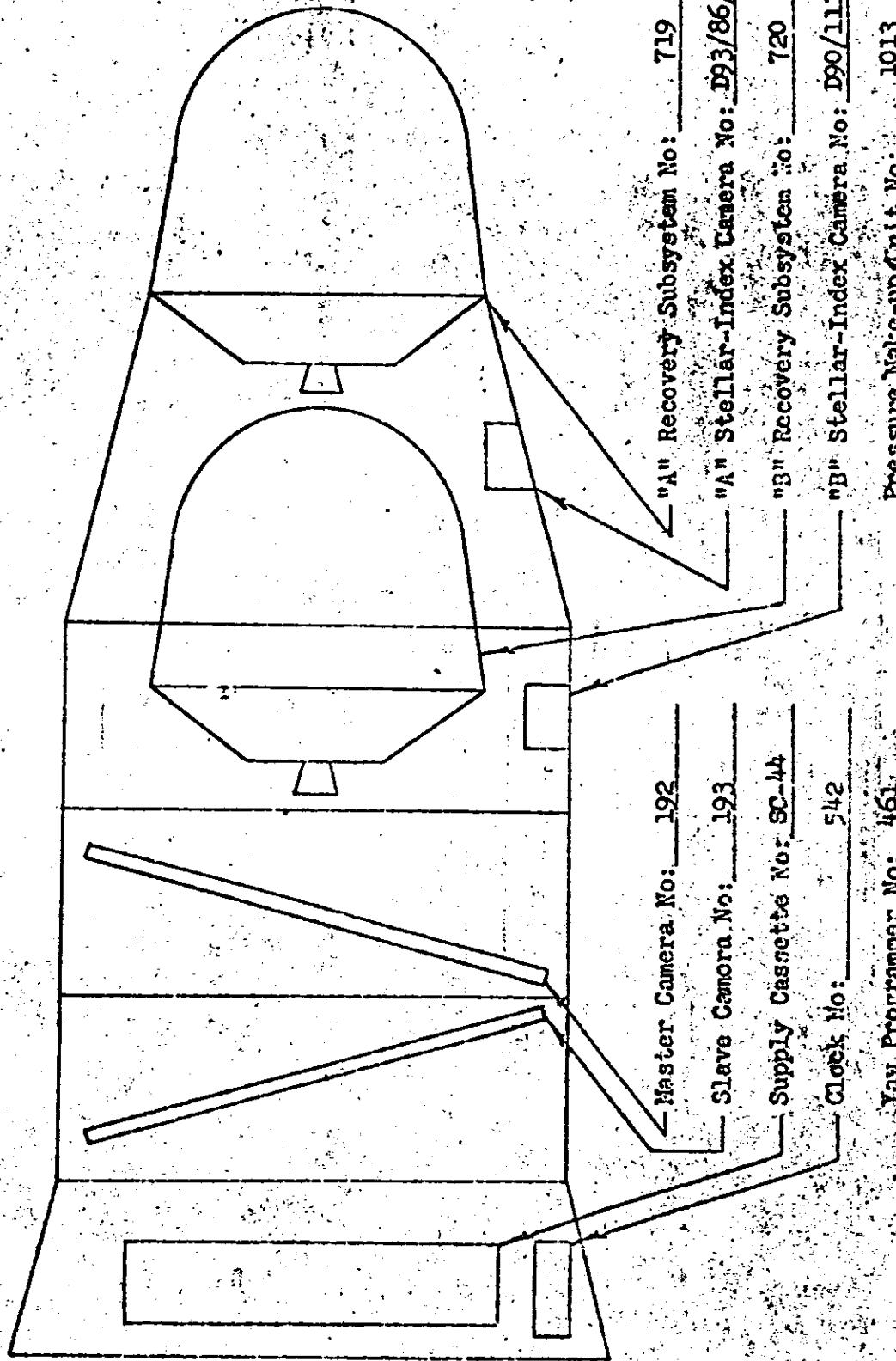


FIGURE 1-1

No. [REDACTED]

SECTION 2

PRE-FLIGHT SYSTEMS TEST

A. ENVIRONMENTAL TESTING

1. Test Objective

As a standard procedure, the J payload systems are subjected to thermal/altitude environmental testing which simulates orbital environment. One of the purposes of this test is to demonstrate the system susceptibility to corona discharge. Such discharge fogs the film thus degrading the operational photography.

2. General Test Information

The J-34 payload system was subjected to an environmental HIVOS chamber test from 1 through 8 March 1966. A complete "J" mission was simulated. The -1 and -2 missions consisted of approximately 3000 cycles per mission per instrument. Camera internal operating pressures ranged from 0.5 microns to 80 microns. This system was the first to incorporate the new "double intermix" command system.

3. Panoramic Camera Performance

The dynamic performance of both instruments was satisfactory for the center of format switch, lens rotation and film transport. The 99/101 percent clutch ratios were 6/6 on both instruments.

Instrument cycle periods were fast during the first 8 orbits of the -1 mission and excessive coasting was noted. These anomalies occurred when temperatures were high (100°F).

The slave instrument payload had scratches which were traced to the payload riding on a roller flange after manual handling during a test interruption. Both instruments produced material completely free of corona. All recorded pan instrument data was excellent.

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4. Stellar/Index Performance

The S/I instruments operated satisfactorily. Examination of the payload on D-93 showed one instance of stellar shutter malfunction (high density). Instrument D-90 had 26 frames out of 452 frames affected by corona. The corona density was within acceptable density levels. The stellar shutter D-93 was checked over and certified to be acceptable.

This "J" system was the first to provide S/I control from the slave instrument. The smear pulses on the slave payload were normal.

5. Clock Performance

The clock accuracy was satisfactory. The clock readouts were correlated with the IRIG "C" time and the results are shown in Table 2-1.

6. Command and Instrumentation Performance

The instrumentation functioned normally. The command system was satisfactory except the V/H delay stepper switch did not home with brush 14 in orbits 13, 14, 15 and 16. The transfer from -1 to -2 was accomplished by the secure real time command (KZ-38).

7. Pressure Make-Up System Performance

The PMU operated normally. The average gas consumption on both missions was 7.92 psi/min. The pressure ranged from 44 microns during operates to 0.5 microns with the instruments off.

8. Temperature Summary

Average instrument temperatures ($^{\circ}$ F) for several days in both missions are listed below:

-1 Mission		Master		Slave		Beta Angle	
		Day	High	Low	High		
	1		105	70	95	70	53
	2		100	72	100	72	53
	3		100	65	100	66	53
	4		87	64	82	65	53
-2 Mission		Master		Slave		Beta Angle	
		Day	High	Low	High		
	1		84	74	86	76	0
	2		80	58	80	62	0
	3		76	52	78	56	0
	4		100	60	100	60	0

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B. RESOLUTION TEST

Resolution and theodolite tests of the J-34 system were completed on 6 April 1966. Results of the thru-focus resolution tests of pan instruments 192 and 193 show the following characteristics;

MASTER PAN INSTRUMENT NO. 192

Maximum high contrast resolution 169 lines/mm at -0.000 focal position.

Maximum low contrast resolution 112 lines/mm at -0.001 focal position.

SLAVE INSTRUMENT NO. 193

Maximum high contrast resolution 165 lines/mm at -0.001 focal position.

Maximum low contrast resolution 119 lines/mm at -0.001 focal position.

The resolution test data for both instruments as shown in Figures 2-1 and 2-2, has been reviewed and appears normal in all respects. The demonstrated resolution performance meets the system requirements specification.

C. LIGHT LEAK TEST

The first live payload light leak test on the J-34 system was performed on 19 April. This test was the first in which the test exposure was increased to 90 minutes (from 60 minutes). This exposure produces on the high speed test film a density equivalent to the density that would be produced on flight film by about 90 percent of the computed direct solar illumination reaching a system in four orbits. Since portions of the system do not see this solar illumination or are affected primarily by the lower intensity albedo illumination, such conditions are considered in the evaluation of light leak effects. However, this first light leak test showed severe fogging at several points on the film from felt seals on the slave instrument drum, and minor fog from a leak at the teardrop fitting just forward of the fairing access door.

After correcting the teardrop leak and reworking the slave felt seals, a second light leak test was run on 25 April. The 90 minute soak was also used on this test. In addition, the instruments were cycled at high speed to the sit position. This procedure causes the instrument

No.

scan arms (stovepipes) to come to rest near the start of scan position, which is the normal sit position during flight operations. In all previous J system tests, cycling to the sit position has been done at low speed which causes the scan arms to come to rest at the "home" position near the end of scan. The effect of this procedural change is that the scan arm tends to obscure leaks from the output edge of the drum. The results from the second light leak test verified correction of the teardrop leak but showed severe leaks from the felt seal at the input side of the slave instrument. These leaks also produced minor fog on the master film in the area where this film passes the slave instrument. On the basis of our regular evaluation criteria, the system freedom from light leakage was acceptable, except for the slave instrument drum area which was not acceptable.

Light leakage from the drum area of the main instrument has been a continuous problem of the J program. It has been the primary reason for rework and retest for light leakage qualification at A/P. While some minor improvements have been made, the problem will continue until adequate drum seals are provided. Although rework of felt seals has enabled systems to pass the light leakage tests, flight results suggest that creep in the stretched felt seals has effectively undone the rework. As a result of flight performance evaluations, Boston has investigated the problem and concluded that simple modifications to the felt seals will solve the problem. Although a Boston ECO is expected in the near future, no adequate felt seals are presently available.

Since all of the severe fog from the felt seals is within three frames of the end of pass, and since no effective means for correcting the problem is available, the light leakage requirement for the slave instrument was waived and the J-34 system light leakage performance was considered acceptable for flight.

D. J-34 FLIGHT READINESS AND LOADING EVALUATION

The final Flight Readiness Test payload was processed and evaluated on 6 January 1967. The test payload showed that the functioning and data recording of both pan instruments was acceptable for flight loading.

Master Instrument No. 192 showed two notable defects. On some (but not all) frames there was distinct banding of fogged main formats. A check of timing mark spacing showed scan head velocity variations of up to 20 percent in such frames. Such variations in scan velocity occurred frequently in early "J" systems but no direct correlation with quality degradations has been established. Since a check of the scan arm-lens latch mechanism showed normal operation, the banding is considered to be within acceptable limits. Also on the Master Instrument

No.

it was noted that a film rail partially obscured the timing marks for about one-half of the scan. This condition was not expected to have any significant effect on use of the timing record. Payload from Slave instrument No. 193 showed excellent data recording and no indication of functional defects. A slight minus density streak was noted in the pan format from scraped emulsion on the exposure slit. This scraped emulsion was removed before flight loading.

Rail scratching of both pan instrument films was average from inboard (data block) rails and much less than average from out-board rails. Although the system does not have polished rails, the only evidence of emulsion buildup was slight shadows at one or two shrinkage markers on each instrument. It is noted that humidity of the work area was much less than usual (20% to 30% rather than 40% to 50%) during readiness and loading operations. Low humidity appears to be very significant in minimizing rail scratches and accumulations of scraped emulsion.

Since this system is the first to be prepared for flight with full Phase III procedures, this was the first time that a live film bench test of S/I units was made as a part of the Flight Readiness Procedure. Ten cycle samples from each unit were processed and evaluated on 5 January. These samples showed normal operation of Stellar and Index components of S/I units D-93 and D-90, which were installed for flight in the "A" and "B" positions respectively.

Loading of the main instrument supply cassettes proceeded in a routine manner on 8 January. Sensitometric samples of all flight payloads were prepared and showed acceptable characteristics. Since flight payload from the J-38 (PG-2) system had shown abnormally high base-plus-fog density, two simulated "full" processing samples were prepared from both main instrument flight payloads. The flight samples processed at the "full" level showed lower base-plus-fog densities than the manufacturer's control sample. It is also noted that these samples showed the highest maximum densities that we have ever observed on type 3404 film. Maximum densities of 2.81 and 2.85 resulting from one of the full level processes can only be considered advantageous in that they represent an extension of the effective exposure range.

Final tracking and light leak checks were completed on 9 January. During the tracking tests, it was noted that there were distinct and continuous scratch-like streaks in the back coating of both main films. Similar, though less severe, marking has been noted during flight loading on many previous systems. Such marking has

been described by the manufacturer as an unavoidable defect in the manufacturing process which produces no defect whatsoever in image quality. Microscopic examination by Performance Evaluation of both processed and unprocessed flight samples supports the statement that image quality is not affected.

The final light leak checks showed an exceptional freedom from light leakage.

The J-34 system was certified for flight on 9 January.

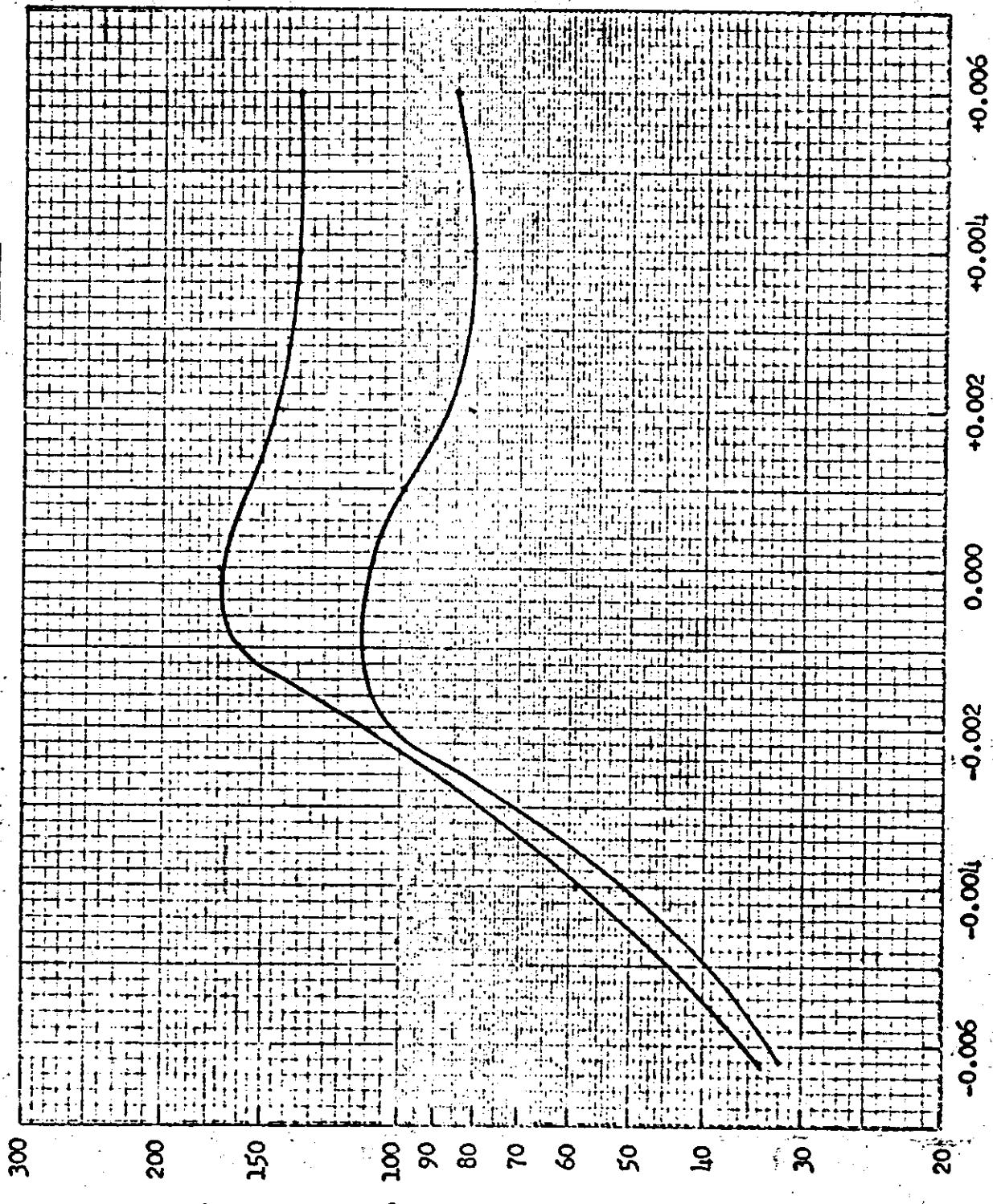
E. COMMENTS

The J-34 payload system was the first system to utilize the full Phase III concept (Factory to Pad).

System testing was completed on 21 June 1966 and the payload system went into Phase II storage. The system was removed from storage on 2 August 1966 for 90 day recycle testing. The system was returned to Phase III storage on 11 August 1966. The system was removed from storage on 7 November 1966 for flight preparations. The flight date was cancelled and the system was returned to a partial Phase III storage on 29 November 1966. The system was removed for flight preparations again on 27 December 1966. All flight preparations were completed and the system was shipped to VAFB on R-3 for launch preparations. All confidence tests were completed and the system was launched on the scheduled date.

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PRE-FLIGHT DYNAMIC RESOLUTION

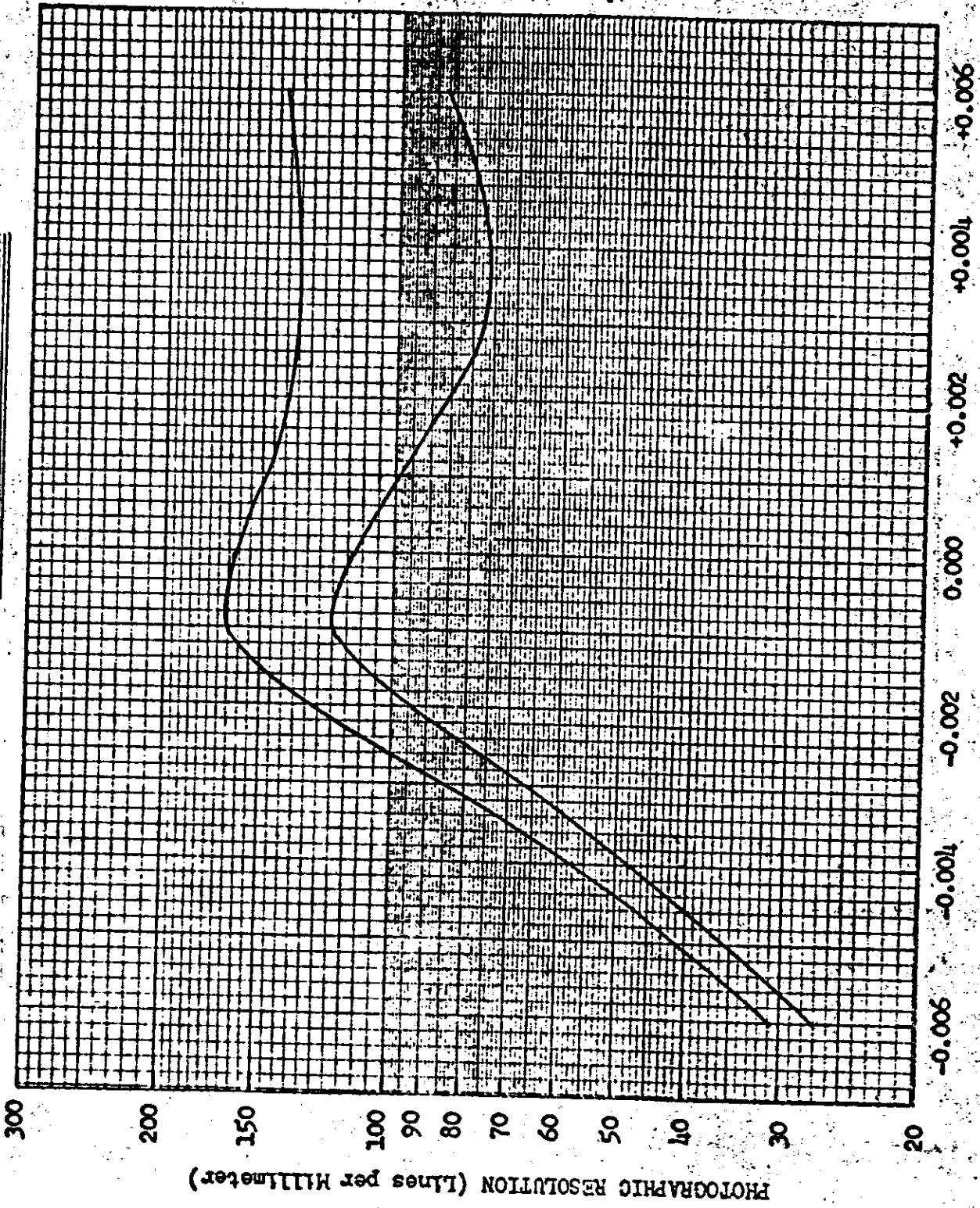


THROUGH FOCUS INCREMENTS (Inches)

Figure 2-1

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PRE-FLIGHT DYNAMIC RESOLUTION



J-34 HIVOS CLOCK CORRELATION

	IRIG REV DAY-HR-MIN-SEC	IRIG SECONDS	CLOCK SECONDS	DELTA IRIG	DELTA CLOCK	ERR OF
1	60 8 31 46.220	5214706.220	47701.828	---	---	---
1	60 9 34 51.220	5218491.220	51486.837	3785.000	3785.009	0.009
2	60 10 1 56.320	5220116.320	53111.931	1625.100	1625.094	-0.006
2	60 11 4 51.380	5223891.380	56886.983	3775.060	3775.052	-0.008
3	60 11 31 41.420	5225501.420	58497.034	1610.040	1610.051	0.011
3	60 12 34 51.320	5229291.320	62286.933	3789.900	3789.899	-0.001
5	60 15 35 46.270	5240146.270	73141.899	10854.950	10854.966	0.016
0	7 4 0.050-DELTA TIME			TOTAL ACCUM.	ERROR	0.021
6	62 8 1 53.040	5385713.040	21273.738	---	---	---
6	62 9 4 58.380	5389498.380	25059.080	3785.340	3785.342	0.002
7	62 10 34 58.280	5394898.280	30458.986	5399.900	5399.906	0.006
8	62 11 1 53.340	5396513.340	32074.045	1615.060	1615.059	-0.001
9	62 12 31 53.380	5401913.380	37474.088	5400.040	5400.043	0.003
9	62 13 34 58.340	5405698.340	41259.049	3784.960	3784.961	0.001
10	62 14 1 53.380	5407313.380	42874.086	1615.040	1615.037	-0.003
11	63 9 4 24.840	5475664.840	111425.559	68551.459	68551.472	0.013
12	63 10 34 25.140	5481265.140	116825.851	5400.300	5400.292	-0.008
13	63 12 4 24.880	5486664.880	122225.595	5399.740	5399.744	0.004
14	63 12 31 19.800	5488279.800	123840.529	1614.920	1614.934	0.014
15	63 14 1 20.220	5493680.220	129240.940	5400.420	5400.411	-0.009
16	64 8 1 0.160	5558460.160	194020.875	64779.940	64779.935	-0.005
16	64 9 4 5.220	5562245.220	197805.932	3785.060	3785.057	-0.003
1	64 9 31 9.000	5563869.000	199429.705	1623.780	1623.773	-0.007
1	64 10 34 22.198	5567662.198	203222.911	3793.198	3793.206	0.008
2	64 11 1 7.200	5569267.200	204827.911	1605.002	1605.000	-0.002
2	64 12 4 12.080	5573052.080	208612.805	3784.880	3784.894	0.014
3	64 13 34 12.130	5578452.130	214012.848	5400.050	5400.043	-0.007
4	64 14 1 7.140	5580067.140	215627.856	1615.010	1615.008	-0.002
4	64 15 4 12.200	5583852.200	219412.916	3785.060	3785.060	0.000
2	7 2 19.160-DELTA TIME			TOTAL ACCUM.	ERROR	0.017
5	65 8 0 17.240	5644817.240	17925.564	---	---	---
5	65 9 3 22.540	5648602.540	21710.848	3785.300	3785.284	-0.016
6	65 10 33 22.260	5654002.260	27110.579	5399.720	5399.731	0.011
8	65 12 30 7.610	5661007.610	34115.938	7005.350	7005.359	0.009
8	65 13 33 22.380	5664802.380	37910.698	3794.770	3794.760	-0.010
9	65 14 0 17.350	5666417.350	39525.671	1614.970	1614.973	0.003
10	66 8 4 30.750	5731470.750	104579.153	65053.400	65053.482	0.082
10	66 9 7 35.680	5735255.680	108364.068	3784.930	3784.915	-0.015
12	66 11 4 30.760	5742270.760	115379.161	7015.080	7015.093	0.013
12	66 12 7 35.640	5746055.640	119164.029	3784.880	3784.868	-0.012
13	66 12 34 30.880	5747670.880	120779.279	1615.240	1615.250	0.010
13	66 13 37 35.680	5751455.680	124564.073	3784.800	3784.794	-0.006
14	67 9 5 25.790	5821525.790	154634.202	70070.109	70070.128	0.020
15	67 9 32 20.810	5823140.810	156249.244	1615.020	1615.042	0.022
15	67 10 35 25.920	5826925.920	200034.340	3785.110	3785.096	-0.014
16	67 11 2 20.910	5828540.910	201649.322	1614.990	1614.982	-0.008
16	67 12 5 25.880	5832325.880	205434.294	3784.970	3784.972	0.002
2	4 5 8.640-DELTA TIME			TOTAL ACCUM.	ERROR	0.091

TABLE 2-1

SECTION 3

FLIGHT OPERATIONS

A.

SUMMARY

All launch, ascent, and injection events occurred as programmed; however telemetry Link II failed to turn on during ascent and was inoperative for the remainder of the flight. The orbit achieved was within the 3 sigma dispersions.

Both panoramic cameras operated satisfactorily throughout the flight. Average cycle rates for both cameras deviated from the pre-flight calibrations by less than 2 per cent.

Both the -1 and -2 Stellar/Index cameras and the clock system performed satisfactorily throughout the flight.

The AP command system failed to respond once to real time command (RTC) 11 and once more to RTC 15 during the -2 mission.

The -1 mission thrust cone skin temperature sensor was inoperative for the duration of the mission.

The pressure make-up system operated satisfactorily throughout the flight with 690 PSIA supply remaining.

The on-orbit internal temperatures were slightly higher than recent systems.

Both recovery systems operated normally throughout the flight.

Kik-Zorro 38 (early A to B switchover) was utilized on Rev 72 and performed satisfactorily.

The orbit sine function generator (OSFG) performed normally for the duration of the flight.

The -1 mission was successfully terminated on Rev 81 by air-catch recovery and the -2 mission was completed by air-catch recovery on Rev 193.

No. [REDACTED]

B. PANORAMIC CAMERA PERFORMANCE

Both panoramic cameras operated normally throughout the mission. Cycle period data for the engineering passes monitored are tabulated in Table 3-1. Camera system dynamic operation, 99/101 percent clutch operation, start-up, shut-down, and transport functions were normal for all passes monitored. The cut and wrap operation and transfer to the -2 system occurred as programmed, utilizing the Kik-Zorro 38 (early A to B switchover) command.

The panoramic film was exhausted on Pass 183 frame 55 and frame 71 for the Master and Slave cameras respectively.

Panoramic Film Consumption-Cycles

	<u>Actual</u>	
	<u>Master</u>	<u>Slave</u>
Sample off-Spooling	20	20
Pre-Launch	126	125
-1 Mission	2869	2870
-2 Mission	3029	3034
Total	6044	6049

FMC Match

The V/H ramp to orbit match was acceptable throughout the flight. The following settings for RTC's 6, 8, and 10 were utilized to obtain the optimum FMC match during the flight:

	RTC Commands			Remarks
	<u>6</u>	<u>8</u>	<u>10</u>	
RTC	6	5	6	Launch thru Rev 3
Positions	7	3	7	Rev 4 thru Rev 92
	6	4	7	Rev 93 thru the end of the mission

C. STELLAR/INDEX CAMERA PERFORMANCE

Both the -1 and -2 Stellar/Index cameras operated properly on all engineering passes.

D. INSTRUMENTATION AND COMMAND SYSTEM PERFORMANCE

The instrumentation and command systems operated properly throughout the flight with the following exceptions:

1. The -1 mission thrust cone skin temperature sensor became inoperative between the mating confidence and the launch countdown and remained inoperative for the duration of the -1 mission.
2. The AP command system failed to respond to an RTC on two different occasions during the flight. On both passes the vehicle command tone verification monitors indicated normal vehicle decoder output. Additional commands were issued and the command system responded normally on both occasions to step commands to the desired terminal positions.

The first missed command occurred on pass 169 [REDACTED] RTC 11 was in position 11 and one command was issued to step RTC 11 to position one. However, the command was not executed.

The second missed command occurred on pass 177 [REDACTED] RTC 15 was in position 13 and 12 commands were issued to step RTC 15 to position 9. However, the 9th command was not executed.

E. CLOCK SYSTEM PERFORMANCE

Clock system operation was normal for the duration of the flight. Good correlation between the flight clock and [REDACTED] Tracking Station time was obtained. Table 3-2 contains the correlation data.

F. PRESSURE MAKE-UP SYSTEM PERFORMANCE

Pressure make-up system performance was normal throughout the mission. Average gas consumption was approximately 9.0 psi/min for the 223 minutes of total operate time. The system had a surplus of 690 PSIA at the end of the mission.

G. THERMAL ENVIRONMENT

Temperature data for the [REDACTED] acquisitions are included in Table 3-3. The average instrument temperatures ranged from a high of 97°F on the Master and 89°F on the Slave to a low of 75°F on the Master and 69°F on the Slave.

The average J-34 payload system temperatures were approximately 10°F higher than the J-24 payload system, even though the orbits were the same. The J-34 payload system was launched at 13:38 PST on 14 January 1967 and the J-24 payload system was launched at 14:31 PDT on 22 September 1965. No explanation of this difference in average payload system temperature is available. A detailed analysis is continuing to ascertain the causes of this anomaly.

J-34 FLIGHT 01-13-67

				INST. 192		INST. 193							
REV.	OP.	RAMP TUR	SYSTEM	ACTUAL UNIT	SYSTEM	ACTUAL UNIT	SYSTEM	192					
MODE	R	A	SECS	CALIB.	DEV.	DEV.	DEV.	DEV.	DIF				
009	A	7	3	197	4.714	4.695	1.50F	0.57F	4.525	3.60F	-0.01F	-2.5	
016	A	7	3	1699	2.216	2.236	0.81S	0.84S	2.245	1.32S	-1.20S	0.4	
032	A	7	3	1745	2.215	2.239	0.65S	0.83S	2.245	1.40S	-1.37S	0.4	
047	A	7	3	1771	2.214	2.243	1.30S	1.32S	2.250	1.67S	-1.64S	0.3	
072	B	7	3	250	4.487	4.420	1.61F	1.49F	4.400	1.52F	1.94F	-0.4	
079		7	3	1696	2.212	2.240	1.24S	1.27S	2.250	1.75S	-1.72S	0.4	
095	B	6	4	1930	2.212	2.235	1.03S	1.05S	2.235	1.08S	-1.05S	0.0	
127	B	6	4	1994	2.212	2.240	1.22S	1.24S	2.250	2.17S	-2.15S	0.5	
143	B	6	4	1989	2.212	2.217	0.16S	0.21S	2.223	0.51S	-0.48S	0.2	
168	B	6	4	580	3.534	3.560	0.34S	0.75S	3.540	0.59S	-0.18S	-0.5	

DEV. AND DIFF. ARE IN PERCENT

THE (-) SIGN INDICATES THAT INST 1 IS SLOWER THAN INST 2

F=FAST AND S=SLOW

TABLE 3-1

CLOCK CORRELATION SUMMARY
ORDER FIT ONE

SYS TIME I/P	CL TIME I/P	COMP SYS TM	DELTA ST	REV	ST
0.37988503D 05	0.111283892D 06	0.379885114D 05	-0.0084	9	COO
0.77532875D 05	0.150628281D 06	0.775328938D 05	-0.0188	16	COO
0.32751214D 05	0.192446610D 06	0.327512159D 05	-0.0019	24	COO
0.77687447D 05	0.237382852D 06	0.776874505D 05	-0.0035	32	COO
0.38220978D 05	0.284316381D 06	0.382209718D 05	0.0062	41	COO
0.72300793D 05	0.318396203D 06	0.723007881D 05	0.0049	47	COO
0.32927218D 05	0.365422641D 06	0.329272184D 05	-0.0004	56	COO
0.72371385D 05	0.404866809D 06	0.723713798D 05	0.0052	63	COO
0.32929803D 05	0.451825234D 06	0.329297971D 05	0.0059	72	COO
0.72399046D 05	0.491294483D 06	0.723990396D 05	0.0064	79	COO
0.32945499D 05	0.137003600D 04	0.329454968D 06	0.0022	88	COO
0.72396718D 05	0.408212580D 05	0.723967123D 05	0.0057	95	COO
0.32922016D 05	0.877465650D 05	0.329220115D 05	0.0045	104	COO
0.72357187D 05	0.127181740D 06	0.723571800D 05	0.0070	111	COO
0.32863099D 05	0.174087655D 06	0.328630872D 05	0.0118	120	COO
0.72278669D 05	0.213503243D 06	0.722786687D 05	0.0003	127	COO
0.32766310D 05	0.260390891D 06	0.327663090D 05	0.0010	136	COO
0.72178023D 05	0.299802616D 06	0.721780275D 05	-0.0045	143	COO
0.32630110D 05	0.346654708D 06	0.326301117D 05	-0.0017	152	COO
0.72021855D 05	0.386046460D 06	0.720218572D 05	-0.0022	159	COO
0.32455676D 05	0.432880292D 06	0.324556815D 05	-0.0055	168	COO
0.71829866D 05	0.472254487D 06	0.718298700D 05	-0.0040	175	COO
0.32243911D 05	0.519068546D 06	0.322439212D 05	-0.0102	184	COO

A0=-0.7329536226D 05 A1= 0.999999834740D 00

SIGMA=0.00655 NO. POINTS= 23

RATIO OF CLOCK TIME TO SYS TIME= 0.100000016526D 01

ORDER FIT TWO

SYS TIME I/P	CL TIME I/P	COMP SYS TM	DELTA ST	REV	ST
0.37988503D 05	0.111283892D 06	0.379885008D 05	0.0022	9	COO
0.77532875D 05	0.150828281D 06	0.775328859D 05	-0.0109	16	COO
0.32751214D 05	0.192446610D 06	0.327512106D 05	0.0034	24	COO
0.77687447D 05	0.237382852D 06	0.776874477D 05	-0.0007	32	COO
0.38220978D 05	0.284316381D 06	0.382209712D 05	0.0068	41	COO
0.72300793D 05	0.318396203D 06	0.723007890D 05	0.0040	47	COO
0.32927218D 05	0.365422641D 06	0.329272210D 05	-0.0030	56	COO
0.72371385D 05	0.404866809D 06	0.723713836D 05	0.0014	63	COO
0.32929803D 05	0.451825234D 06	0.329298020D 05	0.0010	72	COO
0.72399046D 05	0.491294483D 06	0.723990451D 05	0.0009	79	COO
0.32945499D 05	0.137003600D 04	0.329455020D 05	-0.0038	88	COO
0.72396718D 05	0.408212580D 05	0.723967184D 05	-0.0004	95	COO
0.32922016D 05	0.877465650D 05	0.329220175D 05	-0.0015	104	COO
0.72357187D 05	0.127181740D 06	0.723571856D 05	0.0014	111	COO
0.32863099D 05	0.174087655D 06	0.328630921D 05	0.0069	120	COO
0.72278669D 05	0.213503243D 06	0.722786727D 05	-0.0037	127	COO
0.32766310D 05	0.260390891D 06	0.327663116D 05	-0.0016	136	COO
0.72178023D 05	0.299802616D 06	0.721780287D 05	-0.0057	143	COO
0.32630110D 05	0.346654708D 06	0.326301110D 05	-0.0010	152	COO
0.72021855D 05	0.386046460D 06	0.720218546D 05	0.0004	159	COO
0.32455676D 05	0.432880292D 06	0.324556762D 05	-0.0002	168	COO
0.71829866D 05	0.472254487D 06	0.718298623D 05	0.0037	175	COO
0.32243911D 05	0.519068546D 06	0.322439104D 05	0.0006	184	COO

A0=-0.7329536171D 05 A1= 0.999999922776D 00

A2=-0.7567429548257D-13

SIGMA=0.00377 NO. POINTS= 23

TABLE 3-2

TABLE 3-3
J-34 TEMPERATURE SUMMARY

SENSOR	Master Camera	ORBITS ACQUIRED																													
		A B		25		72		88		96		104		111		120		127		136		143		152		168		175		184	
Master	2	86	81	80	81	85	82	80	71	78	73	76	72	76	71	76	70	78	85	79	86	77	76	70	78	77	77	70	78		
Camera	3	97	90	92	92	93	92	95	91	93	91	90	89	86	81	81	86	80	86	81	83	79	83	77	77	73	75	77	77	74	
4	98	92	92	89	92	93	95	93	91	93	91	90	87	87	82	86	81	85	80	85	78	83	79	77	73	76	76	76	74		
5	95	91	94	88	91	93	92	90	93	91	90	93	92	93	89	83	87	83	88	81	87	82	87	80	85	79	85	78	78		
6	100	94	96	94	93	97	94	99	98	93	91	90	94	94	91	91	97	91	90	78	85	78	83	79	74	77	74	77	73		
7	104	96	95	95	95	100	96	102	99	95	99	91	97	91	91	97	91	90	78	85	81	87	81	86	79	76	77	71	71		
8	91	94	88	91	93	97	94	99	98	93	90	95	95	88	91	97	91	90	78	85	81	87	81	86	79	76	77	71	71		
9	95	92	88	91	92	92	90	93	90	95	93	90	95	95	95	95	95	95	95	95	95	95	95	95	95	95	95	95	95		
11	101	95	95	88	93	89	88	93	89	95	95	95	92	89	90	90	90	90	90	90	90	90	90	90	90	90	90	90	90		
12	95	92	88	88	92	92	93	91	94	92	93	91	94	92	93	91	94	92	93	91	94	92	93	91	94	92	93	91	94		
Avg.	97	91	90	91	91	94	91	96	94.5	90	94	91	96	94.5	90	94	91	96	94	90	89	89	79	85	80	83	79	83	78	83	
Sleve Camera	3	94	88	87	87	90	87	92	90	88	83	90	85	88	84	83	84	80	74	80	74	83	77	71	71	76	76	76	76	69	
4	91	85	84	85	85	85	87	85	87	84	85	86	90	86	85	84	81	76	82	77	81	77	81	77	76	76	76	76	64		
5	89	84	83	85	85	85	87	85	87	84	87	85	86	86	86	85	84	75	78	76	77	74	77	73	76	76	76	74			
6	91	85	83	85	85	85	87	85	87	84	87	85	86	86	85	85	84	82	78	76	77	74	77	73	76	76	76	74			
7	92	85	85	85	85	85	87	85	87	84	85	86	91	86	92	93	85	85	86	82	78	80	73	71	75	71	75	71	71		
8	90	89	83	83	83	83	87	84	89	89	85	85	89	85	85	83	71	71	81	77	81	77	81	76	80	75	75	75			
9	80	76	76	76	75	75	79	80	79	79	79	79	79	79	79	79	79	79	79	79	79	79	79	79	79	79	79	79	79		
11	96	87	89	87	89	87	92	87	96	94	87	93	86	88	77	83	77	83	77	84	77	81	76	69	66	69	66	69	65		
12	81	77	75	78	78	78	81	79	79	79	79	79	79	79	79	79	79	79	79	79	79	79	79	79	79	79	79	79	79		
Avg.	89	83	83	83	87	84	89	87.5	83	87	84	81	73	79	74	78	74	78	74	78	74	78	74	78	74	78	74	78	74		
Supply Spool	1	71	69	67	71	74	73	74	71	73	74	70	63	67	66	63	67	63	67	66	62	69	62	67	63	66	62	66	62		
2	80	74	75	80	77	81	80	77	80	76	75	66	71	67	66	69	64	67	63	66	62	69	62	67	63	66	62	66	62		

TABLE 3-3

J-34 TEMPERATURE SUMMARY

<u>SENSOR</u>		<u>ORBITS ACQUIRED</u>											
<u>Fair ("A")</u>		<u>A B</u>											
<u>Barrel #1 ("B")</u>		<u>1 2 3 4 5 6 7 8 9 10 11 12</u>											
1	2 6 1 2	9	16	24	32	41	47	56	61	63	72	79	88
2	3 5 4 3	60	44	57	47	54	37	70	77	27	37	67	27
3	4 3 2 1	24	4	20	4	20	4	30	4	33	4	78	49
4	5 1	4	4	0	0	4	4	0	0	10	4	97	88
5		59	49	52	45	52	45	66	59	45	62	81	59
6		84	61	78	65	72	61	87	87	58	84	55	57
Barrel #2		81	78	78	74	68	64	89	92	50	84	50	--
Clock		56	66	49	59	49	56	59	52	52	45	42	28
1	2 3 4	58	85	55	82	51	76	69	65	58	62	58	51
2	1 5	77	89	77	92	74	86	92	72	74	89	83	48
3		70	54	67	54	70	61	80	80	60	83	73	60
4		59	59	56	56	59	62	65	59	61	48	59	54
5													
Conic Adapter													
1		52	42	46	42	42	39	52	52	36	49	29	36
2													
Thrust Cone "A" to "B" SRV													
1		NA	NA	NA	NA	NA	NA	NA	NA	NA	NA	NA	NA
2		74	65	65	62	68	64	69	67	64	67	72	76
Press. Make-Up Bottle													
1													
2		113	96	99	96	105	96	105	101	93	105	87	75
Recovery Battery "B" SRV													
1		124	107	110	107	115	104	115	113	104	113	99	90
Master Cassette "A" SRV													
1		78	79	76	82	83	82	81	82	81	79	76	79
2		74	68	66	68	68	69	70	69	71	70	71	71

SECTION 4

MISSION 1038-1 RECOVERY SYSTEM

SRV #719 was received at A/P on 27 October 1965. The receiving weight was 152.75 pounds. After modifications and incorporation of outstanding E.O.'s, the SRV was delivered to Systems Test for incorporation into the J-34 system.

