

SPECIAL HANDLING

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SPECIAL HANDLING

62 253

SPECIAL PURPOSE PACKAGES FLOWN

<u>PACKAGE</u>	<u>FREQ BAND (MCS)</u>	<u>VEHICLE</u>	<u>DATE LAUNCHED</u>	<u>REMARKS</u>
E-1 SOCTOP	2500 - 3200	1057	10 Aug 60	
E-2 SOCTOP	300 - 900 2500 - 3200	1058	13 Sep 60	
E-3 SOCTOP	300 - 900 1000 - 6000	1061	26 Oct 60	No Agena separation
E-4 SOCTOP	300 - 900	1062	12 Nov 60	
E-5 SOCTOP	300 - 900 1000 - 6000	1105	30 Mar 61	No orbit
E-6 SOCTOP W-7 TAKI	1000 - 6000 160 - 175 100 - 1000	1107	18 Jun 61	
E-8 SOCTOP W-9 WILD BILL	1000 - 6000 40 - 120	1109	7 Jul 61	
E-9 SOCTOP W-10 TAKI	1000 - 6000 160 - 175 100 - 1000	1111	4 Aug 61	No orbit
E-10 TEXAS PINT	100 - 150	1112	30 Aug 61	
E-11 TOPSOC	400 - 800 800 - 1600	1113	12 Sep 61	
E-12 TOPSOC	400 - 800 800 - 1600	1114	18 Sep 61	
E-13 TOPSOC	400 - 800 800 - 1600	1115	13 Oct 61	

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<u>PACKAGE</u>	<u>FREQ BAND (MCS)</u>	<u>VEHICLE</u>	<u>DATE LAUNCHED</u>	<u>REMARKS</u>
E-14TOPSOC		1117 ✓	5 Nov 61	
E-1 GRAPE JUICE		1119 ✓	12 Dec 61	
E-18 S.T. LOW		2203 ✓	21 Dec 61	
E-14 TAKI		1120 ✓	13 Jan 62	No orbit.
E-14 WILD BILL		1123 ✓	27 Feb 62	
E-21 ST. LOW		2204 ✓	7 Mar 62	
E-22 G. B. J. K. H. A.		1124	17 April 62	
E-22 TAXI		1125	19 April 62	
E-24 W. E. J. T. E. A. S. O.		M 31	26 July 62	
P. M. J. A. S. O. C. K.		1125	17 July 62	
E-25 G. B. J. K. H. A.				

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PURPOSE	ANTENNA TRACKING DETERMINATION
RECEIVERS	KITAL VIDEO
PULSE WIDTH	NONE
PULSE REPETITION RATE, BRGZ	ROUGH INDICATION
ANTENNA PATTERN	350 MILES LONG and 1,000 MILES WIDE
	200 mile altitude
DATA ST RATE	BINARY REGISTERS
READOUT	FR/S. TELEMETRY Six commutator points (Three each band)
FREQUENCY	300 - 900 MCS 2500 - 3200 MCS (1000 - 6000 MCS W/FILTRERS OUT)

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Purpose.

To determine whether the Soviets had a radar with a capability to side-track an artificial earth satellite and, specifically whether Apollo vehicles were being so tracked.

Description.

Frequency. - The two-channel system covered the frequency bands 300 to 900 mc and 2500 to 3200 mc (in vehicles 1061 and if a filter was removed to permit coverage from 1000 to 6000 mc). Consideration was given to the feasibility of installing a 70 to 90 mc band but because of space limitations for antennas this band was not explored.

Power. - It was calculated that, in order to track a satellite at an altitude of a few hundred miles, a radar would have to have a beam power (i.e. including antenna gain) of at least 400 megawatts in the 300 to 900 mc interval and 1,000 megawatts in the 2500 to 3200 mc interval.

