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VWZD Official Copy

CLASSIFICATION CHANGED TO  
SECRET  
APR 20 5 1966

TOP SECRET

SECRET TO VANDENBERG AFB CALIF  
AF SPACE SYSTEMS DIVISION LOS ANGELES CALIF

SECRET TO /SATELLITE/ SUBUTYVALE CALIF

LOCKHEED MISSILES & SPACE DIVISION SUBUTYVALE CALIF

LOCKHEED MISSILES & SPACE DIVISION VANDENBERG AFB  
CALIF /COMSEC/ ZEN

DOUGLAS AIRCRAFT COMPANY VANDENBERG AFB CALIF  
/COMSEC/ ZEN

INFO: 1 MISSILE DIV VANDENBERG AFB CALIF /COMSEC/ ZEN

SECRET TO VANDENBERG AFB CALIF /COMSEC/ ZEN

SECRET TO PATRICK AFB FLORIDA

SECRET FROM VWZD-19-4-51-5

LOSA FOR BRID BRIGLIN 6004 HQ FOR COL MOORE BRIGLIN LIND/

SUBUTYVALE FOR TWCA-2 /A.I. BRIGLIN/ BRIGLIN LIND/YAFB

FOR DEPT 65-44 BRIGLIN HQ/YAFB FOR MR. BRIGMAN. INFO CLK

1 MD FOR COMMAND POST AND DOORS BRIGLIN 5445 TW FOR VWZD/

DOWNGRADED AT 3 YEAR INTERVALS  
DECLASSIFIED AFTER 15 YEARS.  
DOD EOE 5800.10

Cy 7 of 8 exp

OFFICE SYMBOL VWZD	ORIGINATOR [REDACTED]	DATE 11 Nov 59	RECEIVED BY LUCAS A. STRETT, JR., Colonel, USAF
NAME (SIGNATURE) [REDACTED]	[REDACTED]	[REDACTED]	[REDACTED]
DATE 11 Nov 59	COORDINATION SHEET		

~~CONFIDENTIAL~~

LAUNCH OPERATIONS SUMMARY

LAUNCH OPERATIONS SUMMARY AND TEST RESULTS FOR THE WEDGALL. SUBJECT

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1. LAUNCH OPERATIONS

LAUNCH OPERATIONS WERE STARTED AT 600 FT ON 1A  
 THE FIRST HOLD WAS CALLED AT T-15 MINUTES TO  
 COMPLETE WORK THAT HAD BEEN DELAYED EARLIER WHEN IT WAS  
 NECESSARY TO REPLACE A LEAKING AGMA ACID FILL COUPLER  
 DURING THE HOLD A FACILITY RELEASED VALVE IN THE ORBITAL  
 STAGE AND THE PRESSURIZATION SYSTEM WAS REPLACED ALSO.  
 DURATION OF THE HOLD WAS 20 MINUTES. THE SECOND HOLD,  
 WHICH LASTED 5 MINUTES, WAS CALLED AT T-10 MINUTES TO  
 ALLOW COMPLETION OF ALL OF THE TERMINAL COUNTDOWN PRE-  
 PARATIONS. TERMINAL COUNTDOWN STARTED AT 1201 FT. THE  
 THIRD AND FINAL HOLD WAS CALLED IN PHASE II OF THE TERMINAL  
 COUNTDOWN TO SOLVE A PROBLEM IN THE TEST NITROGEN  
 PRESSURIZATION SYSTEM. THE HOLD WAS FOR 2 MINUTES AFTER  
 WHICH THE TERMINAL COUNTDOWN PROCEEDED SATISFACTORILY  
 TO LIFT OFF AT 10000 FT.

2. BOOST PHASE

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**CONTINUED FROM PREVIOUS PAGE**

THIS WAS THE FIRST RECOVERED VEHICLE TO BE LAUNCHED FROM THE FRONT OF THE COOLED FLAME-BUCKET CONFIGURATION. ALL OF THE EVENTS ASSOCIATED WITH LIFT-OFF, THE LAUNCH ROLL-OVER (NO. 3 ENG.), AND CLIMB-OUT APPEAR TO HAVE BEEN ACCOMPLISHED SATISFACTORILY. THE SECOND GUIDANCE SYSTEM INITIATED STEERING CORRECTIONS BETWEEN T + 29.75 SECONDS AND T + 34.15 SECONDS, AND THE VEHICLE RESPONSE TO THESE CORRECTIONS APPEARS TO HAVE BEEN PROPER. BOOSTER MAIN ENGINE SHUT-OFF OCCURRED AT T + 149.10 SECONDS AS COMMANDS BY THE GROUND GUIDANCE SYSTEM, AND IT WAS FOLLOWED BY 7.1 SECONDS OF VEHICLE ENGINE SOLO OPERATION. PRELIMINARY CALCULATIONS, BASED ON GUIDANCE SYSTEM DATA, SHOW THAT THE BOOSTER COAST APOGEE ALTITUDE WAS 0.3 M.M. HIGHER THAN THE NOMINAL VALUE AND THE APOGEE VELOCITY WAS 6 FTS LOWER THAN THE NOMINAL VALUE.