The capsule was delivered for shipment to VAFB on 10 January 1967.

The -1 recovery capsule was successfully recovered by air-catch on Rev 81 at 15:31 PST on 19 January 1967. All re-entry sequence of events monitored appeared normal and occurred within tolerance. The capsule impact point was very close to the predicted impact point.

Predicted	$25^{\circ} 59.2' N$	$155^{\circ} 51.3' W$
Actual	$26^{\circ} 03.0' N$	$155^{\circ} 54.0' W$

The re-entry sequence of events is contained in Table 4-1.

MISSION 1038-1RECOVERY SEQUENCE OF EVENTS

<u>Event</u>		<u>Actual</u>	<u>Nominal</u>	<u>Delta Time</u>
*Arm		76.90	77.0 + 1.0	
*Transfer		1.98	2.0 + 0.25	
Electrical Disconnect		0.87	.900 + 0.43 - 0.40	
Separation		--	--	
**Spin		3.31	3.4 + 0.30	
Retro		7.55	7.55 + 0.45	
Despin		10.75	10.75 + 0.59	
T/C Separation		1.53	1.5 + 0.15	
*** "G" Switch Open		479.34	482.8	
Parachute Cover Off		33.61	34.0 + 1.5	
Drogue Chute Deployed		0.58	0.63 + 0.08	
Main Chute Bag Separate		12.21	10.0 + 3.0 - 2.2	
Main Chute Deployed		0.52	0.52 + .13	
Main Chute Disreef		4.61	4.5 + 0.80	

* From Separation

** From Elect. Disc.

*** From Retro

TABLE 4-1

SECTION 5

MISSION 1038-2 RECOVERY SYSTEM

SRV #720 was received at A/P on 27 October, 1965 at a receiving weight of 150.25 pounds. After modification and incorporation of outstanding E.O.'s the capsule was delivered to Systems Test for incorporation into the J-34 system.

The capsule was delivered for shipment to VAFB on 10 January 1967.

The -2 recovery capsule was successfully recovered by air-catch on Rev 193 at 15:38 PST on 26 January 1967. All re-entry sequence of events monitored appeared normal and occurred within tolerance. The capsule impact point was very close to the predicted impact point.

Predicted	23° 25.7' N	169° 12.06' W
Actual	23° 43.0' N	169° 01.0' W

The re-entry sequence of events is contained in Table 5-1.

MISSION 1038-2

RECOVERY SEQUENCE OF EVENTS

<u>Event</u>		<u>Delta Time</u>
	<u>Actual</u>	<u>Nominal</u>
*Arm	N/A	77.0 ± 1.0
*Transfer	2.00	2.0 ± 0.25
Electrical Disconnect	0.74	.900 + 0.43 - 0.40
Separation	--	--
**Spin	3.44	3.4 ± 0.30
Retro	7.68	7.55 ± 0.45
Despin	10.47	10.75 ± 0.59
T/C Separation	1.47	1.50 ± 0.15
***"G" Switch Open	514.66	520.7
Parachute Cover Off	33.81	34.0 ± 1.5
Drogue Chute Deployed	0.57	0.63 ± .08
Main Chute Bag Separate	11.43	10.0 + 3.0 - 2.2
Main Chute Deployed	0.57	0.52 ± 0.13
Main Chute Disreefed	4.47	4.50 ± 0.80
* From Separation		
** From Elect. Disc.		
*** From Retro		

TABLE 5-1

SECTION 6
MISSION 1038 PANORAMIC CAMERAS

A. COMPONENT ASSIGNMENT

<u>Component</u>	<u>Master (Fwd) Serial Number</u>	<u>Slave (Aft) Serial Number</u>
Main Camera	192	193
Main Camera Lens	1792435	1982435
Supply Horizon Camera	284-G6	291-G6
Supply Horizon Camera Lens	E12858	E12857
Take-up Horizon Camera	294-G5	291-G5
Take-up Horizon Camera Lens	E12881	E12852
Supply Cassette	SC-44	SC-44

B. CAMERA DATA AND FLIGHT SETTINGS

Main Camera:

Lens	24" f/3.5	24" f/3.5
Slit Width	0.225"	0.175"
Filter Type	Wratten 23A	Wratten 21
Film Type (Eastman)	3404	3404
Supply Horizon Cameras	<u>Port</u>	<u>Starboard</u>
Lens	55 mm f/6.3	55 mm f/6.3
Aperture Setting	f/6.3	f/8.0
Exposure Time	1/100 second	1/100 second
Filter Type	Wratten 25	Wratten 25

~~TOP SECRET~~
Take-up Horizon Camera

	<u>Starboard</u>	<u>Port</u>
	Master (Fwd)	Slave (Aft)
Lens	55 mm f/6.3	55 mm f/6.3
Aperture Setting	f/8.0	f/6.3
Exposure Time	1/100 second	1/100 second
Filter Type	Wratten 25	Wratten 25

C. POST FLIGHT PERFORMANCE EVALUATION

The overall image quality of the mission was not as good as observed on recent missions. The cause of the generally poorer imagery was the atmospheric conditions. The percentage of clear terrain areas, observed in the index camera photography, was unusually low. In certain isolated instances the quality of the photography approached that of recent missions. This fact tends to substantiate that atmospherics degraded the imagery.

The original negative of the -1 mission was processed by ASFPPF. There was no detectable difference between the two portions of the mission due to processing.

The master film record had a crease beginning on frame 139 pass D-38 and it left the material on frame 141 at the time-track edge. The crease extended one quarter inch into the format. This defect is explained by tension transients caused by a splice located in frame 142.

There are certain characteristic anomalies that are considered inherent to the operation of the J-1 system. The degree of degradation is minor, and they are continually being monitored to prevent increased severity. In this mission rail scratches and dendritic static discharges were more severe than normal. Scan head roller scratches and light leak fog patterns were less than normal.

The starboard horizon photography of the slave instrument was veiled from pass D5 through D-85, at which time a gradual closing begins. The imagery was clear at the end of the mission. The horizon boots were properly painted and checked prior to flight.

SECTION 7

MISSION 1038 STELLAR-INDEX CAMERA

A. COMPONENT ASSIGNMENT

<u>Component</u>	<u>-1 Mission Serial Number</u>	<u>-2 Mission Serial Number</u>
Camera	D-93	D-90
Index Camera Lens	819192	819958
Index Reseau	86	111
Stellar Camera Lens	11910	10742
Stellar Reseau	112	108

B. CAMERA DATA AND FLIGHT SETTINGS

Stellar Camera:

Lens	85 mm f/1.8	85 mm f/1.8
Exposure Time	2 seconds	2 seconds
Filter Type	None	None
Film Type (Eastman)	3401	3401

Index Camera:

Lens	38 mm f/4.5	38 mm f/4.5
Exposure Time	1/500 second	1/500 second
Filter Type	Wratten 21	Wratten 21
Film Type (Eastman)	3400	3400

C. POST FLIGHT EVALUATION

The -1 mission stellar camera operated normally except for a double exposure on frame 229. Approximately 50% of each format is affected by flare. There are an average of 10 stellar images per format. The images are elongated rather than points. The last 13 frames are affected by the usual scratches, gouges, and pinholes associated with film depletion.

The -1 mission index camera produced good imagery. Frame 229 was double exposed.

The -2 mission stellar camera operated satisfactorily. Approximately 20% of each format was affected by flare. There were approximately 10 stellar images per format and they were elongated.

The -2 mission index camera produced good image quality. Frames 4 and 16 are double exposed in this camera and the stellar camera.

SECTION 8

PANORAMIC CAMERA EXPOSURE

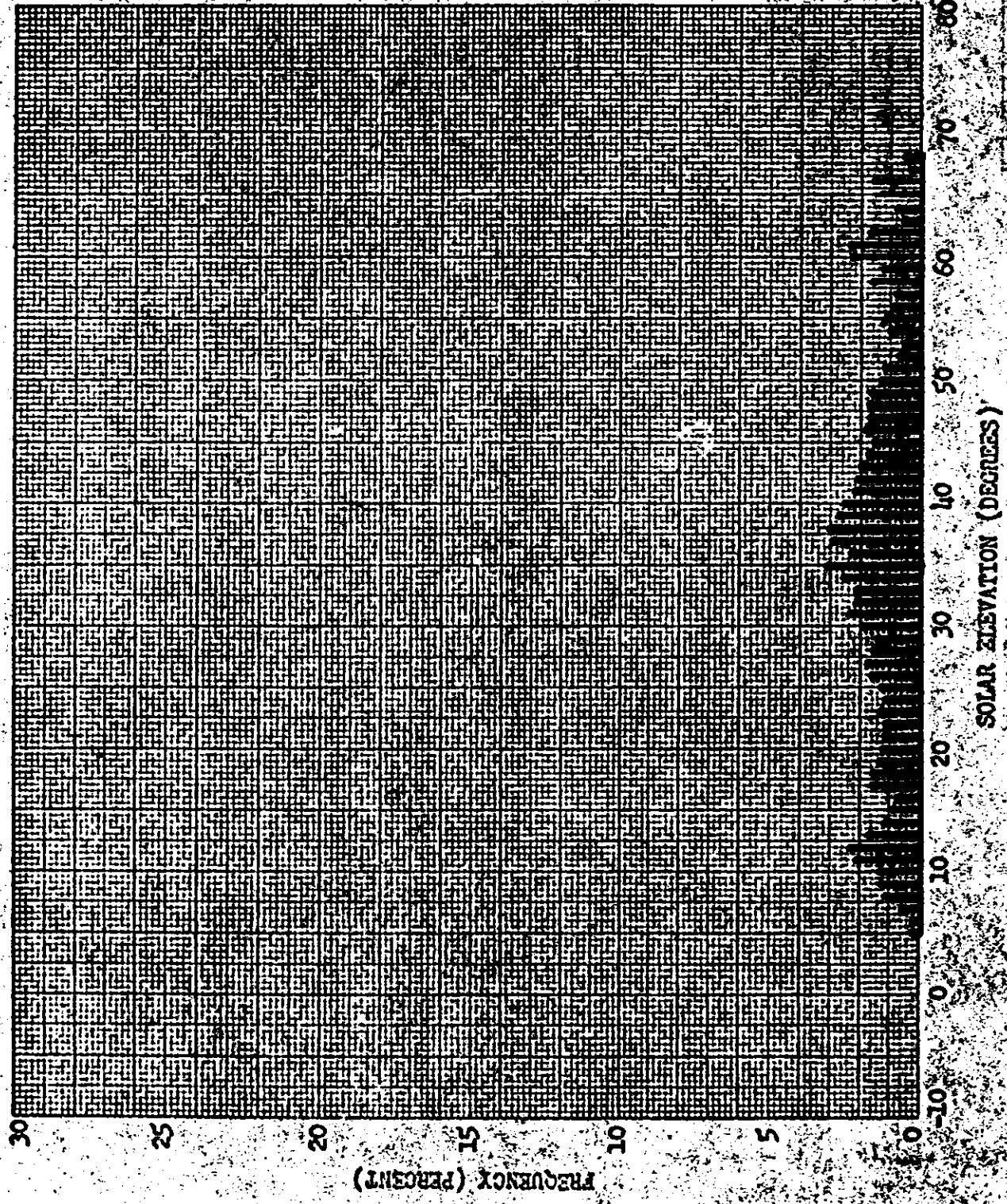
The Master camera contained a 0.225 inch wide slit and a Wratten 23A filter while the Slave camera had a 0.175 inch wide slit and a Wratten 21 filter. These conditions placed the nominal exposure between the full processing and intermediate curve as published by [REDACTED] for 3404 emulsion.

The frequency distributions of the solar elevations and solar azimuths encountered during the photographic operations are shown in Figures 8-1 to 8-4.

The nominal exposure times of the Master and Slave cameras are shown as a function of latitude for passes D-8, D-72, D-136, and D-184 in Figures 8-5 to 8-12. The predicted level of processing for the original negative is based on the in-flight performance estimate and is tabulated below with the processing levels reported by [REDACTED] for -1 mission and -2 mission by AFSPPF.

Mission	Camera		%	%	%	%
			Primary	Int.	Full	Transition
1038-1	Fwd	Predicted	0	21.6	78.4	---
		Reported	2	10	76	12
1038-1	Aft	Predicted	0	22	78	---
		Reported	1	7	81	11
1038-2	Fwd	Predicted	1.4	36.4	62.2	---
		Reported	0	10	81	9
1038-2	Aft	Predicted	3.1	28.9	68	---
		Reported	0	24	70	6

SOLAR ELEVATION FREQUENCY DISTRIBUTION



SOLAR AZIMUTH FREQUENCY DISTRIBUTION

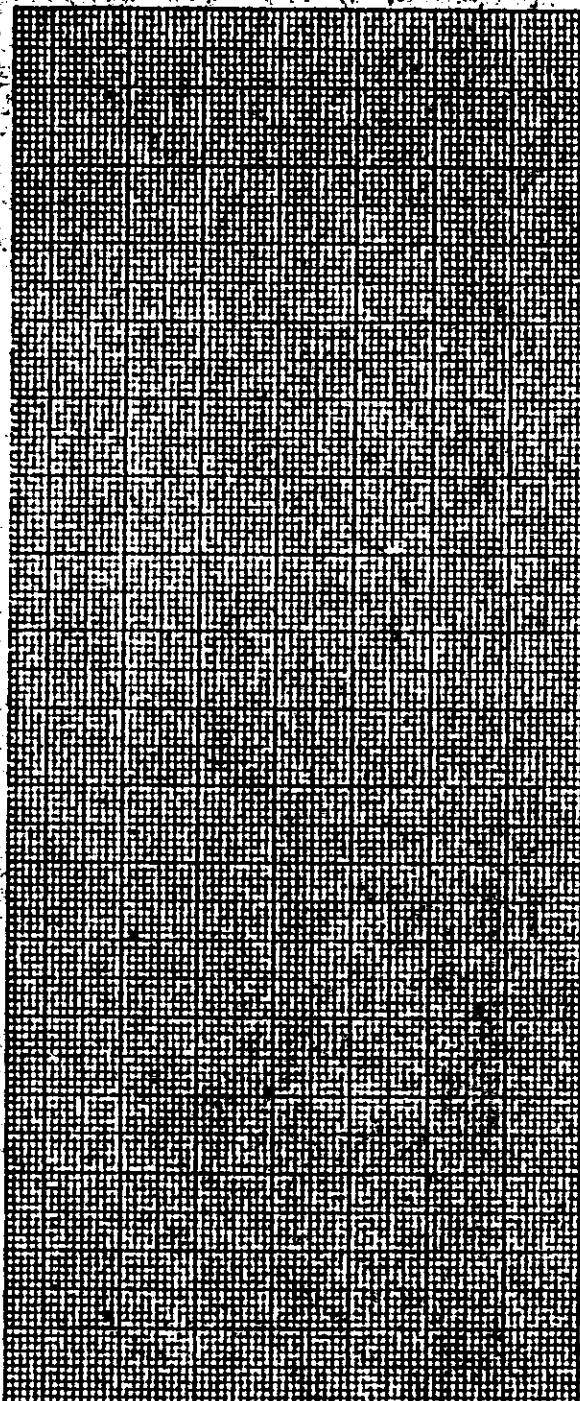
15

10

5

0

FREQUENCY (PERCENT)



Mission No: 1038-1

Payload No: J-34

Camera No: 192

Launch Date: 1-14-67

Launch Time: 2128Z

Inclination: 80°

SIGN NOTATION

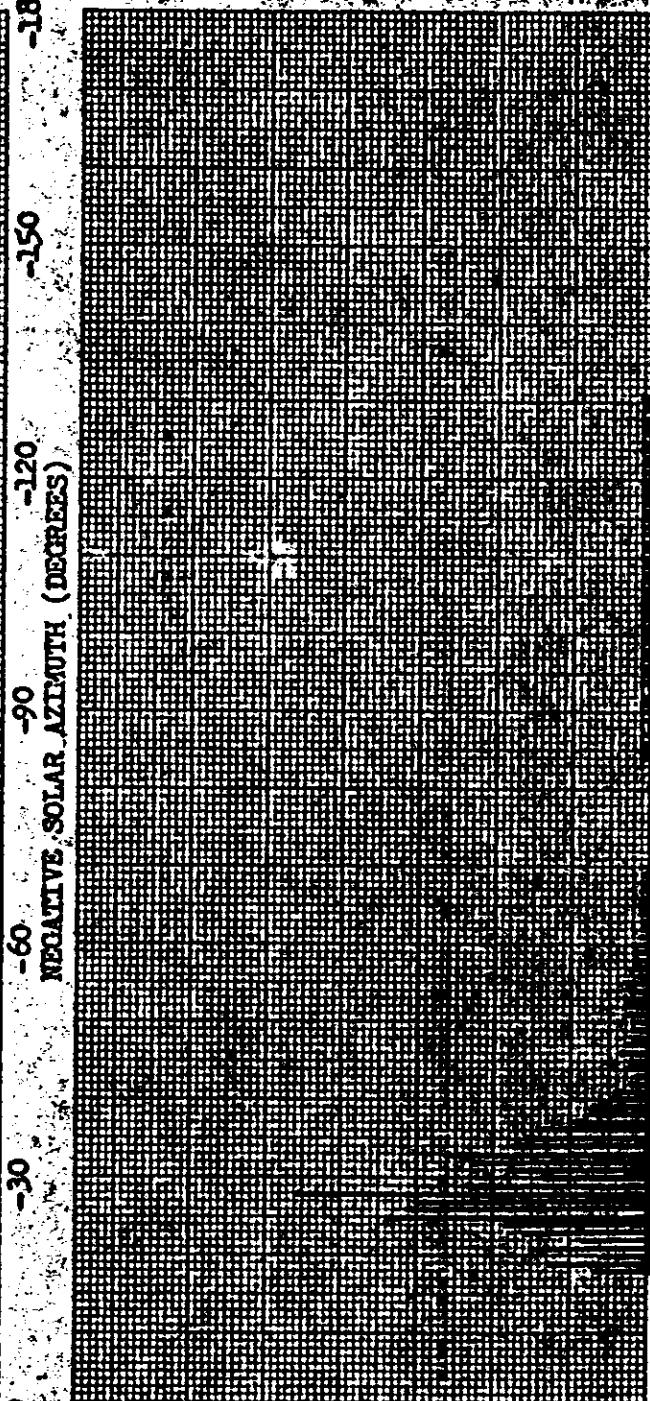
FREQUENCY (PERCENT)

15

10

5

0



NEGATIVE SOLAR AZIMUTH (DEGREES)

-180

-150

-120

-90

-60

-30

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

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30

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90

120

150

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180

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150

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90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

0

30

60

90

120

150

180

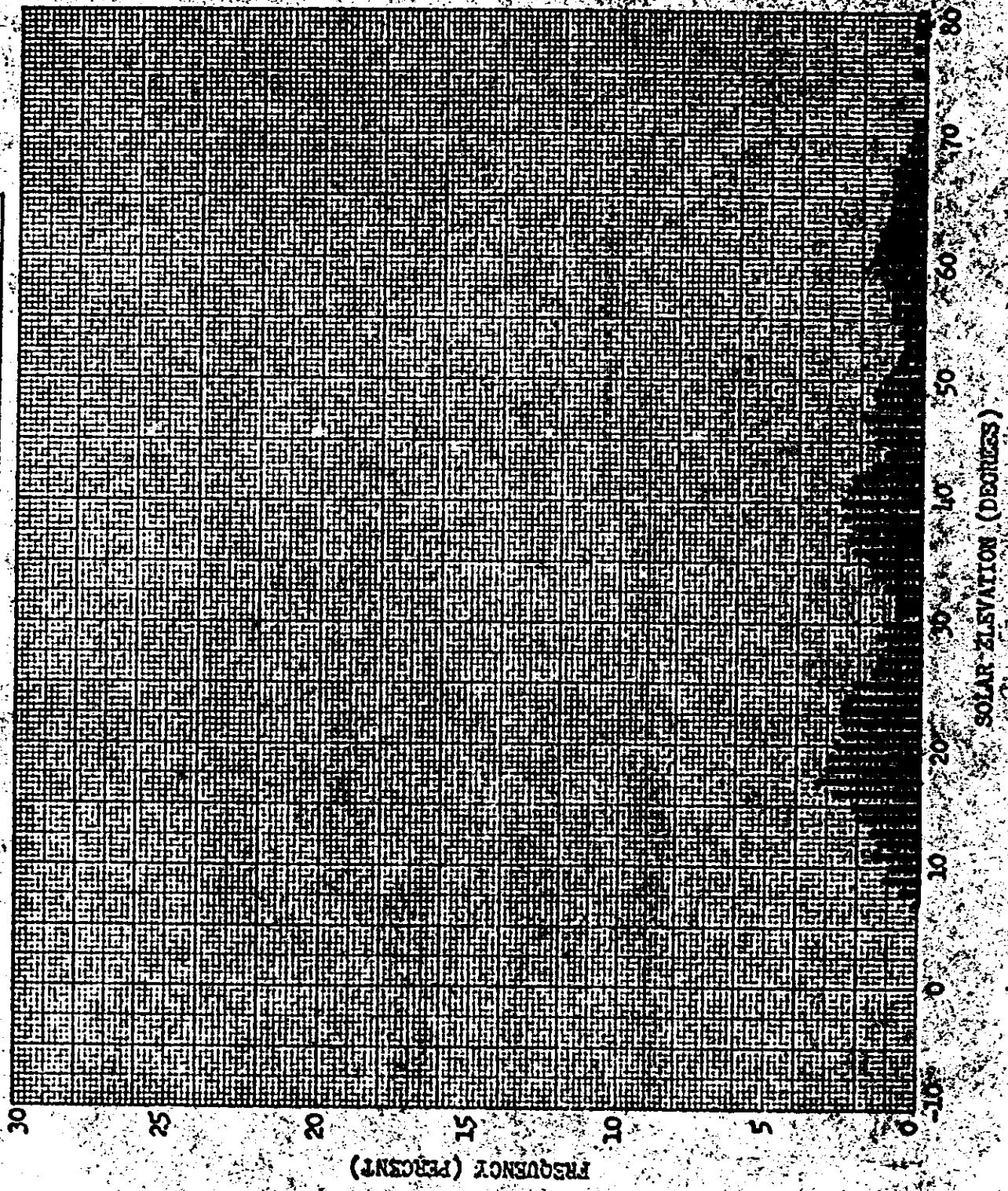
0

30

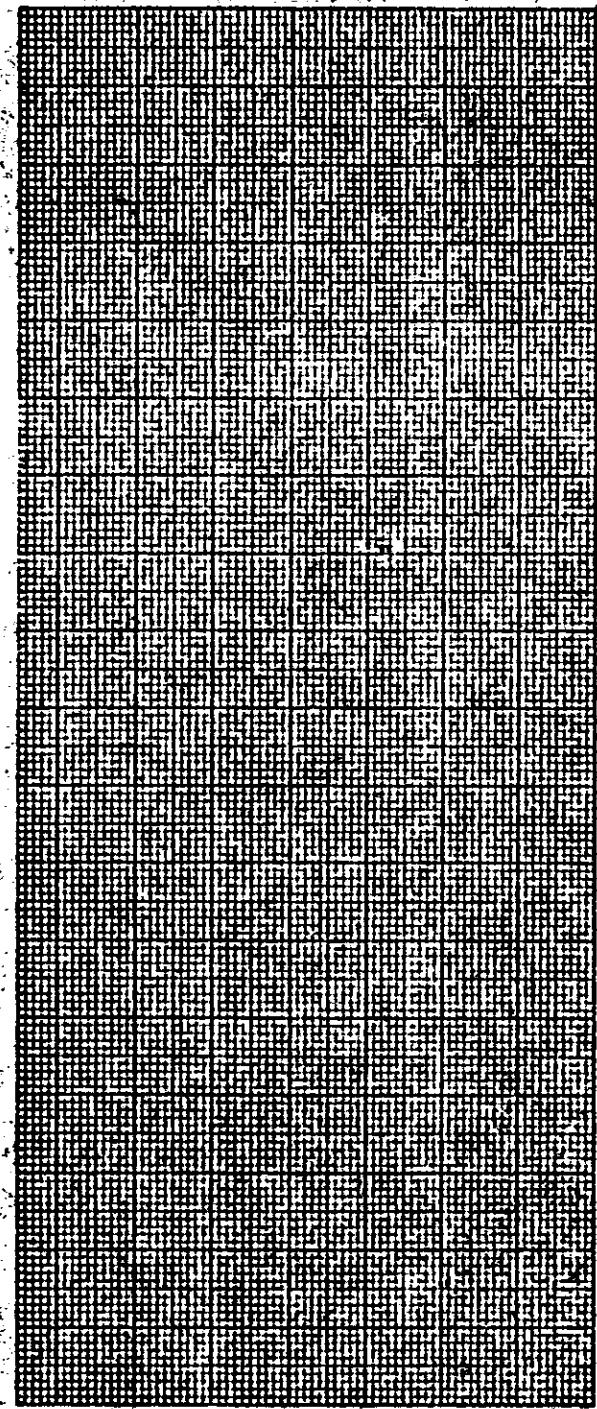
60

90

SOLAR ELEVATION FREQUENCY DISTRIBUTION



SOLAR AZIMUTH FREQUENCY DISTRIBUTION



-180 -150 -120 -90 -60

NEGATIVE SOLAR AZIMUTH (DEGREES)

0 5 10 15

FREQUENCY (PERCENT)