Receivers. - Because of the expected high power levels it was possible to use simple crystal video receivers. Frequency was defined by band-pass filters and the received power criterion was achieved by adjusting the receiver thresholds to -20dbm (300 - 900 mc) and -32dbm (2500 - 3200 mc) respectively.

Pulse Repetition Frequency. - Initially it was assumed that the lowest pulse repetition frequency to be encountered would not be less than 17 pulses per second.

Pulse Counting Circuits. - Each receiver was equipped with two binary registers or accumulators; a counting register and a storage register which

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SOCTOP (Cont'd)

cooperated to provide a coarse measure of pulse arrival rate. The counting register accumulates pulses from the receiver until 255 pulses have been received at which time it sends a single pulse to the storage register and then accepts no more pulses. Every fifteen seconds, however, the counting register is reset to zero so the content of the storage register indicates the number of fifteen second intervals in which 255 or more pulses have been received. The content of the storage register is read out via the telemetry link, after which the register is reset to zero.

Antenna Pattern. - The antenna pattern is such that a ground area approximately 350 miles long by 1,000 miles wide is covered at an altitude of 200 statute miles.

Programming. - The equipment was programmed "on" and "off" by the vehicle stored program command system. This command system is preset prior to launch. Initially it was programmed on over the entire Soviet area; however, on later flights it was programmed on in segments of orbits over the Soviet area as an aid to location of emitters.

Operations.

SOCTOPS were flown on the following Discoverer vehicles:

<u>Vehicle</u>	<u>Date Launched</u>	<u>Bands Flown</u>
1057	10 Aug 60	2500 - 3200 mcs
1058	13 Sep 60	300 - 900 and 2500 - 3200 mcs
1061*	26 Oct 60	300 - 900 and 1000 - 6000 mcs
1062	12 Nov 60	300 - 900 mcs
1105**	30 Mar 61	300 - 900 and 1000 - 6000 mcs
1107	16 Jun 61	1000 - 6000 mcs

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<u>Vehicle</u>	<u>Date Observed</u>	<u>Frequency Band</u>
1107	7 JUL 61	1000 - 6000 cps
1109	7 AUG 61	1000 - 6000 cps

... not observed ... antennas not uncovered.
... in a ... orbit.

... 108 and 109 ... frequency
... 300 - 900 cps; because of the presence of the ... band ...
... tracking information could be derived in ... band.

Vehicle 1053 was tracked in real-time in the ... orbits
... 10, 17 and 25. In orbit 10, over the Pacific, the radar (c)
appeared to have ... one-third to one-fourth of 25.

No indications were obtained of long-range tracking activity over
the ...; however, the occasional contacts obtained in both frequency bands
are consistent with the existence of a northern pointing, fixed-beam
radar north of Moscow.

The high band S ... carried on vehicles 1107 and 1109 experienced
circuit malfunctions.

Conclusions.

The ... answered the question they were designed to ask.

Valuable information can be obtained with extremely modest equipment
and data link requirements --- in the latter case, three commutator points
in a vehicle no carrier.

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SOCTOP (Cont'd)

Recommendations.

The SOCTOPS have served their purpose and accomplished their primary objective; better location of tracking radars should be performed by Project 102 payloads.

Since it will be desirable to determine whether the pending reconnaissance vehicles are being tracked -- and if so, in what frequency interval -- SOCTOPS are being procured to cover the frequency ranges 300 - 900, 900 to S-Band and S-Band to C-Band.

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T A K I

PURPOSE	TALL KING DETECTOR GENERAL ACTIVITY INDICATOR
RECEIVERS	XYTAL VIDEO
PULSE WIDTH	2 - 40 MICROSECONDS
PULSE REPETITION FREQUENCY	20 - 1000 PULSES PER SECOND
ANTENNA PATTERN	APPROXIMATELY 400 NM CIRCLE @ 1000 MCS AND 700 NM CIRCLE @ 100 MCS
DATA STORAGE	VEHICLE TAPE RECORDER
READOUT	FM/FM TELEMETRY Sixty point commutator
FREQUENCY	160 - 175 MCS 100 - 1000 MCS
PEAK POWER DETECTION	-22 to 0 DBM
AVERAGE POWER DETECTION	1 to 8% DUTY CYCLE

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TAKI

Purpose

The TAKI receiver is a special purpose, light weight receiver. The receiver performs two basic functions: (1) it is a TALE KITT detector and (2) it detects and analyzes signal parameters of emitters within its overall frequency range.

