

COMINT AND RAPID REPORTING INTERFEROMETRY EXPERIMENT

"CARRIE"

PROGRAM OVERVIEW



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OBJECTIVES

- **SATISFY KNOWN SIGINT COLLECTION SHORTFALL**

- **DEMONSTRATE ADVANCED TECHNOLOGY**
 - **REPROGRAMMABLE ON-BOARD PROCESSOR**
 - **ALGORITHMS FOR GEOLOCATION AND IDENTIFICATION**
 - **MINIATURE GPS RECEIVER**

- **DEMONSTRATE THE TACTICAL UTILITY OF A DEDICATED OVERHEAD SIGINT SYSTEM**
 - **STREAMLINED TASKING**
 - **ON BOARD GEOLOCATION AND IDENTIFICATION**
 - **DIRECT DOWNLINK TO EPDS VAN**

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<u>REQUIREMENT</u>	<u>NSA REQUEST</u>	<u>CARRIE CAPABILITY</u>
RF Range	0.12 - 2.0 GHz	0.10 - 0.85 GHz
Target Signals	<div style="border: 1px solid black; width: 100%; height: 50px;"></div>	
Pre-detection	B/W up to 250 KHz (several minutes)	B/W 100 KHz (1 spin-digital storage)
Ant. Polrzt. n.	Selectable L,RHCP	RHCP
Throughput	1000 geos / day	
Geo Accuracy	<div style="border: 1px solid black; width: 100%; height: 30px;"></div>	
RF Accuracy	10-100 KHz	50 KHz
Inclination	60 - 70 deg	105 deg
Spin Axis	North-South	North-South
Mission Duration	3 years	1 year

(From Aug 87 NSA P52 letter describing a COMINT Mapping System minimum requirements set.)

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*****-SECRET** **CARRIE PROGRAM**

- **DARPA funded tactical COMINT mapping experiment.**
- **NRO executed reconnaissance spacecraft acquisition program**

- **Product of 1988 DARPA broad area announcement and subsequent source selection:**
 - **Ball Aerospace: spacecraft bus and integration**
 - **ATP 17 Dec 90**
 - **AIL Systems: COMINT mapping payload**
 - **ATP 28 Sep 90**

- **Launch on 1st TAURUS booster with TAOS spacecraft: 15 Oct 92**

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CARRIE DESCRIPTION

- o 400 lb Spacecraft in 280 nm orbit
- o Experimental COMINT mapping mission
 - oo On-board search, ID, geolocation and reporting of select tactical comms

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oo Direct reporting to EPDS van

o AFSCN / [redacted]

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LAUNCH SEGMENT OVERVIEW

- **FIRST TAURUS LAUNCH (PEGASUS + PEACEKEEPER 1ST STAGE)**
- **BOOSTER PROGRAM IS AN EXPERIMENTAL CONCEPT FOR A RAPID REACTION LAUNCH CAPABILITY**
 - Streamline booster launch base processing
 - Spacecraft encapsulated prior to booster integration
- **LAUNCHED FROM NORTH VANDENBERG AFB 3RD QTR CY 92**
- **RIDESHARE WITH TAOS spacecraft**

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GROUND SEGMENT OVERVIEW



- **EPDS vans**
 - **Receive transponds in theatre**
 - **Provide focal point and connectivity to tactical requests**