Mission No: 1038-2

Payload No: J-34

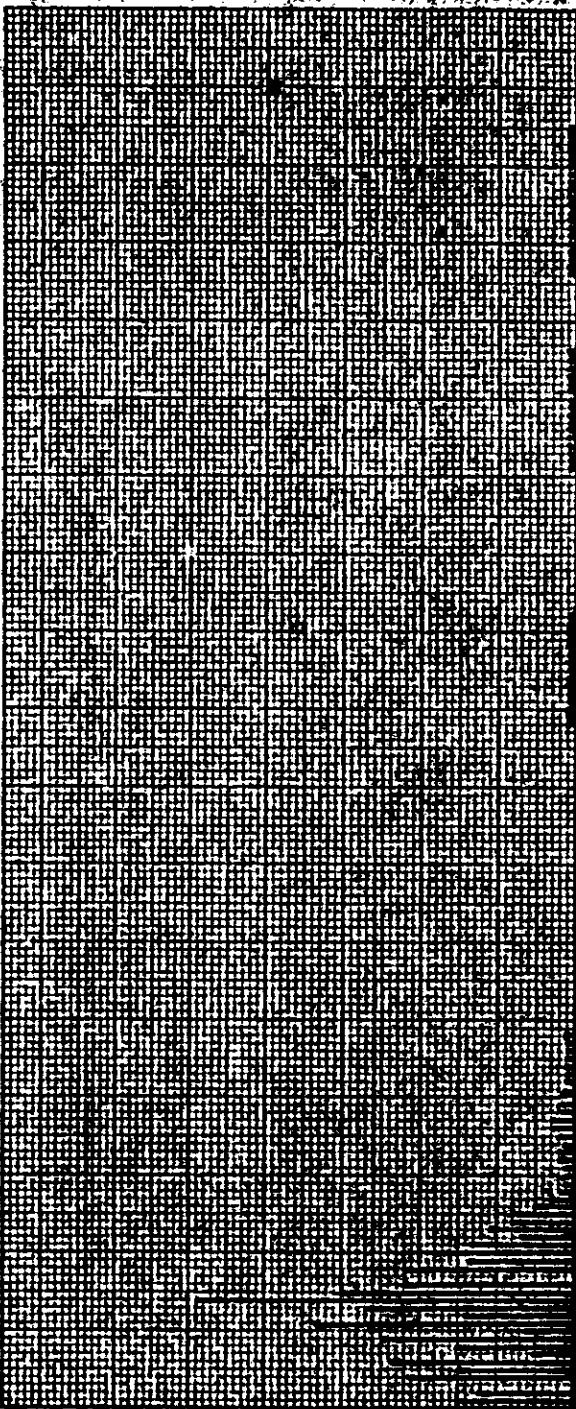
Camera No: 192

Launch Date: 1-14-67

Launch Time: 2128Z

Inclination: 80°

SIGN NOTATION

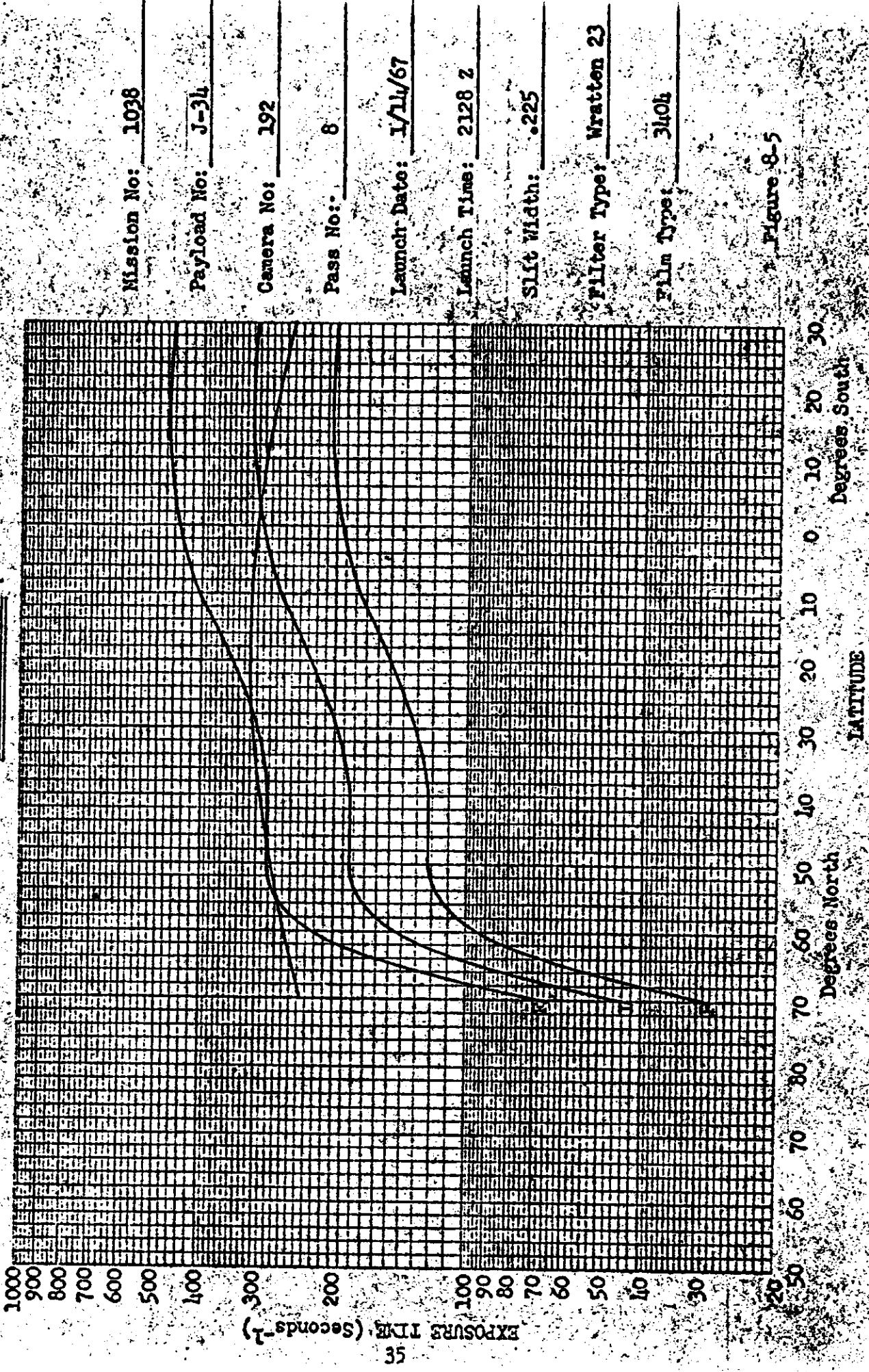


0 30 60 90 120 150

180

Figure 8-4

EXPOSURE POINTS



EXPOSURE POINTS

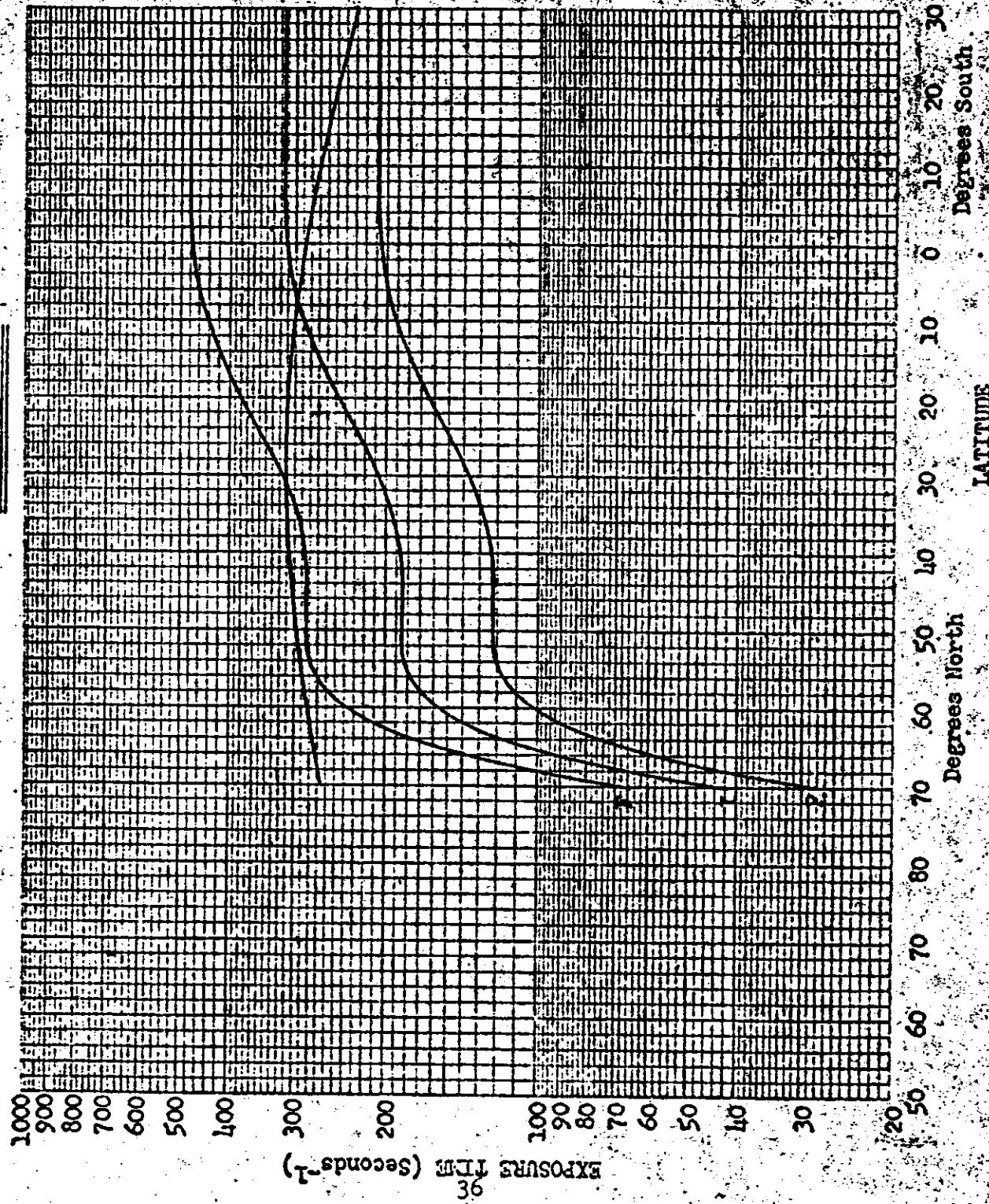


Figure 8-6

EXPOSURE POINTS

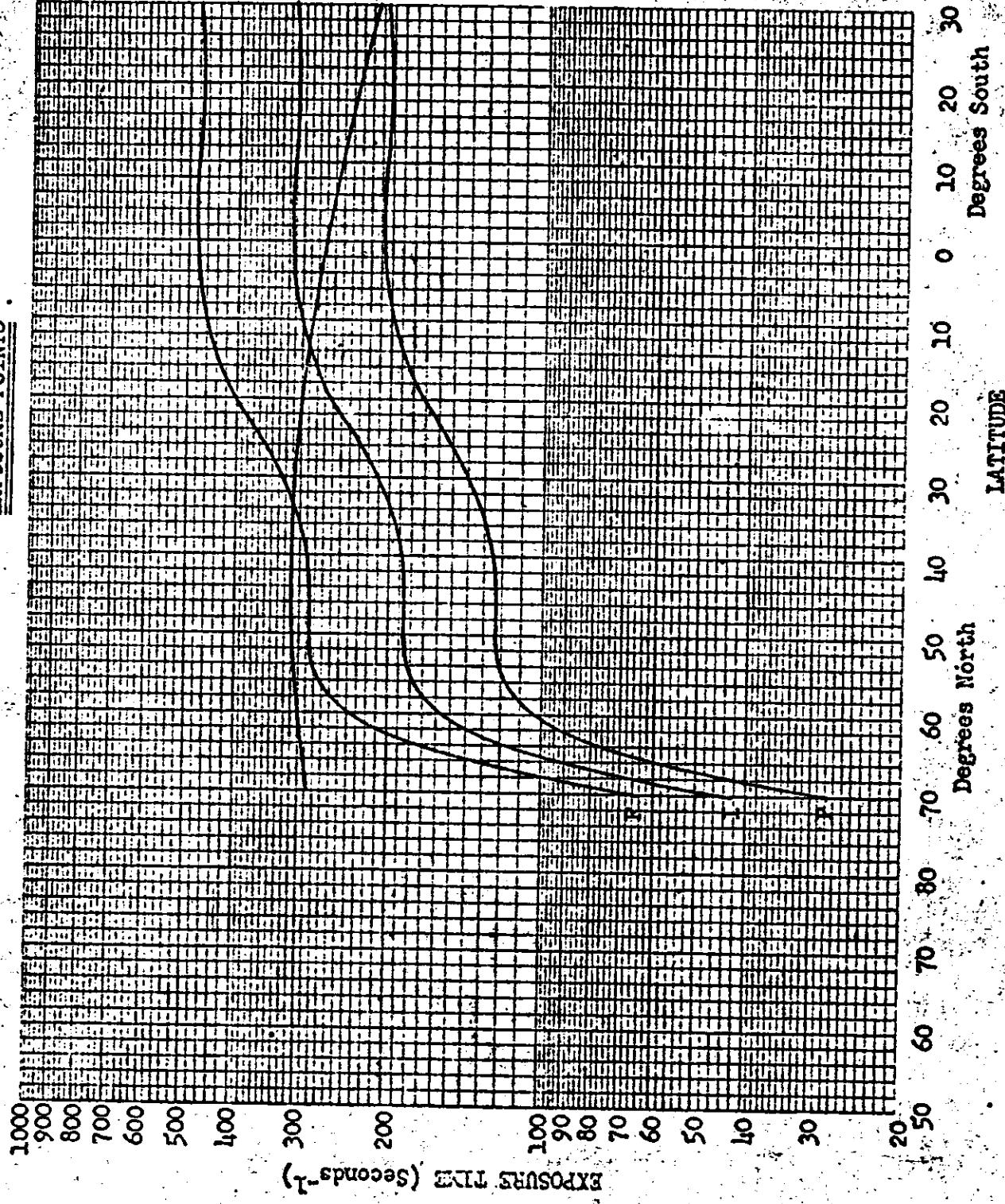


Figure 8-7

EXPOSURE POINTS

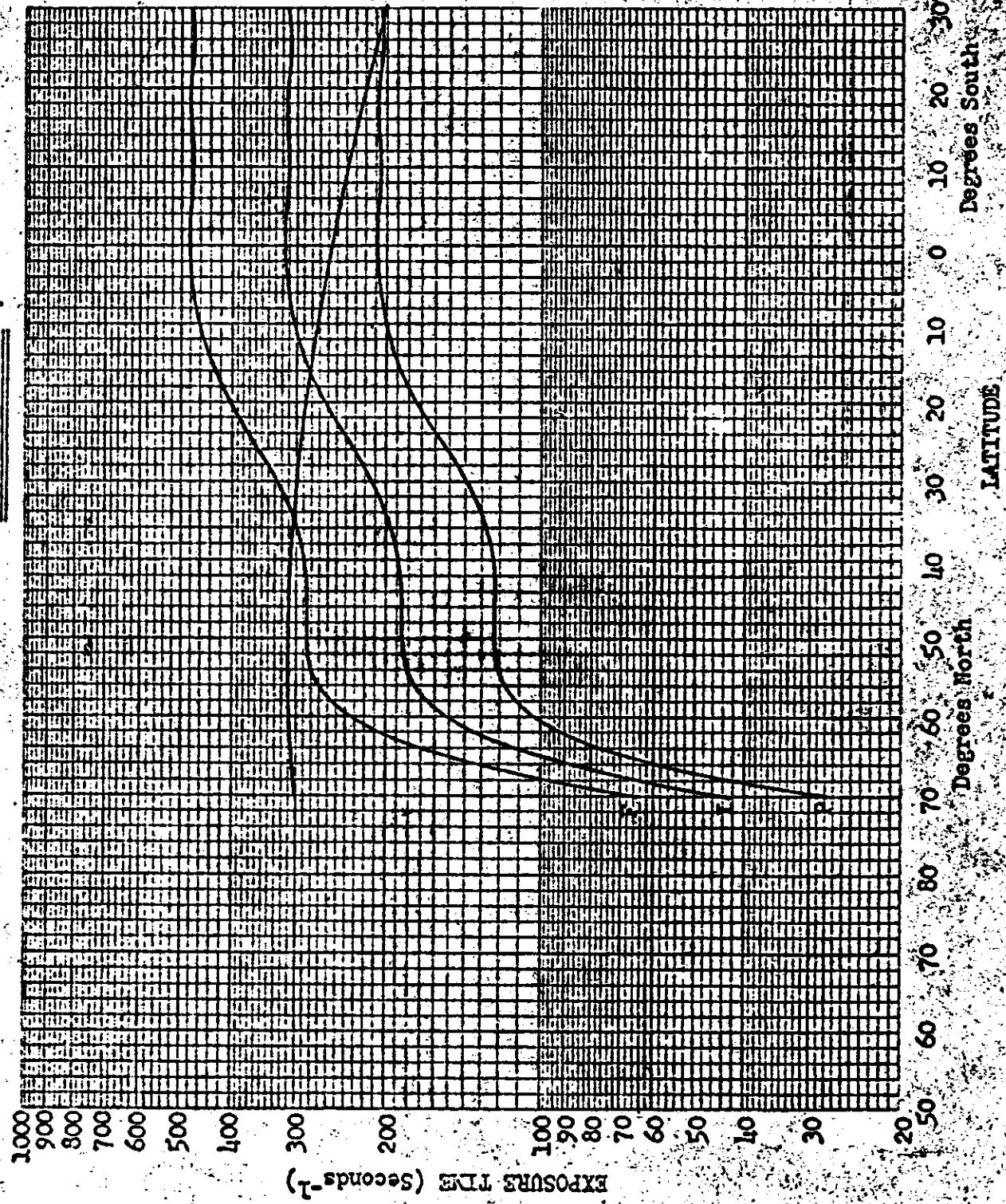


Figure 8-8

Degrees South
LATITUDE

Figure 8-9

EXPOSURE POINTS

1000

900

800

700

600

500

400

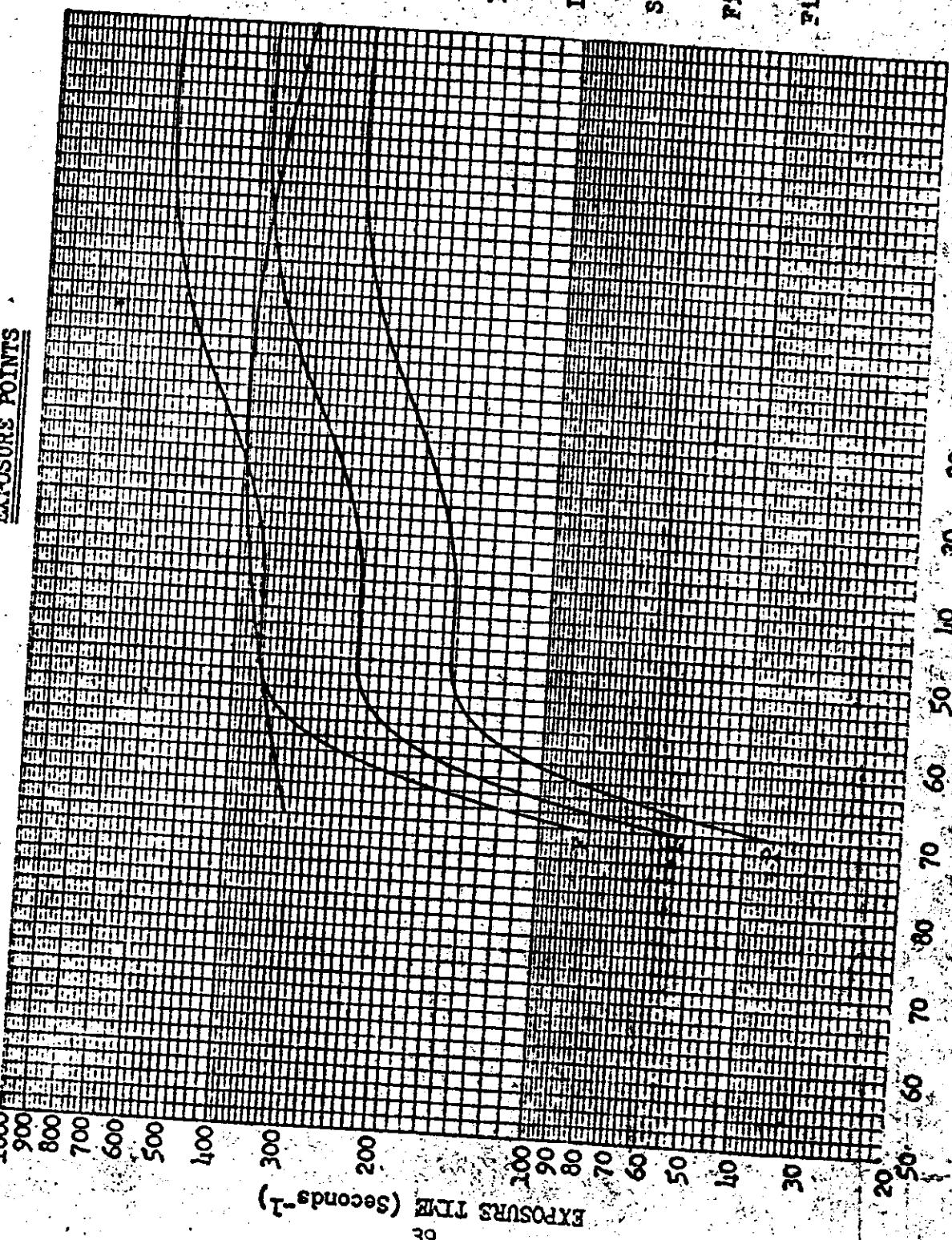
300

200

100

0

EXPOSURES TIME (Seconds-⁻¹)



EXPOSURE POINTS

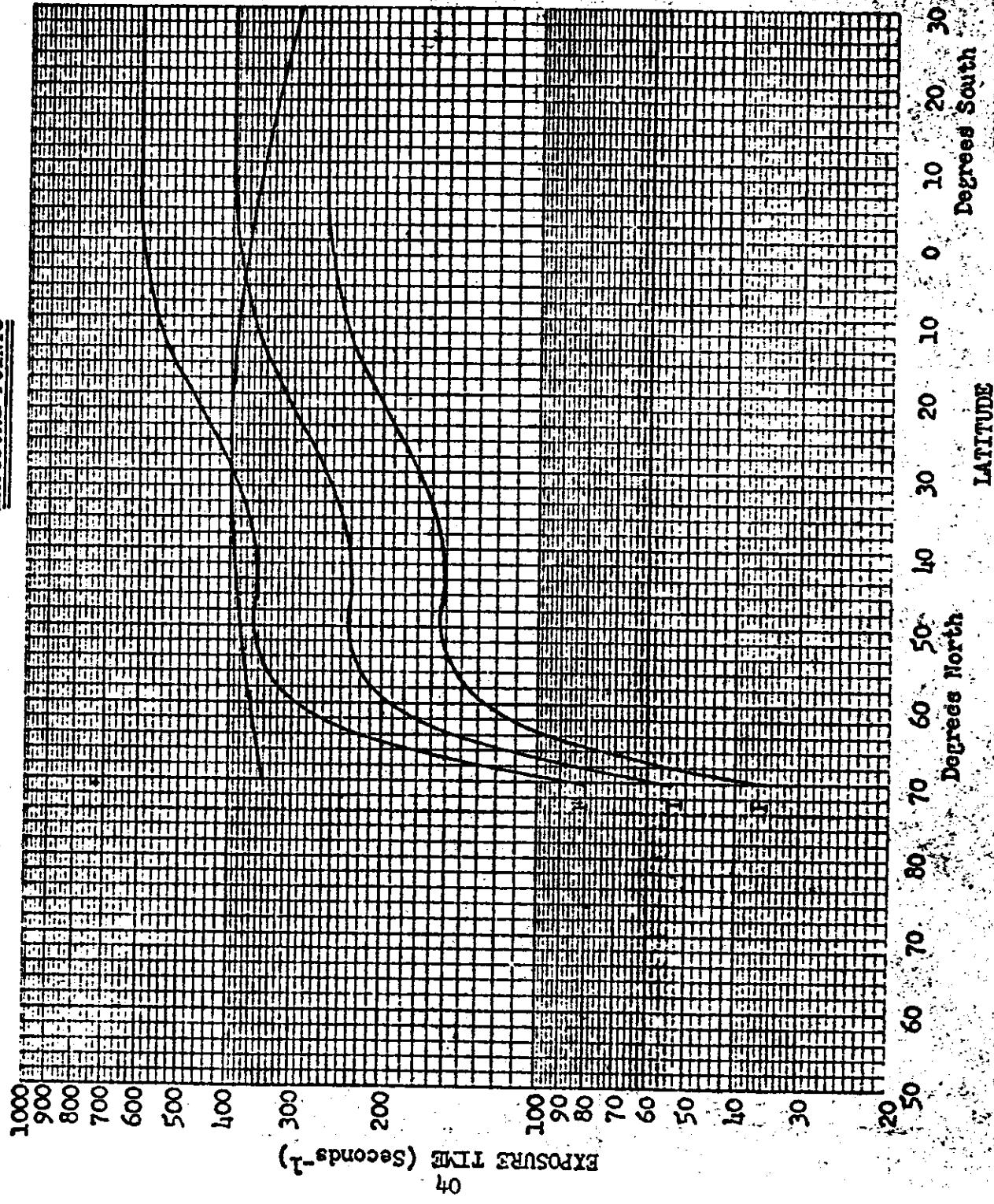


Figure 8-10

TOP SECRET C/

MISSION 1038-1				INSTR - FWD				3/6/67 PROCESSING AND EXPOSURE ANALYSIS			
PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	PROCESSED	UNDER PROCESSED	PROCESSED	CORRECT EXP/PROC	OVER EXPOSED	UNDER PROCESSED	PROCESSED	CORRECT EXP/PROC	OVER EXPOSED
PRIMARY	4	0 PC	25 PC	75 PC	0 PC	0 PC	0 PC	0 PC	0 PC	0 PC	0 PC
INTERMEDIATE	38	0 PC	18 PC	76 PC	5 PC	0 PC	0 PC	0 PC	0 PC	0 PC	0 PC
FULL	214	35 PC	0 PC	63 PC	2 PC	0 PC	0 PC	0 PC	0 PC	0 PC	0 PC
ALL LEVELS	256	29 PC	3 PC	65 PC	3 PC	0 PC	0 PC	0 PC	0 PC	0 PC	0 PC
MISSION 1038-1				INSTR - AFT				3/6/67 PROCESSING AND EXPOSURE ANALYSIS			
PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	PROCESSED	UNDER PROCESSED	PROCESSED	CORRECT EXP/PROC	OVER EXPOSED	UNDER PROCESSED	PROCESSED	CORRECT EXP/PROC	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC	18 PC	18 PC	18 PC	18 PC	18 PC
INTERMEDIATE	34	0 PC	24 PC	62 PC	12 PC	12 PC	19 PC	19 PC	19 PC	19 PC	19 PC
FULL	212	39 PC	0 PC	57 PC	5 PC	5 PC	13 PC	13 PC	13 PC	13 PC	13 PC
ALL LEVELS	246	33 PC	3 PC	57 PC	6 PC	6 PC	0 PC	0 PC	0 PC	0 PC	0 PC
MISSION 1038-2				INSTR - FWD				3/6/67 PROCESSING AND EXPOSURE ANALYSIS			
PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	PROCESSED	UNDER PROCESSED	PROCESSED	CORRECT EXP/PROC	OVER EXPOSED	UNDER PROCESSED	PROCESSED	CORRECT EXP/PROC	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC	18 PC	18 PC	18 PC	18 PC	18 PC
INTERMEDIATE	35	0 PC	11 PC	74 PC	14 PC	14 PC	20 PC	20 PC	20 PC	20 PC	20 PC
FULL	238	27 PC	0 PC	64 PC	19 PC	19 PC	0 PC	0 PC	0 PC	0 PC	0 PC
ALL LEVELS	273	23 PC	1 PC	65 PC	10 PC	10 PC	0 PC	0 PC	0 PC	0 PC	0 PC
MISSION 1038-2				INSTR - AFT				3/6/67 PROCESSING AND EXPOSURE ANALYSIS			
PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	PROCESSED	UNDER PROCESSED	PROCESSED	CORRECT EXP/PROC	OVER EXPOSED	UNDER PROCESSED	PROCESSED	CORRECT EXP/PROC	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC	35 PC	35 PC	35 PC	35 PC	35 PC
INTERMEDIATE	69	4 PC	48 PC	39 PC	6 PC	6 PC	3 PC	3 PC	3 PC	3 PC	3 PC
FULL	198	17 PC	0 PC	62 PC	21 PC	21 PC	1 PC	1 PC	1 PC	1 PC	1 PC
ALL LEVELS	267	14 PC	12 PC	56 PC	17 PC	17 PC	0 PC	0 PC	0 PC	0 PC	0 PC
MISSION 1038-2				BASE FOG				3/6/67 PROCESSING AND EXPOSURE ANALYSIS			
PROCESS LEVEL	BASE FOG	UNDER EXPOSED	PROCESSED	UNDER PROCESSED	PROCESSED	CORRECT EXP/PROC	OVER EXPOSED	UNDER PROCESSED	PROCESSED	CORRECT EXP/PROC	OVER EXPOSED
PRIMARY	0.01-0.09	0.01-0.13	0.14-0.39	0.40-0.90	0.40-0.90	0.40-0.90	0.91-1.34	0.91-1.34	0.91-1.34	0.91-1.34	0.91-1.34
INTERMEDIATE	0.10-0.17	0.01-0.20	0.21-0.39	0.40-0.90	0.40-0.90	0.40-0.90	0.91-1.69	0.91-1.69	0.91-1.69	0.91-1.69	0.91-1.69
FULL	0.18 AND UP	0.01-0.39	0.40-0.90	0.40-0.90	0.40-0.90	0.40-0.90	1.70 AND UP	1.70 AND UP	1.70 AND UP	1.70 AND UP	1.70 AND UP

TABLE 9-1

TOP SECRET C/

J MISSION DENSITY RANGES

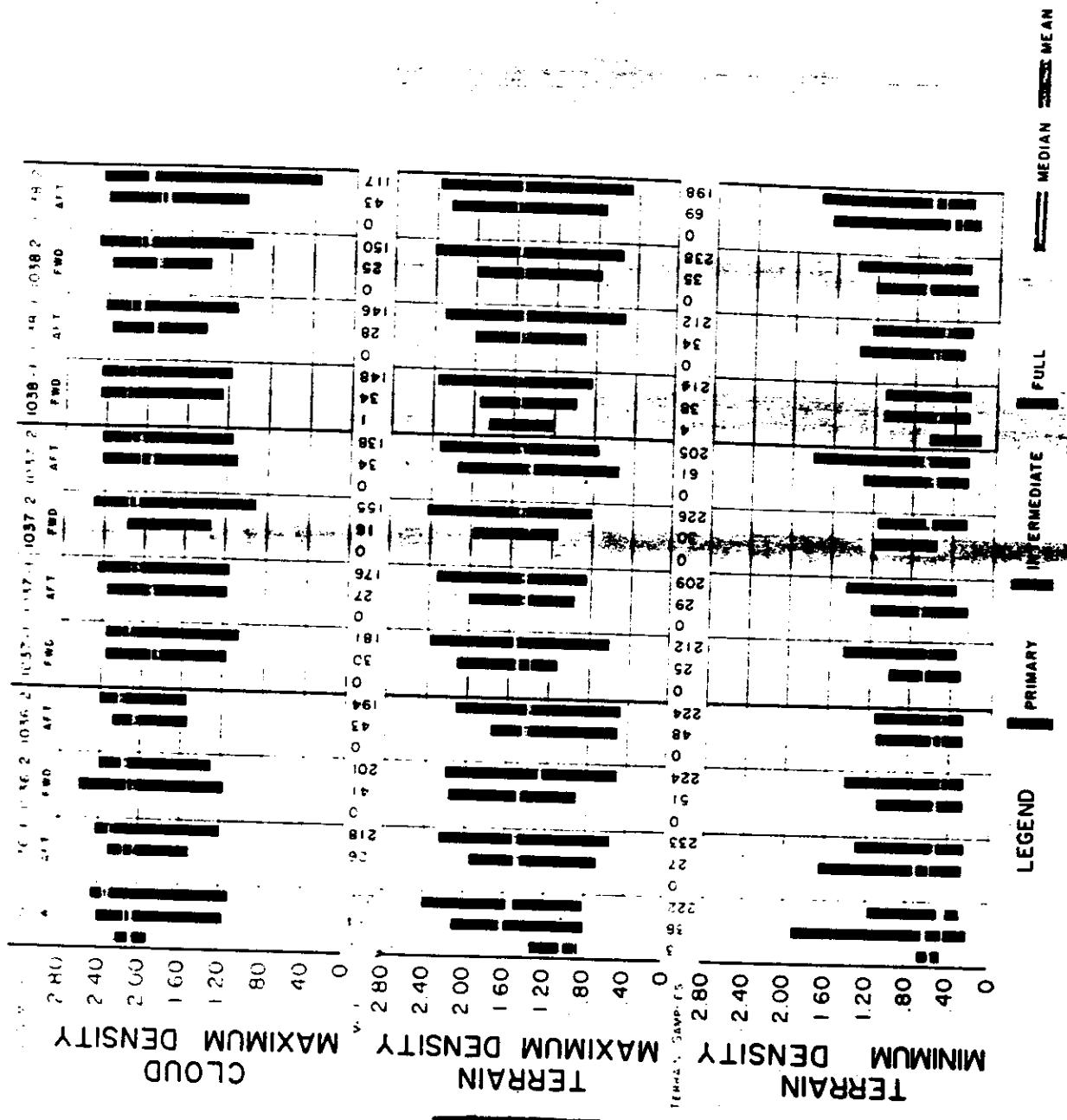


FIGURE 9-1

SECTION 10

PERFORMANCE MEASUREMENTS

The photography acquired by both panoramic cameras during Missions 1038-1 and 1038-2 received a MIP rating of 80. A summary is tabulated below of the average MTF/AIM resolution values measured by AFSPPP. The length of the microdensitometer slit used was 1 x 80 microns.

<u>MISSION</u>	<u>CAMERA</u>	<u>CYCLES/mm</u>	<u>MTF</u>	<u>AIM RESOLUTION</u>
1038-1	Fwd	53	58	19.8
1038-2	Fwd	63		
1038-1	Aft	76	77	15.0
1038-2	Aft	77		

The details of the measurement and computing techniques, complete measured and target locations are fully reported in the evaluation report published by AFSPPP and are therefore not included in this report.

SECTION II

VEHICLE ATTITUDE

The vehicle attitude errors for both Mission 1038-1 and 1038-2 were derived from the reduction of the Stellar camera photography. This attitude data is supplied to A/P by EPIC.

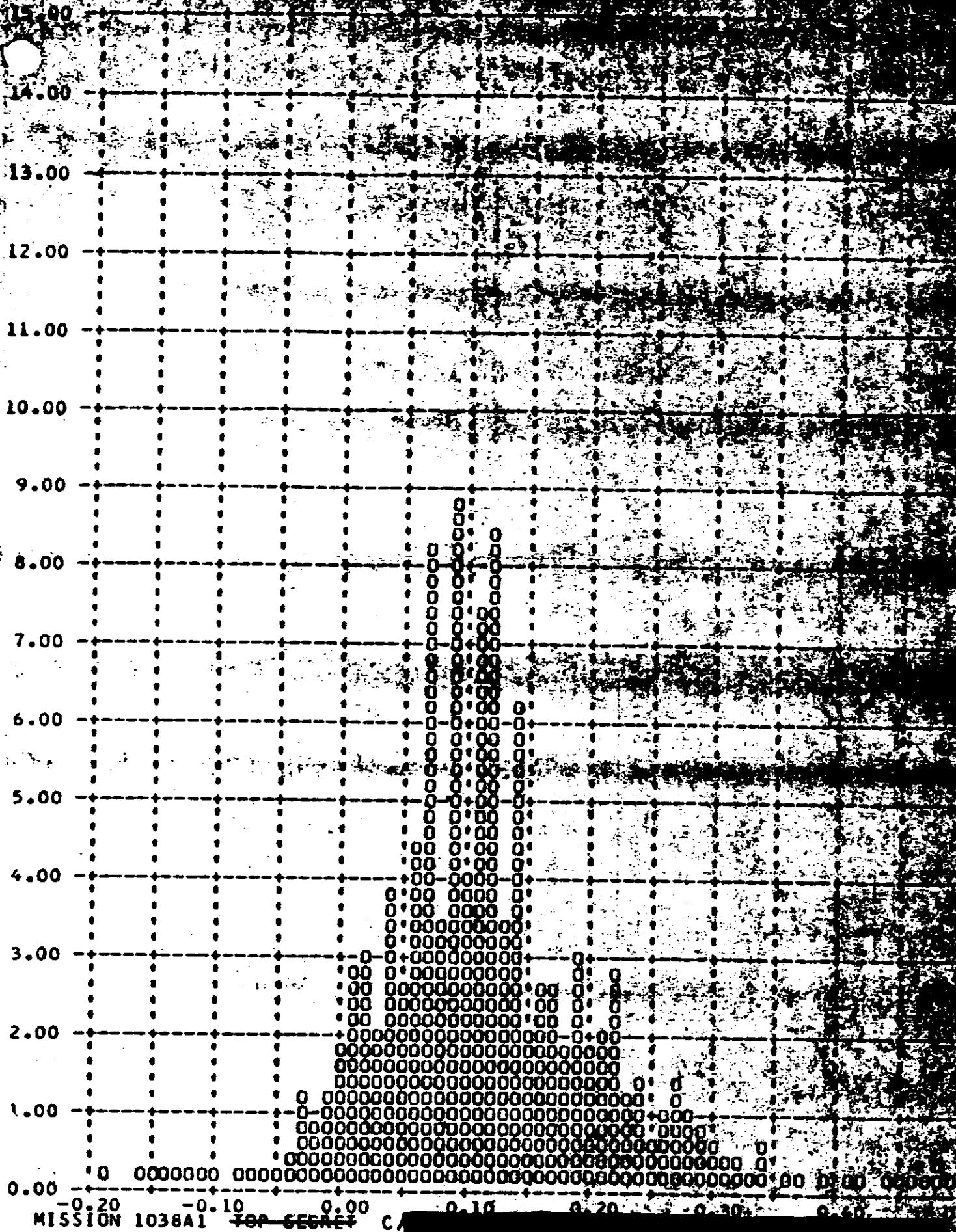
The attitude errors for each frame and the attitude control rates are calculated at the A/P computer facility. The computer also plots the frequency distribution of the rates and errors. Figures 11-1 thru 11-6 show these distributions for Mission 1038-1 and Figures 11-7 thru 11-12 for Mission 1038-2.

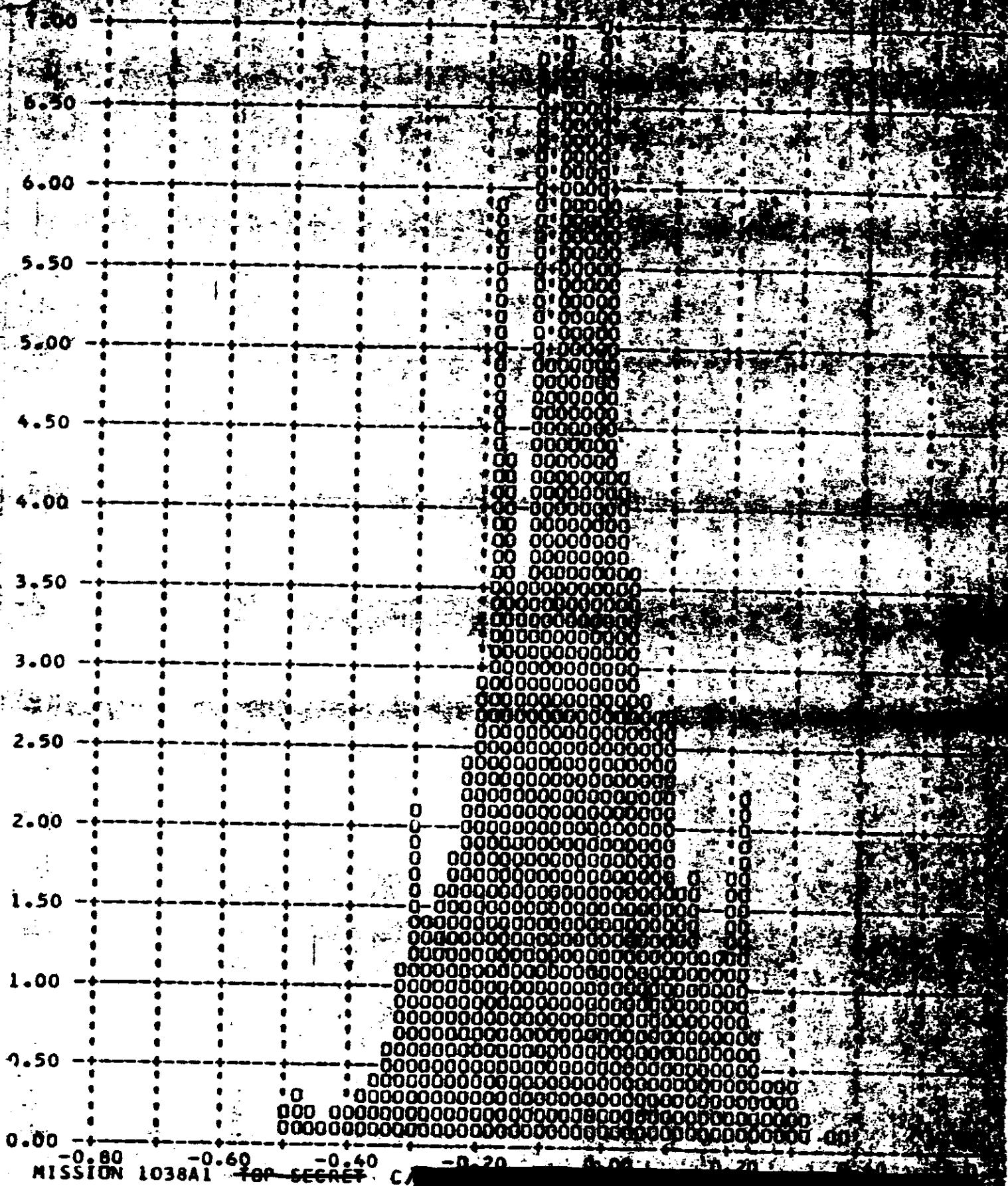
The summary table below lists the maximum attitude errors and rates that were experienced during 90% of the FWD camera photographic operations, excluding the first six frames of each operation, and the total range of the errors and rates.

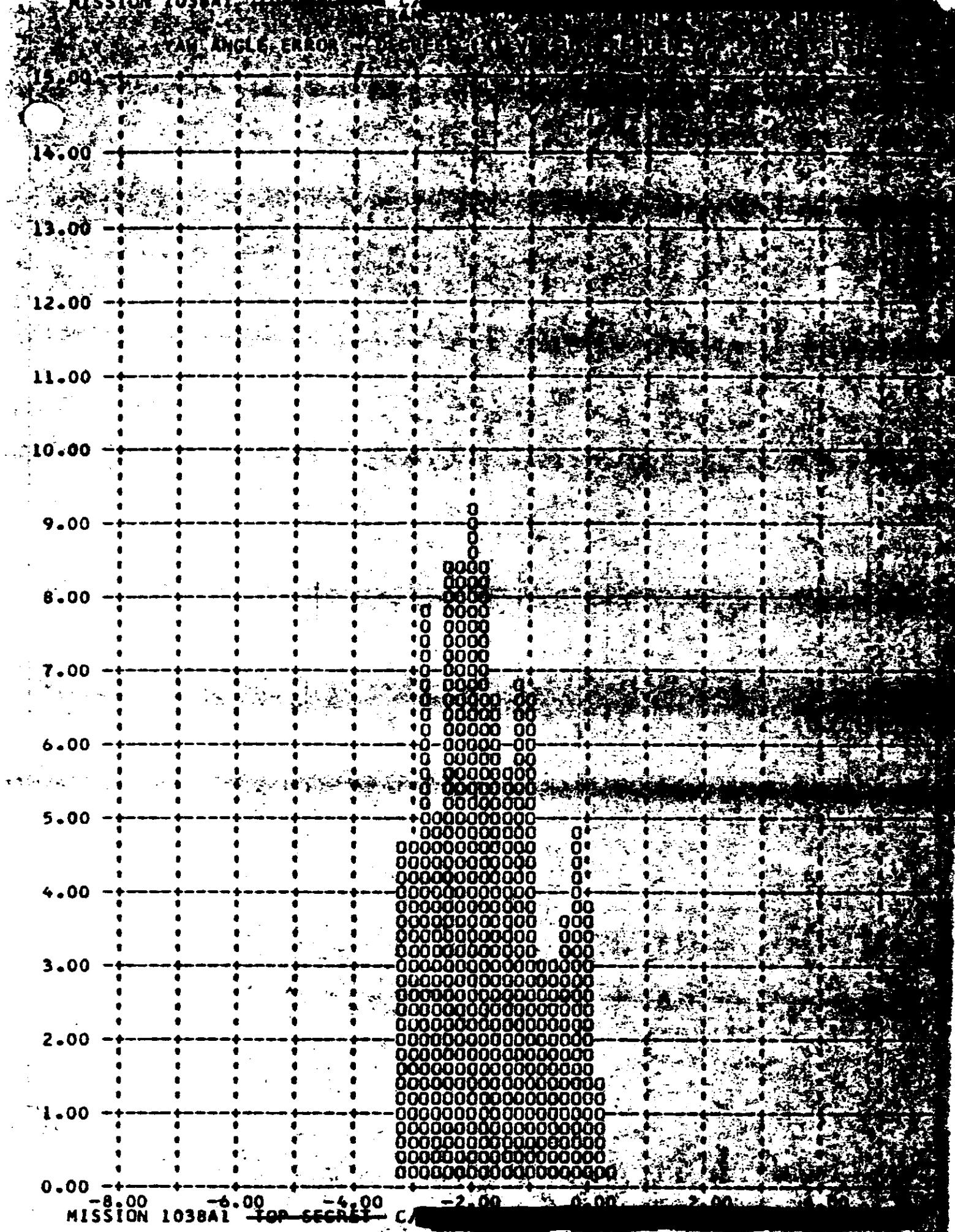
Value	Mission 1038-1		Mission 1038-2	
	90%	Range	90%	Range
Pitch Error ($^{\circ}$)	0.22	-0.19 to +0.49	0.39	+0.22 to +0.71
Roll Error ($^{\circ}$)	0.25	-0.50 to +0.38	0.51	-0.70 to +0.40
Yaw Error ($^{\circ}$)	2.98	-3.2 to +0.6	2.61	-3.2 to +0.6
Pitch Rate ($^{\circ}/hr$)	18.67	-34 to +98	19.98	-36 to +100
Roll Rate ($^{\circ}/hr$)	33.66	-80 to +95	46.65	-95 to +100
Yaw Rate ($^{\circ}/hr$)	39.91	-70 to +40	27.79	-52 to +32

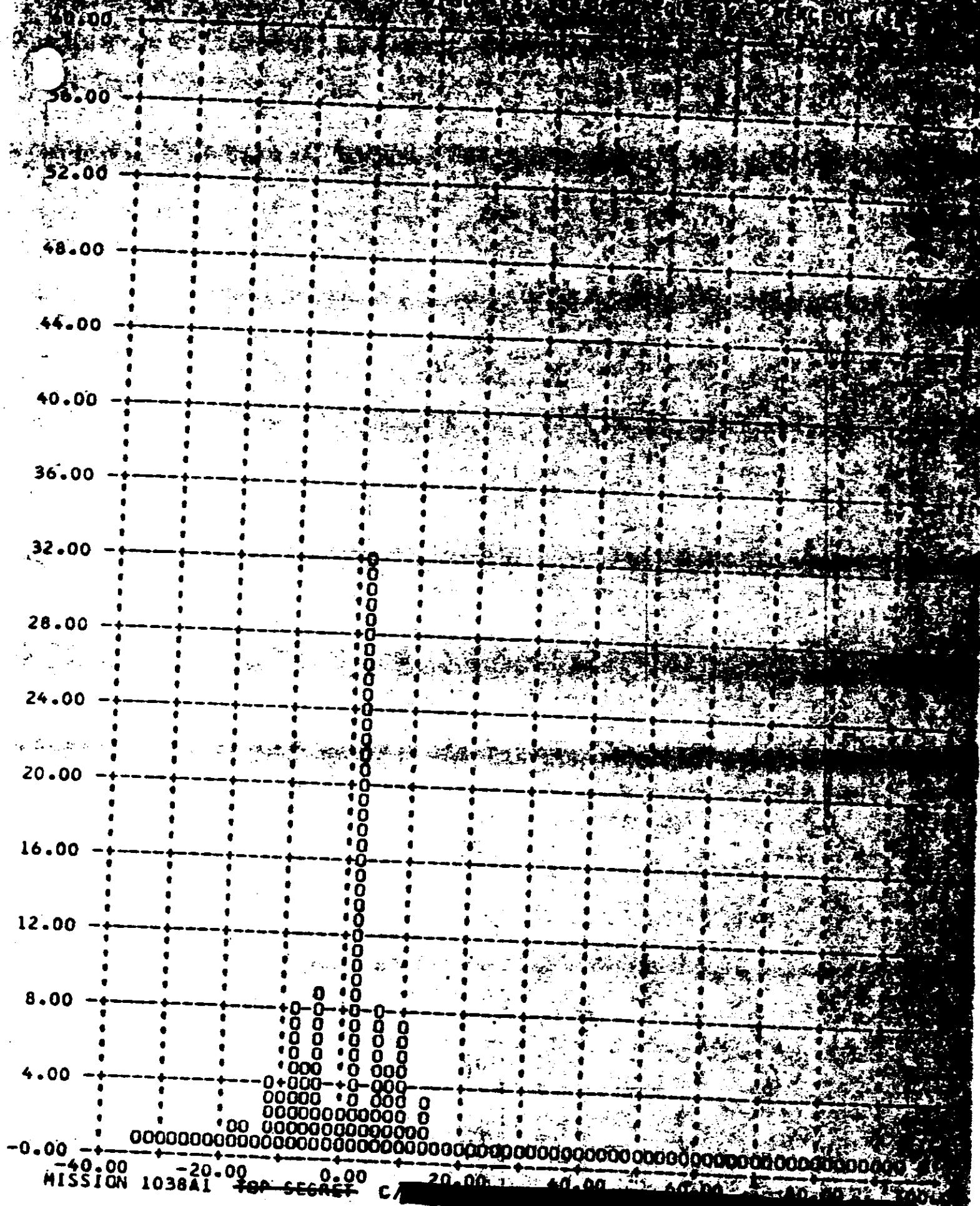
The panoramic photography was not degraded by the attitude control system.