Description

TAKI "A" - The TALE KITT detection portion of the system is a crystal video detector. The signal from the 100 - 1000 mc antenna is fed through a power divider to a 160 - 175 mc band-pass filter. A matched crystal detector and a unit tuned to 170 mc demodulates the video pulses. Video bandwidth is approximately 1 mc. The output of the crystal detector is amplified and passed through a threshold to reduce the number of noise pulses. This threshold is preset at 10 db above the tangential signal level to produce a low false alarm rate. Signals with sufficient signal strength pass through the threshold circuit to a flip-flop circuit, a pulse repetition gate of 95 pulses per second or multiple thereof and a 10 microsecond pulse width gate to three log rate count meters. The PRF and PW gates are accurate to within 1%. These log rate count meters serve as both storage and recount devices. The data from the log rate count meters are computed once each second and stored in the vehicle tape recorder. The data are read out via the Fk/Fm telemetry link.

TAKI "B" - The TAKI "B" portion of the system is a wide open detection system over a frequency band of 100 - 1000 mcs. Video pulse demodulation is accomplished by a broadband matched crystal. The signal is then amplified by a log video amplifier. Eight outputs are obtained from the system.

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TAMI (Cont'd)

Light outputs are:

(1) "A" Counter - The "A" pulse counter counts the number of pulses entering the receiver each second. Its range is from 1 to 100 pulses per second.

(2) "B" Counter - Function is same as the "A" counter. Its range is from 10 to 100 pulses per second.

(3) PRF - The pulse repetition frequency is measured from 20 to 1000 pulses per second.

(4) ITT - The interlaced pulse train detector measures the number of times two pulses arrive in the system within 1 microsecond of each other. Both the PRF and PW measuring circuits are subject to error in the presence of two or more interlaced pulse trains.

(5) PW - Pulse width is measured from 2 to 100 microseconds.

(6) Peak Power Detector - The peak power detector measures the peak input signal power.

(7) Average Power Detector - The average power detector measures the duty cycle of the power received in the system.

(8) Analog - An analog recording is made of a signal by using several consecutive commutator points. A video sample of the peak detector is also sampled on seven consecutive commutator points.

All of the above measurement devices have their own storage system. The input to the storage system is a voltage commensurate with the measurements obtained by each device. The storage circuits are sampled once each second and the data stored in the magnetic tape recorder. The data are read out via the telemetry link.

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TAKI (Cont'd)

Antenna Pattern. - The antenna pattern is approximately a 700 NM circle at 100 mcs and a 400 NM circle at 1000 mcs.

Programming. * The system is programmed "on" and "off" by the vehicle stored program command system. These commands are preset prior to launch of the vehicle.

Operations.

TAKIs have flown on the following Discoverer flights:

<u>Vehicle</u>	<u>Date Flown</u>	
1107	16 Jun 61	
1111	4 Aug 61	(Vehicle failed to orbit)

Results.

On vehicle 1107 flown 16 June 1961 the TAKI was operational through 38 orbits. Failure after the 38th orbit was due to loss of power.

The TAKI was programmed "on" 26 of these 38 orbits. Intercept data were collected on 24 of the 26 programmed orbits. An estimated 8,600 data frames of intercept information was intercepted during the flight.