B. 20 CPS OSCILLATIONS ARE EVIDENT IN VEHICLE ACCELEROMETER DATA AS HAS BEEN NOTED IN PREVIOUS DISCOVERIES.

FLIGHTS. THE MAXIMUM PEAK-TO-PEAK AMPLITUDE OF THE OSCILLATIONS IN AGEMA ACCELEROMETER DATA IS 1.4 G.

**3. COAST PHASE**

A. THE GROUND GUIDANCE SYSTEM TRANSMITTED AN AGEMA

[REDACTED]

... TO THE CORRECTION OF 7.55 SECONDS AND A VELOCITY-  
TO BE GAINED CORRECTION OF 1.50 INCHES. THESE CORRECTIONS  
WAS PROPERLY RECEIVED AND EXECUTED BY THE ORBITAL STAGE  
VEHICLE.

B. SEPARATION OF THE ORBITAL STAGE FROM THE BOOSTER  
WAS COMPLETED BY GUIDANCE AT T = 166.54 SECONDS  
AND THE SEPARATION WAS SATISFACTORILY ACHIEVED BY T = 167  
SECONDS.

4. ORBITAL STAGE BOOST

A. THE ACQUA ENGINE FIRED IN A NORMAL MANNER AT  
T = 212.22 AND OPERATED SATISFACTORILY FOR 242.4 SECONDS.  
ENGINE THRUST AND SPECIFIC IMPULSE APPEAR TO HAVE BEEN AT  
OR SLIGHTLY ABOVE THE PRINTED VALUES FOR THESE PARA-  
METERS. ENGINE SHUTDOWN OCCURRED AT T = 454.56 SECONDS  
IN RESPONSE TO A COMMAND FROM THE INTEGRATOR. THE  
REMARKS VELOCITY GAIN DURING THE ORBITAL STAGE BOOST  
PERIOD WAS SHOWN BY THE INTEGRATOR DATA TO BE 16,324 FTS.

B. THE ACQUA ATTITUDE WAS PROPERLY CONTROLLED DURING  
THE COAST AND ORBITAL STAGE BOOST PHASES AND CONTROL  
GAS EXPENDITURE DURING THESE PHASES WAS NORMAL. PRE-  
SENTLY AVAILABLE RECORDS SHOW THAT THE TEC ELECTRICAL

[REDACTED]



TESTING THE VARIATIONS AND CALIB

POWER SUPPLY VOLTAGE DROPPED SEVERAL VOLTS BELOW

13 JAN 1968

TOLERANCE DURING THE ORBITAL BOOST PERIOD. THE LOW VOL-  
TAGE DOES NOT APPEAR TO HAVE EFFECTED SYSTEMS OPERATION.

8 APR 1968

ALL TIMES CONTROLLED EVENTS THROUGH THE START OF

VEHICLE REORIENTATION AFTER ORBIT INJECTION OCCURRED IN  
THE DESIRED SEQUENCE AND AT PROPER TIMES.

C. THE VEHICLE WAS SATISFACTORILY DIRECTED INTO ORBIT  
WITH NEAR-NOMINAL CONDITIONS OF POSITION AND VELOCITY.

**D. DATA ACQUISITION**

THE VARIOUS TRACKING SYSTEMS RECEIVED THE ALPHA TELE-  
METRY SIGNAL FROM LIFTOFF TO T + 309 SECONDS AND GOOD  
DATA RECORDS HAVE BEEN PRODUCED FOR THE PERIOD. THE

ACTUATION BEACON SIGNAL WAS RECEIVED FROM LIFTOFF TO  
T + 418 SECONDS. THE VARIOUS TRACKING RADARS ACTIVELY TRACKED  
THE ALPHA BEACON FROM LIFTOFF TO T + 418 SECONDS.

SATISFACTORY ORBITAL TRACKING DATA WERE ACQUIRED BETWEEN  
LIFTOFF AND T + 418 SECONDS.

**E. PAD DAMAGE**

DAMAGE TO THE PAD FACILITIES AND EQUIPMENT IS CONSIDER-  
ED TO BE NORMAL.

REF 4.

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