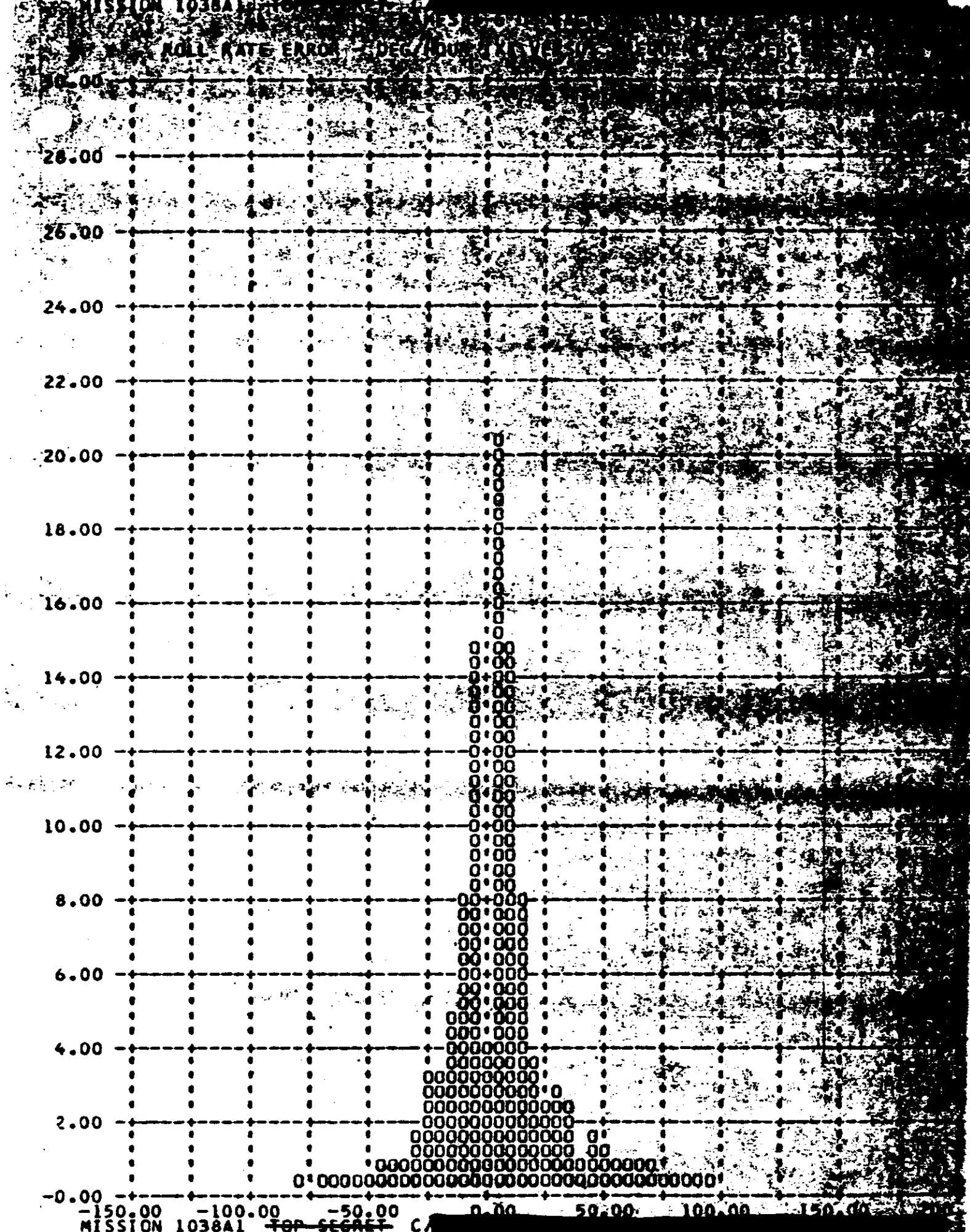
PITCH ANGLE

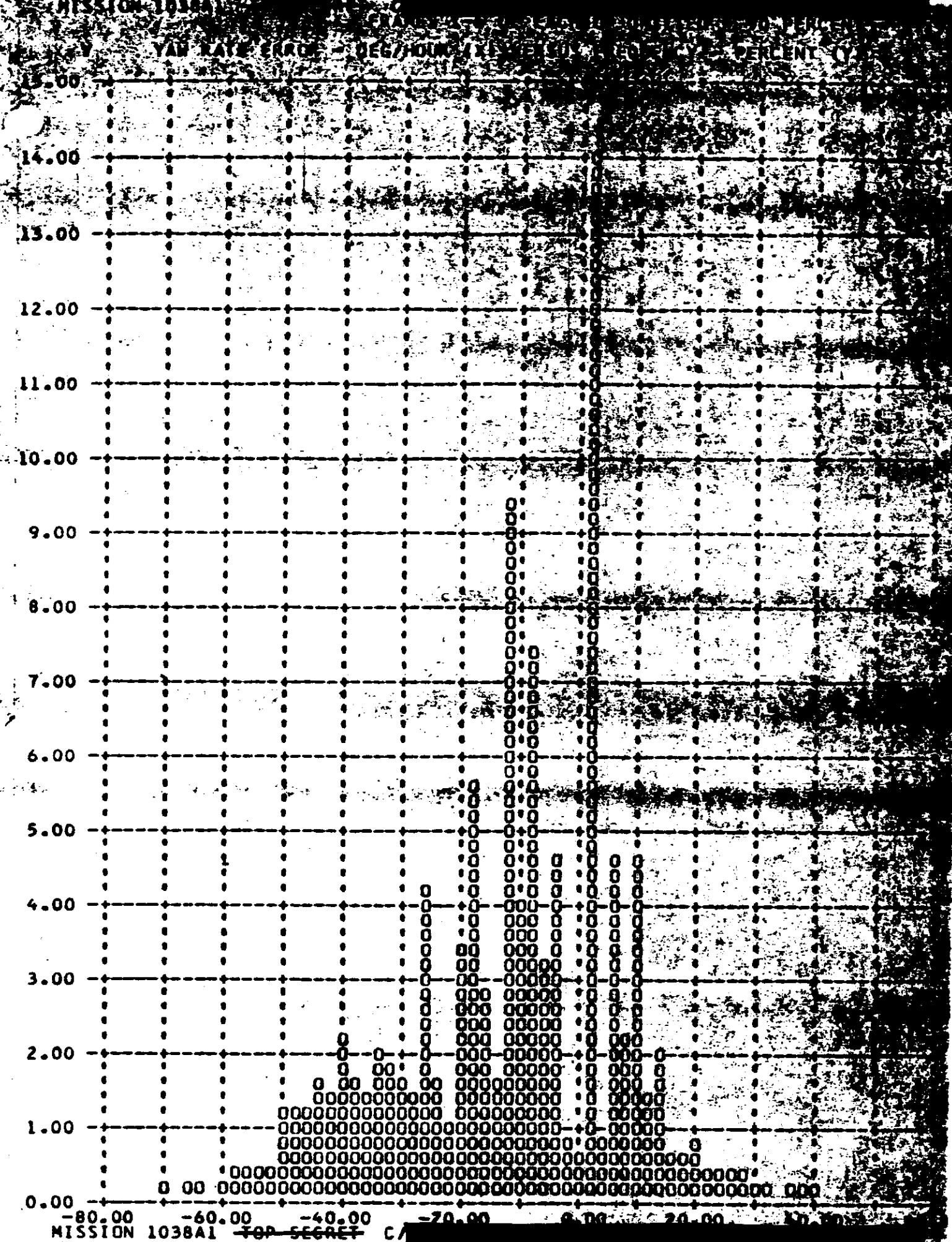






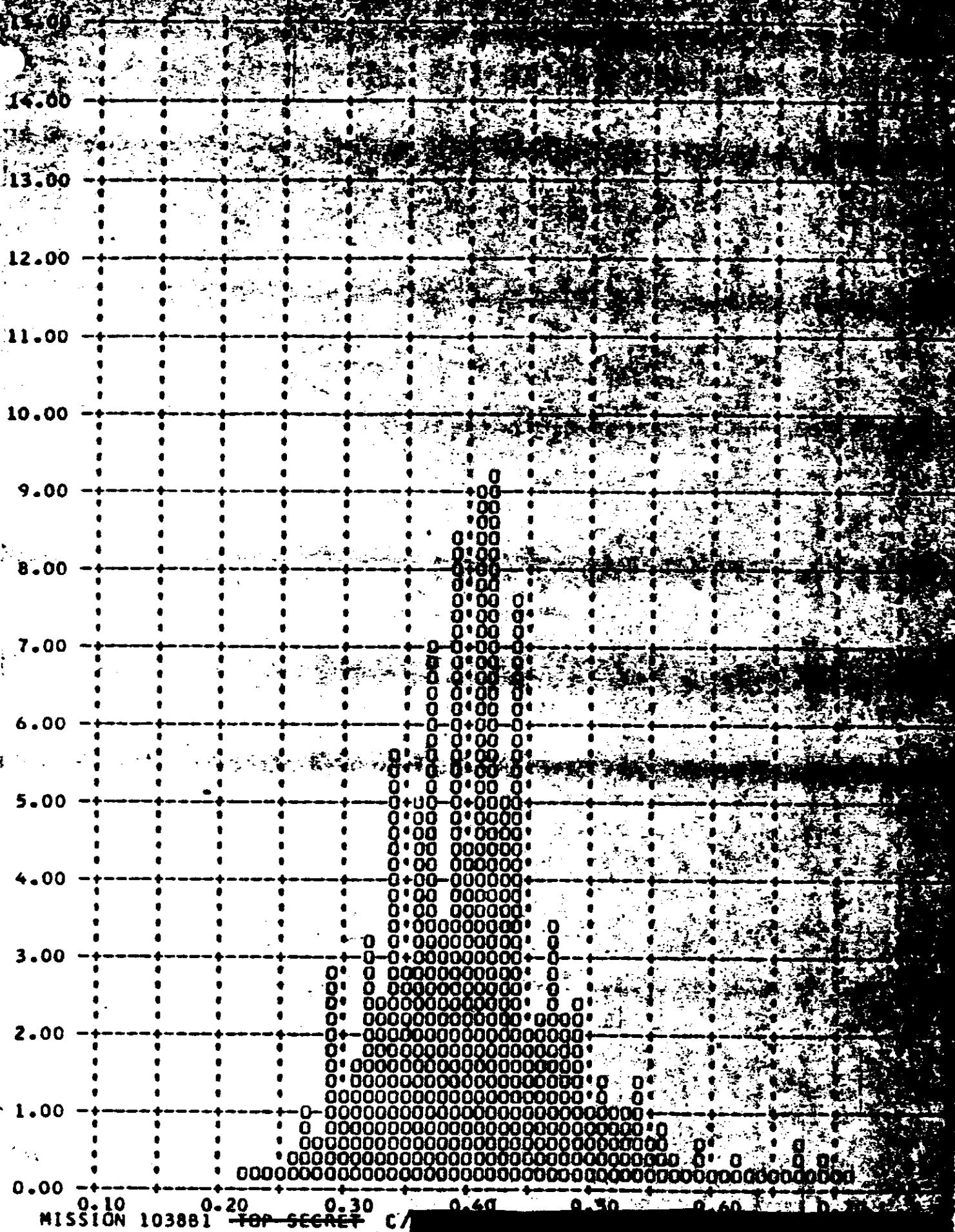




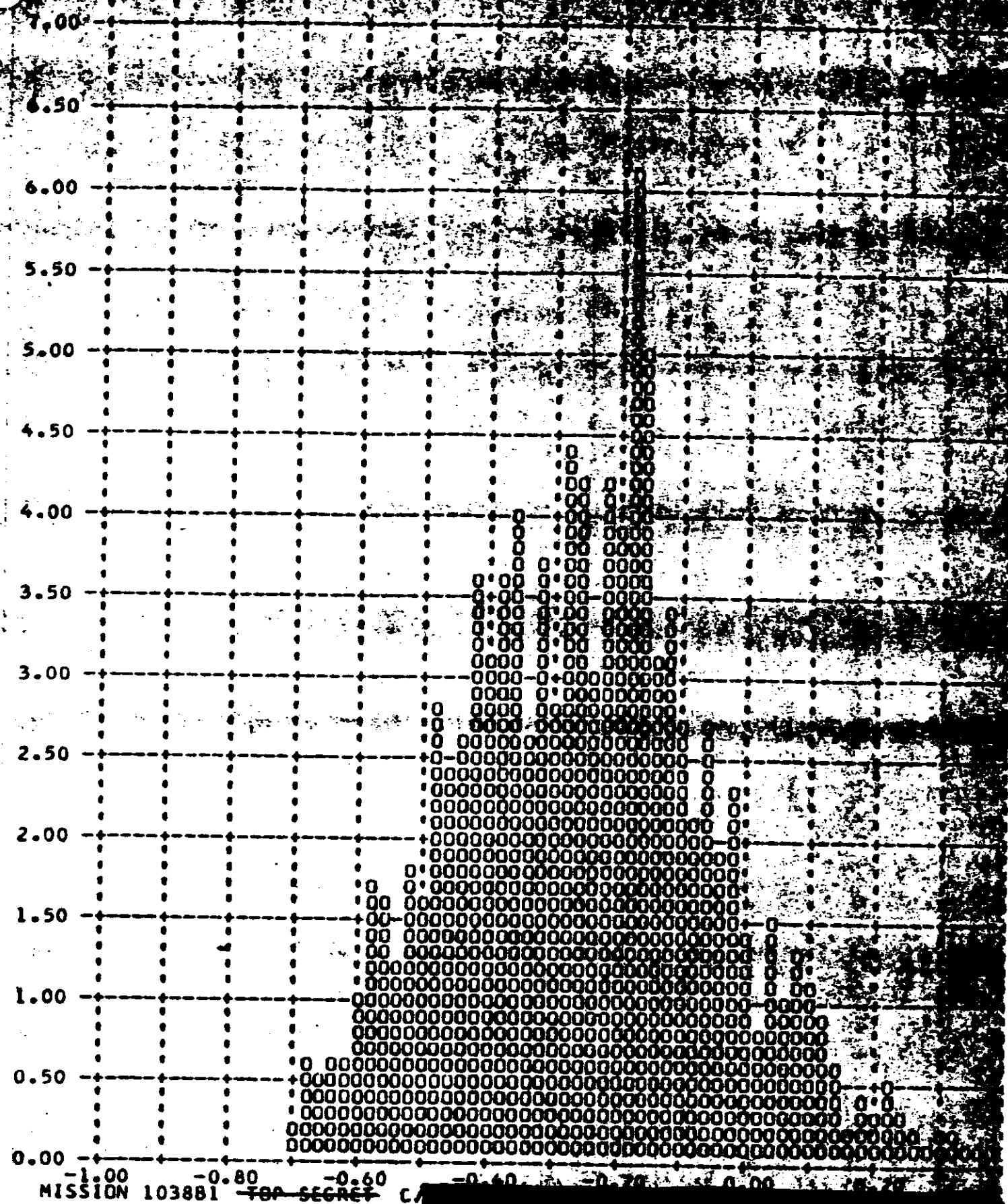


MISSION 103881

PITCH ANGLE FROM THE FIGHTER PILOT'S INSTRUMENT

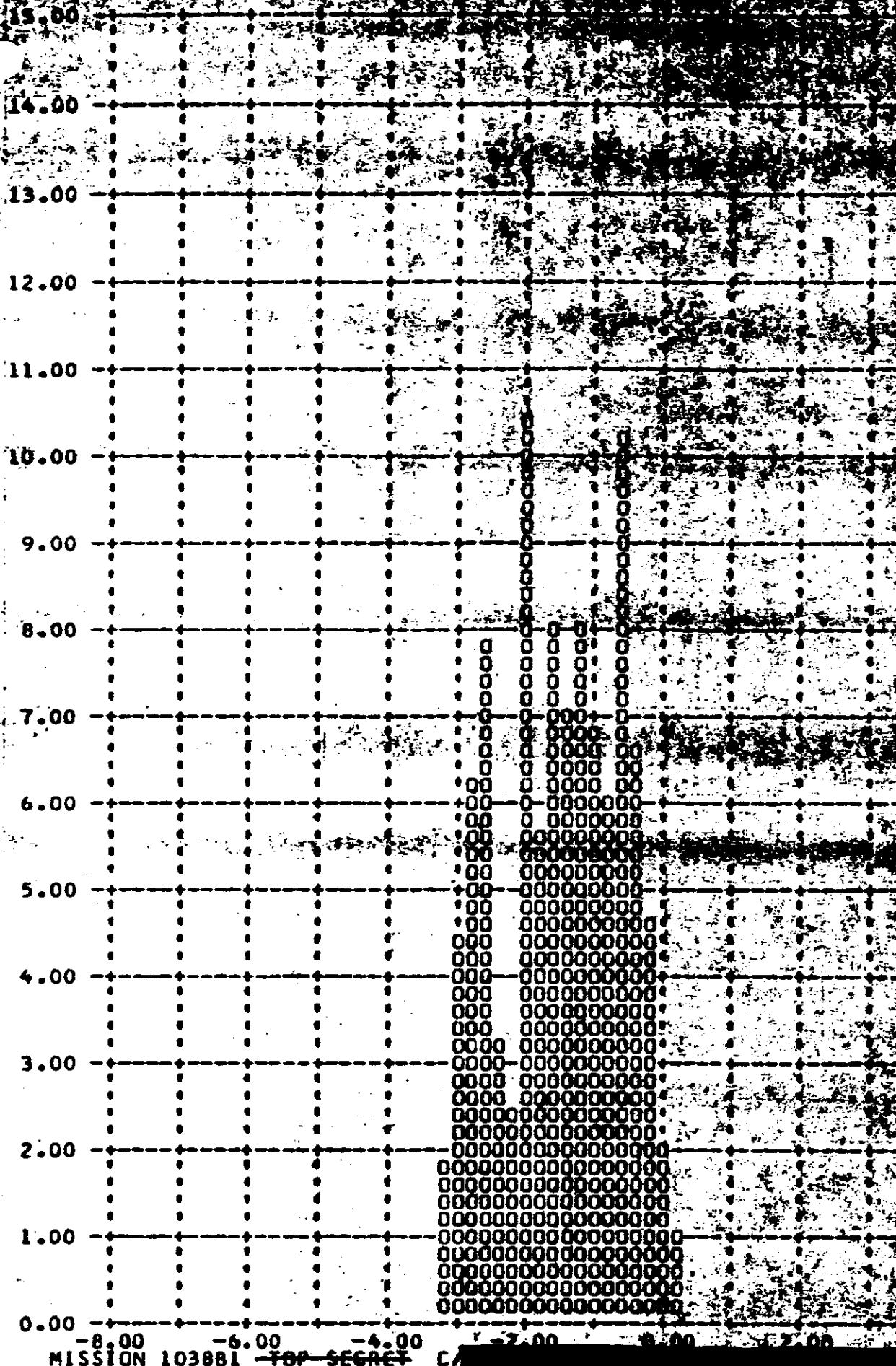


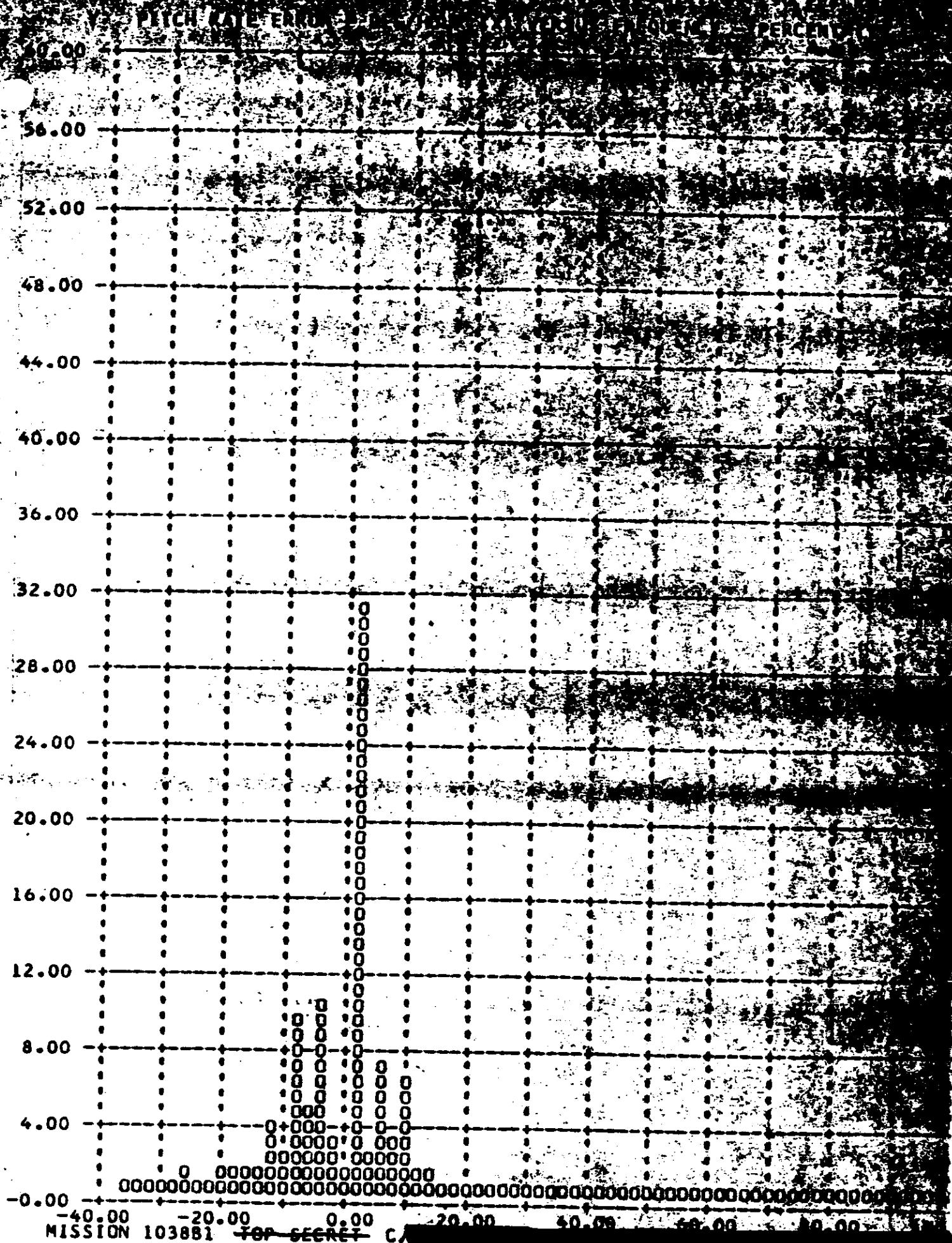
MISSION 103881 TOP SECRET C/ [REDACTED]

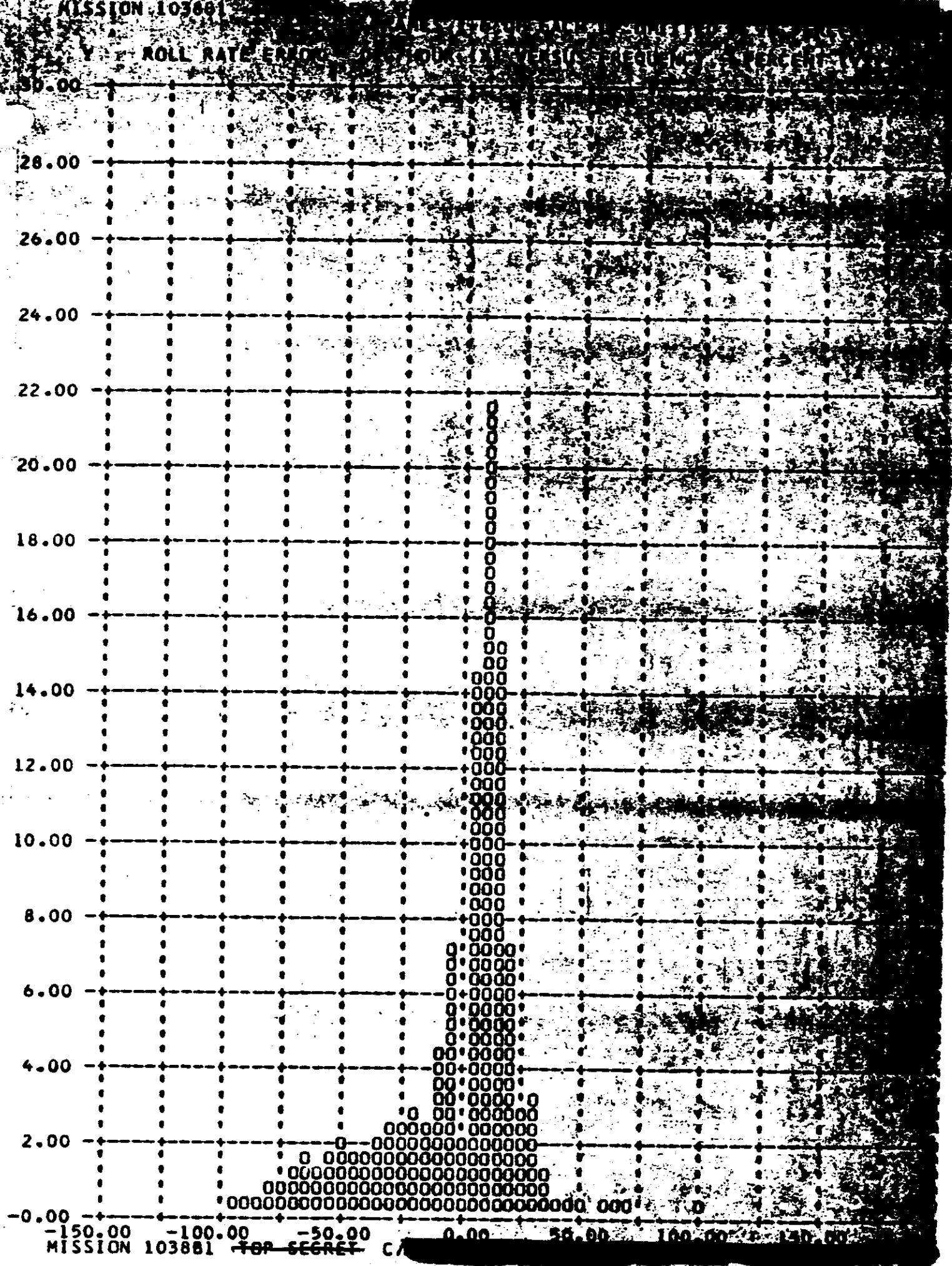


MISSION 103881

YAW ANGLE ERROR (DEGREES) VERSUS TIME (SECONDS)







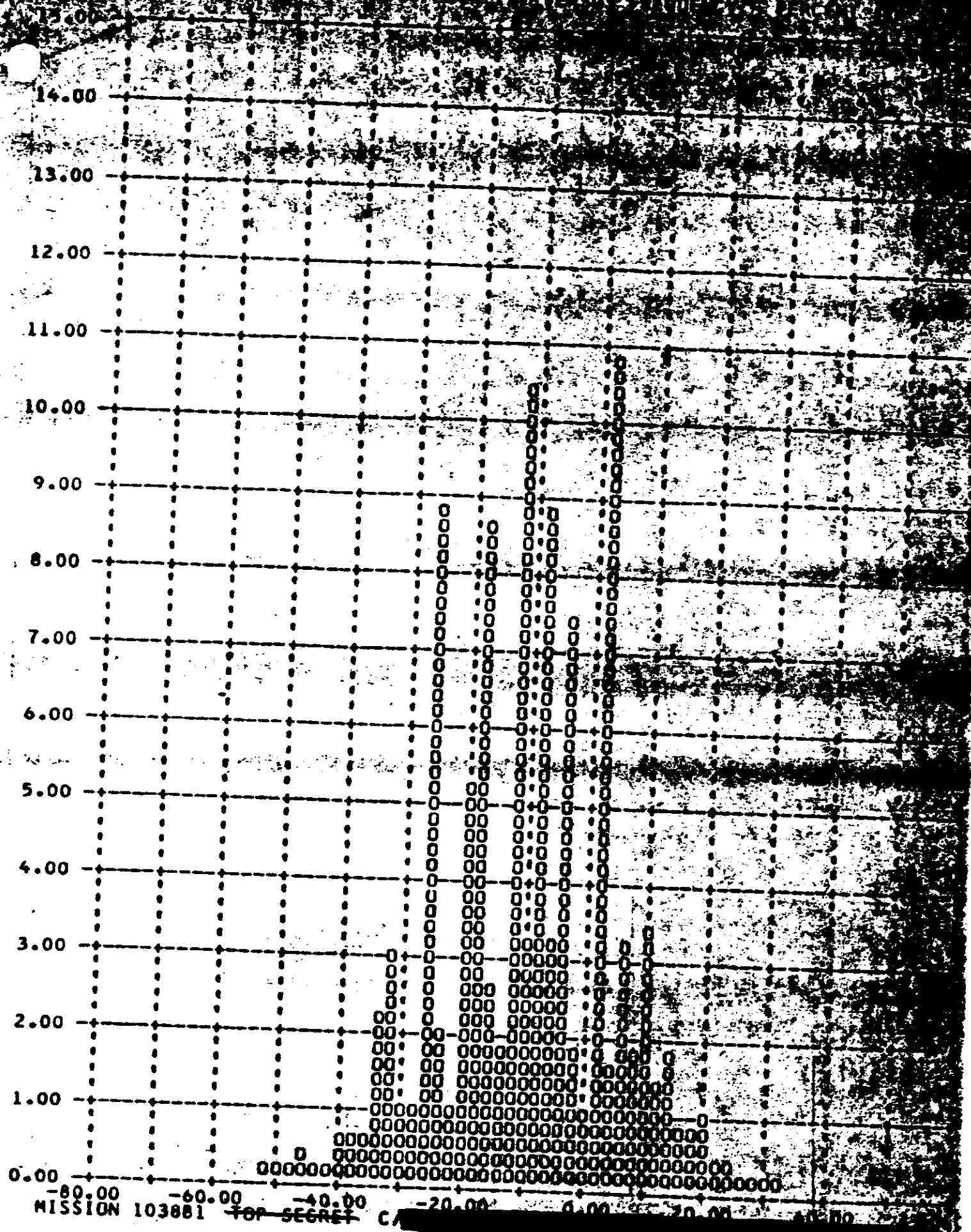


IMAGE SUMMAR ANALYSIS

The frame correlation tape supplied to A/P by MPIO contains binary time word of each frame of photographs. A computer program has been assembled at A/P which calculates the exposure time of each frame and compares the camera cycle rate with the eproms to calculate the V/h mismatch. This data is combined with the vehicle attitude error and rate values of each frame and the crab error caused by earth rotation at the latitude of each frame. The program outputs the total along track and cross track IMC error and the range of ground resolution that can be acquired by a camera regardless of focal length and system capabilities.

The computer rejects the first six frames of all operations as the large V/h error induced by camera start-up is not representative of the overall system operations. The frequency distribution of the V/h errors and resolution limits for each panoramic camera have been calculated and plotted in Figures 12-1 through 12-16. The variation in the data for the panoramic cameras is the result of the different slit widths used during the mission and the resulting slower exposure time in the FWD camera.

The summary shown in Table 12-1 presents the maximum V/h rate errors and resolution limits that existed during 90% of the photographic operations and the total range of values during all operations that were computed.

~~TOP SECRET C~~

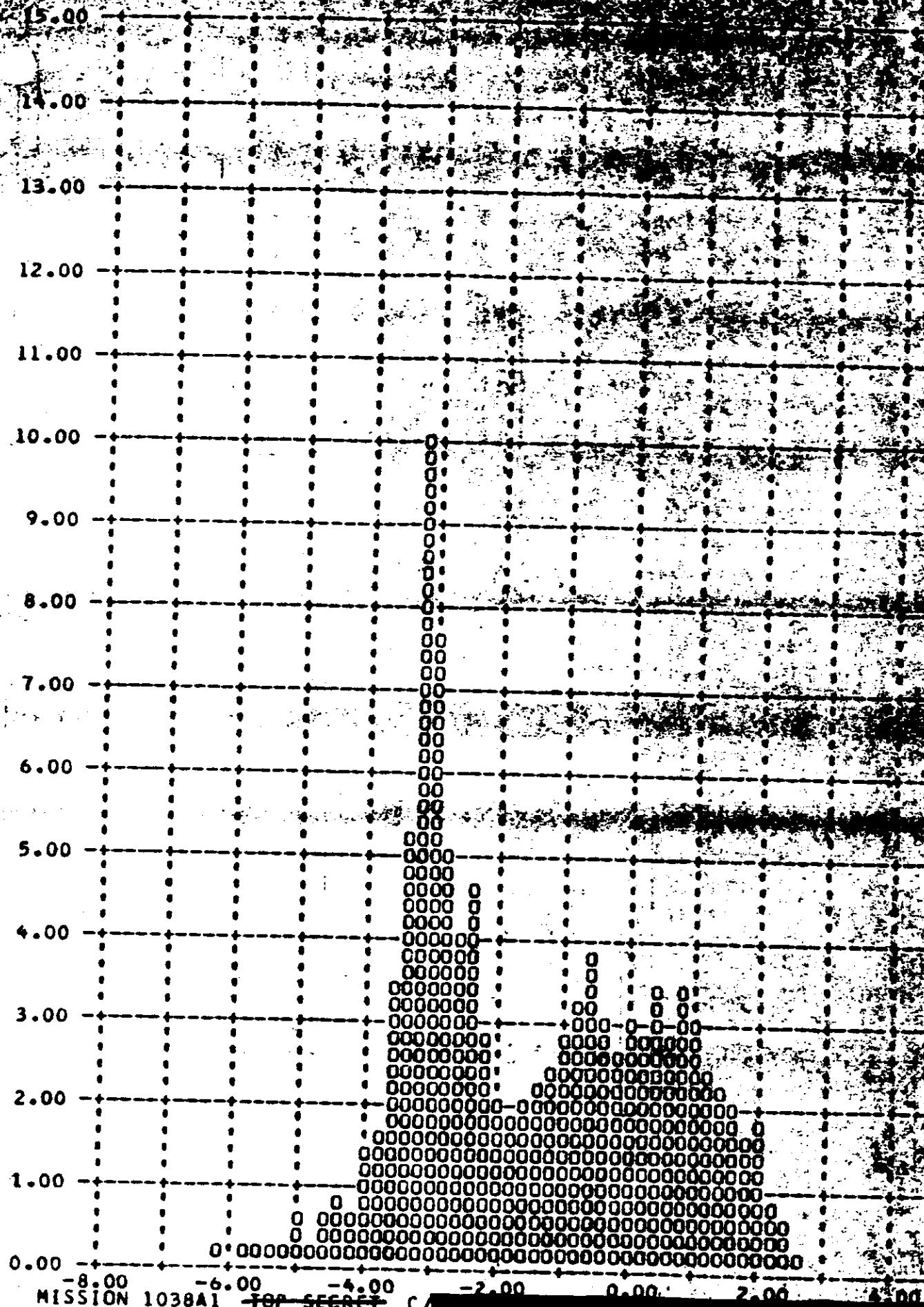
MISSION 1038

V/H RATIO AND RESOLUTION LIMITS

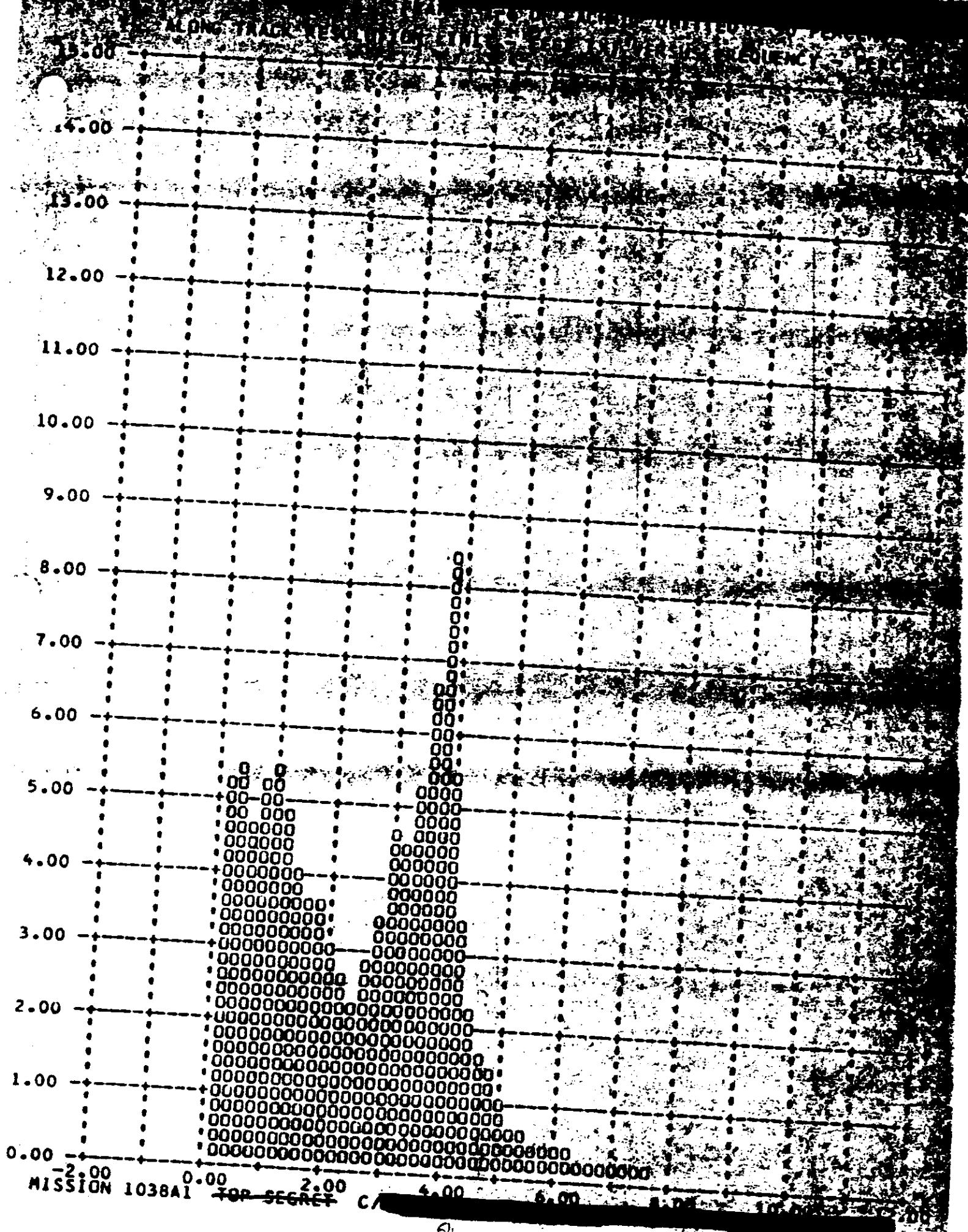
Value	Units	Camera	Mission 1038-1		Mission 1038-2	
			Range	Value	Range	Value
v/h Ratio Error	%	FWD	-3.56 to -12.6	3.40	-5.4 to +2.6	2.0
		APT	-9.0 to -13.6	3.07	-1.0 to +2.6	2.6
Along Track Resolution Limit	Foot	FWD	4.12	0.2 to 6	3.7	0.2 to 6
		APT	4.12	0.3 to 6	3.7	0.2 to 6
Cross Track Resolution Limit	Foot	FWD	0.2	0.2 to 0	0.2	0.2 to 0
		APT	0.2	0.2 to 0	0.2	0.2 to 0

TABLE

V/H RATIO EARTH - PERCENT OF THE EARTH'S GRAVITY - PERCENT



MISSION 1038A1 TOP SECRET C / [REDACTED]