Although the results of this mission have not been completely analyzed, or are not available as of this date, the following results can be stated:

- (a) A high number of intercepts were made on orbit #25. (High signal density)
- (b) There was a higher than average number of intercepts made on orbit #16.
- (c) On orbit #16 an intercept was made of an exceptionally strong signal. This signal had a PRF of less than 100 pulses per second and a pulse width of approximately 22 microseconds.

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(a) The video monitor was inoperative during the flight and the recorder was inoperative for the duration of the flight. The video monitor was inoperative for the duration of the flight.

(b) The video monitor was inoperative during the flight and the recorder was inoperative for the duration of the flight. The video monitor was inoperative for the duration of the flight.

(c) The video monitor was inoperative during the flight and the recorder was inoperative for the duration of the flight. The video monitor was inoperative for the duration of the flight.

(d) The video monitor was inoperative during the flight and the recorder was inoperative for the duration of the flight. The video monitor was inoperative for the duration of the flight.

(e) The video monitor was inoperative during the flight and the recorder was inoperative for the duration of the flight. The video monitor was inoperative for the duration of the flight.

(f) The video monitor was inoperative during the flight and the recorder was inoperative for the duration of the flight. The video monitor was inoperative for the duration of the flight.

The recorder that was originally officially for this project was not received in time for this flight and a substitute recorder was used. This recorder had poor speed characteristics as a result; direct conversion of the data to machine readable form was not possible. However, the manually reduced information has been punched on cards for further machine processing. Also, recent analysis has shown that by the use of appropriate conversion charts reference tags can be placed on intercept information with a fairly high degree of accuracy. Stanford University will publish the results in their 4th quarterly report on SAC Contract AF OR(67)-732. All future data should be amenable to machine processing.

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TEXT ()

Classification:

not to be used in any form for computer processing. The interest

that sensitivity be increased in those circuits where it was found

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10 - 1000
 20 - 1000
 30 - 1000
 40 - 120
 50 - 120
 60 - 120
 70 - 120
 80 - 120
 90 - 120
 100 - 120
 110 - 120
 120 - 120
 130 - 120
 140 - 120
 150 - 120
 160 - 120
 170 - 120
 180 - 120
 190 - 120
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 870 - 120
 880 - 120
 890 - 120
 900 - 120
 910 - 120
 920 - 120
 930 - 120
 940 - 120
 950 - 120
 960 - 120
 970 - 120
 980 - 120
 990 - 120
 1000 - 120

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WILD BILL

Purpose.

A special purpose, light weight receiver designed to detect high powered CW and pulse modulated signals in the VHF region.

Description.

Receiver. - A sweeping filter with a pass band of 10 mc sweeps the frequency band of 40 - 120 mc in approximately 0.8 seconds. Upon detection of a signal the sweeping filter dwells on the signal for the remainder of the sweep period. The video output from the sweeping filter is amplified and sent to the measuring circuits.

Measuring Circuits. - The following measurements are made on signal parameters:

(a) Pulse counter - The pulse counter counts the number of pulses entering the receiver each second.

(b) The interlaced pulse train detector measures the number of interlaced pulse trains which enter the system each second.

(c) PRF - The pulse repetition frequency is measured between 20 and 1000 pulses per second.

(d) PW - Pulse width is measured between 10 microseconds and 4 milliseconds.

(e) Peak Power - The peak power detector will measure input signal power between approximately -50dbm and -37dbm. The AGC circuit will limit the amplitude of all signals greater than -40dbm. This circuit will serve primarily to indicate the effectiveness of this action. In this manner signals not quite strong enough to be accepted by the measuring circuits will be detected.

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TELEVISION (Continued)

(C) Average Detector - The average modulation detector will measure the average modulation in the case of sinusoidally modulated W-F, or the duty-cycle, in the case of a pulse modulated signal.

() Frequency - Intercept frequency is determined by a reference voltage from the sweep filter.

Output. - All of the above data are fed to a computer and sampled at the rate of once each second. Narrow band video and wide band video is also computed. In addition other measurements are computed; i.e. temperature, voltage readings, etc. Timing pulses reset the storage system each second. The data are stored in the vehicle tape recorder and transmitted to ground via the FM/FM telemetry link.