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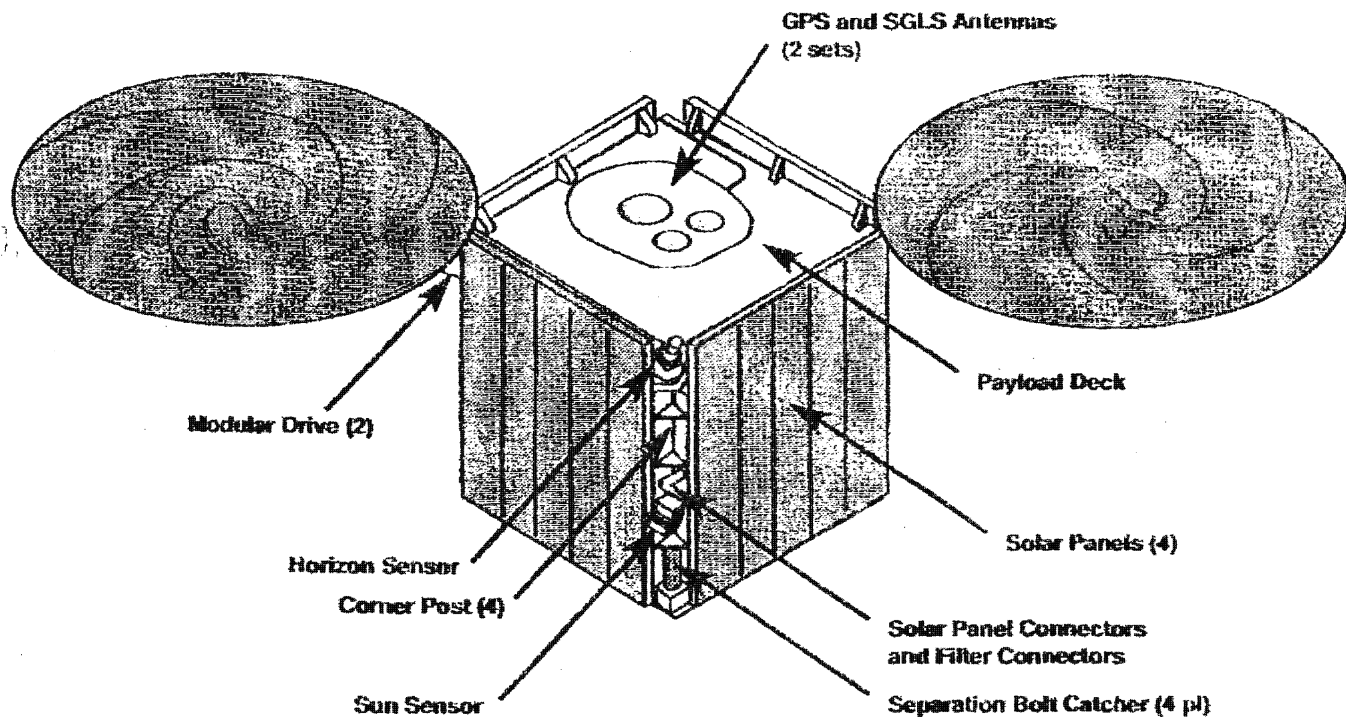
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CARRIE SPACECRAFT