MISSION 1038A1

CROSS TRACK RESUME

14.00

16.00

13.00

12.00

11.00

10.00

9.00

8.00

7.00

6.00

5.00

4.00

3.00

2.00

1.00

0.00

-4.00 -2.00 0.00 2.00 4.00 6.00 8.00

MISSION 1038A1 TOP SECRET C

MISSION 1038A1

INC ERROR - PERCENT INTEGRAL FREQUENCY

15.00

14.00

13.00

12.00

11.00

10.00

9.00

8.00

7.00

6.00

5.00

4.00

3.00

2.00

1.00

0.00

-8.00

-6.00

-4.00

-2.00

0.00

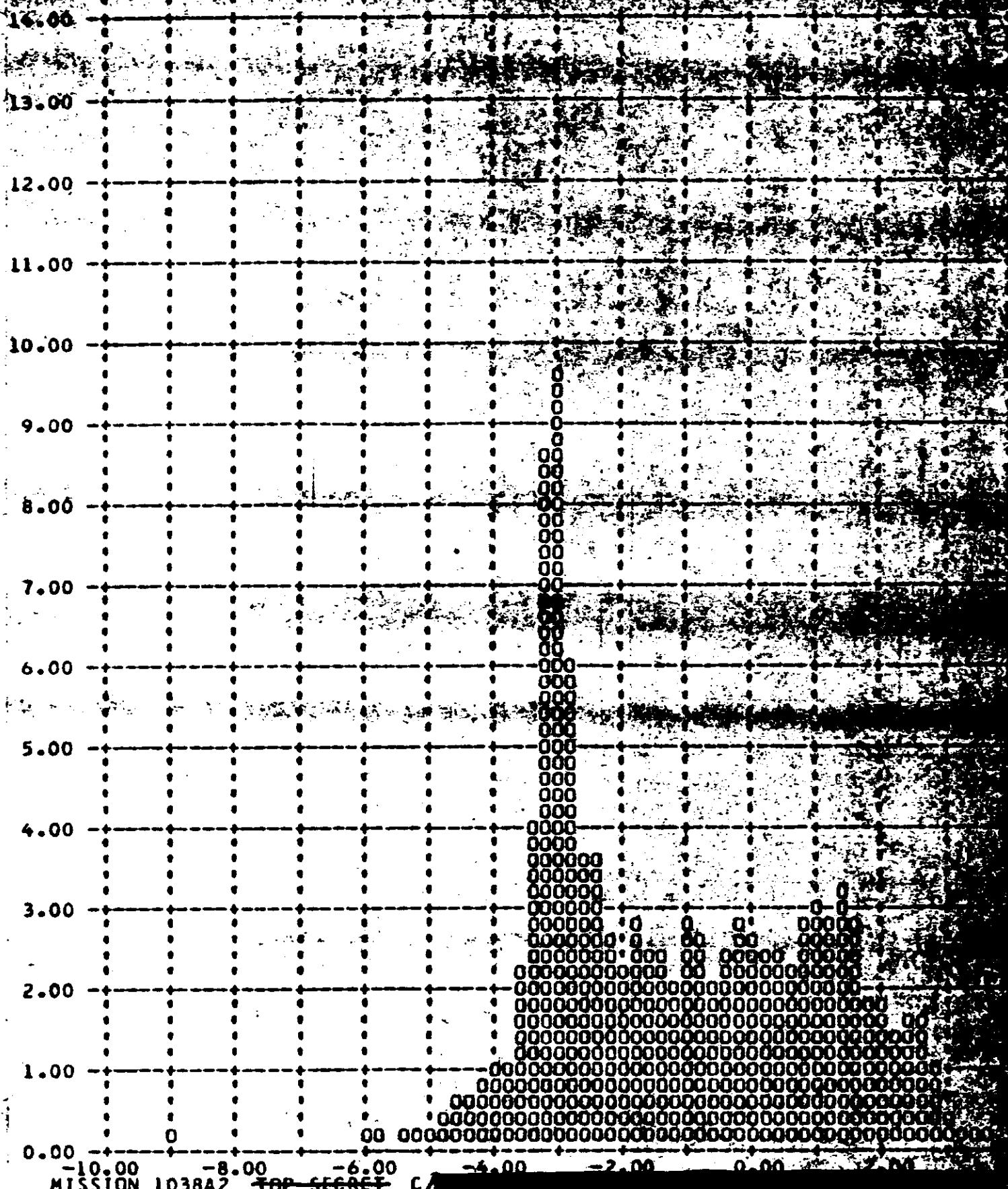
2.00

4.00

MISSION 1038A1 TOP SECRET C

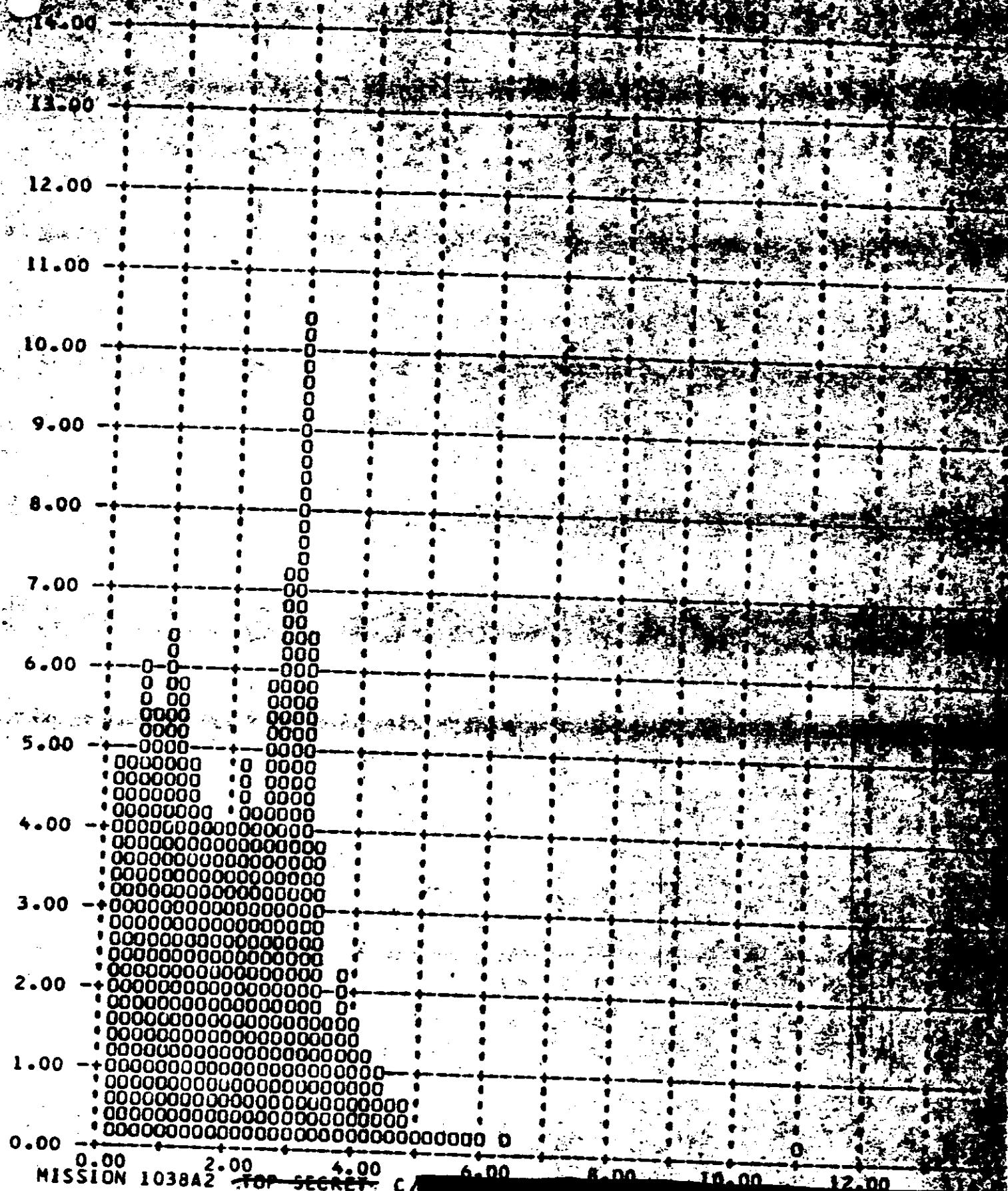
MISSION 1038A2

WINGMAN POSITION REPORT



MISSION 1038A2 TOP SECRET C

Figure 12-5



MISSION 1038A2

CROSS TRACK RESOLUTION

15.00

14.00

13.00

12.00

11.00

10.00

9.00

8.00

7.00

6.00

5.00

4.00

3.00

2.00

1.00

0.00

-4.00

-2.00

0.00

2.00

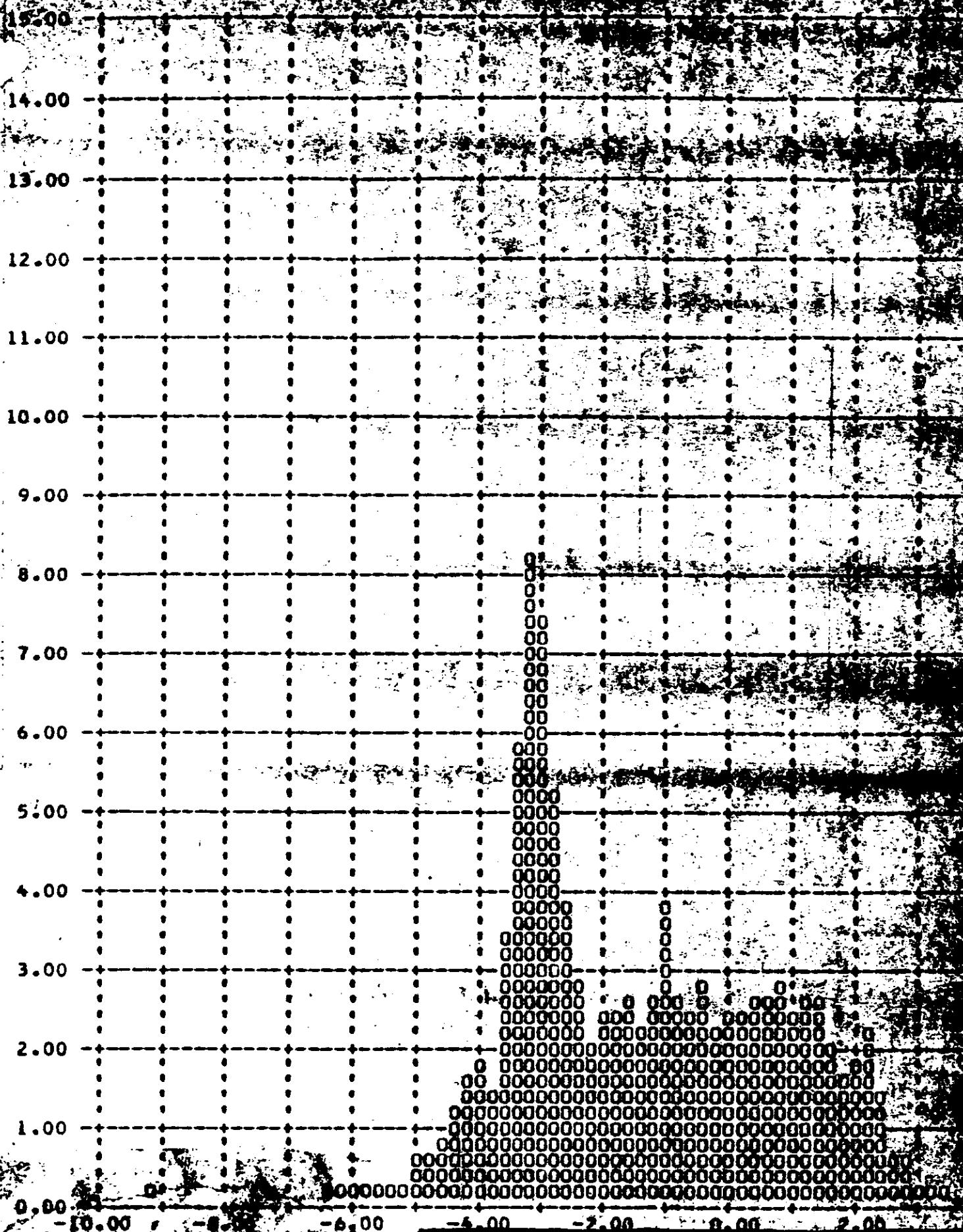
4.00

6.00

8.00

10.00

MISSION 1038A2 TOP SECRET C

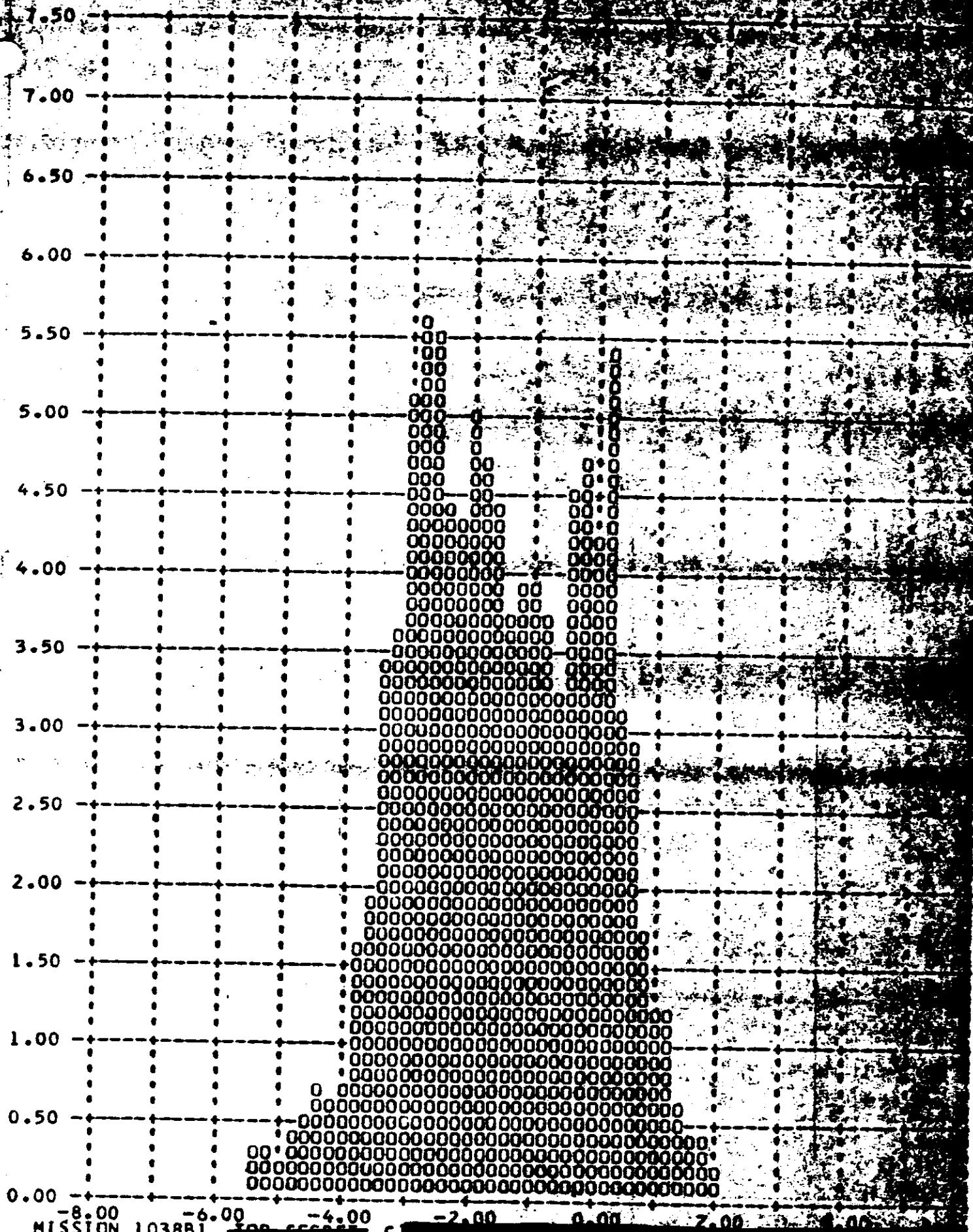
MISSION 103842
IMC ERROR - PERCENT

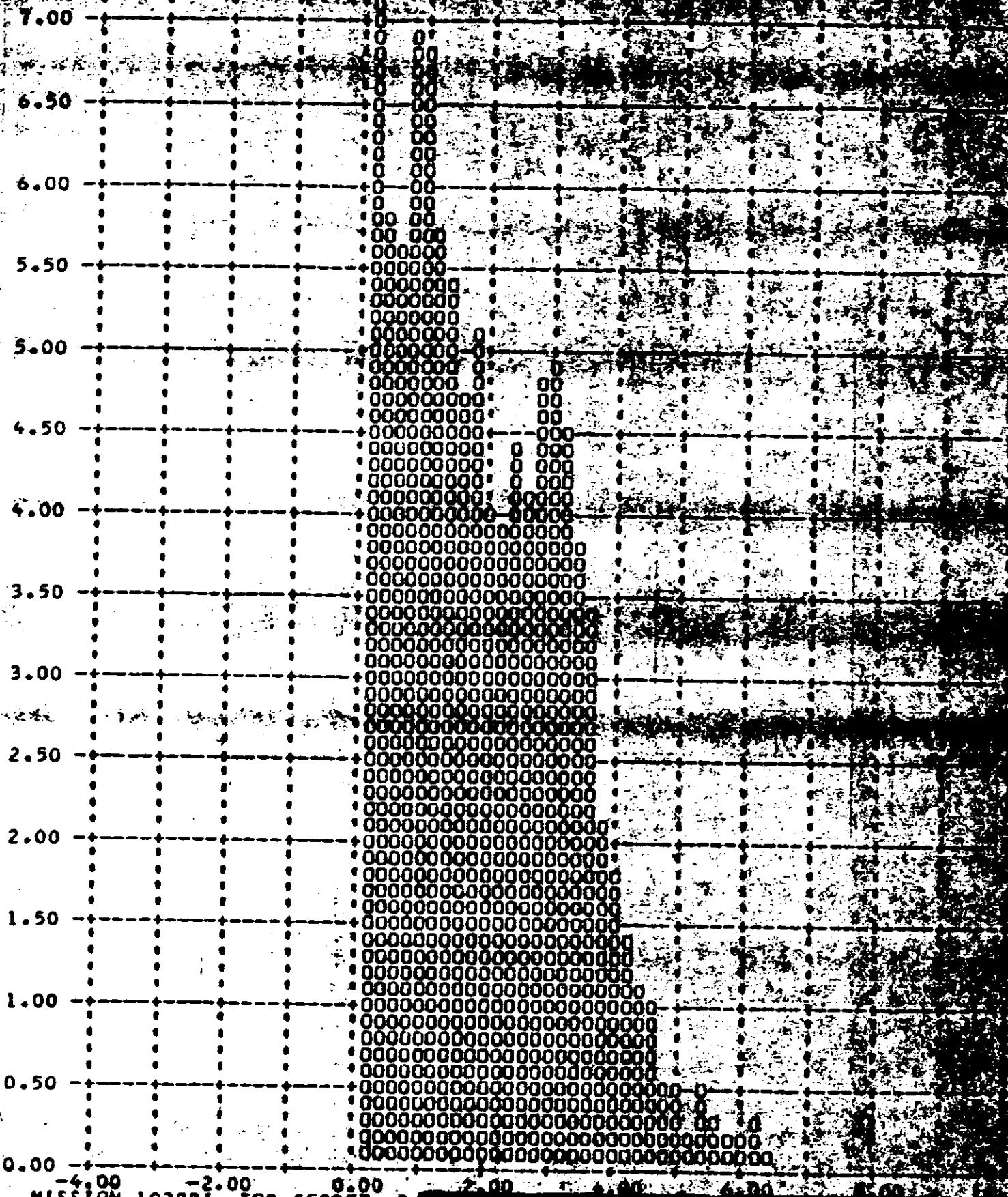
MISSION 103842

TOP SECRET C

MISSION 10388

W/H RATIO ERROR PERCENT X EVER IN FREQUENCY PERCENT





MISSION 1038B1 TOP SECRET C

MISSION 103881

CROSS TRACK RESOLUTION

15.00

14.00

13.00

12.00

11.00

10.00

9.00

8.00

7.00

6.00

5.00

4.00

3.00

2.00

1.00

0.00

-4.00

-2.00

0.00

2.00

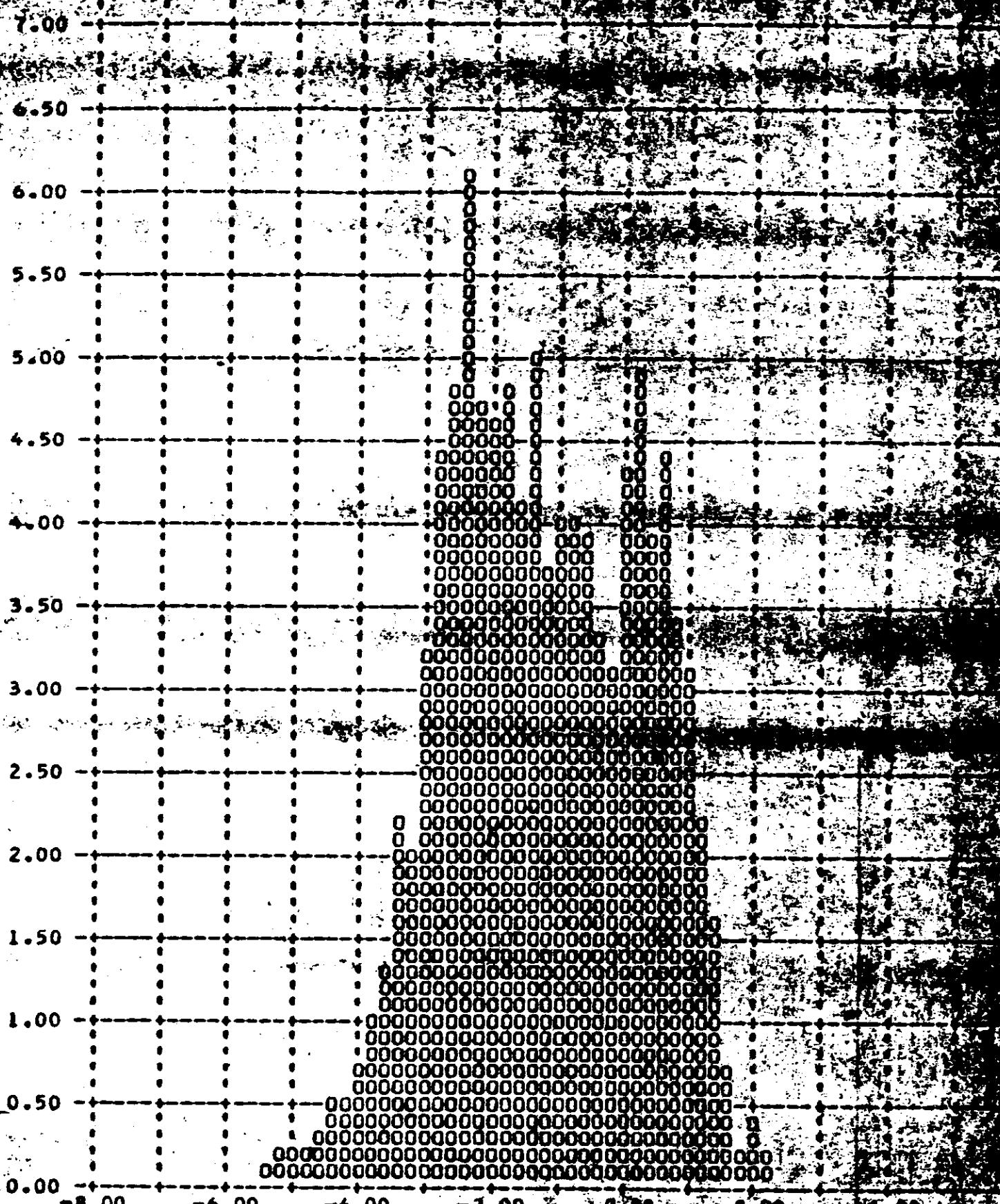
4.00

6.00

8.00

10.00

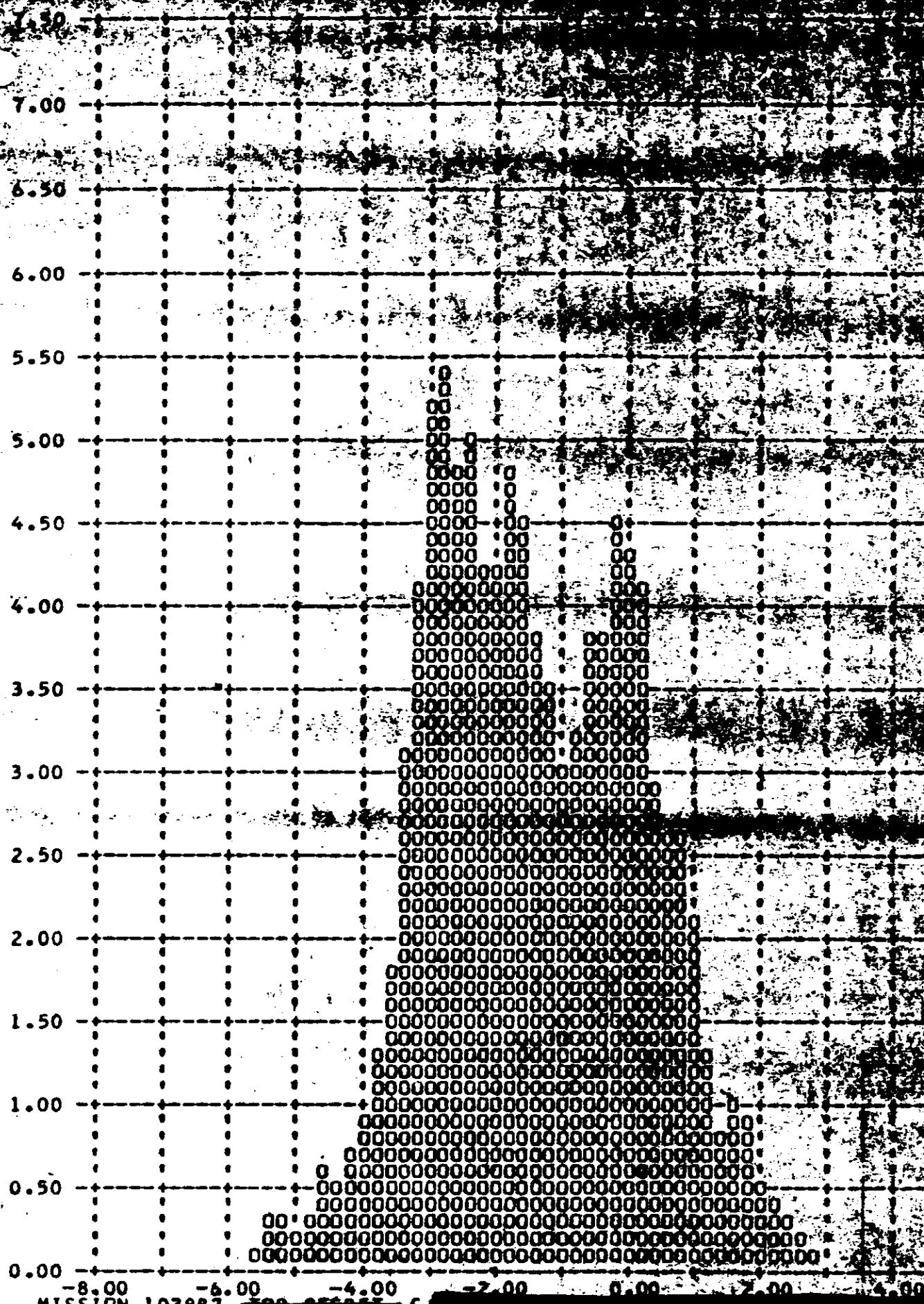
MISSION 103881 TOP SECRET C



MISSION 103881 TOP SECRET C

MISSION 103882

V/H RATE



MISSION 103882

TOP SECRET

MISSION 103882

ALONG TRACK REPORT

1.50

7.00

6.50

6.00

5.50

5.00

4.50

4.00

3.50

3.00

2.50

2.00

1.50

1.00

0.50

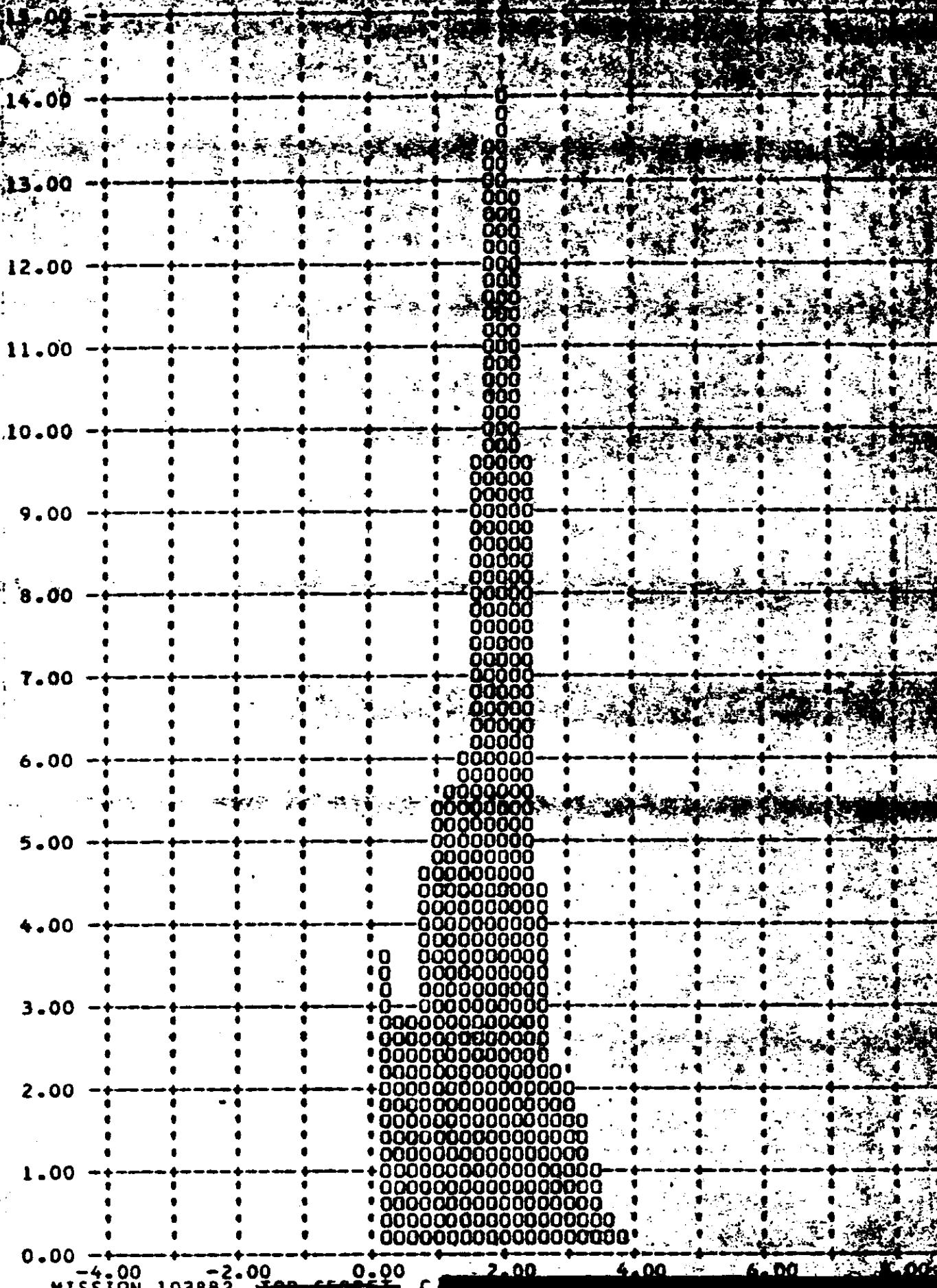
0.00

-4.00 -2.00 0.00 2.00 4.00 6.00 8.00

MISSION 103882

TOP SECRET C

MISSION 1038B2 CROSS TRACK RESOLUTION



MISSION 1038B2

TOP SECRET

CA

MISSION 103882

7.00

6.50

6.00

5.50

5.00

4.50

4.00

3.50

3.00

2.50

2.00

1.50

1.00

0.50

0.00

-8.00

-6.00

-4.00

-2.00

0.00

2.00

4.00

MISSION 103882

TOP SECRET C

RADIATION DOSAGE

Each recovery system flown on a Corona mission contains a sealed packet of Eastman Type 3401 and Royal X Pan emulsions to determine the total radiation received at the take-up cassette. Both film types have been irradiated by IMSC at various levels and the base plus fog densities recorded after controlled processing.

Following recovery the film dosimeter packets are removed at A/P and processed with a pre-flight sample of the same film type and sensitometric control film. The resulting base plus fog density measurement of the dosimeter strips is used to ascertain the total radiation level. The table below presents the base plus fog readings for the dosimeter strips and the radiation level equivalents.

Mission 1038-1

Mission 1038-2

<u>Emulsion</u>	<u>B + F</u>	<u>Density</u>	<u>Radiation</u>	<u>B + F</u>	<u>Density</u>	<u>Radiation</u>
Type 3401		0.18	0.6R		0.22	1.0R
Royal X Pan		0.24	0.4R		0.30	0.6R

This level is well below the level that will degrade the panoramic photography.

SUMMARY

SYSTEM RELIABILITY

Reliability calculations for the payload are based on a sample beginning with M-7. Hence both the major parts of the M-7 program and the J-19 Program are covered in the calculation. For certain auxiliaries, i.e., the stellar-index camera and the horizon cameras, the sample size is changed to recognize incorporation of modified equipment or new designs where reliability was one of the principal reasons for the modification. However, for primary mission function, the sample size is consistent with reliability reporting for the vehicle.

The reliability estimates of this section deal exclusively with the payload. Failures to achieve orbit or vehicle induced failures are thereby excluded. Recoveries before a complete mission has been completed are considered as full missions providing that early termination was caused by reasons not connected with payload operation. Film quality is not considered in the reliability estimate calculation. Hence, only electrical and mechanical functioning are considered.

The reliability estimate is also divided into primary and secondary functions. The primary functions are operation of the panoramic cameras, main camera door operation, operation of the payload clock, and recovery operations. The secondary mission functions are horizon camera operation excluding catastrophic open shutter failure mode, auxiliary data recording, and stellar-index camera operation. A summary of estimated reliability is shown in Table 14-1.

Panoramic Camera Reliability

Sample Size - 167 opportunities to operate.
One failure - S/I Programmer on system J-19.
Assume - 3000 cycles per camera per mission.
Estimated Reliability - 99.0% at 50% confidence level.

Main Camera Door Reliability

Sample Size - 55 vehicles x 2 doors = 110 opportunities to open.
Estimated Reliability = 99.4% at 50% confidence level.

Payload Command & Control Reliability

Sample size: 936 hours operation

Two Failures

Estimated Reliability = 97.3% at 50% confidence level

Payload Clock Reliability

Sample size: 936 hours operation

No Failures

Estimated Reliability = 99.1% at 50% confidence level

Estimated Reliability of Payload Functioning on Orbit: 97.5%
50% confidence level

Recovery System Reliability

75 opportunities to recover

1 failure - improper separation due to rubber band

Estimated Reliability = 97.8% at 50% confidence level

Stellar-Index Camera Reliability

Sample begins with J-3

Sample size = 23,380

Number of failures = 4

Estimated Reliability = 91.9% at 50% confidence level

Horizon Camera Reliability

Sample includes J5 and up

Sample size: 94,500

Estimated Reliability of Single Camera = 98.9% at 50% confidence level.

Estimated Reliability of Four Horizon Cameras at a Parallel Redundant System = 99.9% at 50% confidence level.

ESTIMATED RELIABILITY SUMMARY

(AT 50% CONFIDENCE LEVEL)

MISSION NUMBER	PRIMARY FUNCTIONS	SECONDARY FUNCTIONS		RECOVERY SYSTEM	SIGNALING SYSTEM	WEAPONS SYSTEM	MATERIALS SYSTEM	AIRCRAFT SYSTEM	MISSION SUPPORT
		ON - ORBIT FUNCTIONS	OFF - ORBIT FUNCTIONS						
9008 to 1008	PANORAMIC COMMAND & CONTROL SYSTEM BOOKS	PATROL CLOCK	SAMPLE COLLECTOR	RECOVERY SYSTEM	SIGNALING SYSTEM	WEAPONS SYSTEM	MATERIALS SYSTEM	AIRCRAFT SYSTEM	MISSION SUPPORT
1009	PILOT'S POSITION	AMPLES	PALEOLOGY	RECOVERY SYSTEM	SIGNALING SYSTEM	WEAPONS SYSTEM	MATERIALS SYSTEM	AIRCRAFT SYSTEM	MISSION SUPPORT
1010	PILOT'S POSITION	AMPLES	PALEOLOGY	RECOVERY SYSTEM	SIGNALING SYSTEM	WEAPONS SYSTEM	MATERIALS SYSTEM	AIRCRAFT SYSTEM	MISSION SUPPORT
1011	PILOT'S POSITION	AMPLES	PALEOLOGY	RECOVERY SYSTEM	SIGNALING SYSTEM	WEAPONS SYSTEM	MATERIALS SYSTEM	AIRCRAFT SYSTEM	MISSION SUPPORT
1012	PILOT'S POSITION	AMPLES	PALEOLOGY	RECOVERY SYSTEM	SIGNALING SYSTEM	WEAPONS SYSTEM	MATERIALS SYSTEM	AIRCRAFT SYSTEM	MISSION SUPPORT
1013	PILOT'S POSITION	AMPLES	PALEOLOGY	RECOVERY SYSTEM	SIGNALING SYSTEM	WEAPONS SYSTEM	MATERIALS SYSTEM	AIRCRAFT SYSTEM	MISSION SUPPORT
1014	PILOT'S POSITION	AMPLES	PALEOLOGY	RECOVERY SYSTEM	SIGNALING SYSTEM	WEAPONS SYSTEM	MATERIALS SYSTEM	AIRCRAFT SYSTEM	MISSION SUPPORT
1015	PILOT'S POSITION	AMPLES	PALEOLOGY	RECOVERY SYSTEM	SIGNALING SYSTEM	WEAPONS SYSTEM	MATERIALS SYSTEM	AIRCRAFT SYSTEM	MISSION SUPPORT
1016	PILOT'S POSITION	AMPLES	PALEOLOGY	RECOVERY SYSTEM	SIGNALING SYSTEM	WEAPONS SYSTEM	MATERIALS SYSTEM	AIRCRAFT SYSTEM	MISSION SUPPORT
1017	PILOT'S POSITION	AMPLES	PALEOLOGY	RECOVERY SYSTEM	SIGNALING SYSTEM	WEAPONS SYSTEM	MATERIALS SYSTEM	AIRCRAFT SYSTEM	MISSION SUPPORT

ESTIMATED RELIABILITY SUMMARY (AT 50% CONFIDENCE LEVEL)

ESTIMATED RELIABILITY SUMMARY (AT 50% CONFIDENCE LEVEL)

MISSION NUMBER	PRIMARY FUNCTIONS	SECONDARY FUNCTIONS		STELLAR IMAGE CAMERA	
		IMAGE PROCESSING	IMAGE PROCESSING	SAMPLE	SAMPLE
1033	PANORAMIC CAMERA DOORS	LANDING & CONTROL SYSTEM	PAYOUT CLOCK	ON-BOARD FUNCTIONS	RECOVERY SYSTEM
1034	SAMPLE	SAMPLE	SAMPLE	SAMPLE	SAMPLE
1035	RELIABILITY	FAILURES	FAILURES	FAILURES	FAILURES
1036	RELIABILITY	FAILURES	FAILURES	FAILURES	FAILURES
1037	RELIABILITY	FAILURES	FAILURES	FAILURES	FAILURES
1038	RELIABILITY	FAILURES	FAILURES	FAILURES	FAILURES

SECTION I

SUMMARY DATA

The comparison of the operating parameters and the performance achieved by previous missions has been difficult due to the large volume of data that results from each mission. Some of the pertinent characteristics from prior missions have been summarized in Tables 15-1 through 15-3.

The summary data was started with Mission 1004 as the J-05 camera system was the first to incorporate the major modifications to the titanium drum and scan arm, four roller scan head and Corona J capabilities. Only those missions that culminated in the recovery of some photography have been listed, therefore Missions 1003, 1005, and 1032 are deleted.