Programming. - The system is programmed "on" and "off" by the vehicle stored program command system. These commands are preset prior to launch of the vehicle.

Operations.

WFO 3 UJ flew on Discoverer 1109 on 7 July 1961. The vehicle was successfully orbited, however, the following two phenomena caused a 41% loss of information:

The vehicle was to have carried an eight minute endless-loop tape recorder; last minute failure of the recorder, however, necessitated substitution of a four minute endless-loop recorder so connected that, by using two tracks, eight minutes of storage time was achieved. Readout, however, was destructive and could not be repeated.

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ILD 101 (Cont'd)

... cause of a fault in the, or loose connection in the receiver...
failed to operate, or, all, or a part, of a reconnaissance pass and sometimes
failed to operate, or, all, or a part, of a resident pass. This erratic
behavior not only caused a loss of information but also made extremely difficult
the task of determining precisely whence the data were acquired. Figures ___
and ___ illustrate the behavior during the receiver operating history was
plotted by test.

Results.

In the information that was retrieved, at no time was the sweeping filter
observed to be in a signal, indicating that no signal was encountered which
was enough power to set the level of the automatic gain control. In several
passes, however, it was possible to read peak power measurements.

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RECEIVED: MISSY AND ANDREW'S JUMP DETECTOR

PLATE

RECEIVER

SUB-ANTENNA

DATA STORAGE

VARIABLE RESISTANCE

REAGENT

IN/OUT TELEMETRY LINK
ANALYSIS

ANGINA PATERN
200 mile altitude

250 - 350 IN. CIRCLE

EX JELCY

100 - 300 MDS

800 - 1600 MDS

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TOPSOC

INTRODUCTION.

A signal detection device with a special capability of detecting
UHF-frequency radars [REDACTED]

Description.

Receiver. - TOPSOC is a two-channel detection system. One band of the receiver covers the frequency range 400 - 800 mcs and the other band covers the frequency range 800 - 1600 mcs. Both channels of the receiver are superhetrodyne. The system is capable of intercepting signals with a PRF of 100 to 5000 pulses per second, a pulse width of 5 microseconds to 1 millisecond and a signal strength of -25dbm to -45dbm. Scan time for both bands is 1/2 second. Bandwidth of the receiver is 4 mcs.

Antenna Pattern. - The antenna pattern is approximately a 350 MI circle at 400 mcs and a 250 MI circle at 1600 mcs.

Output. - The system has an analog output. A positive voltage output is used for the amplitude and pulse width of the received signal and a negative voltage is used to determine intercept frequency. Data are returned to the ground on the P./FM telemetry link. The vehicle tape recorder is used to store the information until it can be read out.

Programming. - The system is programmed "on" and "off" by the vehicle stored program command system. Events are preset prior to launch of the vehicle.

Operations.

TOPSOCs have flown on the following Discoverer vehicles:

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TRUCK (Cont'd)

<u>Vehicle</u>	<u>Date Launched</u>
1113	12 Sep 61
1114	13 Sep 61
1115	13 Oct 61

Magnetic tapes have been received from the first two vehicles and indicate that the equipment functioned properly.

Results.

Processing of the analysis records is being performed at NSA and SAC Omaha. It has been confirmed, however, that useful information was obtained.

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TYPE OF TRANSMISSION TYPE DATA LINK SIGNAL (S)

SECURITY CLASSIFICATION

CLASSIFICATION

A.
B.

VEHICLE TRACK INFORMATION

IN/OUT TELEGRAPHIC LINK

APPROXIMATE SPEED 700 KM PER HOUR

100 - 150 HZ

MODE

MODEM

MODEM

DATA STORAGE

MODEM

MODEM PART-RN

FREQUENCY

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ORBITAL RECONNAISSANCE
EXPERIMENT
TEXAS PINT

Purpose.