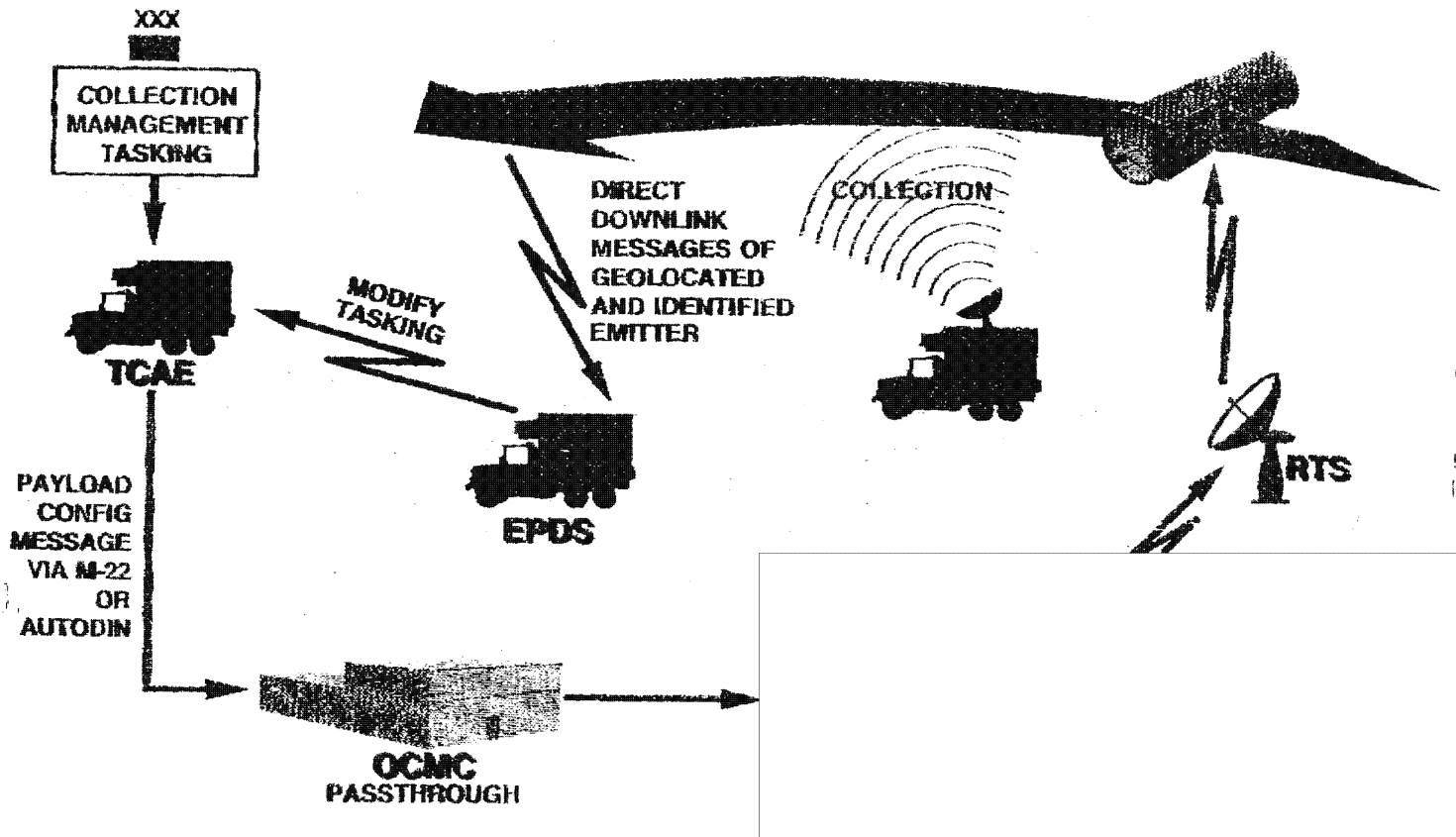


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CARRIE TASKING SCENARIO



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CARRIE SCHEDULE

Month	FY 91					FY 92					FY 93																		
	S	O	B	D	J	F	M	A	M	J	J	A	S	O	N	D	J	F	M	A	M	J	J	A	S	O	N	D	J
P/L ATP	+																												
BUS ATP				+																									
P/L PDR									+																				
BUS PDR										+																			
P/L CDR																													
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LAUNCH																													

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this is a placeholder, we'll send a more attractive one first thing next week.

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SUMMARY

- **OBJECTIVES**

- Satisfy Real Mission Requirement
- Demonstrate tacsat concepts
- Demonstrate new processing technologies

- **PLANNED CAPABILITIES**

- Onboard COMINT geolocation & identification
- Downlink directly to EPDS van [redacted]

- **OPERATIONAL TIME FRAME**

- Launch - Oct 1992
- Available for Demos - Oct 92 through Sep 93

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- **ADVANCED TECHNOLOGY**

BACK UPS

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- **PAYLOAD PROCESSOR CPU (TI TMS320C30)**
 - Commercially available, rad hard, 12 MIP machine
 - Designed for intensive high speed digital signal processing
 - Reprogrammable by electronically erasable (E²) PROM
- **PAYLOAD ALGORITHMS**
 - Co-channel interference determination
 - Sets thresholds without knowing noise floor
 - Real-time phase reconstruction ("on the fly")
- **GPS RECEIVER**
 - DARPA funded, six channel system
 - 8 lbs, 15 Watts, RS232 interface
 - OBP accomplished without ground generated ephemeris

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CARRIE TASKING SCENARIO

- DATES AND LOCATIONS OF DEMONSTRATIONS ARE PREPLANNED
 - Streamlined tasking chain authorized by prior agreement
 - Dates, time of passes through target are determined in advance by orbital mechanics
 - [redacted]
 - [redacted]
 - [redacted]
 - User must request payload configuration from "menu" at least 90 min prior to RTS contact preceding AOI visibility

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