MISSION SUMMARY

MISSION NUMBER	PAYLOAD NUMBER	VEHICLE NUMBER	LAUNCH DATE	LAUNCH TIME	ORBIT INCLINATION (°)	PERIGEE LATITUDE (NM)	APOGEE LATITUDE (NM)	RECOVERY LOCATION (NM)	RECOVERY PASS	CARRIER TYPE	CARRIER NUMBER	CARRIER SLAT	CARRIER SITE	STILL TIME	CARRIER NUMBER	CARRIER SLAT	CARRIER SITE
1004	2-05	1174	8/8/94	2100 E	74.9	95.9	29.0	49	112	14	0-20	0-01	120	0-00	120	0-00	120
1005	2-06	1175	8/24/94	2000 E	74.9	94.0	91.2	69	120	140	0-00	0-01	140	0-00	140	0-00	140
1007	2-07	1176	8/28/94	2010 E	80.0	99.2	91.0	69	120	144	0-00	0-02	145	0-00	145	0-00	145
1008	2-08	1177	7/10/94	2100 E	80.0	99.4	40.4	49	112	160	0-00	0-01	160	0-00	160	0-00	160
1009	2-09	1178	8/13/94	2100 E	80.0	99.4	40.4	49	112	160	0-00	0-01	160	0-00	160	0-00	160
1010	2-10	1179	8/17/94	2100 E	80.0	99.4	40.4	49	112	160	0-00	0-01	160	0-00	160	0-00	160
1011	2-11	1170	8/24/94	2000 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1012	2-12	1171	8/25/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1013	2-13	1172	8/27/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1014	2-14	1173	8/28/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1015	2-15	1174	8/29/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1016	2-16	1175	8/30/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1017	2-17	1176	8/31/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1018	2-18	1177	9/1/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1019	2-19	1178	9/2/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1020	2-20	1179	9/3/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1021	2-21	1170	9/4/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1022	2-22	1171	9/5/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1023	2-23	1172	9/6/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1024	2-24	1173	9/7/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1025	2-25	1174	9/8/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1026	2-26	1175	9/9/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1027	2-27	1176	9/10/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1028	2-28	1177	9/11/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1029	2-29	1178	9/12/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1030	2-30	1179	9/13/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1031	2-31	1170	9/14/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1032	2-32	1171	9/15/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1033	2-33	1172	9/16/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1034	2-34	1173	9/17/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1035	2-35	1174	9/18/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1036	2-36	1175	9/19/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1037	2-37	1176	9/20/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1038	2-38	1177	9/21/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1039	2-39	1178	9/22/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1040	2-40	1179	9/23/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1041	2-41	1170	9/24/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1042	2-42	1171	9/25/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1043	2-43	1172	9/26/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1044	2-44	1173	9/27/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1045	2-45	1174	9/28/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1046	2-46	1175	9/29/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1047	2-47	1176	9/30/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1048	2-48	1177	9/31/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1049	2-49	1178	10/1/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1050	2-50	1179	10/2/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1051	2-51	1170	10/3/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1052	2-52	1171	10/4/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1053	2-53	1172	10/5/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1054	2-54	1173	10/6/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1055	2-55	1174	10/7/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1056	2-56	1175	10/8/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1057	2-57	1176	10/9/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1058	2-58	1177	10/10/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1059	2-59	1178	10/11/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1060	2-60	1179	10/12/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1061	2-61	1170	10/13/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1062	2-62	1171	10/14/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1063	2-63	1172	10/15/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1064	2-64	1173	10/16/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1065	2-65	1174	10/17/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1066	2-66	1175	10/18/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1067	2-67	1176	10/19/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1068	2-68	1177	10/20/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1069	2-69	1178	10/21/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1070	2-70	1179	10/22/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1071	2-71	1170	10/23/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1072	2-72	1171	10/24/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-00	162
1073	2-73	1172	10/25/94	2100 E	80.0	99.4	42.1	69	144	162	0-00	0-01	162	0-00	162	0-0	

MISSION SUMMARY

MISSION NUMBER	PAYLOAD NUMBER	VEHICLE NUMBER	LAUNCH DATE	LAUNCH TIME	ORBIT INCLINATION [°]	ALTITUDE LOCATION [MM]	MASTER ORBITAL DATA		SLAVE NUMBER	SLAVE TYPE	SLAVE INCLINATION [°]	SLAVE ALTITUDE LOCATION [MM]
							RECOVERY PASS NUMBER	CARRIER NUMBER [°]				
1029	J-87	1023	2/1/98	2112 E	78.14	99.8	22.6	170	0.275	W-28	179	0.178
1030	J-88	1028	2/1/98	2202 E	79.0	97.5	16.7	161	0.276	W-19	163	0.179
1031	J-89	1031	2/4/98	2202 E	79.0	104.6	23.3	115	0.228	W-224	161	0.166
1032	J-90	1035	2/5/98	1901 E	—	—	—	—	0.110	W-21	161	0.180
1033	J-91	1036	2/7/98	2018 E	80.1	102.0	50.7	88	0.200	W-21	160	0.181
1034	J-91	1038	2/7/98	2131 E	80.1	105.4	16.2	91	0.300	W-224	161	0.178
1035	J-92	1039	2/9/98	2114 E	80.0	99.8	50.1	160	0.188	W-224	160	0.179
1036	J-92	1040	2/9/98	2046 E	100.0	102.4	26.0	108	0.122	W-224	160	0.180
1037	J-93	1042	2/10/98	1937 E	100.0	91.6	14.5	162	0.200	W-224	162	0.178
1038	J-94	1043	2/14/98	2008 E	80.0	94.9	29.8	162	0.178	W-224	162	0.179

PERFORMANCE SUMMARY

PERFORMANCE SUMMARY

PERFORMANCE SUMMARY

MISSION NUMBER	CAMERA	SERIAL NUMBER	R.I.P. VALUE	ARPPP MTF/AIM		90% ATTITUDE ERROR (°)		90% ALTITUDE RATES (YAW)		90% V/H ERROR (°)	
				AVERAGE SLIT (μ)	STDEV	PITCH	ROLL	YAW	PITCH	ROLL	YAW
1034-1	FWD	100	90	75	80	0.80	0.19	0.98	10.3	20.4	24.0
1034-1	APT	107	90	74	80	0.80	0.19	0.98	10.3	20.4	24.0
1034-2	FWD	100	93	80	85	0.34	0.36	0.33	21.1	29.0	16.2
1034-2	APT	107	93	80	85	0.34	0.36	0.33	21.1	29.0	16.2
1035-1	FWD	100	95	64	60	0.16	0.17	0.15	24.0	27.9	25.9
1035-1	APT	107	95	60	62	0.16	0.17	0.15	24.0	27.9	25.9
1035-2	FWD	100	95	61	61	0.16	0.17	0.15	24.0	27.9	25.9
1035-2	APT	107	95	61	62	0.16	0.17	0.15	24.0	27.9	25.9
1036-1	FWD	100	95	63	64	0.76	0.76	0.76	31.2	36.0	29.5
1036-1	APT	107	95	63	64	0.76	0.76	0.76	31.2	36.0	29.5
1036-2	FWD	100	95	73	64	0.94	0.94	0.94	31.0	36.0	29.5
1036-2	APT	107	95	73	64	0.94	0.94	0.94	31.0	36.0	29.5
1037-1	FWD	100	95	60	60	0.25	0.25	0.25	22.0	40.0	29.3
1037-1	APT	107	95	60	60	0.25	0.25	0.25	22.0	40.0	29.3
1037-2	FWD	100	95	61	61	0.24	0.24	0.24	26.0	36.0	31.4
1037-2	APT	107	95	61	61	0.24	0.24	0.24	26.0	36.0	31.4
1038-1	FWD	100	90	60	60	0.27	0.27	0.27	26.0	36.0	31.4
1038-1	APT	107	90	60	60	0.27	0.27	0.27	26.0	36.0	31.4
1038-2	FWD	100	90	60	60	0.29	0.29	0.29	26.0	36.0	31.4
1038-2	APT	107	90	60	60	0.29	0.29	0.29	26.0	36.0	31.4

EXPOSURE - PROCESSING SUMMARY

EXPOSURE-PROCESSING SUMMARY

EXPOSURE - PROCESSING SUMMARY

~~SECTION A~~

~~APPENDIX~~

-TOP SECRET C [REDACTED]

DENSITY VALUE	PRIMARY		INTERMEDIATE		SECONDARY		TERTIARY		DENSITY FREQUENCIES		
	MIN	MAX	LTH	MIN	MAX	LTH	MIN	MAX	LTH	MIN	MAX
0.01	0	0	0	0	0	0	0	0	0	0	0
0.02	0	0	0	0	0	0	0	0	0	0	0
0.03	0	0	0	0	0	0	0	0	0	0	0
0.04	0	0	0	0	0	0	0	0	0	0	0
0.05	0	0	0	0	0	0	0	0	0	0	0
0.06	0	0	0	0	0	0	0	0	0	0	0
0.07	0	0	0	0	0	0	0	0	0	0	0
0.08	0	0	0	0	0	0	0	0	0	0	0
0.09	0	0	0	0	0	0	0	0	0	0	0
0.10	0	0	0	0	0	0	0	0	0	0	0
0.11	0	0	0	0	0	0	0	0	0	0	0
0.12	0	0	0	0	0	0	0	0	0	0	0
0.13	0	0	0	0	0	0	0	0	0	0	0
0.14	0	0	0	0	0	0	0	0	0	0	0
0.15	0	0	0	0	0	0	0	0	0	0	0
0.16	0	0	0	0	0	0	0	0	0	0	0
0.17	0	0	0	0	0	0	0	0	0	0	0
0.18	0	0	0	0	0	0	0	0	0	0	0
0.19	0	0	0	0	0	0	0	0	0	0	0
0.20	0	0	0	0	0	0	0	0	0	0	0
0.21	0	0	0	0	0	0	0	0	0	0	0
0.22	0	0	0	0	0	0	0	0	0	0	0
0.23	0	0	0	0	0	0	0	0	0	0	0
0.24	0	0	0	0	0	0	0	0	0	0	0
0.25	0	0	0	0	0	0	0	0	0	0	0
0.26	0	0	0	0	0	0	0	0	0	0	0
0.27	0	0	0	0	0	0	0	0	0	0	0
0.28	0	0	0	0	0	0	0	0	0	0	0
0.29	0	0	0	0	0	0	0	0	0	0	0
0.30	0	0	0	0	0	0	0	0	0	0	0
0.31	0	0	0	0	0	0	0	0	0	0	0
0.32	0	0	0	0	0	0	0	0	0	0	0
0.33	0	0	0	0	0	0	0	0	0	0	0
0.34	0	0	0	0	0	0	0	0	0	0	0
0.35	0	0	0	0	0	0	0	0	0	0	0
0.36	0	0	0	0	0	0	0	0	0	0	0
0.37	0	0	0	0	0	0	0	0	0	0	0
0.38	0	0	0	0	0	0	0	0	0	0	0
0.39	0	0	0	0	0	0	0	0	0	0	0
0.40	0	0	0	0	0	0	0	0	0	0	0
0.41	0	0	0	0	0	0	0	0	0	0	0
0.42	0	0	0	0	0	0	0	0	0	0	0
0.43	0	0	0	0	0	0	0	0	0	0	0
0.44	0	0	0	0	0	0	0	0	0	0	0
0.45	0	0	0	0	0	0	0	0	0	0	0
0.46	0	0	0	0	0	0	0	0	0	0	0
0.47	0	0	0	0	0	0	0	0	0	0	0
0.48	0	0	0	0	0	0	0	0	0	0	0
0.49	0	0	0	0	0	0	0	0	0	0	0
0.50	0	0	0	0	0	0	0	0	0	0	0
SUBTOTAL	17	17	0	129	129	0	147	147	0	0	0

TOP SECRET C [REDACTED]

TABLE A-1

DENSITY	PRIMARY VALUE	MIN MAX LIN	INTERMEDIATE MIN MAX LIN	FINAL MIN MAX LIN	ALL LEVEL MIN MAX
0.51					
0.52					
0.53					
0.54					
0.55					
0.56					
0.57					
0.58					
0.59					
0.60					
0.61					
0.62					
0.63					
0.64					
0.65					
0.66					
0.67					
0.68					
0.69					
0.70					
0.71					
0.72					
0.73					
0.74					
0.75					
0.76					
0.77					
0.78					
0.79					
0.80					
0.81					
0.82					
0.83					
0.84					
0.85					
0.86					
0.87					
0.88					
0.89					
0.90					
0.91					
0.92					
0.93					
0.94					
0.95					
0.96					
0.97					
0.98					
0.99					
1.00					
SUBTOTAL		3	0	20	1
				0	83
				0	9
				0	0
				106	10

~~TOP SECRET C~~ [REDACTED]

TABLE A-1

MISSION 2036 COMMUNITY

DENSITY VALUE	PRIMARY MIN MAX LIN	INTERMEDIATE MIN MAX LIN	SECONDARY MIN MAX LIN
1.01	0	0	0
1.02	0	0	0
1.03	0	0	0
1.04	0	0	0
1.05	0	0	0
1.06	0	0	0
1.07	0	0	0
1.08	0	0	0
1.09	0	0	0
1.10	0	0	0
1.11	0	0	0
1.12	0	0	0
1.13	0	0	0
1.14	0	0	0
1.15	0	0	0
1.16	0	0	0
1.17	0	0	0
1.18	0	0	0
1.19	0	0	0
1.20	0	0	0
1.21	0	0	0
1.22	0	0	0
1.23	0	0	0
1.24	0	0	0
1.25	0	0	0
1.26	0	0	0
1.27	0	0	0
1.28	0	0	0
1.29	0	0	0
1.30	0	0	0
1.31	0	0	0
1.32	0	0	0
1.33	0	0	0
1.34	0	0	0
1.35	0	0	0
1.36	0	0	0
1.37	0	0	0
1.38	0	0	0
1.39	0	0	0
1.40	0	0	0
1.41	0	0	0
1.42	0	0	0
1.43	0	0	0
1.44	0	0	0
1.45	0	0	0
1.46	0	0	0
1.47	0	0	0
1.48	0	0	0
1.49	0	0	0
1.50	0	0	0
SUBTOTAL			

~~TOP SECRET~~

Bible A.J.

~~TOP SECRET C~~

TABLE A-1

MISSION #10 - **THE ENDLESS JUNGLE**

DENSITY PRIMARY INTERMEDIATE
VALUE MIN MAX LIM MIN MAX

2.01
2.02
2.03
2.04
2.05
2.06
2.07
2.08
2.09
2.10
2.11
2.12
2.13
2.14
2.15
2.16
2.17
2.18
2.19
2.20
2.21
2.22
2.23
2.24
2.25
2.26
2.27
2.28
2.29
2.30
2.31
2.32
2.33
2.34
2.35
2.36
2.37
2.38
2.39
2.40
2.41
2.42
2.43
2.44
2.45
2.46
2.47
2.48
2.49
2.50
JBTOTAL

SUBTOTAL

卷之三

卷之三

卷之三

~~TOP SECRET~~ C

PAGE A-1

MISSOURI STATE HISTORICAL SOCIETY

DENSITY VALUE	PRIMARY			INTERMEDIATE			SECONDARY			TERTIARY		
	MIN	MAX	LIMIT	MIN	MAX	LIMIT	MIN	MAX	LIMIT	MIN	MAX	LIMIT
2.51	0	0	0	0	0	0	0	0	0	0	0	0
2.52	0	0	0	0	0	0	0	0	0	0	0	0
2.53	0	0	0	0	0	0	0	0	0	0	0	0
2.54	0	0	0	0	0	0	0	0	0	0	0	0
2.55	0	0	0	0	0	0	0	0	0	0	0	0
2.56	0	0	0	0	0	0	0	0	0	0	0	0
2.57	0	0	0	0	0	0	0	0	0	0	0	0
2.58	0	0	0	0	0	0	0	0	0	0	0	0
2.59	0	0	0	0	0	0	0	0	0	0	0	0
2.60	0	0	0	0	0	0	0	0	0	0	0	0
2.61	0	0	0	0	0	0	0	0	0	0	0	0
2.62	0	0	0	0	0	0	0	0	0	0	0	0
2.63	0	0	0	0	0	0	0	0	0	0	0	0
2.64	0	0	0	0	0	0	0	0	0	0	0	0
2.65	0	0	0	0	0	0	0	0	0	0	0	0
2.66	0	0	0	0	0	0	0	0	0	0	0	0
2.67	0	0	0	0	0	0	0	0	0	0	0	0
2.68	0	0	0	0	0	0	0	0	0	0	0	0
2.69	0	0	0	0	0	0	0	0	0	0	0	0
2.70	0	0	0	0	0	0	0	0	0	0	0	0
SUBTOTAL	0	0	0	0	0	0	0	0	0	0	0	0
TOTAL	4	4	1	38	38	34	216	214	148	256	256	19

-POP-SCIENCE- C

WINTER 11

1

TOP SECRET C

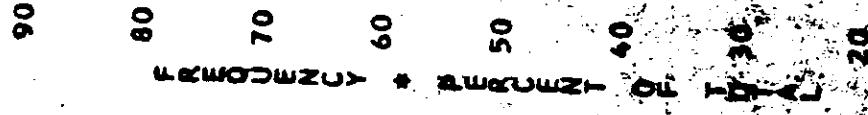
MISSION * 1038-1 * INSTR * FWD * 3/6/67 PLOT OF D MIN * TERRAIN * PROCESSING * PRIMARY
AIRTH MEAN * 0.51 * MEDIAN * 0.63 * STD DEV * 0.23 * RANGE * 0.16 TO 0.64 WITH 4 SAMPLES



MAXIMUM > * MAXIMUM ON HOME

~~SECRET C~~

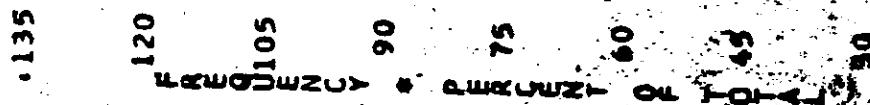
MISSION * 1038-1 * INSTR * FWD * 3/6/67 PLOT OF D MAX * TERRAIN * PROCESSING * PRIMARY
AIRTH MEAN * 1.42 * MEDIAN * 1.31 * STD DEV * 0.29 * RANGE * 1.21 TO 1.84 WITH 4 SAMPLES



AB

TOP SECRET C

MISSION * 1038-1 * INSTR * FWD * 3/6/67 PLOT OF D MAX * CLOUD * PROCESSING * PRIMARY
AIRTH MEAN * 1.50 * MEDIAN * 1.50 * STD DEV * 0.00 * RANGE * 1.50 TO 1.50 WITH 1 SAMPLES



FWD 1038-1 * PLOT OF D MAX * CLOUD * PROCESSING * PRIMARY

~~SECRET~~ C

MISSION * 1038-1 * INSTR * FWD * 3/6/67 PLOT OF 0 MIN * TERRAIN * PROCESSING * INTERMEDIATE
AIRTH MEAN * 0.59 * MEDIAN * 0.55 * STD DEV * 0.21 * RANGE * 0.26 TO 1.11 WITH 39 SAMPLES

20

18

16

14

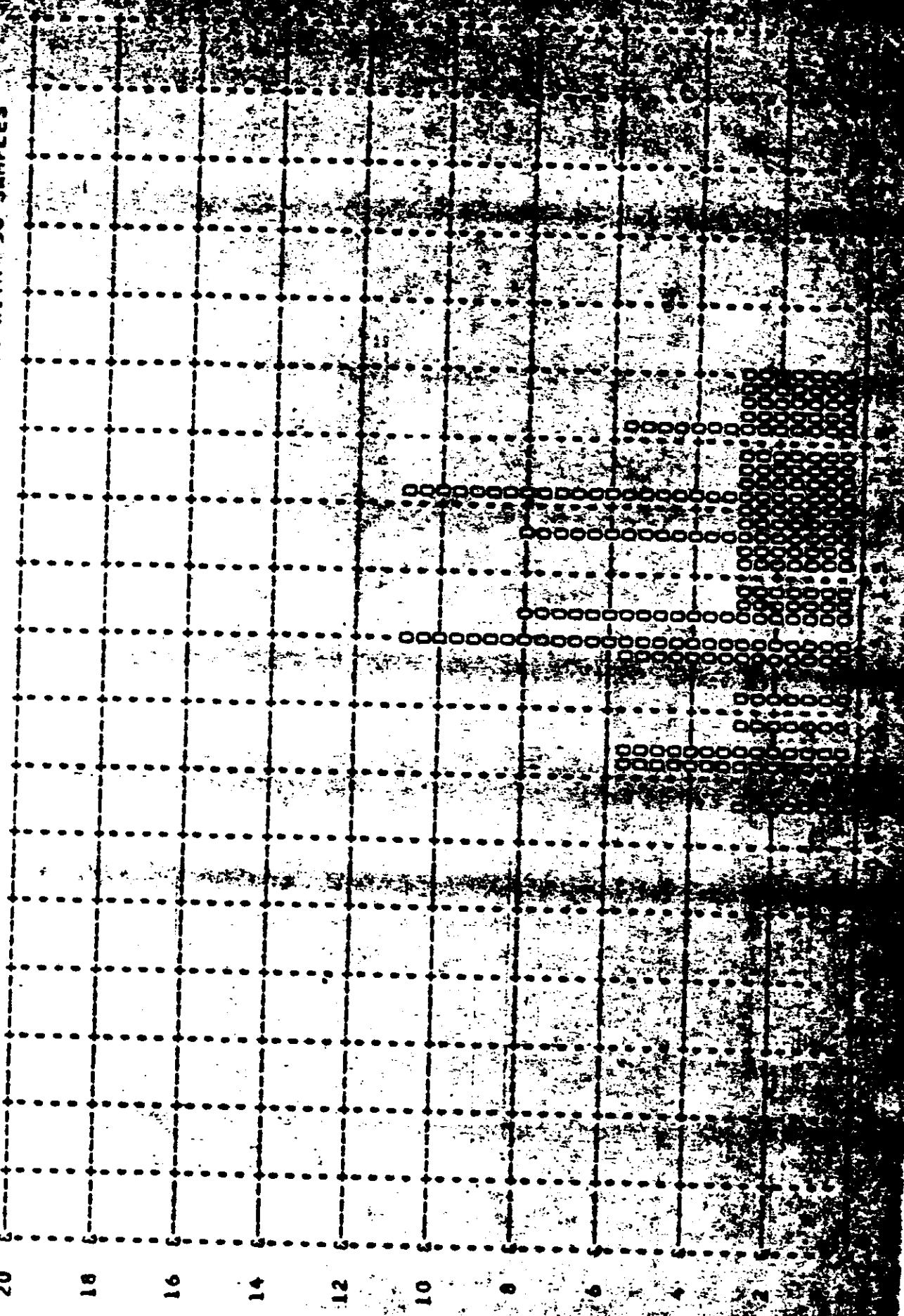
12

10

FREQUENCY UP * FREQUENCY DOWN

TOP SECRET C

MISSION * 1038-1 * INSTR * FWD * 3/6/67 PLOT OF D MAX * TERRAIN * PROCESSING * INTERMEDIATE
AIRTH MEAN * 1.50 * MEDIAN * 1.54 * STD DEV * 0.26 * RANGE * 0.99 TO 1.94 WITH 38 SAMPLES



FREQ'DIST > * 10 12 14 16 18 20

TOP SECRET C

MISSION * 1038-1 * INSTR * FWD * 3/6/67 PLOT OF D MAX * CLOUD * PROCESSING * INTERMEDIA
EARTH MEAN * 2.13 * MEDIAN * 2.20 * STD DEV * 0.29 * RANGE * 1.25 TO 2.46 WITH 34 SAMPLES

20

18

16

14

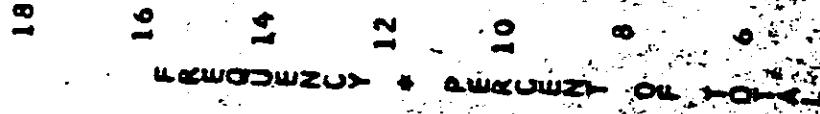
12

10

MAXIMUM * 2.46 * MINIMUM * 1.25

TOP SECRET C

MISSION * 1038-1 * INSTR * FWD * 3/6/67 PLOT OF D MIN * TERRAIN * PROCESSING * FULL
EARTH MEAN * 0.50 * MEDIAN * 0.45 * STD DEV * 0.17 * RANGE * 0.26 TO 1.09 WITH 214 SAMPLES



TOP SECRET C

MISSION * 1038-1 * INSTR * FWD * 3/6/67 PLOT OF D MAX * TERRAIN * PROCESSING * FULL
EARTH MEAN * 1.57 * MEDIAN * 1.52 * STD DEV * 0.37 * RANGE * 0.84 TO 2.36 WITH 214 SAMPLES



TOP SECRET C

-TOP SECRET C

MISSION * 1038-1 * INSTR * FHO * 3/6/67 PLOT OF D MAX * CLOUD * PROCESSING * FULL
AIRTH MEAN * 2.11 * MEDIAN * 2.19 * STD DEV * 0.28 * RANGE * 1.18 TO 2.45 WITH 149 SAMPLES

10

9

8 7 6 5 4 3 2 1

MAXIMUM & MINIMUM ON PAGE

TOP SECRET C

MISSION * 1038-1 * INSTR * FWD * 3/6/67 PLOT OF D MIN * TERRAIN * PROCESSING * ALL LEVELS
AIRTH MEAN * 0.51 * MEDIAN * 0.46 * STD DEV * 0.18 * RANGE * 0.16 TO 1.11 WITH 256 SAMPLES

16 14 12 10

LEADERSHIP * LEADERSHIP ON HOME

MISSION * 1038-1 * INSTR * FWD * 3/6/67 PLOT OF D MAX * TERRAIN * PROCESSING * ALL LEVELS
EARTH MEAN * 1.56 * MEDIAN * 1.52 * STD DEV * 0.36 * RANGE * 0.84 TO 2.36 WITH 256 SAMPLES

TOP SECRET C

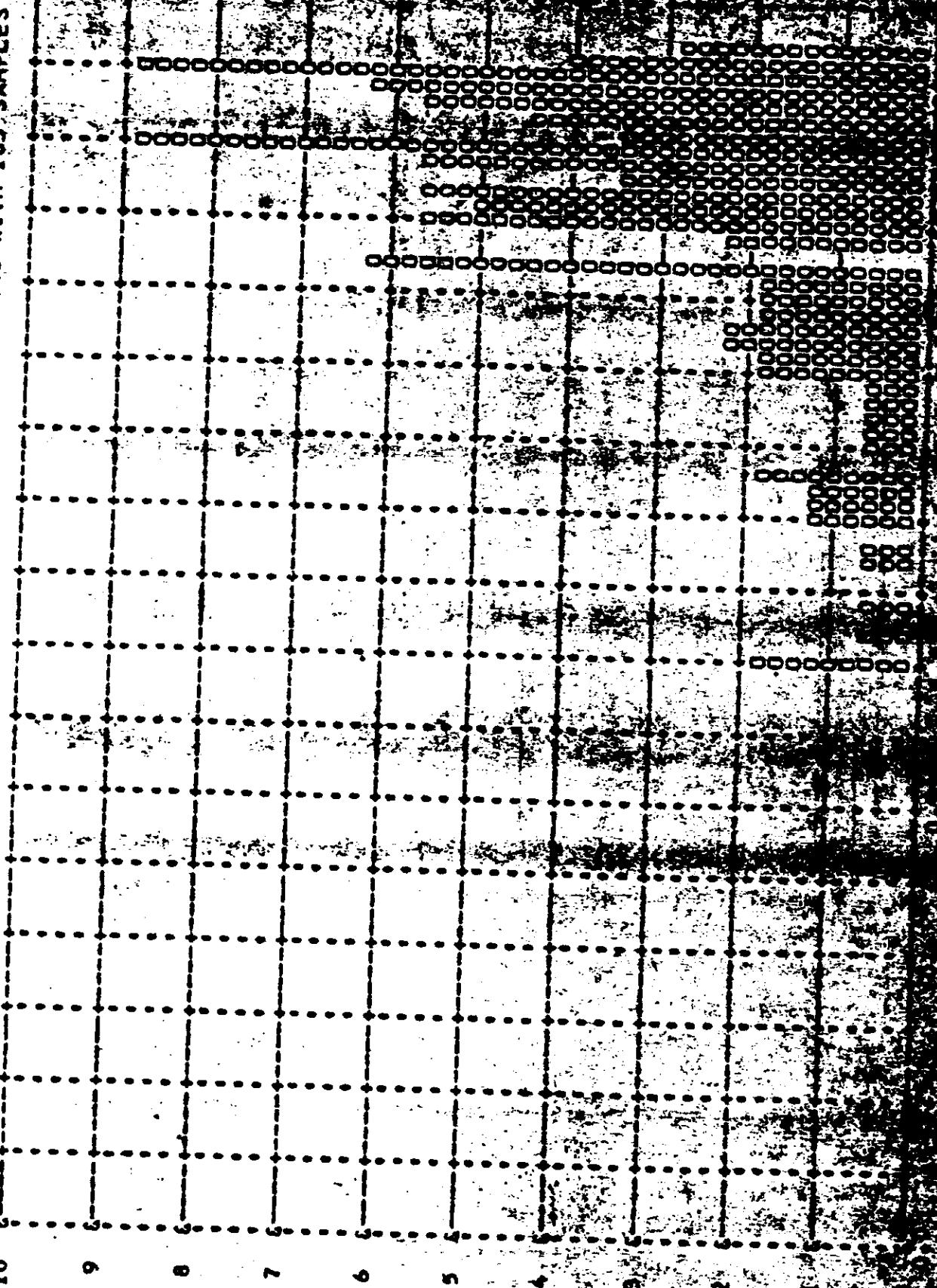
TOP SECRET C

9 8 7 6 5

4.2 4.0 3.8 3.6 3.4 3.2 3.0 2.8 2.6 2.4 2.2 2.0 1.8 1.6 1.4 1.2 1.0 0.8 0.6 0.4 0.2 0.0

TOP SECRET C

MISSION * 1038-1 * INSTR * FWD * 3/6/67 PLOT OF D MAX * CLOUD * PROCESSING * ALL LEVELS
AIRTH MEAN * 2.11 * MEDIAN * 2.19 * STD DEV * 0.28 * RANGE * 1.18 TO 2.46 WITH 183 SAMPLES



LEWISBURG * PAWUWZI - OM - 101 <

MISSION # 1038 INSTRUMENTATION

DENSITY VALUE	PRIMARY		INTERMEDIATE		SECONDARY		THERMOCOUPLE		DENS VAL		NEUTRON FLUX DENS	
	MIN	MAX	MIN	MAX	MIN	MAX	MIN	MAX	MIN	MAX	MIN	MAX
0.01												
0.02												
0.03												
0.04												
0.05												
0.06												
0.07												
0.08												
0.09												
0.10												
0.11												
0.12												
0.13												
0.14												
0.15												
0.16												
0.17												
0.18												
0.19												
0.20												
0.21												
0.22												
0.23												
0.24												
0.25												
0.26												
0.27												
0.28												
0.29												
0.30												
0.31												
0.32												
0.33												
0.34												
0.35												
0.36												
0.37												
0.38												
0.39												
0.40												
0.41												
0.42												
0.43												
0.44												
0.45												
0.46												
0.47												
0.48												
0.49												
0.50												
SUBTOTAL												

~~TOP SECRET C~~ [REDACTED]

TABLE A-2

TABLE A-2

~~TOP SECRET C~~

MISSION * 1038-1 * INSTRUMENT * AFT

3/6/67

DENSITY

DENSITY VALUE	PRIMARY		INTERMEDIATE		FULL		ALL LEVELS	
	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX
1.01	0	0	0	0	0	0	0	0
1.02	0	0	0	0	0	0	0	0
1.03	0	0	0	0	0	0	0	0
1.04	0	0	0	0	0	0	0	0
1.05	0	0	0	0	0	0	0	0
1.06	0	0	0	0	0	0	0	0
1.07	0	0	0	0	0	0	0	0
1.08	0	0	0	0	0	0	0	0
1.09	0	0	0	0	0	0	0	0
1.10	0	0	0	0	0	0	0	0
1.11	0	0	0	0	0	0	0	0
1.12	0	0	0	0	0	0	0	0
1.13	0	0	0	0	0	0	0	0
1.14	0	0	0	0	0	0	0	0
1.15	0	0	0	0	0	0	0	0
1.16	0	0	0	0	0	0	0	0
1.17	0	0	0	0	0	0	0	0
1.18	0	0	0	0	0	0	0	0
1.19	0	0	0	0	0	0	0	0
1.20	0	0	0	0	0	0	0	0
1.21	0	0	0	0	0	0	0	0
1.22	0	0	0	0	0	0	0	0
1.23	0	0	0	0	0	0	0	0
1.24	0	0	0	0	0	0	0	0
1.25	0	0	0	0	0	0	0	0
1.26	0	0	0	0	0	0	0	0
1.27	0	0	0	0	0	0	0	0
1.28	0	0	0	0	0	0	0	0
1.29	0	0	0	0	0	0	0	0
1.30	0	0	0	0	0	0	0	0
1.31	0	0	0	0	0	0	0	0
1.32	0	0	0	0	0	0	0	0
1.33	0	0	0	0	0	0	0	0
1.34	0	0	0	0	0	0	0	0
1.35	0	0	0	0	0	0	0	0
1.36	0	0	0	0	0	0	0	0
1.37	0	0	0	0	0	0	0	0
1.38	0	0	0	0	0	0	0	0
1.39	0	0	0	0	0	0	0	0
1.40	0	0	0	0	0	0	0	0
1.41	0	0	0	0	0	0	0	0
1.42	0	0	0	0	0	0	0	0
1.43	0	0	0	0	0	0	0	0
1.44	0	0	0	0	0	0	0	0
1.45	0	0	0	0	0	0	0	0
1.46	0	0	0	0	0	0	0	0
1.47	0	0	0	0	0	0	0	0
1.48	0	0	0	0	0	0	0	0
1.49	0	0	0	0	0	0	0	0
1.50	0	0	0	0	0	0	0	0
SUBTOTAL	0	0	0	0	0	0	0	0

~~TOP SECRET C~~

TABLE A-2

MISSION P-1036-1 INSTRUMENTATION

DENSITY VALUE	PRIMARY			INTERMEDIATE			SECONDARY		
	MIN	MAX	LIN	MIN	MAX	LIN	MIN	MAX	LIN
1.51	0	0	0	0	0	0	0	0	0
1.52	0	0	0	0	0	0	0	0	0
1.53	0	0	0	0	0	0	0	0	0
1.54	0	0	0	0	0	0	0	0	0
1.55	0	0	0	0	0	0	0	0	0
1.56	0	0	0	0	0	0	0	0	0
1.57	0	0	0	0	0	0	0	0	0
1.58	0	0	0	0	0	0	0	0	0
1.59	0	0	0	0	0	0	0	0	0
1.60	0	0	0	0	0	0	0	0	0
1.61	0	0	0	0	0	0	0	0	0
1.62	0	0	0	0	0	0	0	0	0
1.63	0	0	0	0	0	0	0	0	0
1.64	0	0	0	0	0	0	0	0	0
1.65	0	0	0	0	0	0	0	0	0
1.66	0	0	0	0	0	0	0	0	0
1.67	0	0	0	0	0	0	0	0	0
1.68	0	0	0	0	0	0	0	0	0
1.69	0	0	0	0	0	0	0	0	0
1.70	0	0	0	0	0	0	0	0	0
1.71	0	0	0	0	0	0	0	0	0
1.72	0	0	0	0	0	0	0	0	0
1.73	0	0	0	0	0	0	0	0	0
1.74	0	0	0	0	0	0	0	0	0
1.75	0	0	0	0	0	0	0	0	0
1.76	0	0	0	0	0	0	0	0	0
1.77	0	0	0	0	0	0	0	0	0
1.78	0	0	0	0	0	0	0	0	0
1.79	0	0	0	0	0	0	0	0	0
1.80	0	0	0	0	0	0	0	0	0
1.81	0	0	0	0	0	0	0	0	0
1.82	0	0	0	0	0	0	0	0	0
1.83	0	0	0	0	0	0	0	0	0
1.84	0	0	0	0	0	0	0	0	0
1.85	0	0	0	0	0	0	0	0	0
1.86	0	0	0	0	0	0	0	0	0
1.87	0	0	0	0	0	0	0	0	0
1.88	0	0	0	0	0	0	0	0	0
1.89	0	0	0	0	0	0	0	0	0
1.90	0	0	0	0	0	0	0	0	0
1.91	0	0	0	0	0	0	0	0	0
1.92	0	0	0	0	0	0	0	0	0
1.93	0	0	0	0	0	0	0	0	0
1.94	0	0	0	0	0	0	0	0	0
1.95	0	0	0	0	0	0	0	0	0
1.96	0	0	0	0	0	0	0	0	0
1.97	0	0	0	0	0	0	0	0	0
1.98	0	0	0	0	0	0	0	0	0
1.99	0	0	0	0	0	0	0	0	0
2.00	0	0	0	0	0	0	0	0	0
SUBTOTAL							19	16	82
									3

~~TOP SECRET C~~ [REDACTED]

TABLE A-2

-TOP SECRET C

-DRAFT A-2

八四

~~TOP SECRET~~

MISSION * 1038-1 * INSTR * AFT * 3/6/67 PLOT OF D MIN * TERRAIN * PROCESSING * INTERMED AT
AIRTH MEAN * 0.62 * MEDIAN * 0.55 * STD DEV * 0.26 * RANGE * 0.33 TO 1.37 WITH 34 SAMPLES

20

18 16 14 12 10

FRMADWZU * 0.12500000000000001 ON FORM

TOP SECRET

MISSION * 1038-1 * INSTR * AFT * 3/6/67 PLOT OF D MAX * TERRAIN * PROCESSING * INTERMED
AIRTH MEAN * 1.51 * MEDIAN * 1.55 * STD DEV * 0.28 * RANGE * 0.90 TO 2.00 WITH 34 SAMPLES