To recognize the presence of a specific data link signal which employs a keyed, tone-modulated signal.

To record, in suitably small increments, the carrier frequency of the signal.

To record whether the signal carrier was amplitude modulated, frequency modulated or both.

To record the presence of either or both of two possible tone modulations.

Description.

Receiver. - Mechanically scanned superhetrodyne which sweeps the band 100 to 150 mc in approximately 200 seconds. Local oscillator frequency is digitized in increments to provide signal frequency resolution to 0.25 mc or better.

Recognition Circuitry. - Digital logic requires the presence of three pulses within 120 milliseconds, and requires that the pulses be from 3 to 15 milliseconds wide. Audio frequency filters and a discriminator provide the remainder of the recognition.

Output. - Quantized levels between 0 and 5 volts appearing on three pins for application to the telemetry commutator.

Operations.

The Texas Pint was successfully orbited by Vehicle 1112 on 30 Aug 61. On orbit #1 the recorded reconnaissance information was successfully read out at the Kodiak Tracking Station as was the information from real-time operation.

Following orbit #1 a blown fuse in the "command readout" circuit prevented readout at the regular tracking stations; however, an auxiliary command, associated with the Vela Hotel experiment, permitted readout at Ascension

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TEXAS PINT (Cont'd)

Island. The result of this failure was that, after a readout at Ascension, the equipment would follow the preprogrammed timer schedule and collect reconnaissance information until the tape was full; at this point an end-of-tape signal prevented further recording until the next readout had been accomplished.

It is estimated that, because of this failure, a 75% information loss was suffered.

RESULTS.

The commutated wave-train of real time information from Orbit #1 was reduced manually at Lockheed and is presently being more carefully reduced at USAFSS. Almost every possible combination of "recognition" was observed, and it is tentatively concluded that one of the two calibration transmitters, operating at Elmendorf AFB, was observed at least once.

A digital decommutation has been tried on two of the readout tapes from the Ascension Island and has produced data of good quality. It is to be hoped, therefore, that all of the information can be processed by machine.

Tapes have been received from Ascension through orbit #90; how many contain useful information must await data reduction.

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TABLE 1

RECEIVER	DJ200 AN/100-1 TYPE SIGNAL
RECEIVER	SUPERHETERODYNE
RECOGNIZER	LOGIC CHANNELS
STORAGE	VEHICLE TAPE RECORDER Commutator
READOUT	E./S. INDICATING UNIT
ANALOG PATTERN	APPROXIMATELY 700 NA CIRCLES
FREQUENCY	62.5 - 71 MGS

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Function:

To receive an incoming (in frequency) signal of the AM/SS-1 type. To select, usually, the carrier frequency of the signal. To select either of two teletype channels (with priority switching to be in effect if both are present) to extract the information carried thereon. To record the extracted digital information in a form suitable for presentation as voltage or a correlated teletype points.

Description:

The receiver is a conventional AM superhetrodyne, electronically tuned by a variable voltage generator, and feeds detected AM to the demultiplexor-recognition unit. The demultiplexor decomposes the signal into individual channels, performs a recognition function, and feeds the desired signal to the reading and storage circuit, where the teletype portion of the signal is converted into voltages suitable for storage and retransmission.

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VII.

urpose.

To acquire all of the M/ENC-1 type signal of which the GRAPE JUICE equipment acquires only a part.

Description.

The equipment has not been completely defined but will probably be similar to the GRAPE JUICE equipment.

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200 mile altitude

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NEW JERSEY

REMARKS:

to recognize and stop scan on a specific data-link signal -- a sophisticated
equipment to the signal difference. The target of the signal is the signal, for
transmission and retransmission, the signal content. To record a time history of the
signal content of the received signal as an aid to position fixing.

Description.

The equipment is not yet designed, but it will have to be far more complex
than the T-1000.

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