20

18

16

14

12

10

8

6

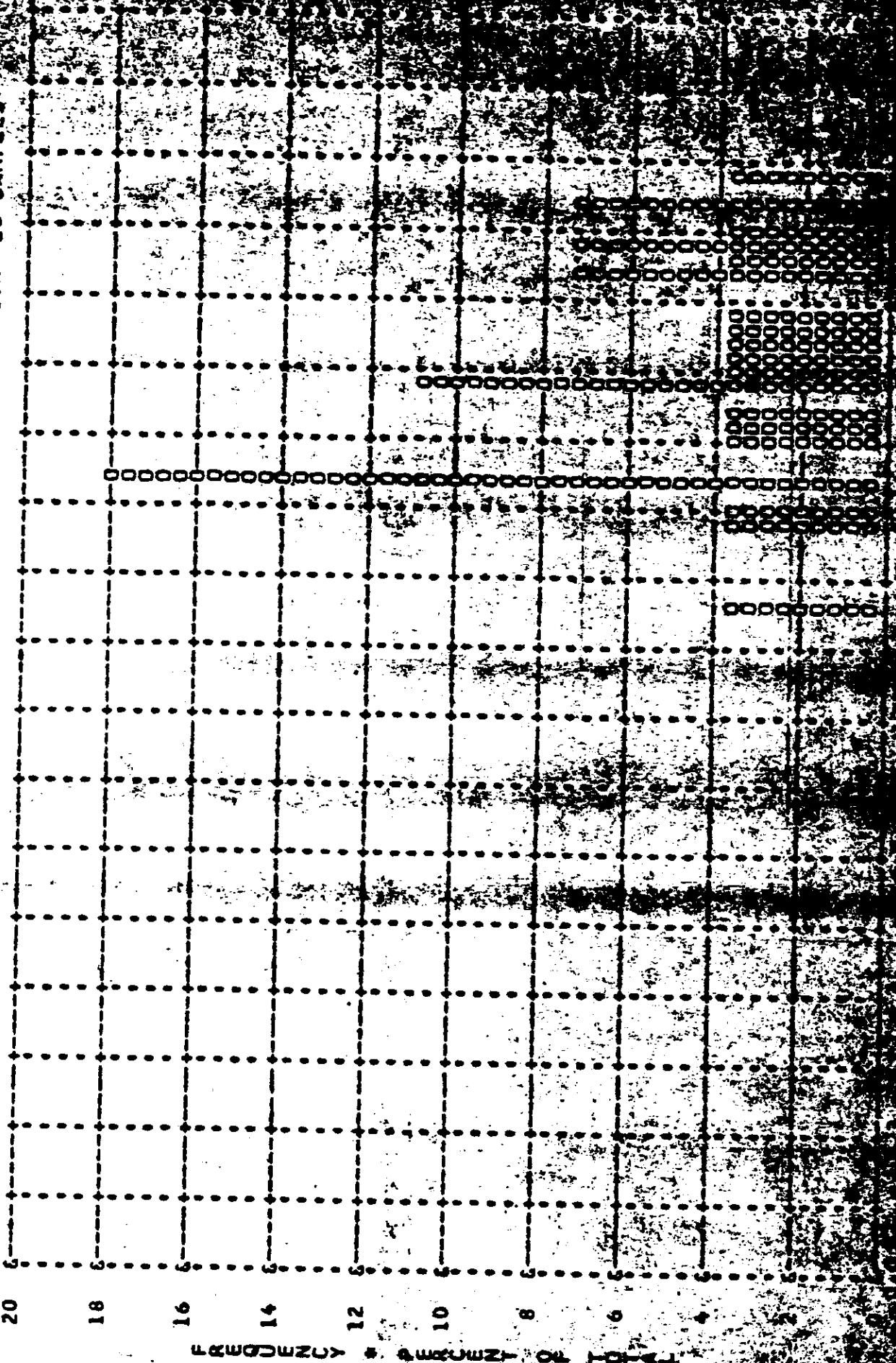
4

2

PERIODICITY * SAWTOOTH ON NOISE

TOP SECRET

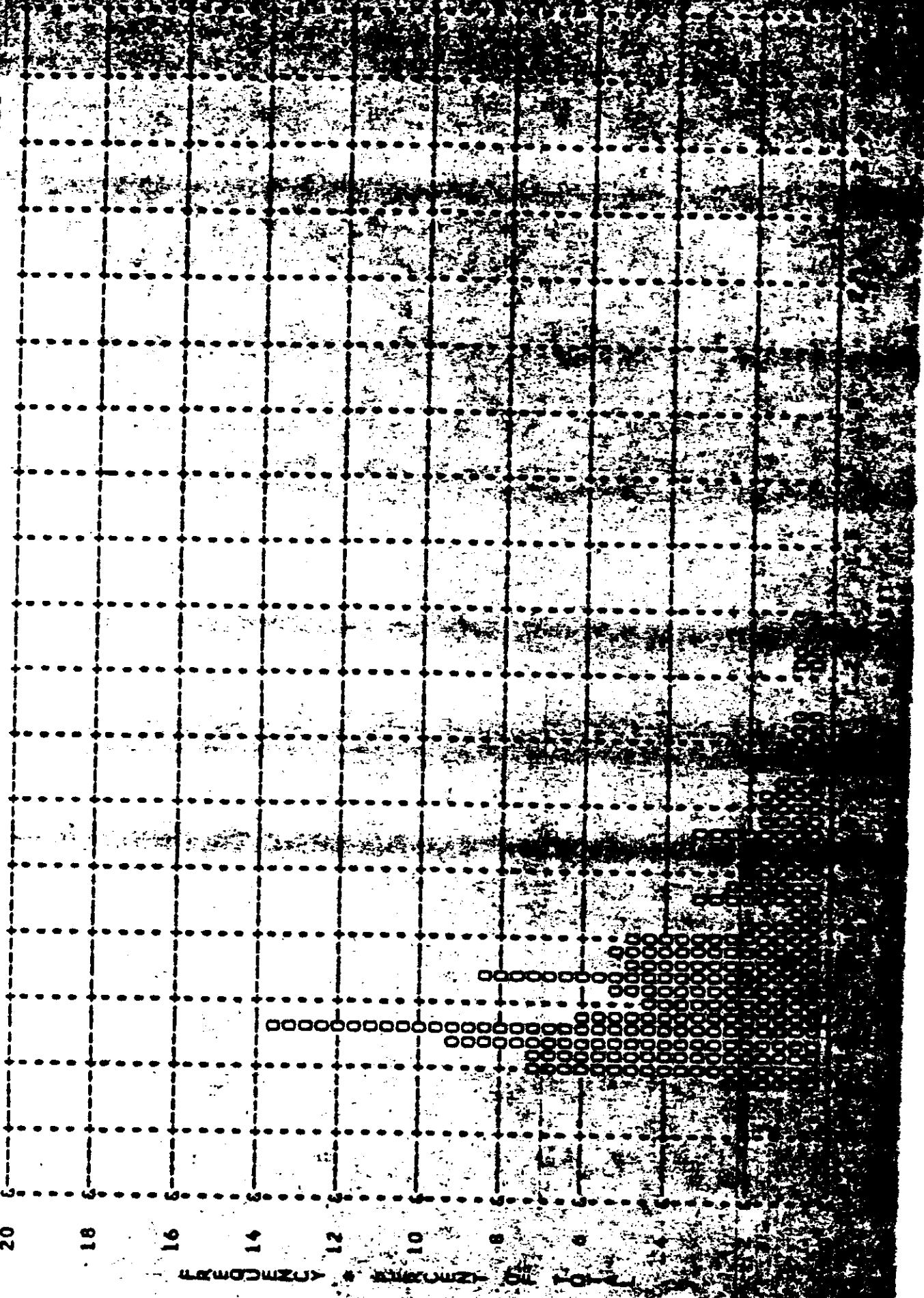
MISSION * 1038-1 * INSTR * AFT * 3/6/67 PLOT OF D MAX * CLOUD * PROCESSING * INTERME
AIRT MEAN * 1.94 * MEDIAN * 1.94 * STD DEV * 0.25 * RANGE * 1.43 TO 2.37 WITH 28 SAMPLES



FORWARDING & QUOTATION

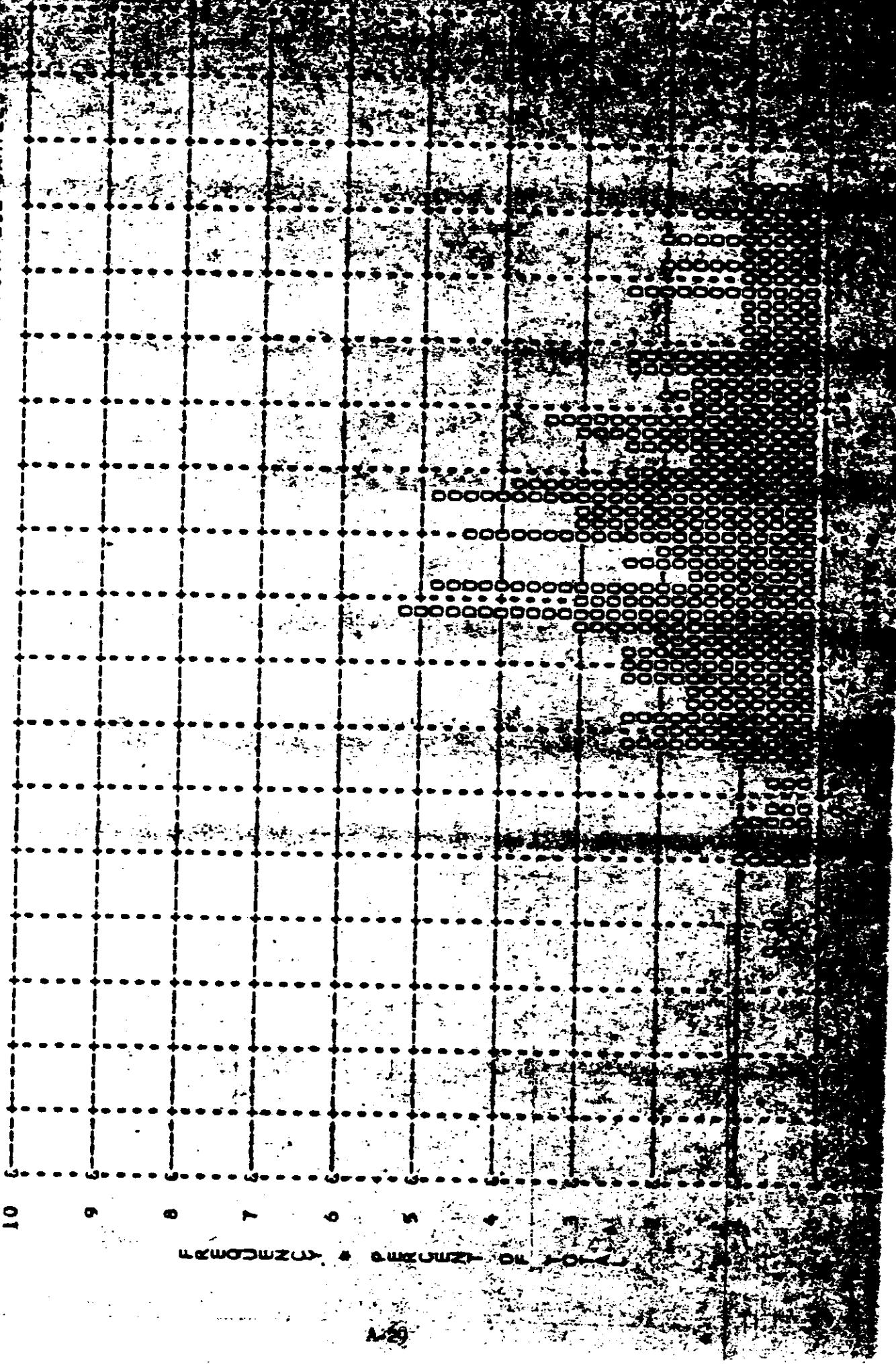
TOP SECRET

MISSION * 1038-1 * INSTR * AFT * 3/6/67 PLOT OF D MIN * TERRAIN * PROCESSING * FULL
AIRTH MEAN * 0.51 * MEDIAN * 0.46 * STD DEV * 0.20 * RANGE * 0.25 TO 1.24 WITH 212 SAMPLES



TOP SECRET

MISSION * 1038-1 * INSTR * AFT * 3/6/67 PLOT OF D MAX * TERRAIN * PROCESSING * FULL
AIRTH MEAN * 1.53 * MEDIAN * 1.51 * STD DEV * 0.37 * RANGE * 0.52 10 2.31 WITH 212 SAMPLES



FOURTH

MISSION # 1038-1 * INSTR * AFT * 3/6/67 PLOT OF D MAX * CLOUD * PROCESSING * FULL
AIRTH MEAN * 2.07 * MEDIAN * 2.18 * STD DEV * 0.30 * RANGE * 1.13 TO 2.43 WITH 146 SAMPLES



TOP SECRET

MISSION * 1038-1 * INSTR * AFT * 3/6/67 PLOT OF 0 MIN * TERRAIN * PROCESSING * ALL LEVELS
AIRTH MEAN * 0.53 * MEDIAN * 0.47 * STD DEV * 0.21 * RANGE * 0.25 TO 1.37 WITH 246 SAMPLES

20

18

16

14

12

10

8

6

0.00000000 * 0.00000000 * 0.00000000

~~TOP SECRET~~

MISSION # 1038-1 * INSTR * AFT * 3/6/67 PLOT OF D MAX * TERRAIN * PROCESSING * ALL LEVELS
AIRTH MEAN # 1.53 * MEDIAN # 1.52 * STD DEV # 0.36 * RANGE # 0.92 TO 2.31 WITH 246 SAMPLES



U.S. GOVERNMENT PROPERTY * EQUITY OWNERSHIP BY THE U.S. GOVERNMENT

TOP SECRET

MISSION * 1038-1 * INSTR * AFT * 3/6/67 PLOT OF D MAX * CLOUD * PROCESSING
AIRTH MEAN * 2.05 * MEDIAN * 2.16 * STD DEV * 0.29 * RANGE * 1.13 TO 2.43 WITH 174 SAMPLES

10

9

E-33

~~TOP SECRET C~~

TABLE A-3

~~TOP SECRET C~~

MISSION #1038-2 INSTRUMENT OF FAILURE

DENSITY VALUE	PRIMARY MIN MAX LTH	INTERMEDIATE MIN MAX LTH	FINAL MIN MAX LTH	ALL MIN MAX LTH
0.51	0	0	0	0
0.52	0	0	0	0
0.53	0	0	0	0
0.54	0	0	0	0
0.55	0	0	0	0
0.56	0	0	0	0
0.57	0	0	0	0
0.58	0	0	0	0
0.59	0	0	0	0
0.60	0	0	0	0
0.61	0	0	0	0
0.62	0	0	0	0
0.63	0	0	0	0
0.64	0	0	0	0
0.65	0	0	0	0
0.66	0	0	0	0
0.67	0	0	0	0
0.68	0	0	0	0
0.69	0	0	0	0
0.70	0	0	0	0
0.71	0	0	0	0
0.72	0	0	0	0
0.73	0	0	0	0
0.74	0	0	0	0
0.75	0	0	0	0
0.76	0	0	0	0
0.77	0	0	0	0
0.78	0	0	0	0
0.79	0	0	0	0
0.80	0	0	0	0
0.81	0	0	0	0
0.82	0	0	0	0
0.83	0	0	0	0
0.84	0	0	0	0
0.85	0	0	0	0
0.86	0	0	0	0
0.87	0	0	0	0
0.88	0	0	0	0
0.89	0	0	0	0
0.90	0	0	0	0
0.91	0	0	0	0
0.92	0	0	0	0
0.93	0	0	0	0
0.94	0	0	0	0
0.95	0	0	0	0
0.96	0	0	0	0
0.97	0	0	0	0
0.98	0	0	0	0
0.99	0	0	0	0
1.00	0	0	0	0
SUBTOTAL	0	0	25	95
			2	12
			0	120
			0	14

~~TOP SECRET C~~

TABLE A-3

MISSION # 1036-27 INSTRUMENT # 1036-27 DENSITY

DENSITY VALUE	PRIMARY MIN MAX LTH	INTERMEDIATE MIN MAX LTH	SECONDARY MIN MAX LTH	TOTAL MIN MAX LTH
1.52	0	0	0	0
1.53	0	0	0	0
1.54	0	0	0	0
1.55	0	0	0	0
1.56	0	0	0	0
1.57	0	0	0	0
1.58	0	0	0	0
1.59	0	0	0	0
1.60	0	0	0	0
1.61	0	0	0	0
1.62	0	0	0	0
1.63	0	0	0	0
1.64	0	0	0	0
1.65	0	0	0	0
1.66	0	0	0	0
1.67	0	0	0	0
1.68	0	0	0	0
1.69	0	0	0	0
1.70	0	0	0	0
1.71	0	0	0	0
1.72	0	0	0	0
1.73	0	0	0	0
1.74	0	0	0	0
1.75	0	0	0	0
1.76	0	0	0	0
1.77	0	0	0	0
1.78	0	0	0	0
1.79	0	0	0	0
1.80	0	0	0	0
1.81	0	0	0	0
1.82	0	0	0	0
1.83	0	0	0	0
1.84	0	0	0	0
1.85	0	0	0	0
1.86	0	0	0	0
1.87	0	0	0	0
1.88	0	0	0	0
1.89	0	0	0	0
1.90	0	0	0	0
1.91	0	0	0	0
1.92	0	0	0	0
1.93	0	0	0	0
1.94	0	0	0	0
1.95	0	0	0	0
1.96	0	0	0	0
1.97	0	0	0	0
1.98	0	0	0	0
1.99	0	0	0	0
2.00	0	0	0	0
SUBTOTAL	0	0	0	0
		18	11	90
				35
				100

~~TOP SECRET C~~

TABLE A-3

MISSION # 1038-2 PRE-INSTRUMENT

DENSITY VALUE	PRIMARY		INTERMEDIATE		SECONDARY		ALL LEVELS	
	MIN	MAX	MIN	MAX	MIN	MAX	MIN	MAX
2.001	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.002	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.003	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.004	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.005	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.006	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.007	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.008	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.009	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.010	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.011	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.012	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.013	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.014	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.015	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.016	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.017	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.018	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.019	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.020	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.021	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.022	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.023	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.024	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.025	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.026	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.027	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.028	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.029	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.030	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.031	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.032	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.033	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.034	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.035	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.036	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.037	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.038	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.039	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.040	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.041	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.042	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.043	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.044	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.045	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.046	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.047	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.048	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.049	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
2.050	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
SUBTOTAL	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00

~~TOP SECRET C~~

TABLE A-3

MISSION 1038-2

DENSITY VALUE	MIN	MAX	LINE MIN	LINE MAX	LINE AVERAGE
2.51	0	0	0	0	0
2.52	0	0	0	0	0
2.53	0	0	0	0	0
2.54	0	0	0	0	0
2.55	0	0	0	0	0
2.56	0	0	0	0	0
2.57	0	0	0	0	0
2.58	0	0	0	0	0
2.59	0	0	0	0	0
2.60	0	0	0	0	0
2.61	0	0	0	0	0
2.62	0	0	0	0	0
2.63	0	0	0	0	0
2.64	0	0	0	0	0
2.65	0	0	0	0	0
2.66	0	0	0	0	0
2.67	0	0	0	0	0
2.68	0	0	0	0	0
2.69	0	0	0	0	0
2.70	0	0	0	0	0
SUBTOTAL	0	0	0	0	0
TOTAL	0	0	35	35	25
	218	238	150	273	273

-TOP SECRET

TABLE A-3

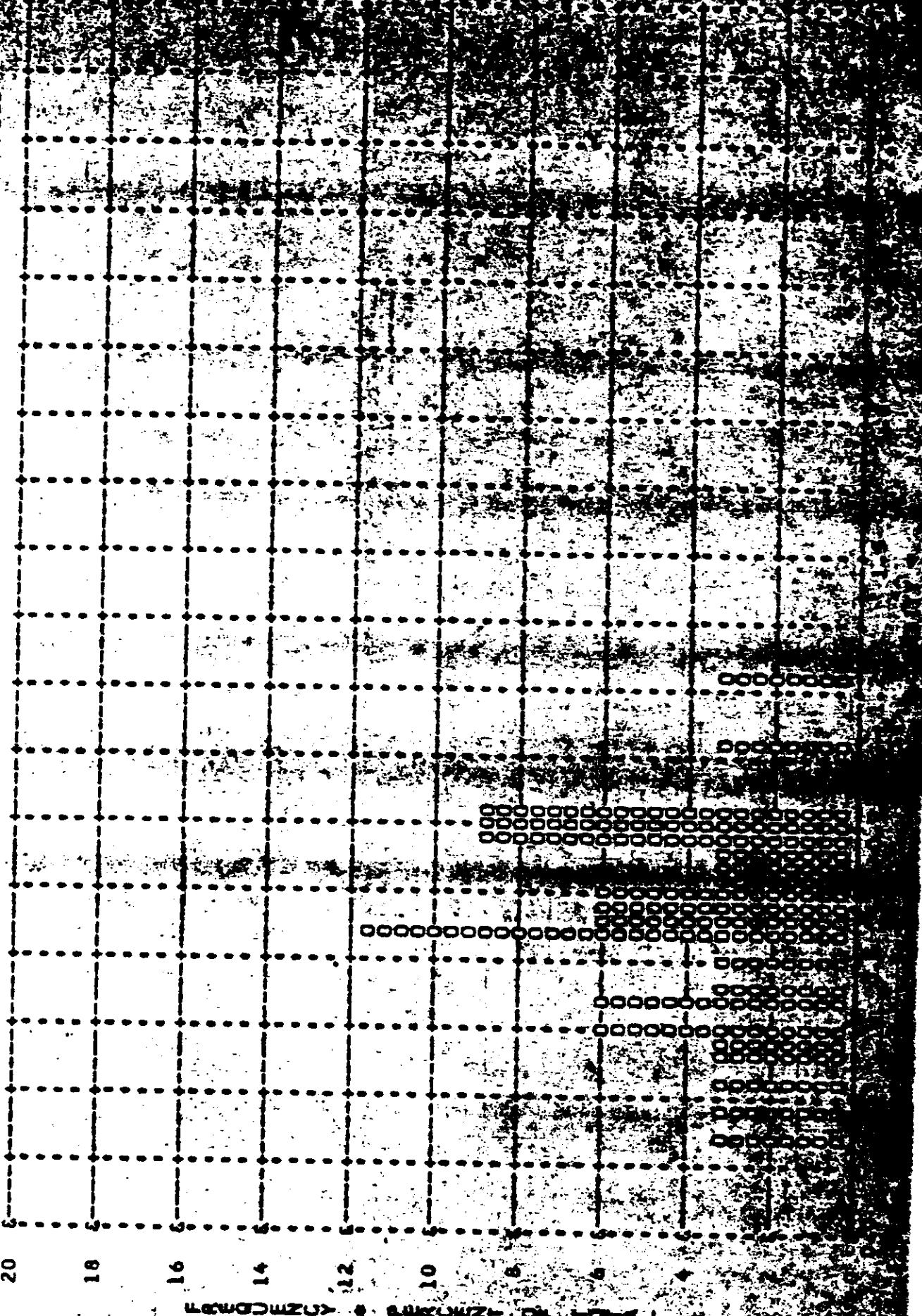
LINE B

LINE C

A-39

TOP SECRET C

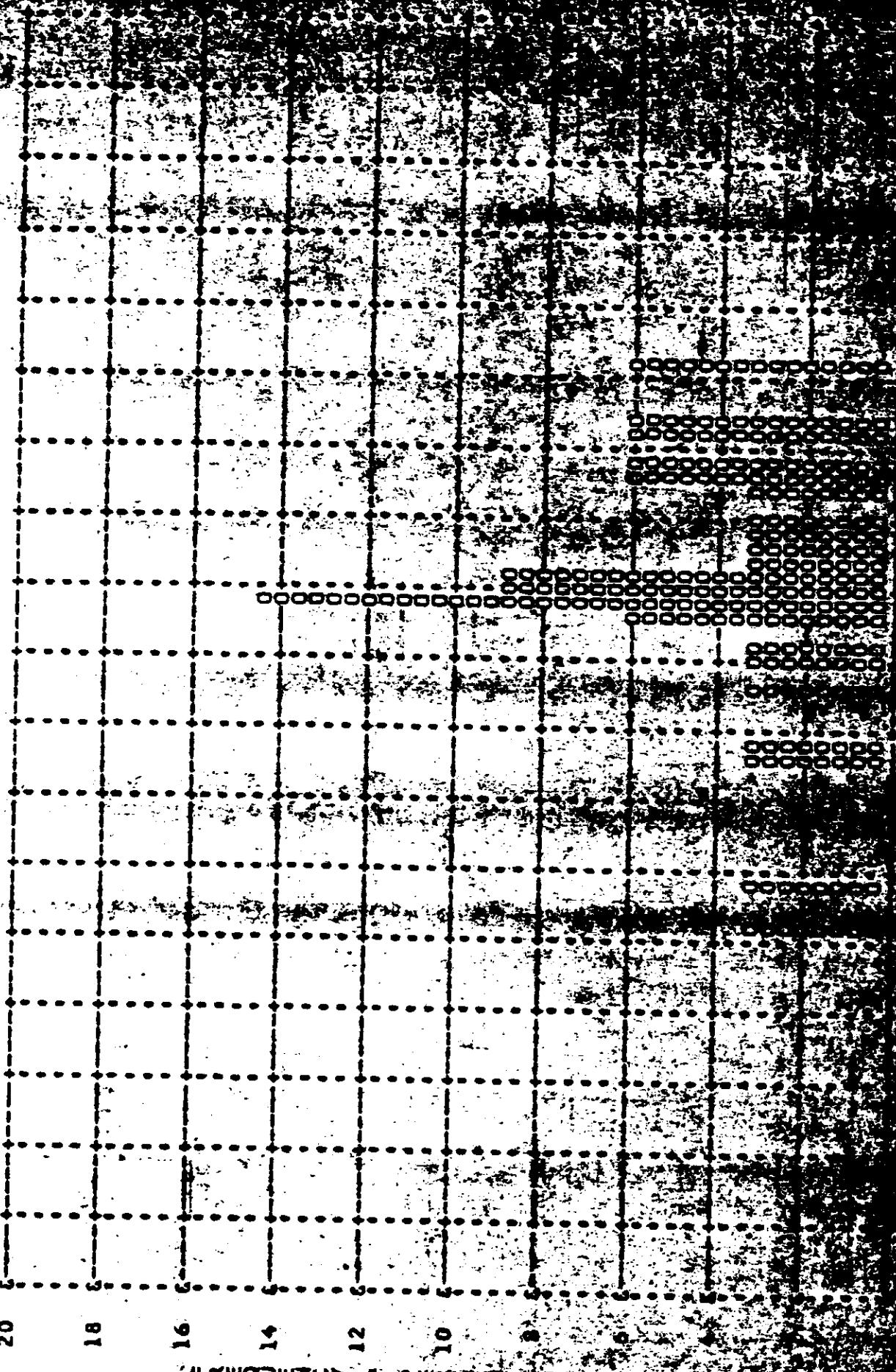
MISSION * 1038-2 * INSTR * FWD * 3/6/67 PLOT OF 0 MIN * TERRAIN * PROCESSING * INTERFERED
AIRTH MEAN * 0.69 * MEDIAN * 0.71 * STD DEV * 0.23 * RANGE * 0.21 TO 1.22 WITH 35 SAMPLES



440

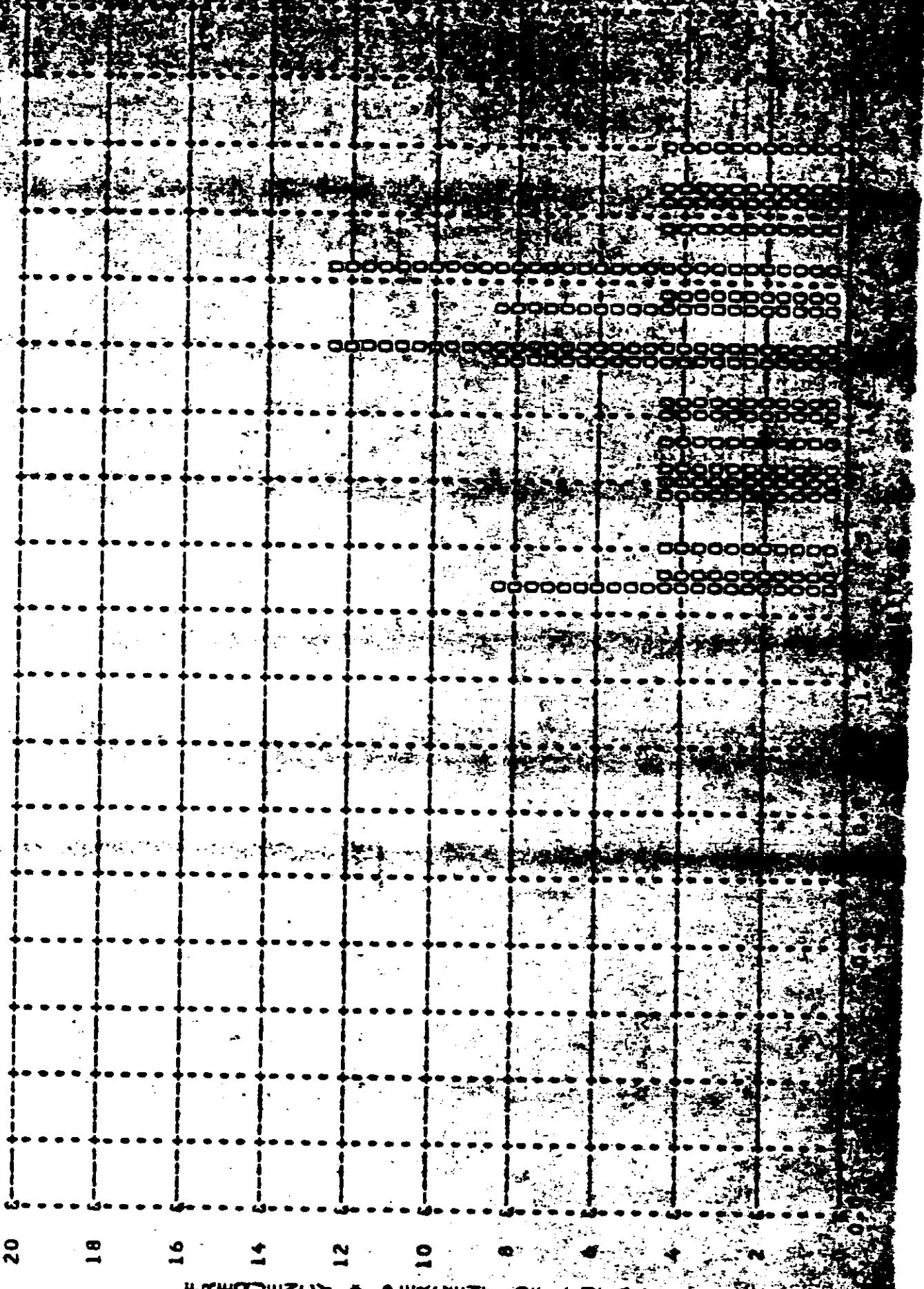
TOP SECRET C

MISSION * 1038-2 * INSTR * FWD * 3/6/67 PLOT OF D MAX * TERRAIN * PROCESSING * INTERMED
AIRTH MEAN * 1.53 * MEDIAN * 1.53 * STD DEV * 0.27 * RANGE * 0.77 TO 1.98 WITH 35 SAMPLES



TOP SECRET C

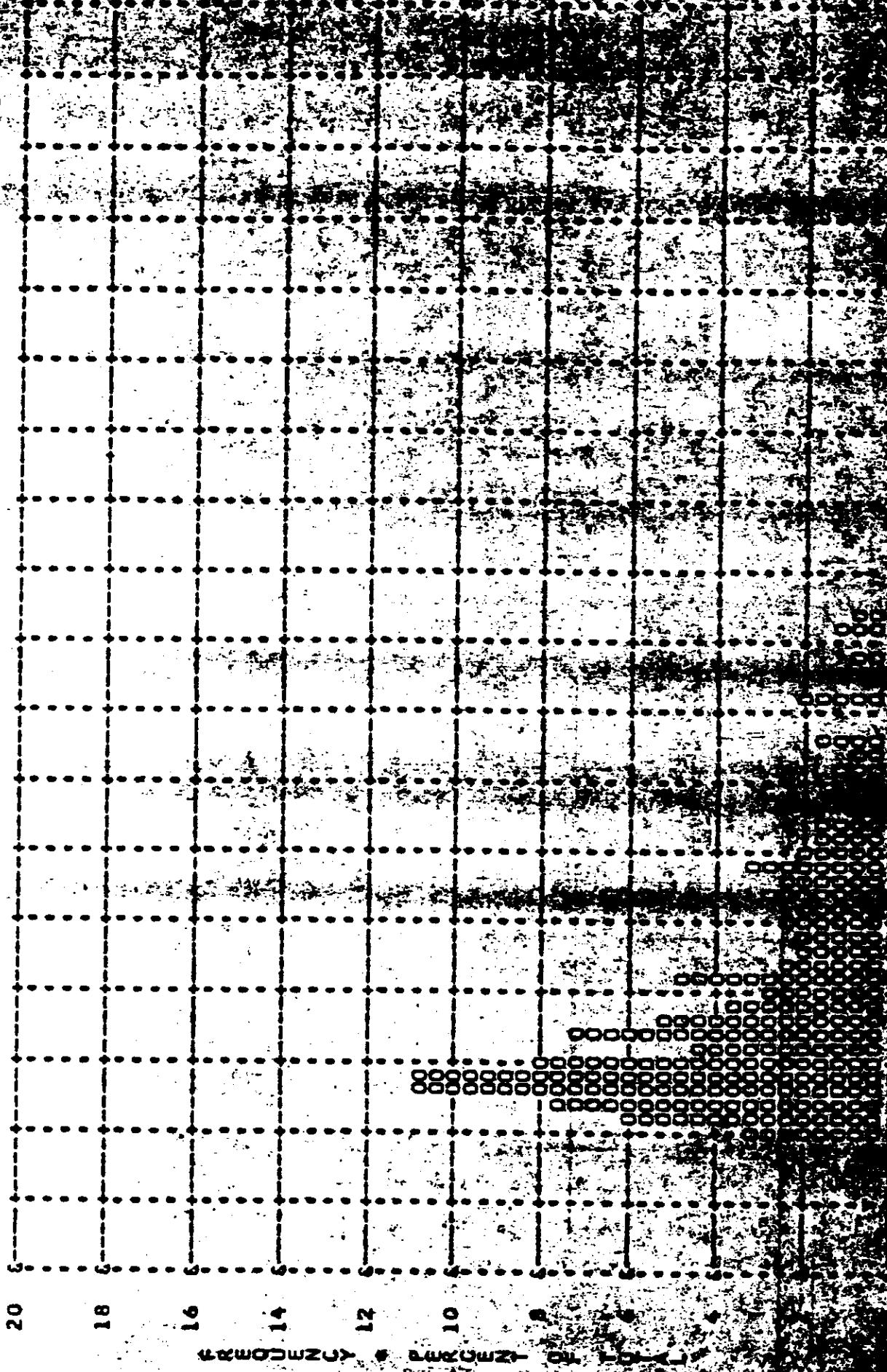
MISSION * 1038-2 * INSTR # FWD * 3/6/67 PLOT OF D MAX * CLOUD * PROCESSING & INTERFACED
AIRTH MEAN * 1.89 * MEDIAN * 1.93 * STD DEV * 0.29 * RANGE * 1.39 TO 2.38 WITH 25 SAMPLES



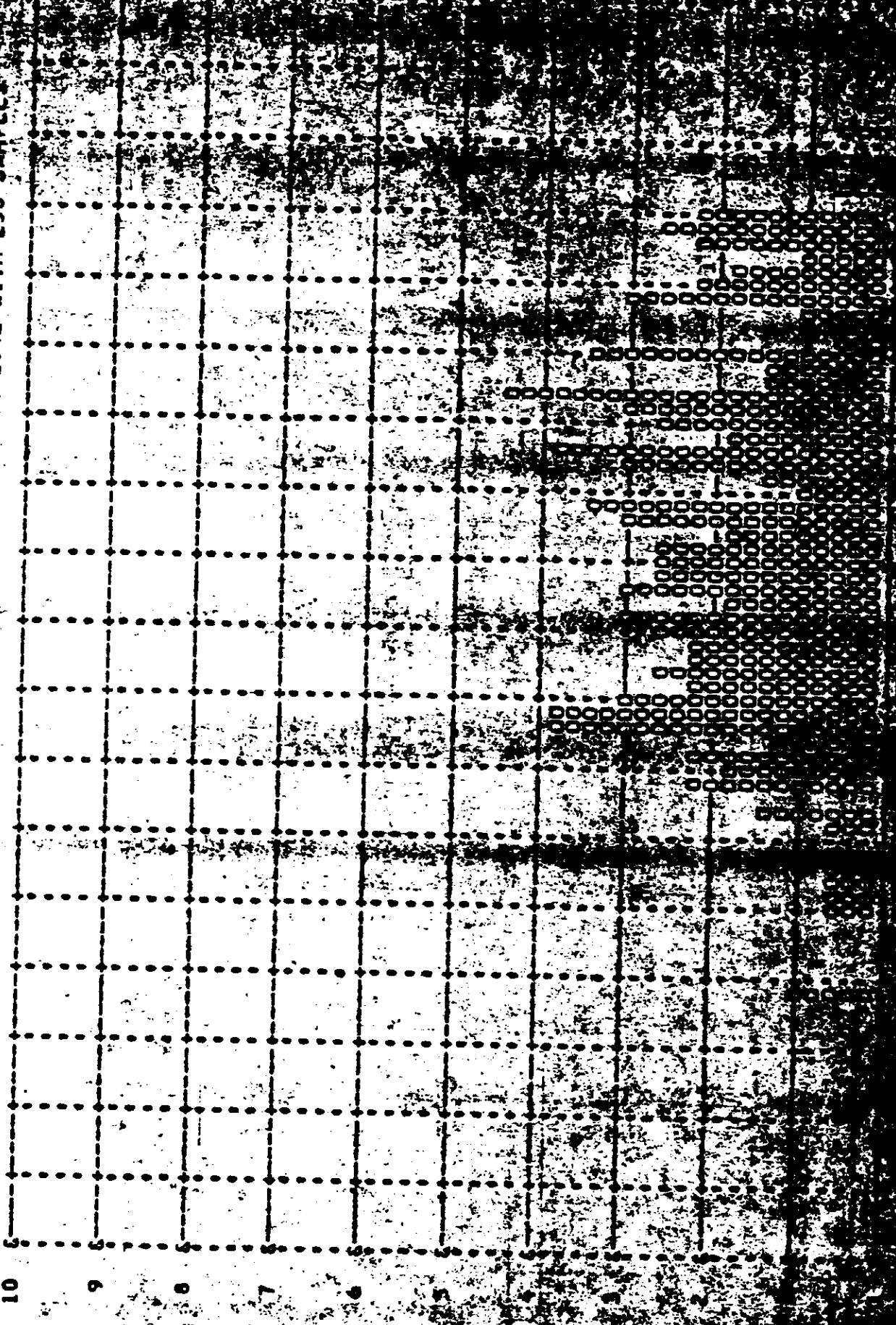
DEWPOINT - °C * TEMPERATURE - °C

TOP SECRET C

MISSION * 1038-2 * INSTR * FWD * 3/6/67 PLOT OF D MIN * TERRAIN * PROCESSING * FULL
AIRTH MEAN * 0.56 * MEDIAN * 0.49 * STD DEV * 0.24 * RANGE * 0.28 TO 1.40 WITH 238 SAMPLES



MISSION * 1038-2 * INSTR * FWD * 3/6/67 PLOT OF D MAX * TERRAIN * PROCESSING * FULL
AIRTH MEAN * 1.57 * MEDIAN * 1.57 * STD DEV * 0.39 * RANGE * 0.35 TO 2.42 WITH 236 SAMPLES



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TOP SECRET C

MISSION * 1038-2 * INSTR * FWD * 3/6/67 PLOT OF D MAX * CLOUD * PROCESSING * FULL
AIRTH MEAN * 2.01 * MEDIAN * 2.09 * STD DEV * 0.35 * RANGE * 1.00 TO 2.50 WITH 150 SAMPLES

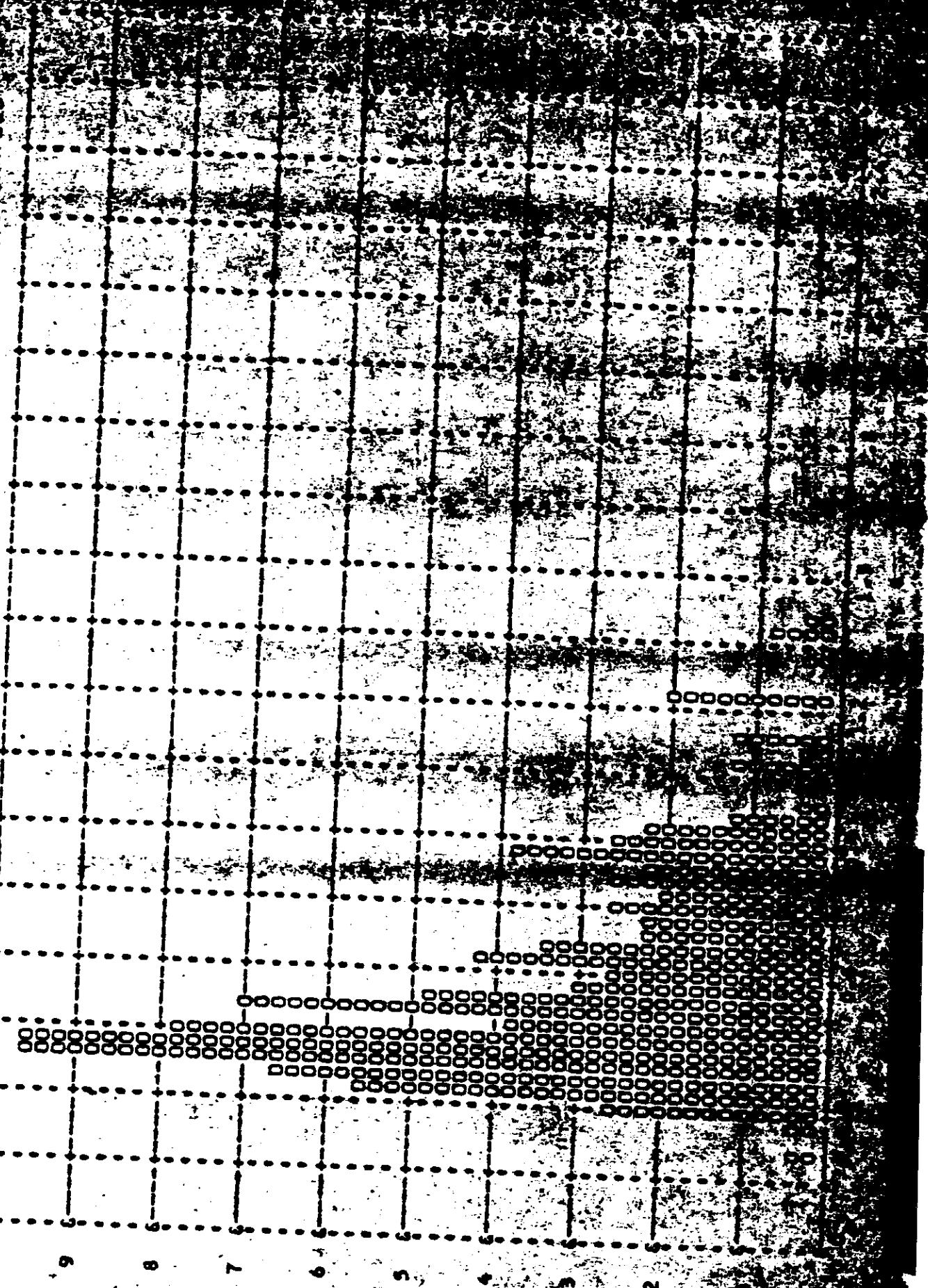
10

9

REMOVED & SECURED BY [redacted]

1005

MISSION # 1038-2 * INSTR * FWD * 3/6/67 PILOT OF D MIN * TERRAIN * PROCESSING ALL LEVELS
AIRTH MEAN * 0.58 * MEDIAN * 0.51 * STD DEV * 0.24 * RANGE * 0.21 TO 1.40 WITH 273 SAMPLES



Il est tout à fait normal que les deux dernières personnes soient absentes de la photo.

SECRET

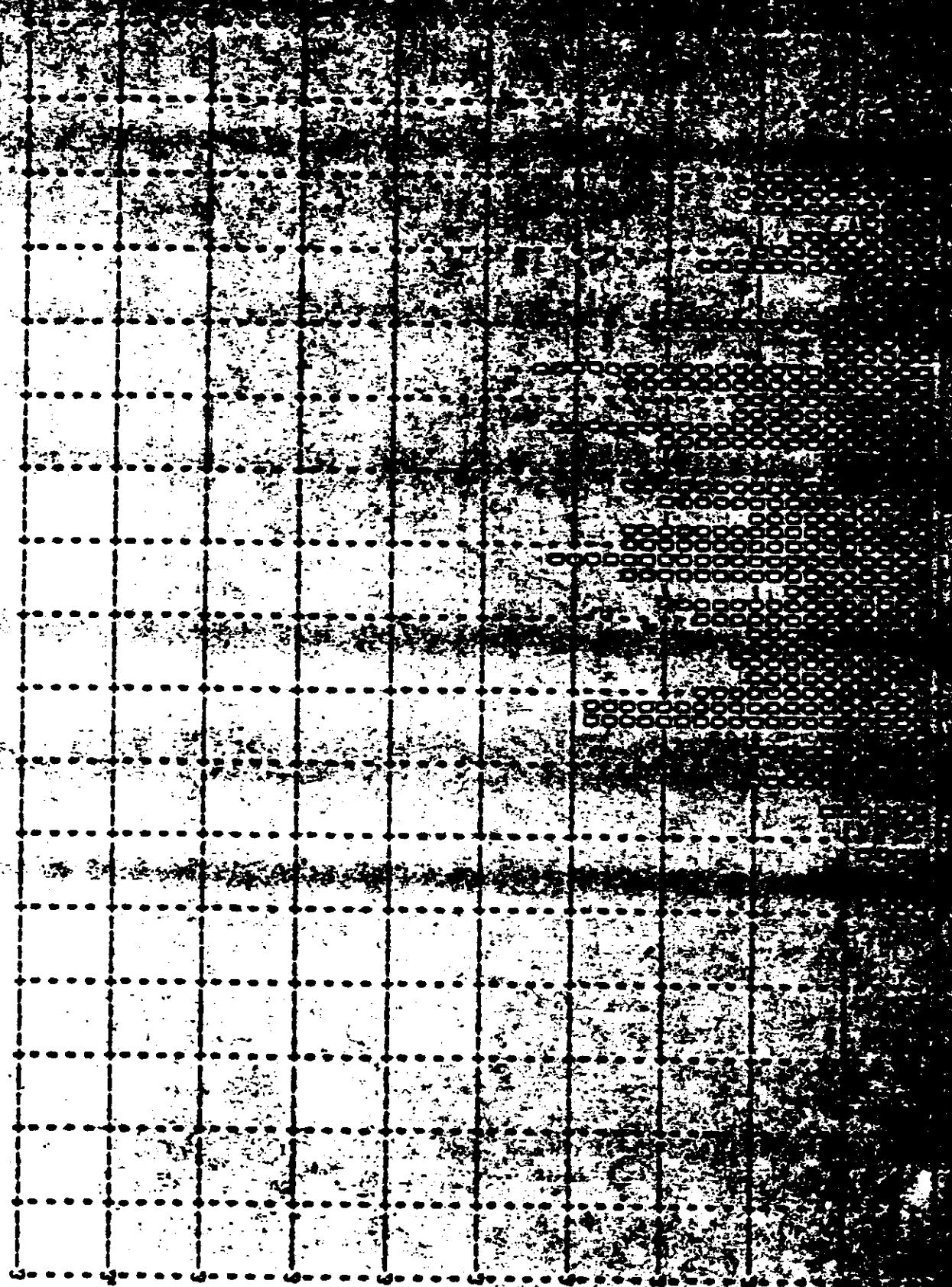
MISSION * 1038-2 * INSTR * FND * 3/6/67 PLOT OF D MAX * TERRAIN * PROCESSING
EARTH MEAN * 1.57 * MEDIAN * 1.54 * STD DEV. * 0.38 * RANGE * 0.55 TO 2.42 MITH 273 JAHK

10

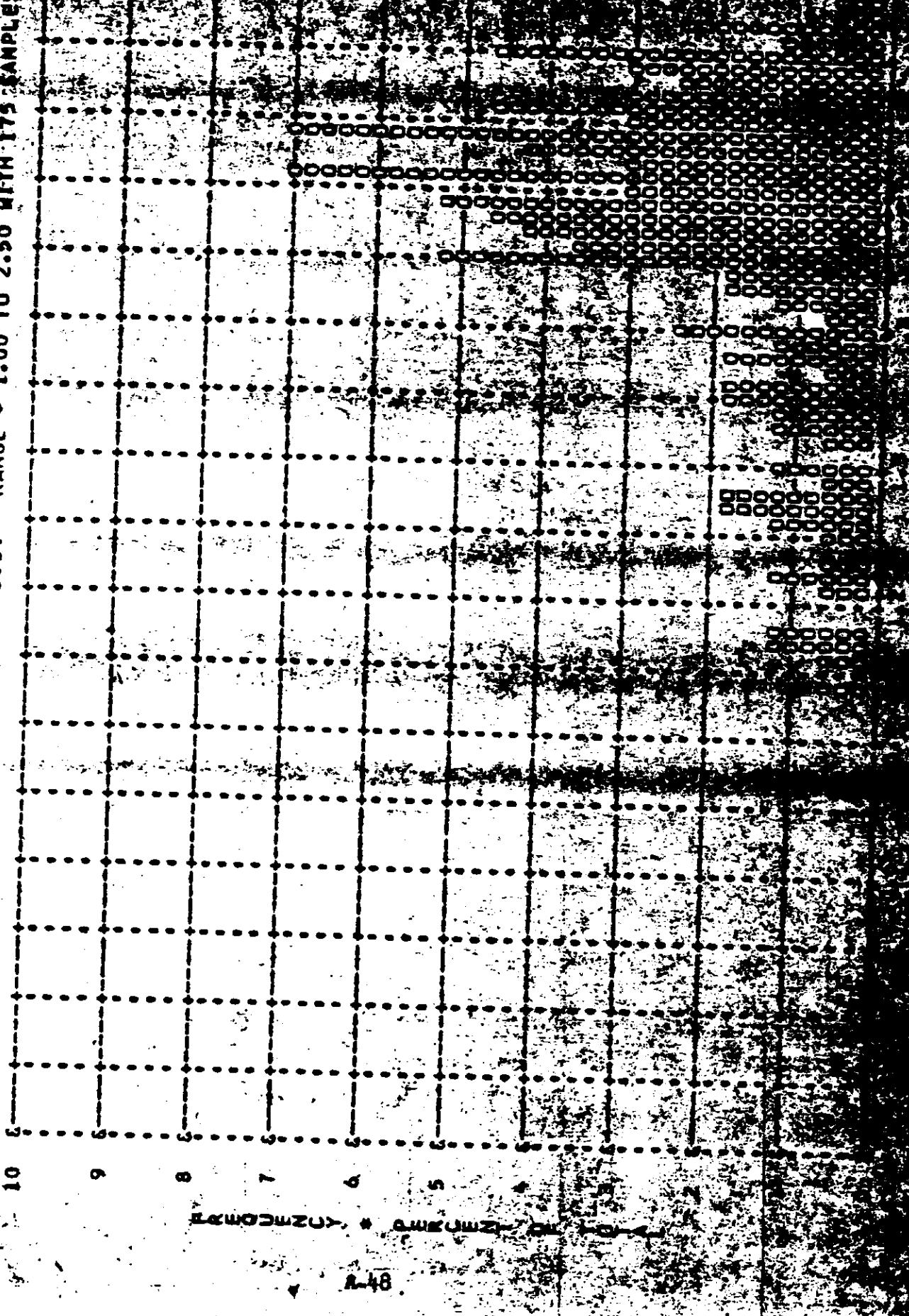
9

8

LEWISBURG * WASHINGTON



MISSION * 1038-2 * INSTR * FWD * * 3/6/67 PLOT OF D MAX * CLOUD * PROCESSING * ALL LEVELS
ALPH MEAN * 1.99 * MEDIAN * 2.06 * STD DEV * 0.34 * RANGE * 1.00 TO 2.50 WITH 175 SAMPLES



DENSITY

VALUES

1.00

0.99

0.98

0.97

0.96

0.95

0.94

0.93

0.92

0.91

0.90

0.89

0.88

0.87

0.86

0.85

0.84

0.83

0.82

0.81

0.80

0.79

0.78

0.77

0.76

0.75

0.74

0.73

0.72

0.71

0.70

0.69

0.68

0.67

0.66

0.65

0.64

0.63

0.62

0.61

0.60

0.59

0.58

0.57

0.56

0.55

0.54

0.53

0.52

0.51

0.50

SUBTOTAL

47

00000

00000

00000

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00000

00000

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~~TOP SECRET C~~

TABLE A-4

A-49

DENSITY VALUE	PRIMARY		INTERMEDIATE		SECONDARY	
	MIN MAX LEM	MIN MAX LEM	MIN MAX LEM	MIN MAX LEM	MIN MAX LEM	MIN MAX LEM
0.51	0.51	0.51	0.51	0.51	0.51	0.51
0.52	0.52	0.52	0.52	0.52	0.52	0.52
0.53	0.53	0.53	0.53	0.53	0.53	0.53
0.54	0.54	0.54	0.54	0.54	0.54	0.54
0.55	0.55	0.55	0.55	0.55	0.55	0.55
0.56	0.56	0.56	0.56	0.56	0.56	0.56
0.57	0.57	0.57	0.57	0.57	0.57	0.57
0.58	0.58	0.58	0.58	0.58	0.58	0.58
0.59	0.59	0.59	0.59	0.59	0.59	0.59
0.60	0.60	0.60	0.60	0.60	0.60	0.60
0.61	0.61	0.61	0.61	0.61	0.61	0.61
0.62	0.62	0.62	0.62	0.62	0.62	0.62
0.63	0.63	0.63	0.63	0.63	0.63	0.63
0.64	0.64	0.64	0.64	0.64	0.64	0.64
0.65	0.65	0.65	0.65	0.65	0.65	0.65
0.66	0.66	0.66	0.66	0.66	0.66	0.66
0.67	0.67	0.67	0.67	0.67	0.67	0.67
0.68	0.68	0.68	0.68	0.68	0.68	0.68
0.69	0.69	0.69	0.69	0.69	0.69	0.69
0.70	0.70	0.70	0.70	0.70	0.70	0.70
0.71	0.71	0.71	0.71	0.71	0.71	0.71
0.72	0.72	0.72	0.72	0.72	0.72	0.72
0.73	0.73	0.73	0.73	0.73	0.73	0.73
0.74	0.74	0.74	0.74	0.74	0.74	0.74
0.75	0.75	0.75	0.75	0.75	0.75	0.75
0.76	0.76	0.76	0.76	0.76	0.76	0.76
0.77	0.77	0.77	0.77	0.77	0.77	0.77
0.78	0.78	0.78	0.78	0.78	0.78	0.78
0.79	0.79	0.79	0.79	0.79	0.79	0.79
0.80	0.80	0.80	0.80	0.80	0.80	0.80
0.81	0.81	0.81	0.81	0.81	0.81	0.81
0.82	0.82	0.82	0.82	0.82	0.82	0.82
0.83	0.83	0.83	0.83	0.83	0.83	0.83
0.84	0.84	0.84	0.84	0.84	0.84	0.84
0.85	0.85	0.85	0.85	0.85	0.85	0.85
0.86	0.86	0.86	0.86	0.86	0.86	0.86
0.87	0.87	0.87	0.87	0.87	0.87	0.87
0.88	0.88	0.88	0.88	0.88	0.88	0.88
0.89	0.89	0.89	0.89	0.89	0.89	0.89
0.90	0.90	0.90	0.90	0.90	0.90	0.90
0.91	0.91	0.91	0.91	0.91	0.91	0.91
0.92	0.92	0.92	0.92	0.92	0.92	0.92
0.93	0.93	0.93	0.93	0.93	0.93	0.93
0.94	0.94	0.94	0.94	0.94	0.94	0.94
0.95	0.95	0.95	0.95	0.95	0.95	0.95
0.96	0.96	0.96	0.96	0.96	0.96	0.96
0.97	0.97	0.97	0.97	0.97	0.97	0.97
0.98	0.98	0.98	0.98	0.98	0.98	0.98
0.99	0.99	0.99	0.99	0.99	0.99	0.99
1.00						
SUBTOTAL				18	7	74

~~TOP SECRET C~~ [REDACTED]

TABLE A-4

DENSITY

VALUE

1.51
1.52
1.53
1.54
1.55
1.56
1.57
1.58
1.59
1.60
1.61
1.62
1.63
1.64
1.65
1.66
1.67
1.68
1.69
1.70
1.71
1.72
1.73
1.74
1.75
1.76
1.77
1.78
1.79
1.80
1.81
1.82
1.83
1.84
1.85
1.86
1.87
1.88
1.89
1.90
1.91
1.92
1.93
1.94
1.95
1.96
1.97
1.98
1.99
2.00

SUBTOTAL

0.00

2 31 1

12

~~TOP SECRET C~~

MISSION - 1035 - INSTRUMENTS - AVERAGE DENSITY

DENSITY VALUE	PRI MARY	INTER MEDIATE	MIN LTH	MAX LTH	MIN MAX
2.01	0	0	0	0	0
2.02	0	0	0	0	0
2.03	0	0	0	0	0
2.04	0	0	0	0	0
2.05	0	0	0	0	0
2.06	0	0	0	0	0
2.07	0	0	0	0	0
2.08	0	0	0	0	0
2.09	0	0	0	0	0
2.10	0	0	0	0	0
2.11	0	0	0	0	0
2.12	0	0	0	0	0
2.13	0	0	0	0	0
2.14	0	0	0	0	0
2.15	0	0	0	0	0
2.16	0	0	0	0	0
2.17	0	0	0	0	0
2.18	0	0	0	0	0
2.19	0	0	0	0	0
2.20	0	0	0	0	0
2.21	0	0	0	0	0
2.22	0	0	0	0	0
2.23	0	0	0	0	0
2.24	0	0	0	0	0
2.25	0	0	0	0	0
2.26	0	0	0	0	0
2.27	0	0	0	0	0
2.28	0	0	0	0	0
2.29	0	0	0	0	0
2.30	0	0	0	0	0
2.31	0	0	0	0	0
2.32	0	0	0	0	0
2.33	0	0	0	0	0
2.34	0	0	0	0	0
2.35	0	0	0	0	0
2.36	0	0	0	0	0
2.37	0	0	0	0	0
2.38	0	0	0	0	0
2.39	0	0	0	0	0
2.40	0	0	0	0	0
2.41	0	0	0	0	0
2.42	0	0	0	0	0
2.43	0	0	0	0	0
2.44	0	0	0	0	0
2.45	0	0	0	0	0
2.46	0	0	0	0	0
2.47	0	0	0	0	0
2.48	0	0	0	0	0
2.49	0	0	0	0	0
2.50	0	0	0	0	0
SUBTOTAL	0	0	0	11	0
	0	0	0	0	20
	0	0	0	0	67
	0	0	0	0	31
	0	0	0	0	62

~~TOP SECRET C~~

TABLE A-4

~~TOP SECRET-C~~

MISSION - 203A-2 INSTRUMENTS

DENSITY VALUE	PRIMARY		INTERMEDIATE		ALL		
	MIN	MAX	BIN	MAX	LIN	MIN	MAX
2.51	0	0	0	0	0	0	0
2.52	0	0	0	0	0	0	0
2.53	0	0	0	0	0	0	0
2.54	0	0	0	0	0	0	0
2.55	0	0	0	0	0	0	0
2.56	0	0	0	0	0	0	0
2.57	0	0	0	0	0	0	0
2.58	0	0	0	0	0	0	0
2.59	0	0	0	0	0	0	0
2.60	0	0	0	0	0	0	0
2.61	0	0	0	0	0	0	0
2.62	0	0	0	0	0	0	0
2.63	0	0	0	0	0	0	0
2.64	0	0	0	0	0	0	0
2.65	0	0	0	0	0	0	0
2.66	0	0	0	0	0	0	0
2.67	0	0	0	0	0	0	0
2.68	0	0	0	0	0	0	0
2.69	0	0	0	0	0	0	0
2.70	0	0	0	0	0	0	0
SUBTOTAL	0	0	0	0	0	0	0
TOTAL	0	0	0	69	69	43	100

~~TOP SECRET-C~~

TABLE A-6

A-54

SECRET C

MISSION * 103842 * INSTR * AFT * 3/6/67 PLOT OF D MIN * TERRAIN * PROCESSING
AIRTH MEAN * 0.48 * MEDIAN * 0.38 * STD DEV * 0.30 * RANGE * 0.19 TO 1.67 WITH 4000 POINTS

20

10

0

-10

-20

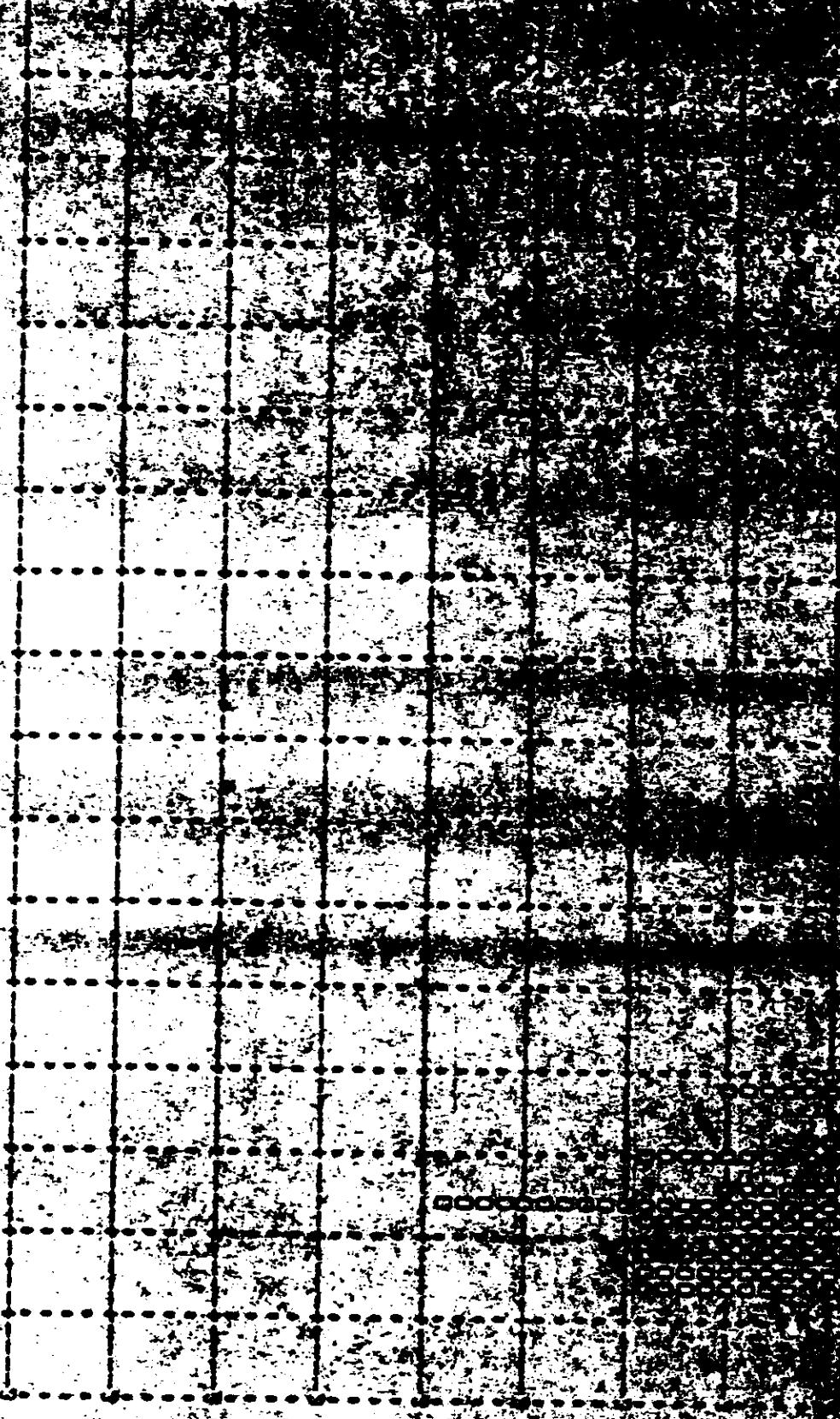
-30

-40

-50

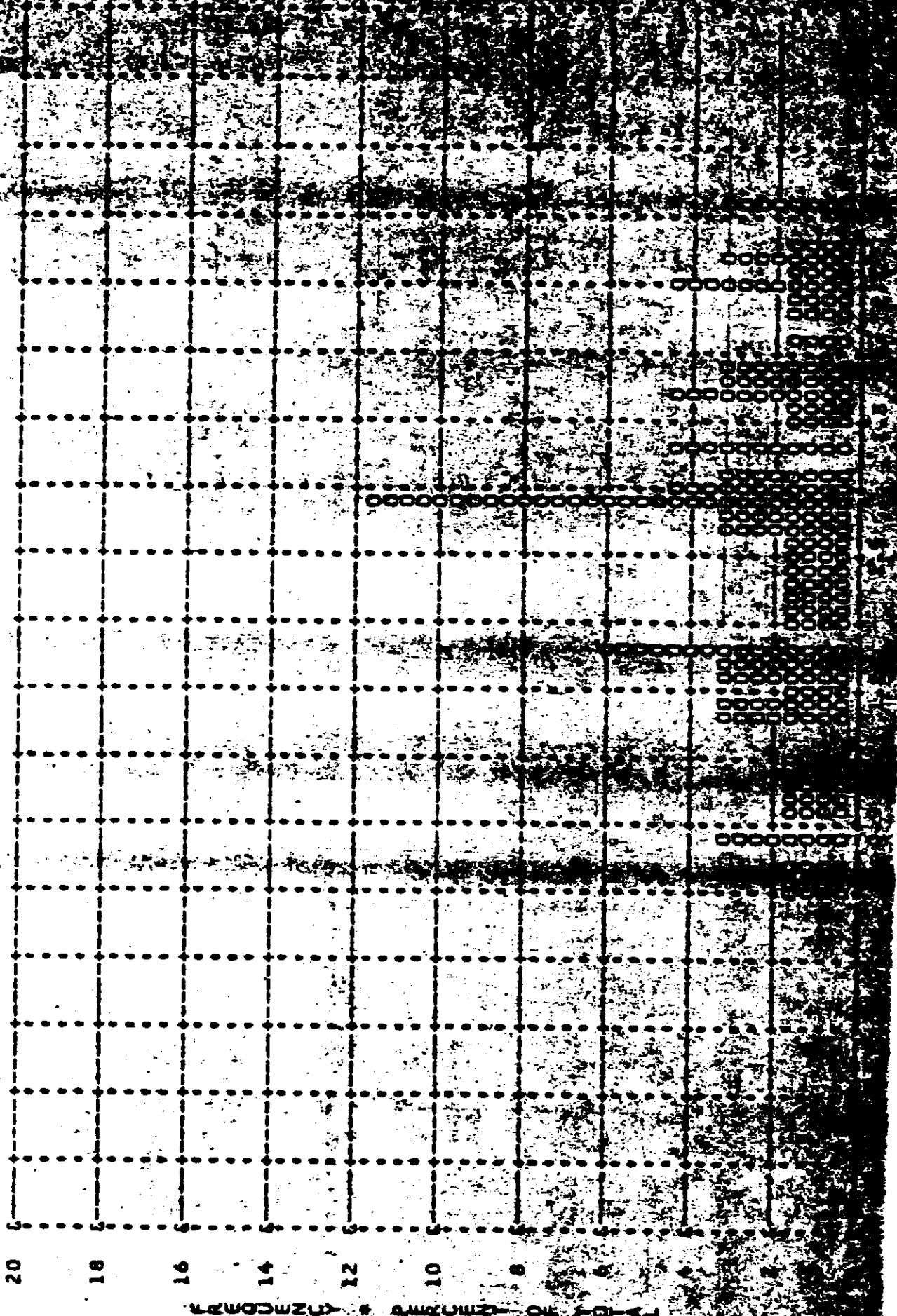
-60

-70



TOP SECRET C

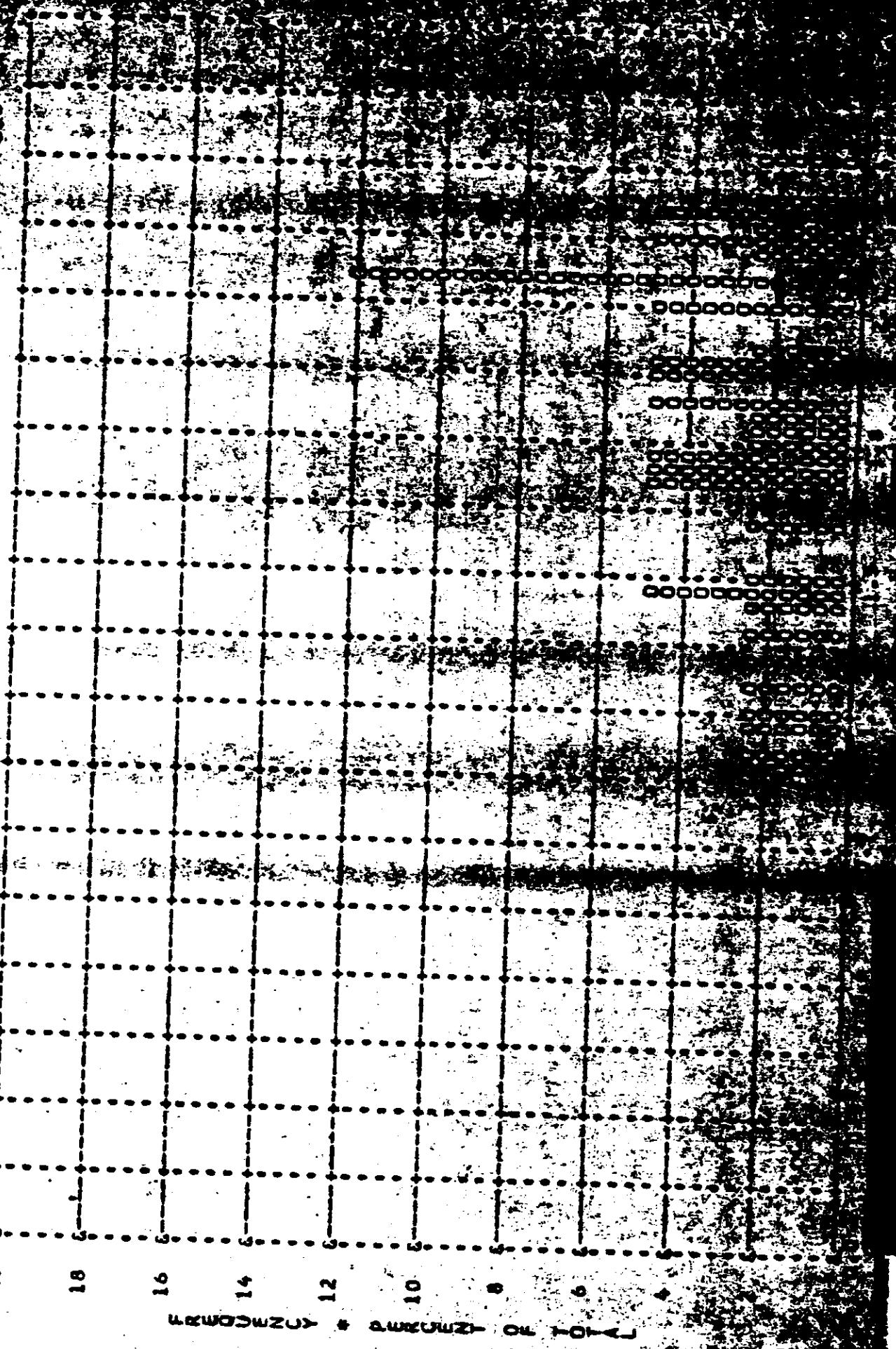
MISSION * 1038-2 * INSTR * AFT * 3/6/67 PLOT OF D MAX * TERRAIN * PROCESSING * INTERMEDIATE
AIRTH MEAN * 1.56 * MEDIAN * 1.61 * STD DEV * 0.39 * RANGE * 0.73 TO 2.27 WITH 69 SAMPLES



LEWISBURG * 04-01-67

TOP SECRET C

MISSION # 1038-2 * INSTR * AFT * 3/6/67 PLOT OF D MAX * CLOUD * PROCESSING & INTERNE
 AIRTH MEAN * 1.83 * MEDIAN * 1.89 * STD DEV * 0.37 * RANGE * 1.06 TO 2.42 WITH 43 SAMPLES



SECRET C

MISSION # 1038-2 * INSTR * AFT * 3/8/67 PLOT OF D MIN * TERRAIN * PROCESSING TIME
AIRTH MEAN = 0.66 * MEDIAN = 0.53 * STD DEV = 0.30 * RANGE = 0.26 TO 1.78 WITH 1.78 AND 1.78

20

10

A-38

~~SECRET~~ C

MISSION * 103B-2 * INSTR * AFT * 3/6/67 PLOT OF D MAX * TERRAIN * PROCESSING * FULL
AIRTH MEAN * 1.55 * MEDIAN * 1.56 * STD DEV * 0.35 * RANGE * 0.47 TO 2.38 WITH 198 SAMPLES

10

9

8

7

6

REWORDZUS * EWEUWZT OR HOI<

REF ID: A60100

MISSION * 1038-2 * INSTR * AFT * 3/6/67 PLOT OF D MAX * CLOUD * PROCESSING * FULL
AIRTH MEAN * 2.00 * MEDIAN * 2.03 * STD DEV * 0.37 * RANGE * 0.35 TO 2.48 WITH 117 SAMPLES

10

9

8

7

6

5

4

3

2

1

0

0.0 0.2 0.4 0.6 0.8 1.0 1.2 1.4 1.6 1.8 2.0 2.2 2.4 2.6 2.8 3.0 3.2 3.4 3.6 3.8 4.0

SECRET C

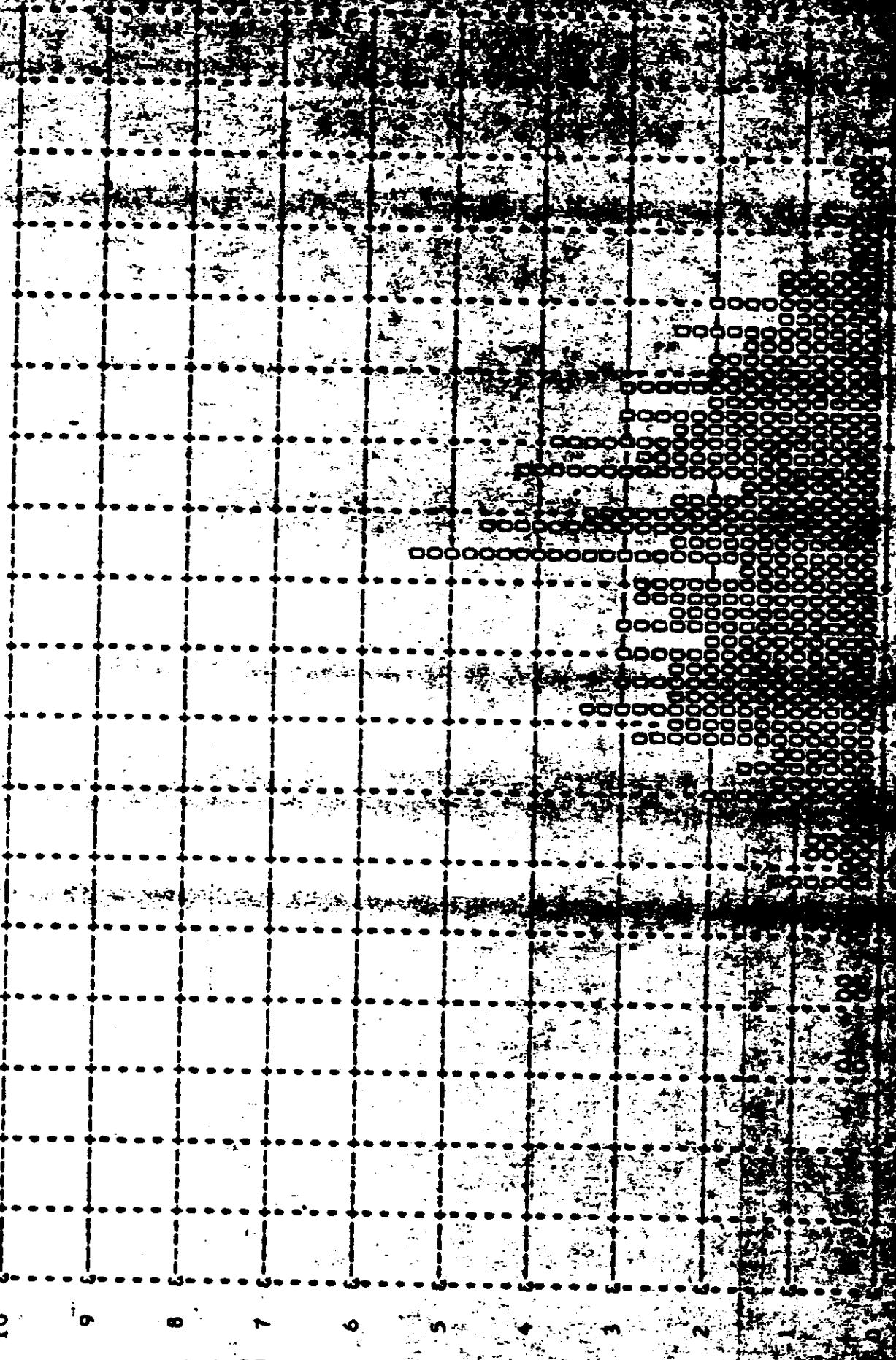
MISSION * 1038-2 * INSTR * AFT * 3/6/67 PLOT OF D MIN * TERRAIN * PROCESSING * ALL LEVELS
AIRTH MEAN * 0.61 * MEDIAN * 0.49 * STD DEV * 0.31 * RANGE * 0.19 TO 1.78 WITH 267 SAMPLES



RECORDED ON 12/20/1967 BY KUZMCDMA

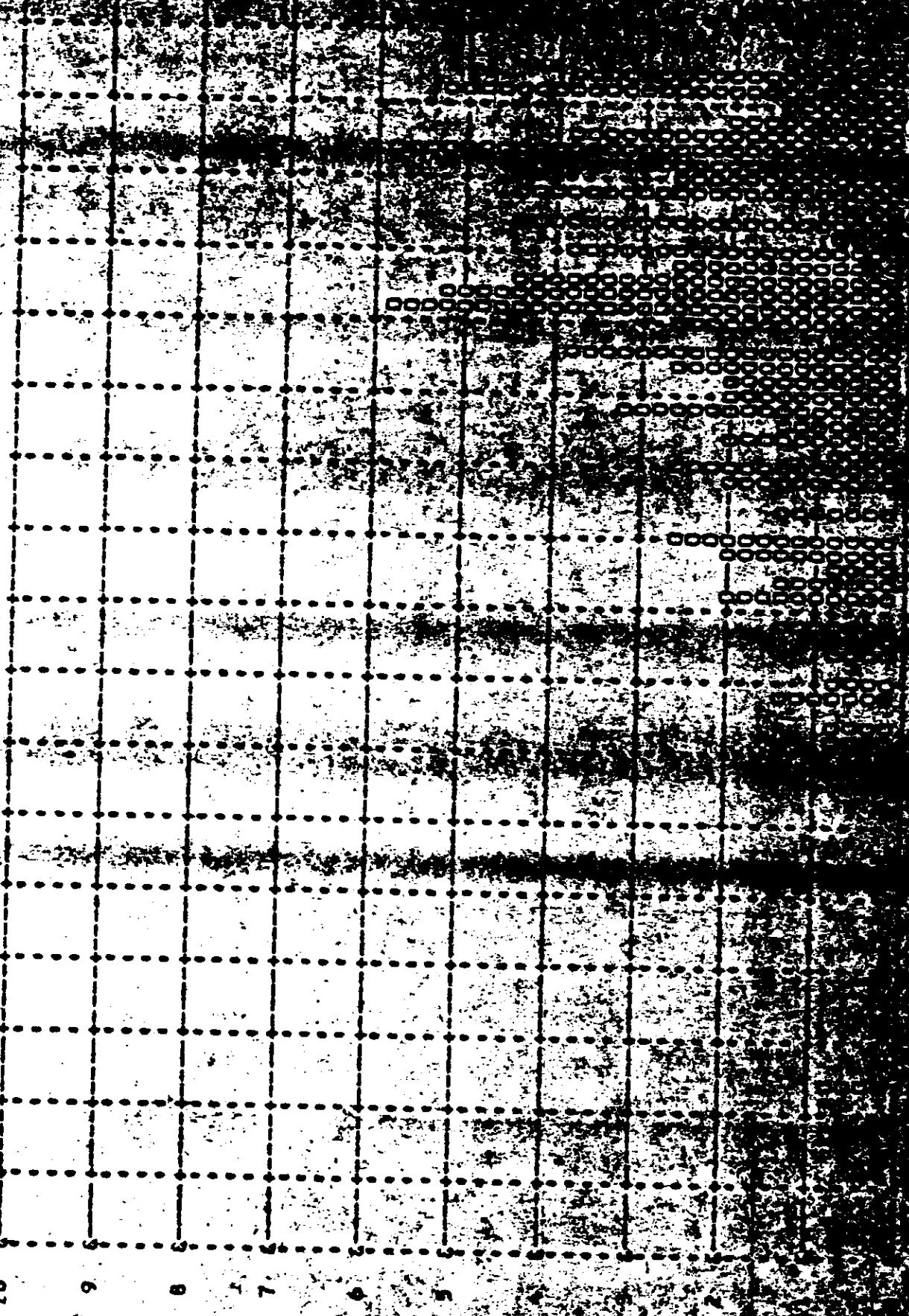
1996-CEAT C

MISSION # 1038-2 * INSTR * AFT * 3/6/67 PLOT OF D MAX * TERRAIN * PROCESSING AT ALL LEVELS
AIRTH MEAN * 1.56 * MEDIAN * 1.57 * STD DEV * 0.36 * RANGE * 0.47 TO 2.38 WITH 267 SAMPLES



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MISSION # 1038-2 * INSTR * AFT * 3/6/67 PLOT OF D MAX * CLOUD * PROCESSING ALL LEVELS
AIRTH MEAN * 1.96 * MEDIAN * 2.01 * STD DEV * 0.36 * RANGE * 0.35 TO 2.48 WITH 160 SAMPLES



~~TOP SECRET C~~

Distribution:

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